

## Numerical modelling of interactions of waves and sheared currents with a surface piercing vertical cylinder

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### Highlights

- Two different approaches, 2-D OpenFOAM and Lagrangian wave-current simulations, are used to model focussed wave groups and sheared currents simultaneously in a controlled manner, and produce input conditions for 3-D OpenFOAM models to investigate wave-current-structure interactions.
- Good agreement between numerical results and experimental data is obtained, indicating that both approaches are capable of replicating experimental wave-current flows, and accurately modelling interactions between surface piercing cylinders and focussing waves on sheared currents.
- The performance of both approaches is evaluated in terms of accuracy and computational effort required.
- It is found that the method of coupling 3-D CFD and Lagrangian models is computationally slightly cheaper and slightly more accurate because of the use of a smaller computationally domain and the iterative wave-current generation in the faster Lagrangian model.

### Abstract

Vertical surface piercing cylinders, such as typical coastal wind turbine foundations and basic elements of many coastal structures, are often exposed to combined loading from waves and currents. Accurate prediction of hydrodynamic loads on a vertical cylinder in a combined wave-current flow is a challenging task. This work describes and compares two different approaches for numerical modelling of the interaction between focussed wave groups and a sheared current, and then their interactions with a vertical piercing cylinder. Both approaches employ an empirical methodology to generate a wave focussed at the location of the structure in the presence of sheared currents and use OpenFOAM, an open source Computational Fluid Dynamics (CFD) package. In the first approach, the empirical wave-on-current focussing methodology is applied directly in the OpenFOAM domain, replicating the physical wave-current flume. This approach is referred to as the Direct Method. In the second approach, a novel Lagrangian model is used to calculate the free surface elevation and flow kinematics, which are then used as boundary conditions for a smaller 3-D OpenFOAM domain with shorter simulation time. This approach is referred to as the Coupling Method. The capabilities of the two numerical methods have been validated by comparing with the experimental measurements collected in a wave-current flume at UCL. The performance of both approaches is evaluated in terms of accuracy and computational effort required. It is shown that both approaches provide satisfactory predictions in terms of local free surface elevation and nonlinear wave loading on the vertical cylinders with an acceptable level of computational cost. The Coupling Method is more efficient

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because of the use of a smaller computational domain and the application of the iterative wave-current generation in the faster Lagrangian model. Additionally, it is shown that a Stokes-type perturbation expansion can be generalized to approximate cylinder loads arising from wave groups on following and adverse sheared currents, allowing estimation of the higher-order harmonic shapes and time histories from knowledge of the linear components alone.

## Keywords

Focussed wave groups; sheared currents; wave-on-current focussing methodology; Lagrangian wave-current flume; Harmonic reconstruction; OpenFOAM;

## 1. Introduction

The review articles by Peregrine and Jonsson (1983a; b), Thomas and Klopman (1997) and Wolf and Prandle (1999) have shown that wave-current interaction is one of the important physical processes in coastal waters. The presence of a background current modifies the wave dispersion, wave-induced velocities and shear stress near the seabed etc., so has an effect on wave loads on structures and wave propagation near coastlines. Coastal engineering applications, such as the design of coastal protection and structures as well as the evaluation of sediment transport and coastal erosion, would benefit from an enhanced knowledge of this complex process and its effect on coastal structures.

In existing design methods, the current profile is usually assumed to be uniform with depth. The uniform current approximation may apply for large-scale ocean currents and deep tidal flows, but it fails to model wind-driven currents and tidal flows in shallow coastal waters that exhibit some degree of variation in the vertical direction (Chakrabarti, 1996; Forristall and Cooper, 1997; Stacey et al., 1999; Gunn and Stock-Williams, 2013). Previous studies demonstrated that the velocity shear modifies the wave dispersion relation (Swan et al., 2001a), produces changes in water-surface elevation (Tsao, 1959; Brink-Kjaer, 1976; Kishida and Sobey, 1988), and causes significant effect on the tendency of surface waves to break (Peregrine and Jonsson, 1983; Yao and Hu, 2005) in a different way when compared to depth-uniform currents. This work considers a current profile which varies with depth so has a significant depth-varying vorticity distribution. Such a profile is a more realistic representation of a current flow in some regions in the open sea.

The vorticity dynamics due to wave-shear current interaction can be described by the vorticity transport equations, which are obtained by taking the curl of the momentum equations. Analytical solutions of the vorticity transport equations exist only for the constant-vorticity case (the current is linearly sheared) (Thomas, 1981; 1990; Nwogu, 2009). For more realistic profiles that vary arbitrarily with depth, the computation is more difficult because of the changing vorticity field in space and time. For initially uniform vorticity, Kelvin's circulation theory applies and the vorticity remains uniformly distributed. Then the wave motion can be treated as an irrotational disturbance, as described by Teles Da Silva and Peregrine (1988). Approximations are necessary if analytical solutions are to be sought for the cases with arbitrary vorticity. Various techniques have been developed (Kirby and Chen, 1989; Swan and James, 2001; Ko and Krauss, 2008; Smeltzer and Ellingsen, 2017), yet these have limited range of applicability; the wave is linear or weakly nonlinear, and the current strength lies within a certain range (either weak, moderate or strong). The difficulties inherent to problems associated with strongly sheared currents have necessitated the use of Computational Fluid Dynamics (CFD), which is a promising tool for modelling the interactions between waves and current, and both with structures.

Much previous numerical work based on CFD has primarily concentrated on regular wave interactions with currents (Santo et al., 2017; Zhang et al., 2014; Markus et al., 2013; Li et al., 2007; Park et al., 2001). However, Tromans et al. (1991) suggested the use of NewWave-type focussed

wave groups as design waves representing individual extreme events in random seas. Jonathan and Taylor (1997), Taylor and Williams (2004), Santo et al. (2013), Christou and Ewans (2014), among others, confirmed that this theory is applicable to a wide range of wave conditions. The original NewWave theory was developed for deep water waves. Later it was demonstrated that it can be applied to waves on shallow water (Whittaker et al., 2016). The use of NewWave-type wave groups for wave-structure interaction has been demonstrated by Zang et al. (2006, 2010) for a ship-shaped fixed body and for a surface piercing cylinder, respectively. Further work using wave groups on cylinders was described in the papers by Fitzgerald et al. (2014) and Chen and co-workers (2014, 2016, 2018), and for jacket-type structures in Santo et al. (2018).

Wind turbines with cylindrical foundations are likely to be located in areas with severe wave conditions, with intermediate and shallow water depths and with significant currents generated by tides, storm wind shear etc. Thus, the interaction of focussed wave groups propagating on either following or adverse sheared currents with surface-piercing cylinders has direct practical applications.

The primary challenge in the numerical modelling of focussed wave groups on sheared currents is the simultaneous and controlled generation of focussed wave groups on flow with non-uniform vorticity. The co-existence of waves and currents alters both the evolution of the waves and the profile of the currents in a way unpredictable by existing analytical approaches. As such, neither the point of focus nor the elevation of the wave and the underlying flow field are known a-priori.

Various approaches are used to achieve wave focussing at a particular location and time in the absence of currents, including a dispersive focussing method and various iterative techniques. The dispersive focussing method calculates the initial phase shift of each wave component based on linear wave theory. This inevitably results in a shift of the actual focus position due to non-linear wave-wave interactions (Rapp and Melville, 1990; Baldock et al., 1996; Johannessen and Swan, 2001). The iterative methods reconcile this issue by iteratively correcting either only the initial phases (Chaplin, 1996; Yao and Wu, 2005) or both the initial phases and the amplitudes (Schmittner et al., 2009; Fernandez et al., 2014; Buldakov et al., 2017) of different wave frequency components in a wave group. The iterative approach derived in Buldakov et al. (2017) calculates the corrected input for the wavemaker considering only the linearized part of wave spectrum and therefore it differs from any previous methodology. This approach has been successfully applied to physical experiments of focussed wave groups on sheared currents (Stagonas et al., 2018a). The wave focussing methodologies discussed previously were mainly used in physical experiments; however, its application to a numerical wave flume is straightforward and can be implemented in a similar way to that in a physical flume (Stagonas et al., 2018b).

For either 2-D or 3-D CFD simulations, a computationally expensive fine grid is necessary to accurately resolve the non-linear evolution of focussed wave groups, and the complex flow-structure interaction. Applying empirical wave-on-current focussing techniques in CFD-based models, even in 2-D, may yield substantial increases of the computational effort required. To accommodate this, a faster numerical model may be used alternatively to produce the input wave-current kinematics for CFD-based models. This work describes and compares two CFD modelling approaches building on the widely used open-source CFD platform OpenFOAM. In the first approach, the wave-on-current focussing methodology (Stagonas et al., 2014; 2018a; 2018b) is applied directly to a CFD numerical wave flume, replicating the physical wave-current flume. 2-D simulations are performed first to calculate iteratively the boundary conditions required to produce focussed wave groups on different flow conditions - namely, quiescent flow without a current, adverse and following sheared current – and the interaction with the structure is then modelled in 3-D. This approach is referred to as the Direct Method hereafter unless otherwise stated.

In the second approach, a novel Lagrangian model (Buldakov et al., 2015) is coupled with the CFD model. Differentiating it from recent one-way ‘online’ coupling approaches used to, e.g., model the

interaction of waves with cylinders (Paulsen et al. 2014a; 2014b), the time histories of the surface elevation and flow kinematics are pre-computed using the Lagrangian model and are then used as inlet boundary conditions for the CFD model. In this ‘offline’ coupling, all reflections are dealt by the CFD simulation, eliminating the need for simultaneous computation and exchange of information between the two models. This approach is referred to as the Coupling Method. We note that such a method of domain decomposition, i.e. one-way coupling of simpler models with more advanced models, was also applied by Biaisser et al. (2004), Drevard et al. (2005), Christensen et al. (2009), among others for various flow problems but excluding the effect of flow currents. Here, the 3-D numerical flume used in the Coupling Method is considerably shorter than that of the Direct Method, and the iterative wave-current generation is applied in the faster Lagrangian model. The performance of both approaches is validated against experimental measurements and is evaluated in terms of accuracy and computational effort. The rest of the paper is organized as follows. The physical experiments on wave-sheared current-cylinder interactions are described in Section 2. Details of the CFD and Lagrangian models are provided in Section 3. The results of both numerical modelling methodologies are compared with the experimental results in Section 4. Section 5 reconstructs the higher order harmonic forces using linear components alone. Conclusions are given in Section 6.

## 2. Experimental setup and methodology

A set of experiments on wave-sheared current interactions with a vertical surface-piercing cylinder of two different sizes was carried out and used to validate the proposed two CFD-based numerical models in this work. This section describes the experimental setup and the applied methodology briefly.

### 2.1 Experimental setup

All experiments were conducted in a 20 m long, 1.2 m wide and 1 m deep recirculating wave-current flume at University College London (UCL) with a water depth of 0.5 m. Two Edinburgh Design Limited (EDL) force-feedback ‘piston-type’ wavemakers, one at each end of the facility, were used to generate and actively absorb the waves. The flow entered vertically into the working section of the flume with the inlet and outlet located approximately 1 m in front of each wavemaker, as shown in Figure 1. A Cartesian coordinate system  $Oxz$  is introduced in both physical and numerical wave flumes such that the origin  $O$  is the plane of the undisturbed free surface,  $x = 0$  is the focus point, and  $z$  positive upwards.

The critical challenge of generating controlled and stable sheared currents was addressed through the use of two carefully designed flow conditioners/profilers installed on top of the inlet and the outlet. The conditioners/profilers consisted of 0.5 m long, 1.2 m wide and 0.88 m deep box sections consisting of vertically and horizontally placed cylindrical elements. Each cylindrical element had a diameter of 8 cm and was constructed using a 5 cm porous galvanised wire mesh, see Figure 2. Compared to previous work, the flow shaping approach used here has the comparative advantage of producing sheared currents with variable vorticity distribution without considerable interference to the generation of waves, see for example Steer et al. (2017) and for more details see Stagonas et al. (2018a).

Flow kinematics were measured with a high speed, time resolved Particle Image Velocimetry (PIV) system produced by TSI Incorporated. The system employs a 5 W water cooled Argon Ion laser operated at a pulsating frequency of 1 kHz. A light arm was used to direct the laser sheet upwards through the bottom of the wave flume (the bed) and measurements were taken at the focus point (FP in Figure 1) and at a distance of approximately 27 cm from the side wall; these were also the locations of the free surface elevation measurements. The flow was seeded with 50  $\mu\text{m}$  polyamide particles and PIV images with a resolution of 1024 $\times$ 1024 pixels were recorded at a frame rate of 250 fps. An example of the kinematics measured for adverse and following currents will be given in the following

section. The velocity measurements are available from still water level ( $z = 0$  m) to approximately 15 cm from the bed, beyond which the camera view was blocked by the support structure of the flume. Surface elevation measurements are used not only for validation but also for providing the inlet boundary conditions for the numerical models.

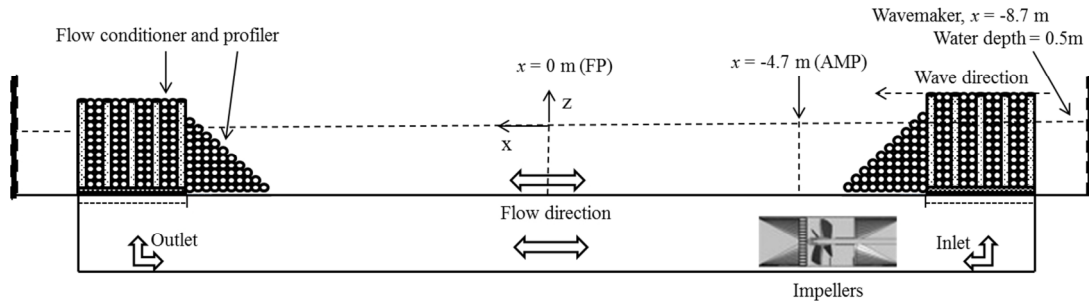


Figure 1. Schematic side view of the UCL wave-current flume showing two wavemakers at each end of the flume, and locations of inlet and outlet of the current discharge. FP stands for Focus Point, and AMP means the location for amplitude matching.

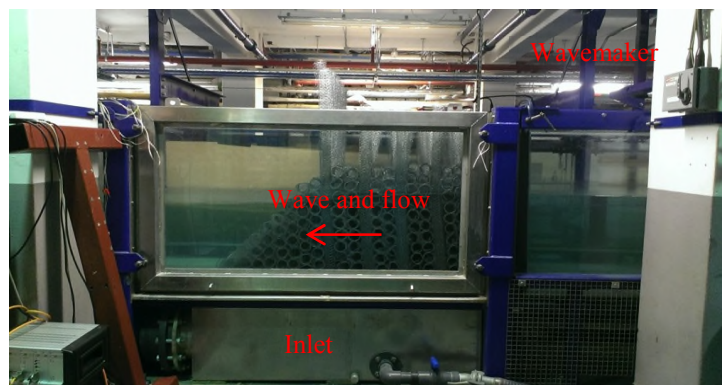


Figure 2 Photograph of the conditioning and profiling system

Focussed wave groups were produced using a Gaussian target spectrum on a water depth of 0.5 m. The same target spectrum was used for waves on adverse and following currents and without a current. The peak frequency was set to 0.6 Hz and the point of focus was 8.7 m from the wavemaker (FP in Figure 1). The phases of different components in a wave group were forced to come to focus at the focus point and the amplitudes were matched to the target spectrum at a distance of 4 m upstream of the focus point (AMP in Figure 1). In this way, focussed wave groups with the same spectrum at a relatively short distance (1 m) from the inlet were produced for all flow conditions. The evolution to focus was measured in the physical wave-current flume using a set of wave gauges, providing the means to validate the numerical results not only at the focus point but also in terms of the evolution of the wave group along the flume.

Free surface elevations in the flume were measured using 7 twin-wire resistance-type wave gauges positioned at  $x = -4.7$  m,  $-3$  m,  $-1.8$  m,  $-1$  m,  $-0.5$  m,  $-0.25$  m, and  $0$  m, and sampled at 100 Hz. A return period of 128 s and a focus time of 64 s were selected for the wave generation. Discrete input spectra consisting of 256 frequency components with  $\Delta f = 1/128$  Hz were used as input to the wavemaker. For simplicity, the wave groups produced were categorized based on the linear sum of the target amplitude components,  $A_L$ . Only the results of nonlinear wave groups with  $A_L = 0.07$  m are used in the present work. The methodology employed to generate these wave groups and sheared currents both in the physical and numerical wave flumes will be described in the following subsection.

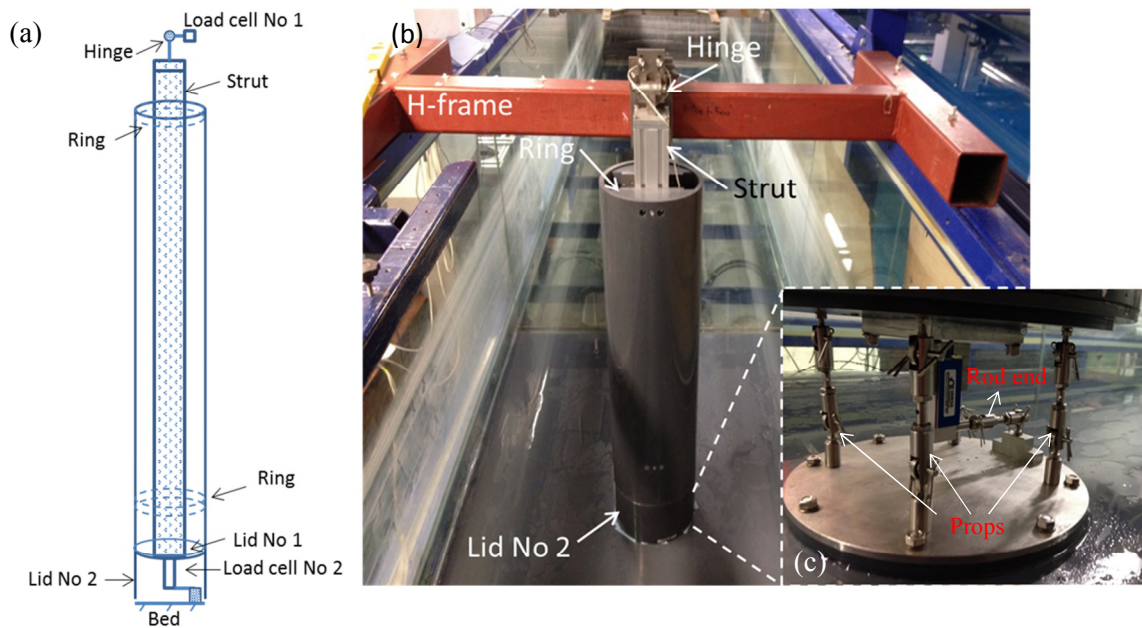


Figure 3 (a) Schematic diagram of side view of the cylinder model illustrating the location of the two load cells used to measure horizontal loads. (b) Photograph of the installed cylinder model. (c) Photograph showing the four props supporting the weight of the cylinder and the second/bottom load cell connected to the bed of the flume.

Experiments were also conducted with two different cylinders positioned at  $x = 0$  m; for these cases an additional wave gauge was placed at the front face of the cylinder. The smaller cylinder had a diameter of 0.165 m and the larger one 0.25 m. For the smaller cylinder, flow induced loads were measured using the load cell set-up described and used in Santo et al. (2017). However, in order to more effectively support the weight of the larger cylinder, a different arrangement was developed to measure the fluid induced horizontal force. The larger cylinder was a polyvinyl chloride (PVC) tube with a diameter of  $D = 0.25$  m. The PVC tube/cylinder was connected to an aluminium rectangular column, marked as 'strut' in Figure 3, via circular rings. The strut had dimensions of 0.09 m (breadth)  $\times$  0.09 m (width)  $\times$  1 m (height), and was connected to a load cell rated at 100 kg from the top through a hinge. This load cell, labelled as Load cell No 1, was rigidly fixed on the steel H-frame that was in turn tightly fixed on the flume walls. Another load cell, labelled as Load cell No 2, was located approximately 10 cm above the flume's floor and was tightly fixed on the bottom of the cylinder/strut, see Figures 3(a) and (b). The opposite end of the second load cell was connected through rod end bearings to an aluminium base, which was in turn fixed on the bed of the flume. Four props, also made using rod ends, supported the weight of the structure resulting in a preload-free cell, see Figure 3(c). The overall arrangement consisting of a strut, connecting rings and two load cells was mounted on rather than suspended from a steel H-frame. The latter arrangement was used in this study for the smaller cylinder as aforementioned and in previously reported tests by Santo et al. (2017).

A piece of PVC was used to model the bottom of the cylinder, labelled as Lid No 1, which was approximately 10 cm above the bed of the flume. Another piece of PVC, labelled as Lid No 2, was used to extend the model cylinder down to approximately 5 mm from the bed and compartmentalise the model cylinder, see Figures 3 (a) and (b). The compartment below the Lid No 1 was flooded and therefore a water-resistant load cell (Load cell No 2 in Figure 3) was used. Both load cells were sampled at 1 kHz and the experimental apparatus was calibrated for both tension and compression using dead weights at the beginning and the end of every testing cycle.

Surface elevation measurements recorded at  $x = -4.7$  m (AMP in Figure 1) for different test cases illustrate a satisfactory level of repeatability. Representative results for experiments with waves on an adverse current with the small (solid line) and the large (dashed line) cylinder in place are presented in Figure 4. The repeatability in load cell measurements for the same testing conditions was also tested. Standard deviations of 0.11 N/0.2 N were calculated from 15 horizontal force records acquired in consecutive repeat tests with the smaller/larger cylinder exposed to waves on the adverse current. These horizontal force results are representative of all the cases considered. It is to be noted that an iterative methodology is used in both physical and numerical wave flumes to generate focussed wave groups and sheared currents in a controlled manner. In the following sections, the iterative methodology is presented first and then the numerical flumes and implementation are described.

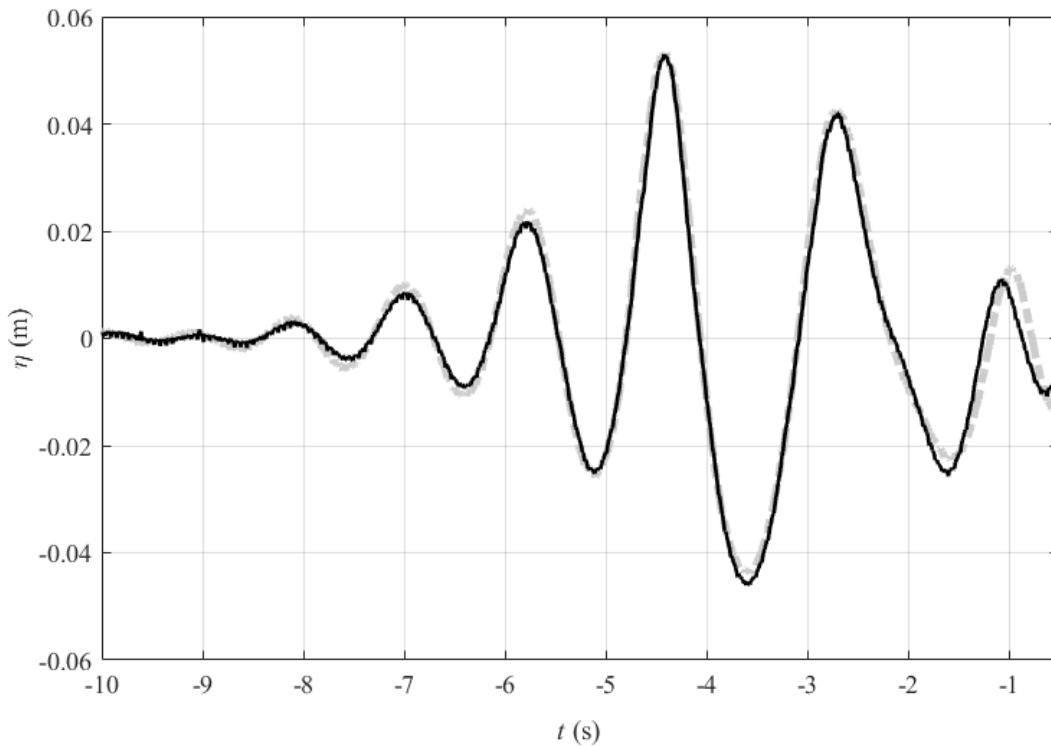


Figure 4 Example of free surface elevation time histories recorded at  $x = -4.7$  m, for  $U = -0.2$  m/s and  $A_L = 0.07$  m. Solid line: free surface elevation profile measured with the larger cylinder installed in the flume. Dashed line: free surface elevation profile measured with the smaller cylinder installed in the flume.

## 2.2 Generation of focussed waves on adverse and following currents

A methodology to accurately generate focussed waves without a current is described in Buldakov et al. (2017) and for waves on sheared currents in Stagonas et al. (2018b). The linearized part of the wave spectrum is isolated by linearly combining four non-linear free surface elevation time histories measured in the wave flume. Initially, a crest focussed wave is produced in the flume and the remaining three wave groups are generated with phase shifts of  $\pi$ ,  $\pi/2$  and  $3\pi/2$ . The measured spectrum (written as a complex variable  $a+ib$ ) is then decomposed as

$$\begin{aligned}
S_0 &= \frac{S_0 + S_1 + S_2 + S_3}{4} \\
S_1 &= \frac{S_0 - iS_1 - S_2 + iS_3}{4} \\
S_2 &= \frac{S_0 - S_1 + S_2 - S_3}{4} \\
S_3 &= \frac{S_0 + iS_1 - S_2 - iS_3}{4}
\end{aligned} \tag{1}$$

where,  $s_n$  are complex spectra of the fully nonlinear surface elevation signals with  $0, \pi/2, \pi$  and  $3\pi/2$  phase shifts.  $S_0$  is the complex spectrum of the 2<sup>nd</sup> order difference components and  $S_1, S_2$  and  $S_3$  are complex spectra of nonlinear super-harmonics for 1<sup>st</sup> (linear), 2<sup>nd</sup> (+) and 3<sup>rd</sup> harmonic, respectively.

New input amplitudes are then calculated based on the measured and the target amplitudes. In the same way input phases are also calculated

$$\begin{aligned}
a_{in}^n(f_i) &= a_{in}^{n-1}(f_i) a_{tgt}(f_i) / a_{out}^{n-1}(f_i) \\
\phi_{in}^n(f_i) &= \phi_{in}^{n-1}(f_i) + (\phi_{tgt}(f_i) - \phi_{out}^{n-1}(f_i))
\end{aligned} \tag{2}$$

where  $a_{in}^n(f_i)$  and  $\phi_{in}^n(f_i)$  are the amplitude and phase of an input spectral component at frequency  $f_i$ , respectively.  $a_{out}^n(f_i)$  and  $\phi_{out}^n(f_i)$  are the amplitude and phase of the corresponding spectral components of the measured/recorded output spectrum, respectively. The superscript  $n$  indicates the  $n$ -th iteration.  $a_{tgt}(f_i)$  and  $\phi_{tgt}(f_i)$  are set by the preselected target spectrum.

Iterations continue until the measured linearized amplitude spectrum matches the target amplitude spectrum, and the phases of the linearized part are zero at the desired location in the flume. By matching the measured amplitude spectrum to the target spectrum, NewWave-type focussed wave groups are generated in either physical or numerical wave flumes. The methodology has also been successfully applied to generate breaking waves by focussing in a CFD wave flume (Stagonas et al., 2018b) and in the present work it is applied to a CFD-based numerical model and a Lagrangian numerical flume with following and adverse sheared currents.

### 3. Numerical setup

Two approaches are used to replicate wave-current conditions generated in the physical flume, thus providing input conditions for the 3-D CFD model with the structure in place. In the first approach, the iteration scheme in physical experiments described in Section 2.2 is applied directly in the 2-D CFD model, while in the second approach, a Lagrangian model (Buldakov et al., 2015) is used to provide input conditions for the 3-D CFD model to reduce the size of the 3-D numerical CFD flume and shorten the simulation time.

In this section, we first present a general description of OpenFOAM-based numerical models and then the methodologies used for replicating the wave-current flow generated in the physical wave-flume are detailed. The accuracy and the efficiency of the methodologies are validated by comparing with the experimental measurements.

#### 3.1 OpenFOAM-based numerical model

The CFD model based on OpenFOAM solves the Reynolds-averaged Navier-Stokes (RANS) equations coupled with the continuity equation for the two-phase combined flow of water and air with the incompressibility assumption,



$$\nabla \cdot \mathbf{u} = 0 \quad (3)$$

$$\frac{\partial \rho \mathbf{u}}{\partial t} + \nabla \cdot (\rho \mathbf{u} \mathbf{u}) - \nabla \cdot (\mu \nabla \mathbf{u}) = -\nabla p^* - \mathbf{g} \cdot X \nabla \rho + \nabla \cdot (\rho \boldsymbol{\tau}) + \sigma \kappa \nabla \alpha \quad (4)$$

where  $\rho$  and  $\mu$  are the density and the dynamic viscosity of the mixed fluid, respectively, which are calculated following the equation (6) based on the Volume-of-Fluid (VOF) technique, which will be discussed below.  $\mathbf{u} = (u, v, w)$  is the fluid velocity field in Cartesian coordinates, and  $p^*$  is the pressure in excess of hydrostatic pressure, defined as  $p^* = p - (\mathbf{g} \cdot X)\rho$ .  $\mathbf{g}$  is the acceleration due to gravity and  $X = (x, y, z)$  is the position vector. The usual Reynolds stress tensor  $\boldsymbol{\tau} = (2/\rho)\mu_t \mathbf{S} - (2/3)k\mathbf{I}$  where  $\mu_t$  is the dynamic eddy viscosity,  $\mathbf{S}$  is the strain rate tensor ( $1/2(\nabla \mathbf{u} + (\nabla \mathbf{u})^T)$ ) and  $k$  is the turbulent kinetic energy per unit mass.

Various turbulence closure models are implemented in OpenFOAM to account for turbulence generation and dissipation, including  $k$ - $\varepsilon$ ,  $k$ - $\omega$  and  $k$ - $\omega$  SST (Brown et al., 2016). These two-equation turbulence models solve equations for the turbulent kinetic energy  $k$ , and either the turbulence dissipation rate  $\varepsilon$  or  $\omega$ , the specific rate of dissipation of the turbulence kinetic energy, to estimate the Reynolds stress. Nevertheless, the laminar flow model of OpenFOAM-2.4.0 is used in all computations reported here as both the external wave fields and the wave force on the cylinder are dominated by inertial (potential flow) effects (Chen et al., 2014; 2018). The reasonably good agreement between the numerical and experimental data shown in the following sections indicates that the consequences of viscosity and flow turbulence on the free surface elevation and wave forces on the cylinder that are of interest in this study are negligible as expected and supports the use of the laminar flow model. It is useful to note that turbulence modelling may be important if drag forces and the formation of wakes are significance (Santo et al., 2015).

Additionally, it is worth mentioning that a mesh scheme with the widths of near-wall cells being ten times smaller than those away from the cylinder surface was used to study the effect of the boundary layer. The wall-normal mesh size is selected to ensure that the dimensionless wall distance ( $y^+$ ) is smaller than 5 based on the flat-plate boundary layer theory. The mesh dimensions for the regions away from the cylinder are determined by convergence tests to ensure that there are sufficient cells per wavelength to resolve propagating incident waves and wave-current-structure interactions; this will be discussed in more detail in the following section. It is found that further refining the mesh inside the boundary layer has negligible influence on the flow-induced forces on the cylinder. The Reynolds number  $Re (= \omega \eta_m^2 / \nu)$  and the maximum local Keulegan-Carpenter number  $KC (= 2 \omega \eta_m / D)$  in this study are approximately  $8.5 \times 10^4$  and 2.3, respectively.  $\omega$  is the peak wave angular frequency,  $\nu$  is the kinematic viscosity and  $\eta_m$  is the maximum free surface elevation which is about 0.15 m in this study.

The last term on the right-hand side of equation (4) is the effect of surface tension in which  $\sigma$  is the surface tension coefficient and  $\kappa$  is the curvature of the interface. The presence of surface tension is found to have minor effects in most civil engineering applications (Jacobsen et al., 2012; Larsen, 2018), thus,  $\sigma = 0$  is used in this study.

The Volume-of-Fluid (VOF) technique is applied in OpenFOAM to locate and track the free surface (interface between air and water), with the following transport equation,

$$\frac{\partial \alpha}{\partial t} + \nabla \cdot (\alpha \mathbf{u}) + \nabla \cdot (\alpha(1 - \alpha) \mathbf{u}_\alpha) = 0 \quad (5)$$

in which  $\alpha$  is the volume fraction function of water within each computational cell. This equation is similar to that proposed in Hirt and Nichols (1981), but with an additional compression technique (the last term on the left-hand side in which  $\mathbf{u}_\alpha$  is an artificial compression term) to limit the numerical diffusion of the interface profile. The compression technique is developed by OpenCFD, and details can be found in Berberovic et al. (2009).

The properties of the fluid at each cell are then calculated by weighting with the VOF function  $\alpha$ , which ranges from 0 (if there is no traced fluid inside a cell) to 1 (when the cell is full of the traced fluid),

$$\rho = \alpha\rho_{\text{water}} + (1 - \alpha)\rho_{\text{air}}; \mu = \alpha\mu_{\text{water}} + (1 - \alpha)\mu_{\text{air}} \quad (6)$$

The equations (3) – (5) are solved with the finite volume method in which the whole computational domain is discretized into a number of cells (Ferziger et al., 2002). The merged Pressure Implicit Splitting Operator (PISO) algorithm is then applied for each cell to decouple pressure from the momentum equation (Issa, 1986).

During any CFD simulation, numerical schemes for various terms in the equations (3) – (5) and the iterative solvers/algorithm settings need to be specified, which may have significant effects on the performance of the CFD solvers in terms of accuracy and efficiency (Larsen et al., 2018). The possible choices of numerical settings in OpenFOAM are extensive; users are recommended to start with the default settings given with the OpenFOAM tutorial, and the best choice can then be determined on a case-by-case basis by numerical experimentation.

In this study, the time derivatives are calculated by the forward Euler scheme, and linear schemes are utilized for other terms in the governing equations excluding divergences of momentum and mass fluxes. A linear-upwind divergence scheme is applied for the momentum flux in which the upwind interpolation weights are employed with an explicit correction based on the local cell gradient. The total variation diminishing (TVD) scheme that combines upwind and central differencing schemes is used for the divergence of mass flux. It is noted that the orthogonality correction is applied for the surface normal gradient due to mesh non-orthogonality. The equations for pressure are solved by preconditioned (bi-) conjugate gradient (PCG/PBiCG) solvers with the diagonal incomplete-Cholesky/incomplete-LU (DIC/DILU) preconditioner. Smooth solvers are utilized to solve the equations for velocity and the volume fraction function  $\alpha$  with the symmetric Gauss-Seidel smoother. The scheme and solver choices used in this study are discussed in more detail in Appendix 1 and are summarized in Table A1. The combination of these choices has proved to work well and yield good results when applied to nonlinear wave interactions with a vertical cylinder for ranges of flow conditions studied in this work.

### 3.2 Direct application of the iterative wave generation methodology in CFD models

This study uses and extends the toolbox ‘waves2Foam’ developed and released by Jacobsen et al. (2012) to realize wave generation and absorption in numerical wave flumes in OpenFOAM. The boundary conditions for generating waves are given analytically according to the linear wave theory, i.e. corresponding velocities and free surface elevations are specified at the input boundary faces. In this study, linear superposition of velocities of the spectral components of a wave group calculated using a desired spectrum (the spectrum of extracted linearized waves used here will be discussed later) is used to generate the focused wave group in the computational domain through a vertical wall.

A new boundary condition is developed within the framework of ‘waves2Foam’ to produce a vertically sheared current. The sheared current profile is defined by a second-order polynomial which is obtained by curve fitting the measured horizontal velocity profile at the model cylinder location. Figure 5 demonstrates the current profiles used in the CFD-based numerical simulations in this paper, and their comparison with measured experimental profiles.

The combined wave and current conditions are then generated by linearly superimposing the focussed wave group and sheared current at the inlet. The boundary condition for generating sheared current is also used at the outlet to ensure mass conservation. The initial conditions and other boundary conditions follow the same set-up as described in Chen et al. (2014) and Santo et al. (2017).

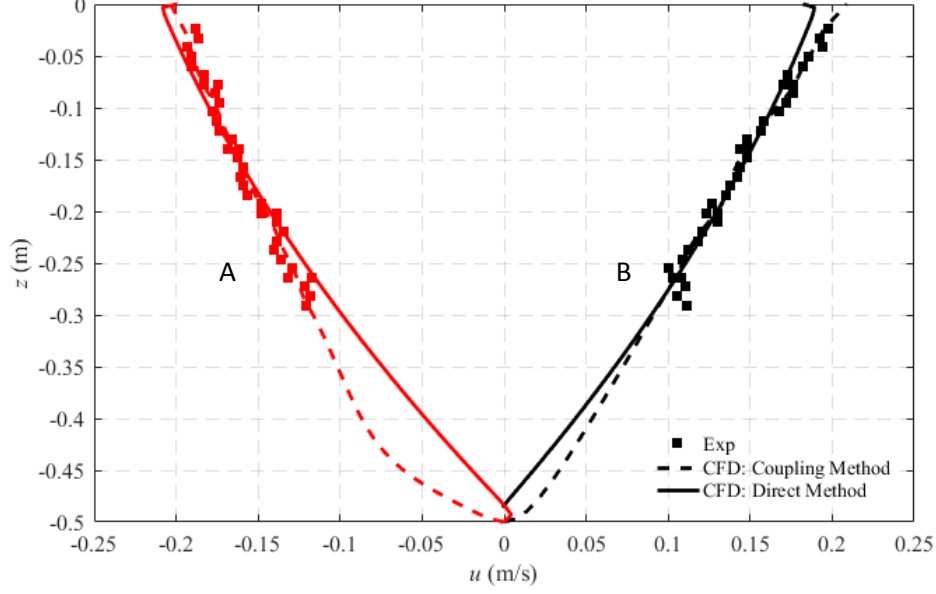


Figure 5 Comparisons of sheared current profile with depth obtained from the experiment and the numerical simulations at the location of the model cylinder for the cases with a sheared current and without waves. A -- Adverse current; B -- following current.

The wave-on-current focussing methodology described in Section 2.2 is now applied to generate focussed waves on various flow conditions in the numerical wave flume. All iterations are performed in a 2-D numerical flume replicating the physical flume at UCL. Although the target spectrum used in the physical wave flume can be used as inputs for the first set of simulations, the linearized spectrum extracted from the actual experimental measurements is used instead to ensure a faster convergence to the experimental measurements (i.e.  $a_{in}^0 = a_{tgt} = a_{Linear}^{exp}$  and  $\phi_{in}^0 = \phi_{tgt} = \phi_{Linear}^{exp}$  in equation 2). The free surface elevations at  $x = -4.7$  m (AMP in Figure 1) and  $x = 0$  m (FP in Figure 1) in the numerical flume are recorded and used for performing the amplitude and phase corrections following equations (1)-(2). Generally, satisfactory/convergent results for all flow conditions considered in this work are obtained within 1 or 2 iterations following the first set of simulations, i.e. in total three sets of 2-D simulations are required. The final corrected set of boundary conditions is then used as input for the 3-D numerical model shown in Figure 6. Previously, the same approach has been successfully used to simulate extreme forces induced by focussed waves on a following uniform current to a jacket structure, see Santo et al. (2018).

The 3-D numerical flume (which is shown in Figure 6) consists of a rectangular domain with a vertical cylinder located at the centre of the flume. The total length of the flume is 13.7 m ( $\sim 4\lambda_p$ ) with a distance of  $L_0$  between the inlet boundary and the vertical cylinder. The last 3 m ( $\sim \lambda_p$ ) of the numerical flume is occupied by the relaxation zone used to minimize wave reflections from the outlet.  $\lambda_p$  is the peak wavelength, which is  $\sim 3.2$  m in this study. The width of the computational domain is 1.2 m, and the water depth  $h$  is 0.5 m, the same as those in the experiments. In the Direct Method,  $L_0 = 8.7$  m ( $\sim 2.7\lambda_p$ ), the same as that in the experiments.

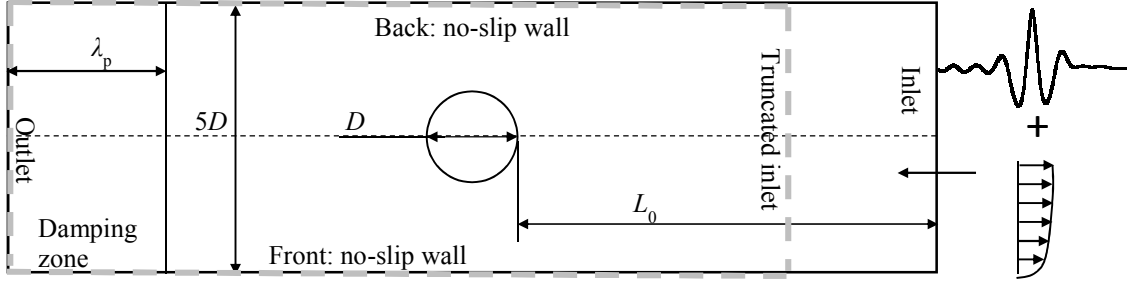


Figure 6 Layout of the computational domain.  $D$  is the diameter of the vertical cylinder and  $\lambda_p$  is the peak wavelength. The truncated inlet demonstrates the inlet boundary for the Coupling Method; the computational flumes for the Coupling and the Direct Methods are bounded by the dashed grey lines and the solid black lines, respectively.

The optimum set-up for the computational domain including its size and the mesh resolution is determined using numerical experimentation (not shown here for brevity but more details are given in Chen et al., 2014). The principle is to have the smallest possible domain size, thus minimum computational effort, while still maintaining the correct flow field around the structure.

Overall, the computational domain is divided into two areas, one with a coarser and one with a finer mesh resolution. In particular, the area near the vertical cylinder and the layers near the air-water interface are resolved with a finer mesh. Horizontal and vertical grid sizes for the coarser mesh are about  $\lambda_p/240$  and  $H_p/12$ , respectively;  $\lambda_p$  and  $H_p$  are the peak wavelength and the peak wave height, respectively. The cell size of the finer mesh is decreased by half, and cell sizes are graded so that the size of the cells between the two areas varies smoothly.

In addition to the spatial/mesh resolution aforementioned, the temporal resolution is also of importance. In OpenFOAM, the time step can either be fixed by specifying a pre-defined value or adjustable such that the maximum Courant number  $Co = u\Delta t/\Delta x$  is maintained throughout the whole domain at all times.  $\Delta t$  is the time step,  $\Delta x$  is the cell size in the direction of the velocity and  $u$  is the magnitude of the velocity at that location (Courant et al., 1967). In this study an adjustable time step is used to achieve  $Co = 0.25$ , which is again determined by numerical experimentation, and not shown here for brevity. For details refer to Larsen et al. (2018).

Table 1 Parameters and computational costs used for two OpenFOAM-based models

Parameters		Direct Method	Coupling Method
Overall length (m)		13.7 ( $\sim 4\lambda_p$ )	10 ( $\sim 3\lambda_p$ )
Overall width (m)		1.2 ( $\sim 5D$ )	1.2 ( $\sim 5D$ )
Distance from inlet to cylinder (m)		8.7 ( $\sim 2.7\lambda_p$ )	5 ( $\sim 1.5\lambda_p$ )
Distance from cylinder to outlet (m)		4.75 ( $\sim 1.5\lambda_p$ )	4.75 ( $\sim 1.5\lambda_p$ )
Length of damping (relaxation) zone (m)		3 ( $\sim \lambda_p$ )	3 ( $\sim \lambda_p$ )
Cell number (million)		$\sim 17.2$	$\sim 12.6$
Maximum Courant number $Co$		0.25	0.25
Computational costs (hrs)	Each 3-D	$\sim 12$	$\sim 15.5$
	Each 2-D	$\sim 1 (\times 12)^*$	--
	Total	$\sim 24^{**}$	$\sim 15.5$

\*In total 3 sets of 2-D simulations are required, and each set of 2-D simulations consists of 4 runs with successive additional phase shifts of  $\pi/2$ , in total 12 2-D simulations are required, and each 2-D simulation requires  $\sim 1$  hour computational time.

\*\*Total time is calculated as the summary of the computational time required to calculate the corrected inlet conditions using either the 2-D CFD model or the Lagrangian model and the time spent for the 3-D simulations.

The simulations were performed using the supercomputing facility at the Pawsey Supercomputing center which supports researchers in Western Australia. Utilising 48 cores for 3-D simulations, the computational time is approximately 12 hours to obtain the results within the time scale of interest, i.e.  $\sim 20$  s of modelled time corresponding to propagation of the wave group and its interaction with the model structure. Each 2-D simulation used to calibrate the incoming wave group takes about 1 hour using 24 cores. The geometric parameters used for 3-D simulations and computational costs required for both 2-D and 3-D simulations are summarized in Table 1.

### 3.3 Generation of the incoming wave-current flow by coupling the Lagrangian and CFD models

The second method of generating wave-current conditions is based on reconstructing experimental surface elevation and kinematics of incoming waves on sheared currents by applying the iterative wave generation methodology (Section 2.2) to a Lagrangian numerical wave-current flume. The Lagrangian kinematics and the free surface elevation are then fed into a truncated numerical CFD wave flume with the cylinder present using an external forcing subroutine built onto waves2Foam and OpenFOAM-based numerical models.

A general Lagrangian formulation for two-dimensional flow of inviscid fluid with a free surface can be found in Buldakov et al. (2006). We consider time evolution of coordinates of fluid particles  $x(a, c, t)$  and  $z(a, c, t)$  as functions of Lagrangian labels  $(a, c)$ . The formulation includes the Lagrangian continuity equation,

$$\frac{\partial(x, z)}{\partial(a, c)} = J(a, c), \quad (7)$$

the Lagrangian form of vorticity conservation,

$$\frac{\partial(x_t, x)}{\partial(a, c)} + \frac{\partial(z_t, z)}{\partial(a, c)} = \Omega(a, c) \quad (8)$$

and the dynamic free-surface condition,

$$x_{tt}x_a + z_{tt}z_a + gz_a \Big|_{c=0} = 0. \quad (9)$$

Functions  $J(a, c)$  and  $\Omega(a, c)$  are given functions of Lagrangian coordinates and are defined by the initial conditions.  $J(a, c)$  is defined by initial positions of fluid particles associated with labels  $(a, c)$ , and  $\Omega(a, c)$  is the vorticity distribution defined by the velocity field at  $t = 0$ . It is convenient to select initial undisturbed positions of fluid particles as Lagrangian labels  $(a, c) = (x_0, z_0)$ . This gives  $J = 1$ . For waves over a flat bed this defines a rectangular Lagrangian domain with  $c = 0$  being the free surface and  $c = -h$  being the bottom, where  $h$  is the undisturbed water depth. The boundary condition at the lower boundary can then be specified as,

$$z(a, -h, t) = -h \quad (10)$$

The presented Lagrangian formulation offers a simple treatment of vortical flows and therefore is suitable for modelling waves on vertically sheared currents. A sheared current can be defined by specifying vorticity depending only on the vertical Lagrangian coordinate  $c$ . For our choice of Lagrangian labels the parallel current can be specified as  $x = a + V(c)t$ ;  $z = c$ , where  $V(c) = V(z_0)$  is the current profile. Substitution to (8) gives,

$$\Omega(a, c) = \Omega(c) = V'(c) \quad (11)$$

Therefore, waves on a sheared current with an undisturbed profile  $V(z_0)$  are described by equations (7,8) with the free surface boundary condition (9), the bottom condition (10) and the vorticity distribution given by (11). Figure 5 demonstrates velocity profiles for adverse and following currents we are using in this paper and their comparison with measured experimental profiles. The current profiles applied for the Coupling Method (CFD: Coupling Method in Figure 5) are obtained from PIV and ADV (Acoustic Doppler Velocimetry) measurements of the current velocity using a Bezier smoothing algorithm.

For convenience and efficiency of numerical realisation, we modify the original problem (7-9) and write it in the following form,

$$\Delta_t \left( \frac{\partial(x, z)}{\partial(a, c)} \right) = 0; \Delta_t \left( \frac{\partial(x_t, x)}{\partial(a, c)} + \frac{\partial(z_t, x)}{\partial(a, c)} \right) = 0 \quad (12)$$

and

$$x_{tt}x_a + z_{tt}z_a + gz_a = RHS(a, t)|_{c=0}, \quad (13)$$

where the operator  $\Delta_t$  denotes the change between time steps and the right-hand side of the dynamic surface conditions includes various service terms. For calculations presented in this paper we use the following additional terms,

$$RHS = \left( \frac{1}{6} \delta_a^2 x_{aa,tt} - \frac{11}{12} g \delta_t^2 z_{a,tt} \right) - k(a) \left( (x_t - V(c))x_a + z_t z_a \right) + P_x(a, t), \quad (14)$$

where  $\delta_a$  and  $\delta_t$  are the numerical mesh step in  $a$ -direction and the time discretization step. The first term in (14) is the dispersion correction term, which increases the accuracy of the numerical dispersion from second to fourth order. The second term enforces dissipation of surface perturbations. It is used for absorbing reflections, and the dissipation strength is regulated by the coefficient  $k(a)$ . The last term in (14) is the prescribed time varying surface pressure gradient which is used for wave generation.

The numerical wave-current flume is created by specifying inlet and outlet boundary conditions, distribution of surface dissipation  $k(a)$  and the surface pressure gradient  $P_x(a, t)$  providing free in- and outflow of the current to and from the computational domain, generation of waves on/over the current and absorption of waves reflected from domain boundaries.

The (purely numerical) dissipation coefficient in the Lagrangian scheme is set to zero in the working section of the flume and gradually grows to a large value near the inlet and outlet boundaries. This results in a steady horizontal free surface at these boundaries which remain at their initial position  $z = 0$  providing parallel inlet and outlet flows. This serves a double purpose. First, reflections from the boundaries are significantly reduced. Second, the boundary conditions at the inlet and outlet can be specified as the undisturbed velocity profile at the inlet and as a parallel flow at the outlet,

$$x_t(a_{in}, c, t) = V(c); z_a(a_{out}, c, t) = 0. \quad (15)$$

Since the Lagrangian scheme assumes the input flow is inviscid but rotational, this justifies the use of free-slip boundary conditions on the side walls and flume bed in this part of the numerical simulations, though of course not on the curved boundary of the cylinder or on the flume sides and floor in OpenFOAM where a no-slip condition is implemented. The wave is generated by creating an area in front of one of the wave absorbers where pressure distribution of a prescribed shape is defined. Time-

varying amplitude of this pressure disturbance is used as a control input for wave generation. The problem is then solved numerically using a finite-difference technique. More details of the numerical method can be found in Buldakov (2013, 2014).

An additional difficulty with numerical realisation of the Lagrangian formulation on sheared currents is continuous deformation of the original physical domain. The accuracy of computations for strongly deformed computational cells reduces considerably. In addition, parts of the deformed physical domain can move outside the region of interest. To avoid these difficulties, we perform sheared deformation of the Lagrangian domain to compensate for the deformation of the physical domain. The deformation takes place after several time steps and moves boundaries of the physical domain back to the original vertical lines. After this Lagrangian labels are re-assigned to new values to preserve the rectangular shape of the Lagrangian computational domain with vertical and horizontal lines of the computational grid.

To reproduce experimental free surface elevation records, we use the iterative procedure described in Section 2.2. Amplitudes and phases of spectral components of a pressure control signal are modified iteratively to match amplitudes and phases of the calculated linearized surface elevation spectrum at selected wave probes with target spectra. Linearized spectra of the actual experimental surface elevation at locations  $x = -4.7$  m (amplitude matching position) and  $x = 0$  m (focus point) are used as targets for the iterative procedure. Each numerical wave is generated with phase shifts of  $n\pi/2$ , with  $n = 0, 1, 2, 3$ . This allows calculation of the linearized output signal of free surface elevation. The linearized output is then compared with the target, and corrections to the input spectrum for next iteration are calculated using the method described in Section 2.2. For further details of the iterative wave matching methodology refer to Buldakov et al. (2017). We apply the procedure to generate incoming waves for experimental cases presented in Section 2.1.

Lagrangian computations of the free surface elevation and flow kinematic time histories closer to the structure are used as boundary conditions for a new, truncated 3-D numerical CFD wave flume (when compared to the CFD domain of the Direct Method; dashed lines in Figure 6). The model cylinder is centrally located in the new domain and the inlet is set at a distance of 5 m upstream from the cylinder. Although Lagrangian calculations cover the full extent of the numerical flume, only the results at the inlet location (truncated inlet in Figure 6) are used in the 3-D CFD model using the Coupling Method. The Lagrangian results are stored every 0.025 s (40 Hz) and are linearly interpolated to match the internal time step of the CFD simulation. The same outlet relaxation zone (damping zone) used in the Direct Method is used in the Coupling Method to minimise wave reflection and absorb outgoing mass fluxes.

In contrast to the Direct Method, all iterations for the Coupling Method are conducted in the Lagrangian wave flume therefore allowing for a shorter CFD wave flume. The layout of the computational domain is also shown in Figure 6. Compared with the Direct Method, the distance between the (truncated) inlet boundary and the vertical cylinder is now 3 m smaller with  $L_0 = 5$  m, reducing total length of the numerical flume from 13.7 m to 10 m. More details about the CFD domains are summarised in Table 1, where it is also seen that the Coupling Method is in total (including the time required for the iterations) approximately 1.5 times faster than the Direct Method despite the fact that 3-D simulations with the former method are found to require more computational time than simulations with the latter method. This increase in computational time is attributed to the additional time required for the communications between the externally provided inlet boundary conditions and the OpenFOAM model. In particular, small fluctuations in inlet boundary conditions require a smaller time step to ensure the stability of the simulations.

### 3.4 Validation of wave-current generation methods

The computational results with both modelling approaches for wave-current interactions without the structure in place are now validated against experimental measurements. Free surface elevation time

histories at  $x = -4.7$  m (amplitude matching position) and at  $x = 0$  m (focus point) with following and adverse sheared currents and without a current are presented in Figure 7. The outputs of the Lagrangian numerical model are also included and are referred to as LaNM. An overall good agreement between experimental results and results from both the Direct and Coupling Methods is observed, with slightly larger differences being found for the Direct Method. As discussed previously, wave-current generation is different between the two numerical methods (Direct and Coupling Methods) and between numerical methods and experiments, and thus the generation of different spurious waves is expected. This explains the main differences between the methods and between calculations and experiments. The generation of spurious long waves will be discussed in more detail in the following section.



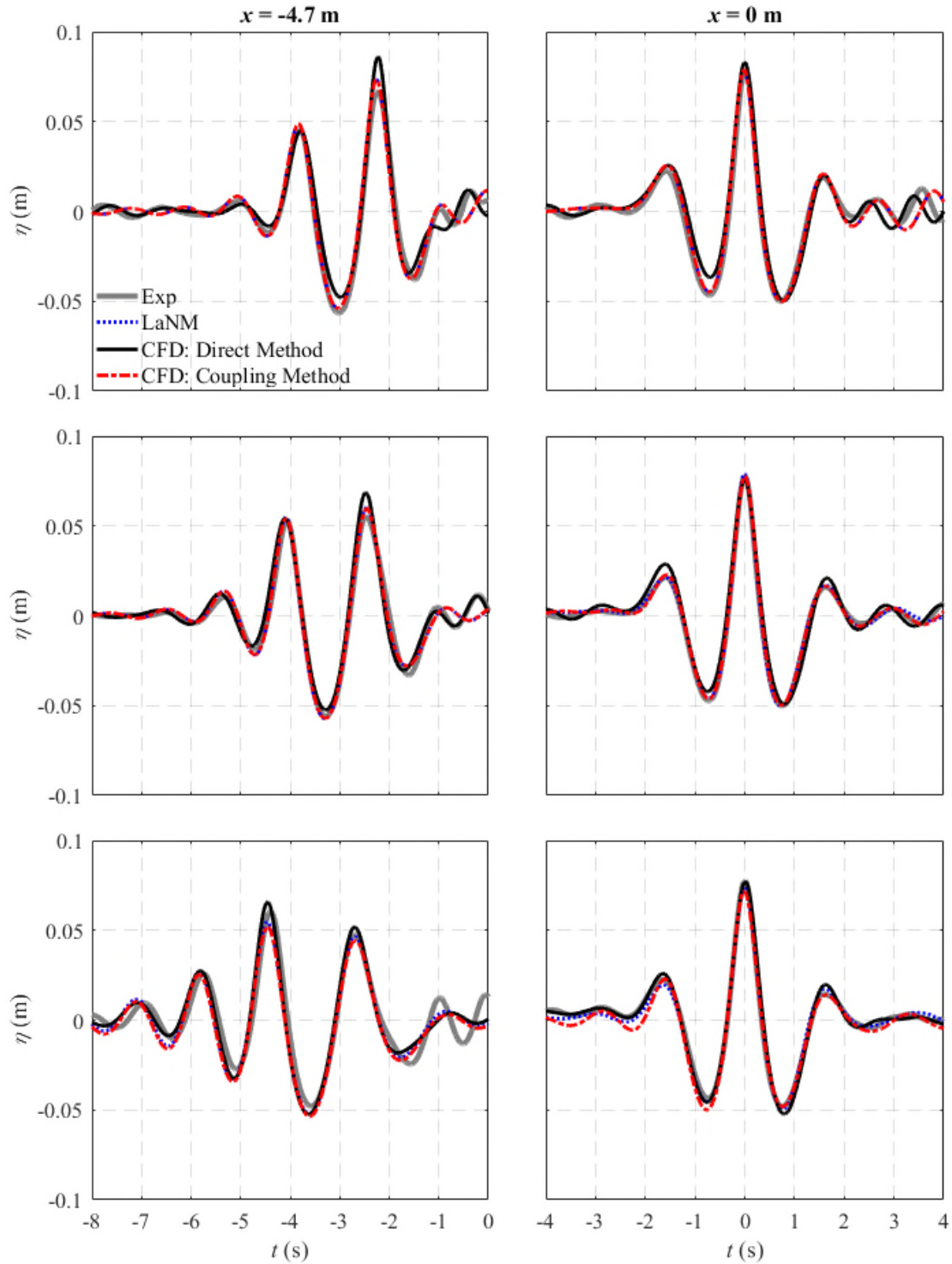


Figure 7 Comparisons of the free surface elevation time histories for cases with and without a sheared current. Left:  $x = -4.7$  m (amplitude matching position); Right:  $x = 0$  m (phase focus position). From top to bottom: following current, no current, and adverse current. All the results presented consider cases without the structure in place.

It is clear from Figure 7 that the wave shapes at the phase focussing position (right panels) for the cases with and without sheared currents are similar to each other as a result of the carefully controlled wave generation. The linearized spectrum at the focus point is the same for the experiments and computations for all the current cases considered (following, no-current and adverse current), while nonlinear contributions from higher order harmonics of the focussed wave group for different current

cases are rather different. This leads to the differences in the main focused wave crests and the following crests at the focussed position. Further analysis on the harmonic structure of the free surface elevation will be presented in the following section. Additionally, it can be seen that the wave shapes at the amplitude matching position (left panels) are rather different from each other for all the current cases considered. This is due to the fact that the dispersion relations are different for waves on following, zero and adverse currents.

Flow kinematics computed at and below the peak of the main wave crest at focus are compared with PIV measurements in Figure 8. It is worth noting that the bottom of the wave flume can either be modelled as a no-slip or a free-slip wall in CFD models. The use of a no-slip wall ensures zero normal and tangential velocities at the boundaries, while for free-slip conditions, the boundary velocity gradient is zero. The results calculated using free-slip boundary conditions are shown in Figure 8 for the sake of convenience for comparison to the Lagrangian computations. The CFD results calculated using no-slip boundary conditions are shown in Appendix 2. It can be seen that there is essentially no difference in terms of flow kinematics outside the boundary layer and thus the forces on the cylinder do not depend on the bed or side wall conditions.

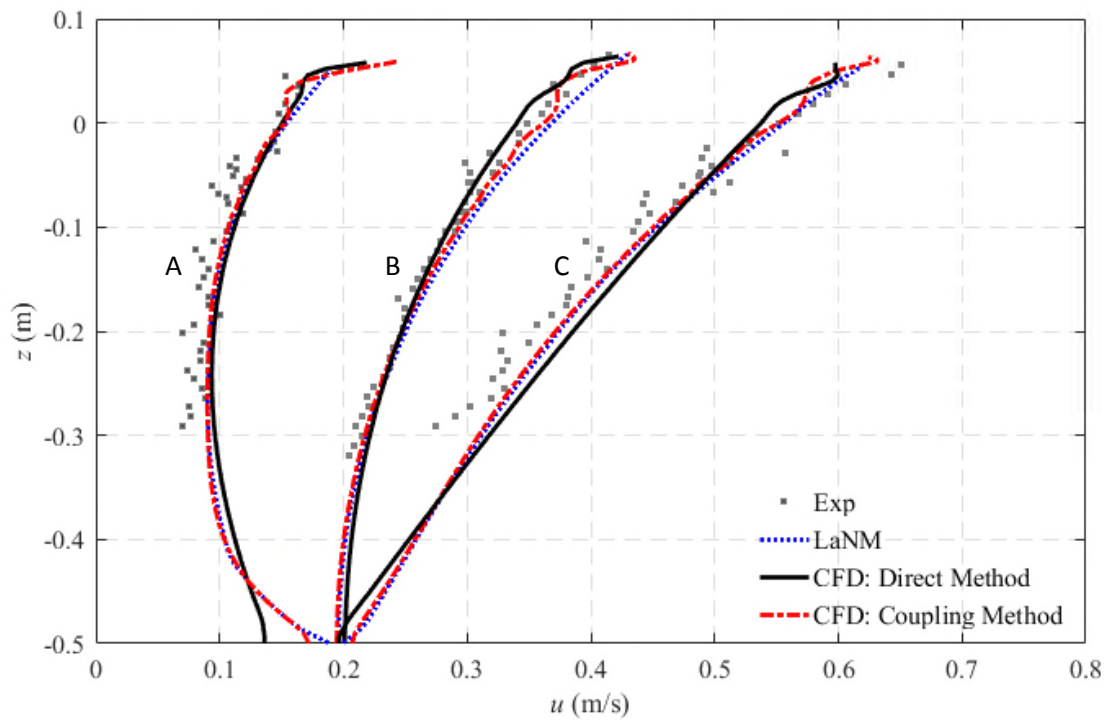


Figure 8 Velocity profiles under the wave crest for focussed wave groups for all three cases considered. Numerical calculations and experimental measurements are included. A -- Adverse current; B -- no current; C -- following current.

It can be seen from Figure 8 that the two CFD models (Direct and Coupling Methods) and the Lagrangian model all provide very good predictions for the flow kinematics below still water level (0 m). The largest discrepancies between the two CFD models (Direct and Coupling Methods), the Lagrangian and experimental measurements are seen to occur in the vicinity of wave crests. This difference in the velocity profile is partly caused by the inaccuracy of the numerical velocity profile (Figure 5). The limitation of the VOF method in reconstructing very steep and sharp free-surfaces is also responsible for this difference around the interface (Wroniszewski et al., 2014). There is a sharp discontinuity of density at the interface, and the density-weighted velocity of air using the VOF factor

$\alpha$  above the interface is close to zero, the velocity across the small interface between air and water is smeared accordingly.

#### 4. Wave-current-structure interactions

Wave-current input conditions generated by the Direct Method and the Lagrangian model are now used to simulate the wave-current-structure interaction using CFD-based models. We consider six cases, including waves on following and adverse currents and without a current interacting with cylinders of two diameters  $D = 0.25$  m and  $D = 0.165$  m.

Comparisons between computed and measured time histories of the horizontal load on the cylinder and the free surface elevation at the front of the cylinder are presented in Figures 9 and 10. Results for maximum free surface elevation and peak forces are summarized in Table A2 of Appendix 3.

Considering the cases with the larger cylinder (Figure 9), the time histories of the non-linear elevation and horizontal force are predicted sufficiently well by either of the two approaches and the differences are observed mostly in the amplitude of the first and the main crests which are also illustrated in Figure 8. The peak free surface elevation and horizontal force are generally very slightly under-predicted by both approaches.

An equally good comparison between experimental and numerical results is reported in Figure 10 for the cases with the smaller cylinder. Differences in computed elevations are relatively larger than those for the larger cylinder, but differences in peak force predictions are as small as those for the larger cylinder. In all six cases considered the highest discrepancies between experimental and numerical force results are seen for the cases with adverse currents and in particular for the smaller cylinder. Computational results presented so far demonstrate a sufficient capacity of both CFD approaches (Direct and Coupling Methods) to model wave-current-structure interactions.

In the same time, CFD model cross-comparisons, by referring to the predicted elevation and force profiles, Figures 8 and 9, and the peak elevation and force, Table A2, show a good agreement and neither of the two approaches appears to be clearly superior to the other. Nevertheless, to further explore the source of the small differences observed between computations and between both computations and measurements, the fully non-linear elevation and force time histories are decomposed into their linear and non-linear components using the methodology described in Section 2.2.

The decomposed spectrum and the inverse Fourier transformation of each spectral part (e.g. time histories of the 2<sup>nd</sup> order difference, linearized, 2<sup>nd</sup> order sum parts etc.) are shown in Figures 11-14. The root mean square error for each spectral part is calculated as:

$$\text{RMSE} = \sqrt{\frac{\sum_{i=1}^N (a_{pi} - a_{mi})^2}{N}} \quad (16)$$

where  $a_{pi}$  and  $a_{mi}$  are the spectral amplitudes of the  $i^{\text{th}}$  ( $i = 0, 1, 2, 3$ ) frequency predicted by the computations, and measured in the experiments, respectively.  $N$  is the number of frequencies considered in the calculations of the RMS error.  $N$  varies from 256 to 80, being larger for the linearized part and decreasing for the nonlinear part. The range of frequencies considered for calculating the RMSE is  $0 < f/f_p < 1$  for the 2<sup>nd</sup> order difference part ( $S_0$ ),  $0 < f/f_p < 3$  for the linearized part ( $S_1$ ),  $1 < f/f_p < 3$  for the 2<sup>nd</sup> order sum part ( $S_2$ ), and  $1.5 < f/f_p < 3.5$  for the third order part ( $S_3$ ). It is noted that each frequency range was selected to include frequency components with non-negligible energy. As such, the integral spectral error calculated with equation (16) is used as an integral measure to evaluate the level of agreement between experimental and numerical results. The RMS errors for both methods and for all test cases are shown in Figures 11 to 14 and they are summarized in Table A3 of the Appendix 3.

Considering the 2<sup>nd</sup> order difference harmonics, discrepancies are seen in the inverse Fourier time histories of surface elevation and horizontal force on both cylinders. Given that waves are generated linearly in the physical flume the occurrence of, e.g., the 2<sup>nd</sup> order spurious wave crests at approximately  $-3 \text{ s} < t < 0 \text{ s}$  in Figures 11 and 13 is not surprising. It is also worth noting that the same methodology (Section 2.2) was used to reproduce the experimental results in the Coupling Method and the Direct Method. As such, the presence of spurious wave crests in the numerical results is also not surprising. The Coupling Method is seen to somehow reproduce more closely 2<sup>nd</sup> order difference harmonics with the experimental results, especially for the tests without currents. Given the variability in wave generation methods between the flumes, and since the 2<sup>nd</sup> order wave generation is not employed, the differences in the elevation of the 2<sup>nd</sup> order difference harmonic are expected.

The best agreement between experimental and numerical results is observed for the linearized part of the spectra. This is an expected outcome since with the iterative methodology the computations are forced to match the linearized part extracted from the experimental spectrum. However, it is illustrated by the time histories in Figures 11 to 14 and the RMS errors in Table A3, the Coupling Method is more efficient in reproducing the experimental results.

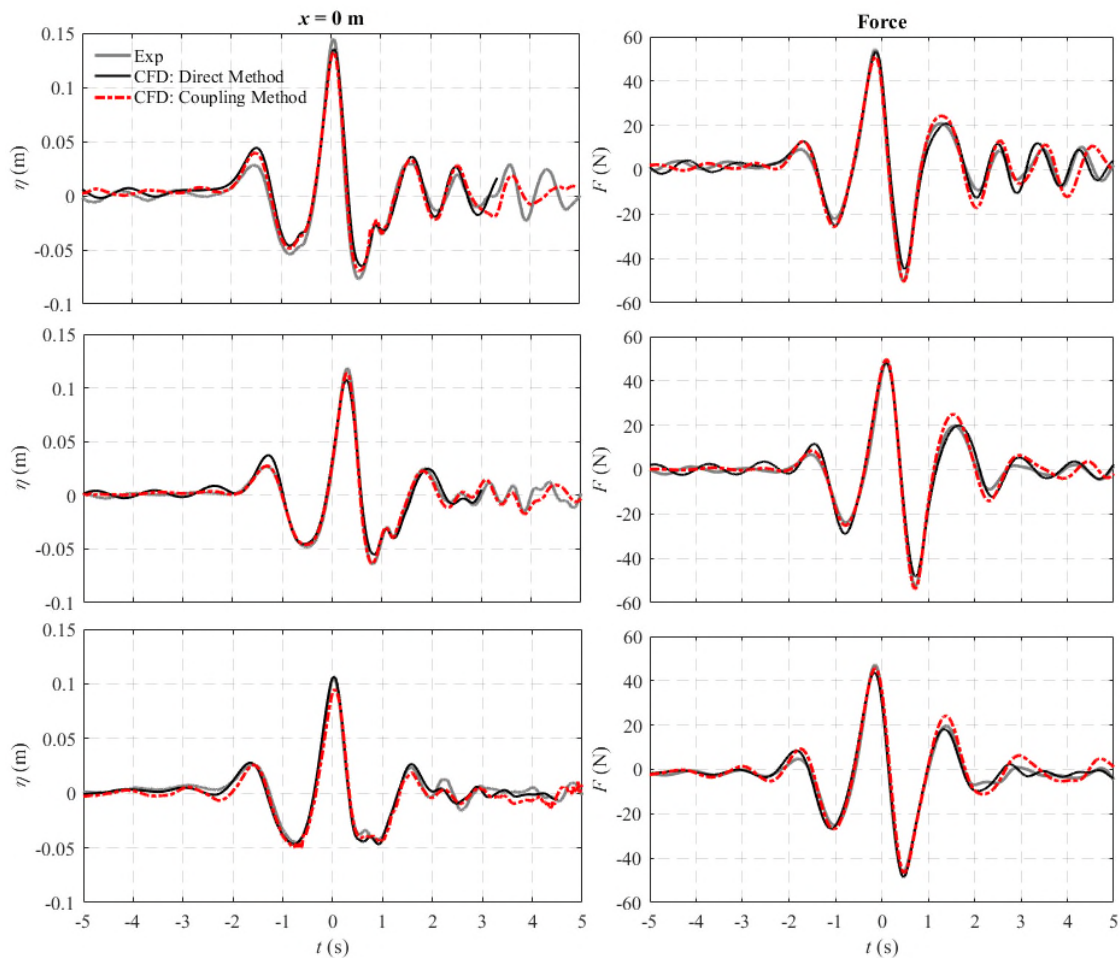


Figure 9 Comparisons of the free surface elevation time histories at the front of the cylinder (left) and the horizontal forces on the cylinder (right) for the larger cylinder ( $D = 0.25 \text{ m}$ ). From top to bottom: following current; no current; adverse current.

In contrast to spurious long waves (2<sup>nd</sup> order difference harmonic), spurious short-wave components (2<sup>nd</sup> order sum harmonic) travel with a celerity smaller than that of the wave group and thus they arrive at and interact with the structure after the focused wave. As a result, the agreement between

experimental (elevation and force) measurements and computations for the 2<sup>nd</sup> order sum harmonics improves, see for example  $S_2$  for  $-1 \text{ s} < t < 1 \text{ s}$  in Figures 11 to 14. Particularly, for tests with the smaller cylinder, the RMS error for the forces predicted by the Coupling Method is smaller but once again the difference with the errors calculated for the Direct Method is not significant.

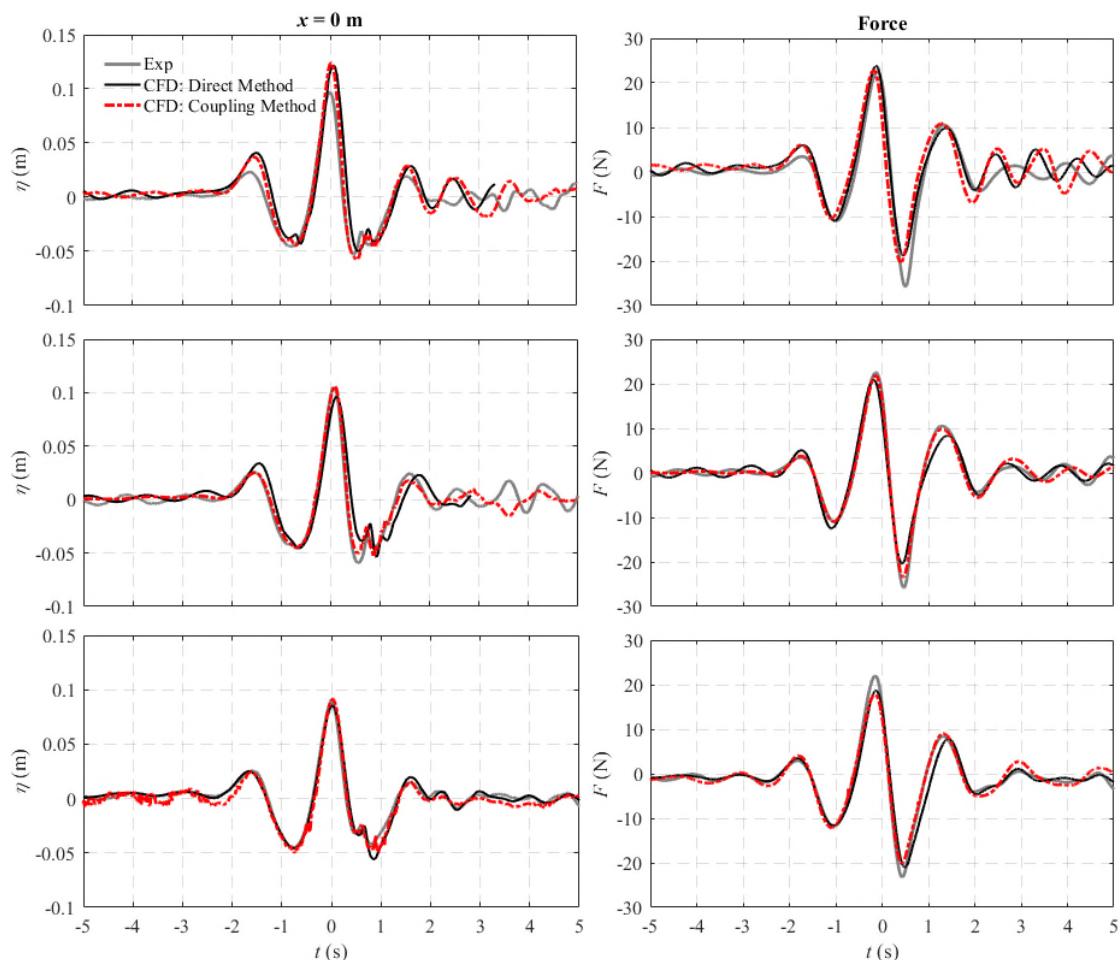


Figure 10 Comparisons of the free surface elevation time histories at the front of the cylinder (left) and the horizontal forces on the cylinder (right) for the smaller cylinder ( $D = 0.165 \text{ m}$ ). From top to bottom: following current; no current; adverse current.

Similar conclusions are drawn from Figures 11 to 14 and Table A3 about the 3<sup>rd</sup> order sum harmonics albeit the agreement between the 3<sup>rd</sup> order horizontal forces is not as impressive as the agreement between experimental and numerical free surface elevation. Although the combination of four phase shifted elevation/force signals is sufficient to efficiently isolate the 3<sup>rd</sup> order harmonics (Buldakov et al., 2017), the very small amplitude of the 3<sup>rd</sup> order harmonics challenges the accuracy limits of experimental measurements. With this in mind, the performance of both Methods is considered to be satisfactory with the Coupling Method results being slightly closer to the experiments. Despite the small amplitudes, the 3<sup>rd</sup> order force harmonics are still important since they are often related to the ‘ringing’ phenomenon.

With regards to the inter-comparison of the two numerical approaches, Figures 11 to 14 reveal no significant differences and neither model is seen to outperform the other. The small differences in the performance of the Coupling and the Direct Method are likely due to the fact that the Lagrangian model reconstructs the experimental input conditions with slightly higher precision; see also Figure 7.

Small discrepancies in the 2<sup>nd</sup> order difference components can be attributed to the different wave generation methods adopted, but they are not seen to result in significant discrepancies in the overall computations of free surface elevation and force time series, e.g. Figures 9 and 10. In general, RMS errors for the Coupling Method tend to be smaller than those for the Direct Method. This in combination with the smaller computational effort required (Table 1) shows an advantage in favour of the Coupling Method.

## 5. Force decomposition

We have demonstrated in Figures 12 and 14 that the harmonic structure of forces on a cylinder in waves and sheared currents can be accurately decomposed into harmonic contributions using the four-phase based decomposition method in Section 2.2. Chen et al. (2018) showed that the harmonic structure of force on a vertical cylinder in a wave group without current can be adequately modelled based on only the linear component as follows. We write the linear component in time as

$$F_1 = F_1 f_1 \quad (17)$$

where  $F_1$  is the peak of the envelope of  $F_1$  in time and  $f_1$  carries all the phase information and group structure in time. Then the assumed form of the total force in time is

$$\begin{aligned} \frac{F}{\rho g R^3} = & \frac{F_1}{\rho g R^3} [f_1] + S_{FF2} \left( \frac{F_1}{\rho g R^3} \right)^2 [\alpha_{FF2} (f_1^2 - f_{1H}^2) + \beta_{FF2} (2f_1 f_{1H})] \\ & + S_{FF3} \left( \frac{F_1}{\rho g R^3} \right)^3 [\alpha_{FF3} f_1 (f_1^2 - 2f_{1H}^2) + \beta_{FF3} (f_{1H} (3f_1^2 - f_{1H}^2))] \\ & + S_{FF4} \left( \frac{F_1}{\rho g R^3} \right)^4 [\alpha_{FF4} (f_1^4 - 6f_1^2 f_{1H}^2 + f_{1H}^4) + \beta_{FF4} (4f_1 f_{1H} (f_1^2 - f_{1H}^2))] + \dots \end{aligned} \quad (18)$$

The force approximation contains Stokes-like amplitude terms  $F_1 / \rho g R^3$  based on the peak amplitude of the linear force component, non-dimensional force coefficients at each order  $S_{FFn}$  and phase coefficients  $(\alpha_{FFn}, \beta_{FFn})$  with  $\alpha_{FFn}^2 + \beta_{FFn}^2 = 1$ .  $R = D/2$  is the radius of the cylinder. The subscript  $H$  denotes the Hilbert transform of the  $f_1$  function in time, and the increasingly complicated products of  $f_1$  and  $f_{1H}$  denote the shape of the  $n$ th harmonic in time. The coefficients  $S_{FFn}$  and  $\alpha_{FFn}, \beta_{FFn}$  are estimated by weighted fits, as described in Chen et al. (2018). Chen et al. (2018) showed that this approximate form works well for all the harmonics up to the 5<sup>th</sup> but that the 3<sup>rd</sup> harmonic fits are less good.

Here we briefly demonstrate that these decompositions work equally well for forces from waves on sheared currents, and that the form of the current affects the force coefficients  $S_{FFn}$  significantly, but the phase terms  $(\alpha_{FFn}, \beta_{FFn})$  only slightly. The coefficient values are given in Table A4 of Appendix 3.

The reconstructed harmonics up to the 4<sup>th</sup> harmonic are compared to the extracted experimental harmonics in Figure 15 for the larger cylinder, and in Figure 16 for the smaller cylinder. The experimental harmonics are extracted with the four phase decomposition method of Section 2.2 and Fitzgerald et al. (2016) and the 4<sup>th</sup> sum harmonic is separated from the 2<sup>nd</sup> order difference term by digital filtering. It can be seen from the figures that the reconstructions of the 2<sup>nd</sup> and 4<sup>th</sup> harmonics work well, and for both cylinders the amplitudes of the harmonics are largest for the following current and smallest for the adverse current. These bracket the case with no current. The 3<sup>rd</sup> harmonic contributions are fitted less well with significant structure outside the time range of the (linear envelope)<sup>3</sup> as discussed by Chen et al. (2018) for cases without current. That is, obvious wiggles outside the envelopes of 3<sup>rd</sup> harmonics are observed as shown in Figures 15 and 16; the envelopes of 3<sup>rd</sup> harmonics are approximated by raising the linear envelope to the power three, and then scaled to

fit the measured envelopes of the 3<sup>rd</sup> harmonic component by a least-squares method (Chen et al., 2018). Further analysis of the forces and scattered waves is left for a follow-on paper.

## 6. Conclusions

Two approaches are proposed and used in this numerical study to generate nonlinear focussed wave groups propagating on a sheared current so as to allow an investigation of complex interactions between a combined wave-current flow and a vertical surface piercing cylinder, with applications to problems in coastal engineering. Both approaches employ an iterative wave-on-current focussing methodology to ensure controlled wave-current generation. In the first approach, i.e. the Direct Method, the iterative methodology is applied directly in a 2-D OpenFOAM model to provide input conditions for a 3-D OpenFOAM model, while in the second approach, i.e. the Coupling Method, the input wave-current kinematics of the 3-D OpenFOAM model is created in a faster numerical model. In this study, a Lagrangian numerical wave-current flume is used as the fast model for reconstructing experimental surface elevation and kinematics of incoming focussed waves on sheared currents. There is no necessity to have such a long distance between the wavemaker and the structure to ensure a full development of the combined wave-current flow before the complex interactions with the structure. Thus using the Coupling Method allows a smaller 3-D computational domain and shorter simulation time for modeling wave-current-structure interactions when compared to the Direct Method.

It is worth noting that the wave-on-current focussing methodology applied in this study considers only the linearized part of wave group spectrum, and phase and amplitude corrections are performed at different locations to improve the effectiveness and convergence of the iterative procedure; the phases are corrected at the pre-selected focus location, and amplitudes are corrected at a location well before the focus position.

Good agreement between the experimental and numerical results demonstrates that both numerical methods are capable of replicating experimental wave-current flows, and then accurately modelling interactions between surface piercing cylinders and focussing waves on sheared currents. It is found that the Coupling Method is computationally cheaper due to the application of the iterative wave-on-current focusing methodology in the faster Lagrangian model. More specifically, for the simulations considered in this study the computational efficiency is increased by a factor of approximately 1.5. Overall, both approaches can be recommended as practical methods for studies of wave-current interactions with structures, especially the Coupling Method that has a higher computational efficiency. It is worth mentioning that the Lagrangian model can be coupled with various models and solvers, and is thus applicable for a wide range of wave-current-structure interaction problems.

It is also found that the Stokes-wave perturbation expansion of Chen et al. (2018) can be generalized to cylinder loads arising from wave groups on adverse and following currents and without a current. The higher-order harmonic shapes can be estimated from knowledge of the linear components alone, and the actual time history at each harmonic can be reconstructed to a reasonable approximation from the linear component time history, using an amplitude coefficient and a phase angle at each harmonic. The 2<sup>nd</sup> and 4<sup>th</sup> harmonic force coefficients are found to be the largest on a following current, and the smallest on an adverse current. The results for waves without a current sit in between. The 3<sup>rd</sup> harmonic forces fit the simple expansion less well, as observed by Chen et al. (2018) for the case of no current. The application of this reconstruction method to a wide range of wave-current conditions will be considered in future work.

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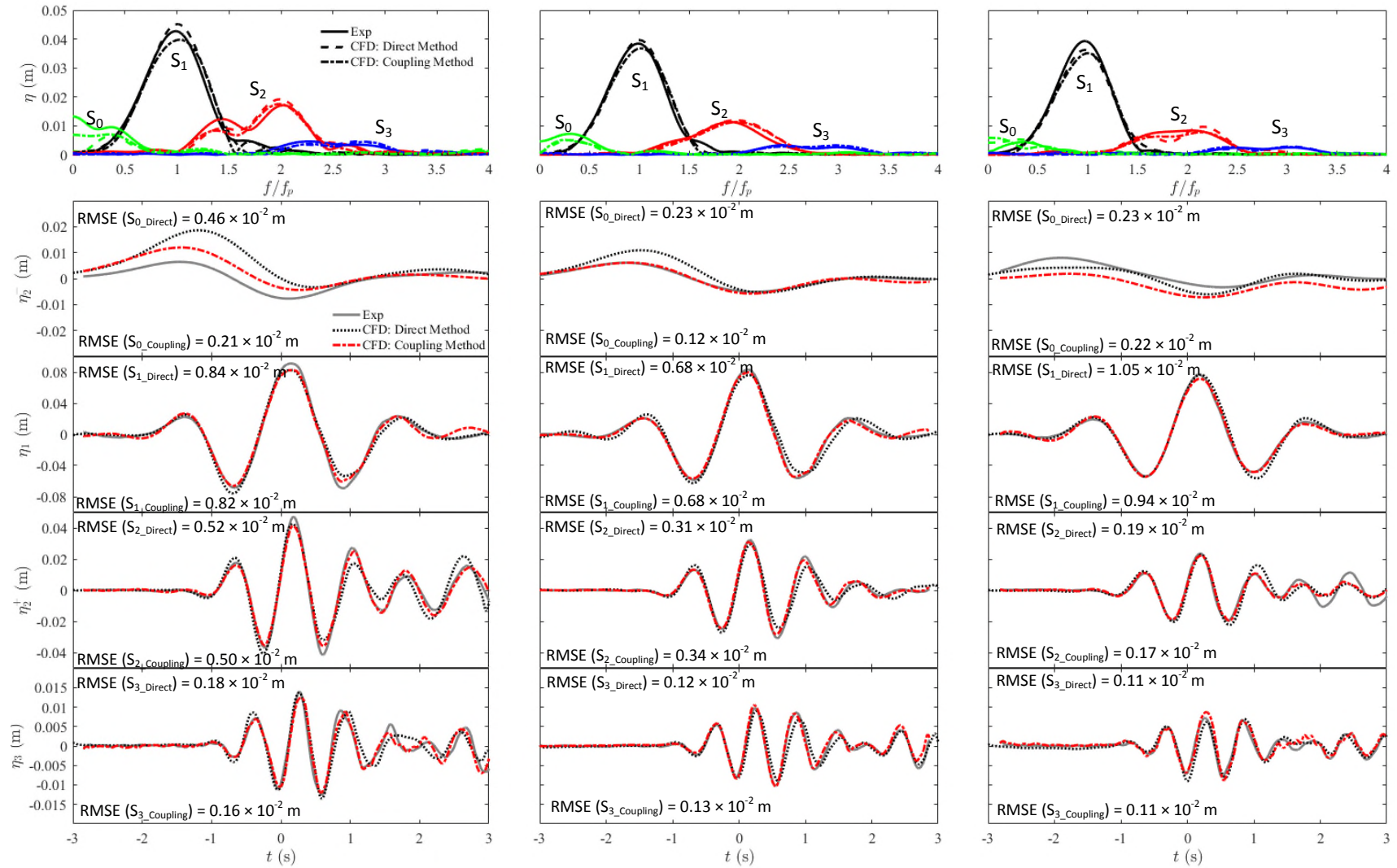


Figure 11 Harmonic components of the free surface elevation at the front face of the larger cylinder ( $D = 0.25$  m). From top to bottom: Amplitude spectra of the free surface elevation, 2<sup>nd</sup> order difference harmonic, linear harmonic, 2<sup>nd</sup> order sum harmonic, and 3<sup>rd</sup> harmonic. From left to right: following current; no current; adverse current.

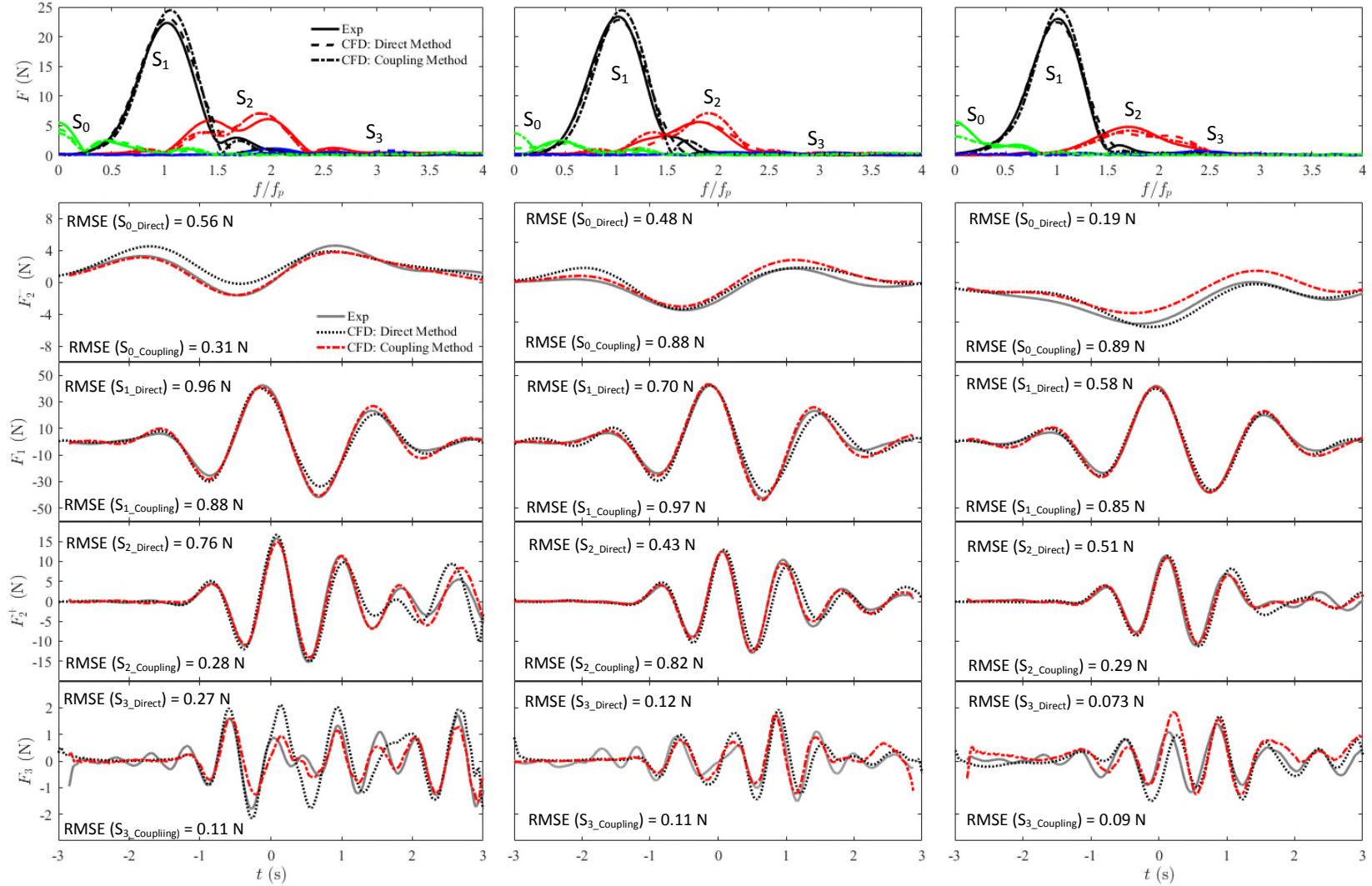


Figure 12 Harmonic components of the wave loading on the larger cylinder ( $D = 0.25$  m). From top to bottom: Amplitude spectra of the force, 2<sup>nd</sup> order difference harmonic, linear harmonic, 2<sup>nd</sup> order sum harmonic, and 3<sup>rd</sup> order harmonic. From left to right: following current; no current; adverse current.

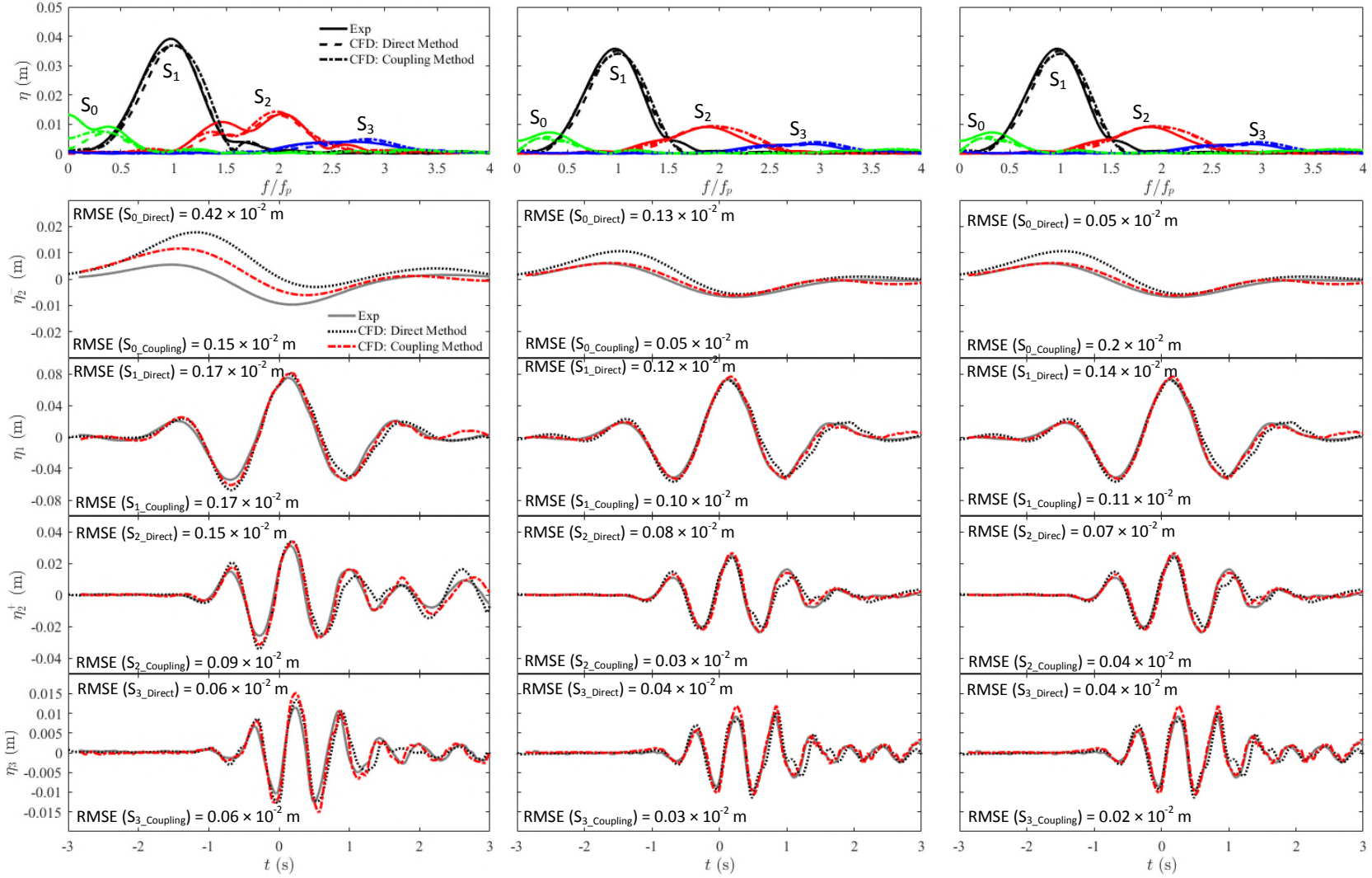


Figure 13 Harmonic components of the free surface elevation at the front face of the smaller cylinder ( $D = 0.165$  m). From top to bottom: Amplitude spectra of the free surface elevation, 2<sup>nd</sup> order difference harmonic, linear harmonic, 2<sup>nd</sup> order sum harmonic, and 3<sup>rd</sup> order harmonic. From left to right: following current; no current; adverse current.

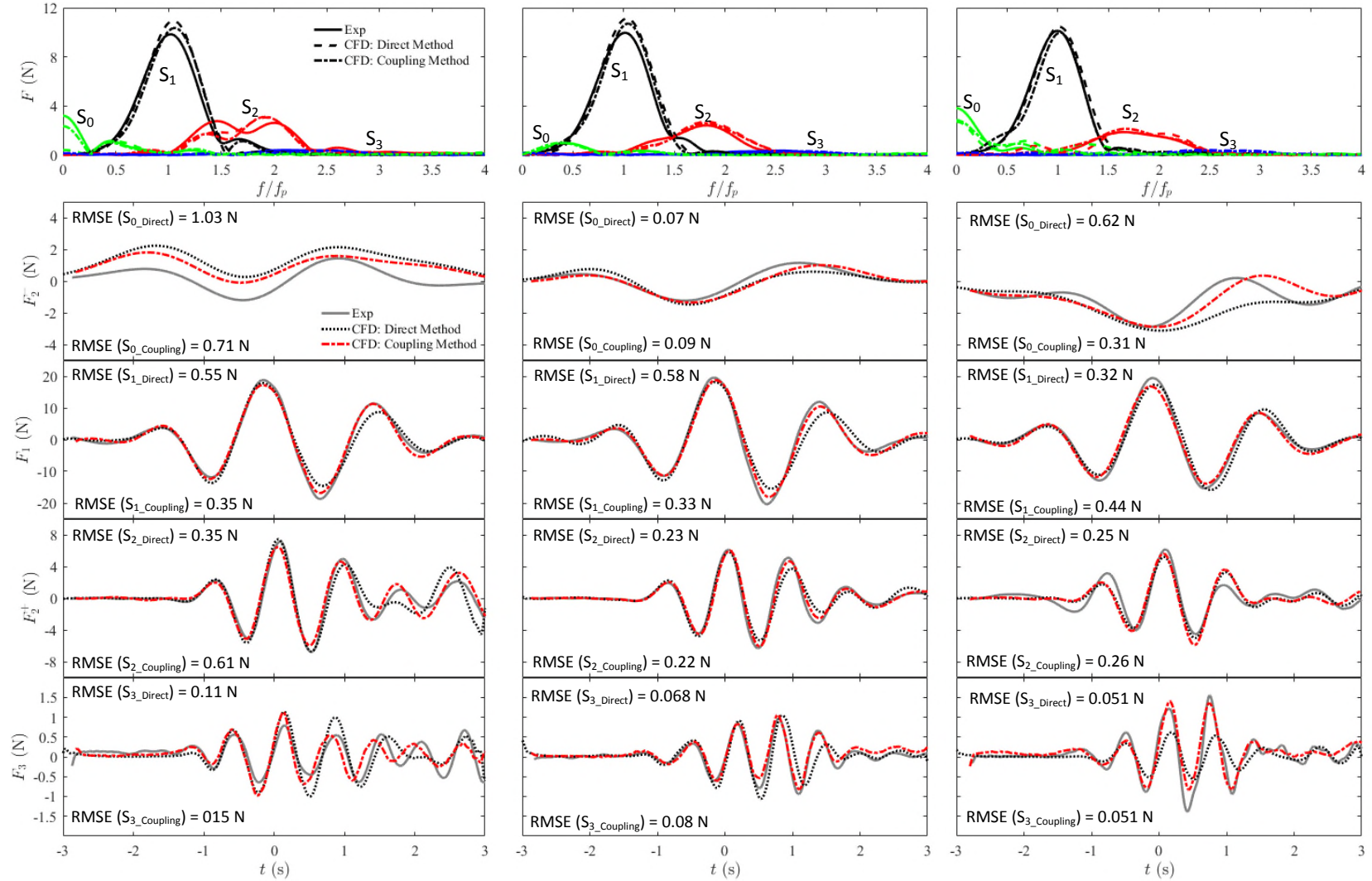


Figure 14 Harmonic components of the wave loading on the smaller cylinder ( $D = 0.165$  m). From top to bottom: Amplitude spectra of the force, 2<sup>nd</sup> order difference harmonic, linear harmonic, 2<sup>nd</sup> order sum harmonic, and 3<sup>rd</sup> order harmonic. From left to right: following current; no current; adverse current.

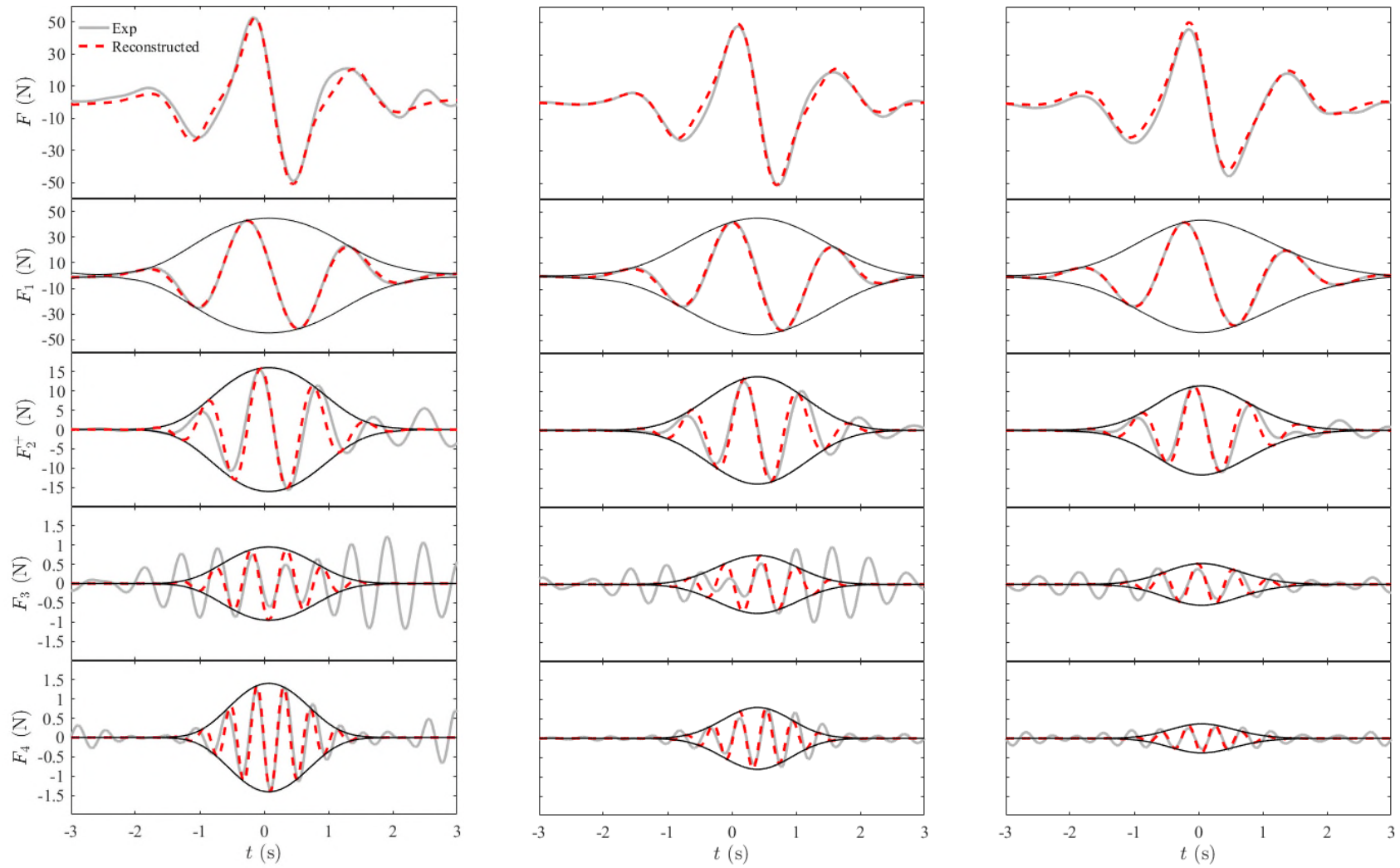


Figure 15 The reconstruction of horizontal wave loading on the larger cylinder ( $D = 0.25$  m). From top to bottom: Total force, linear harmonic, 2<sup>nd</sup> order sum harmonic, 3<sup>rd</sup> order harmonic, and 4<sup>th</sup> order harmonic. From left to right: following current; no current; adverse current.

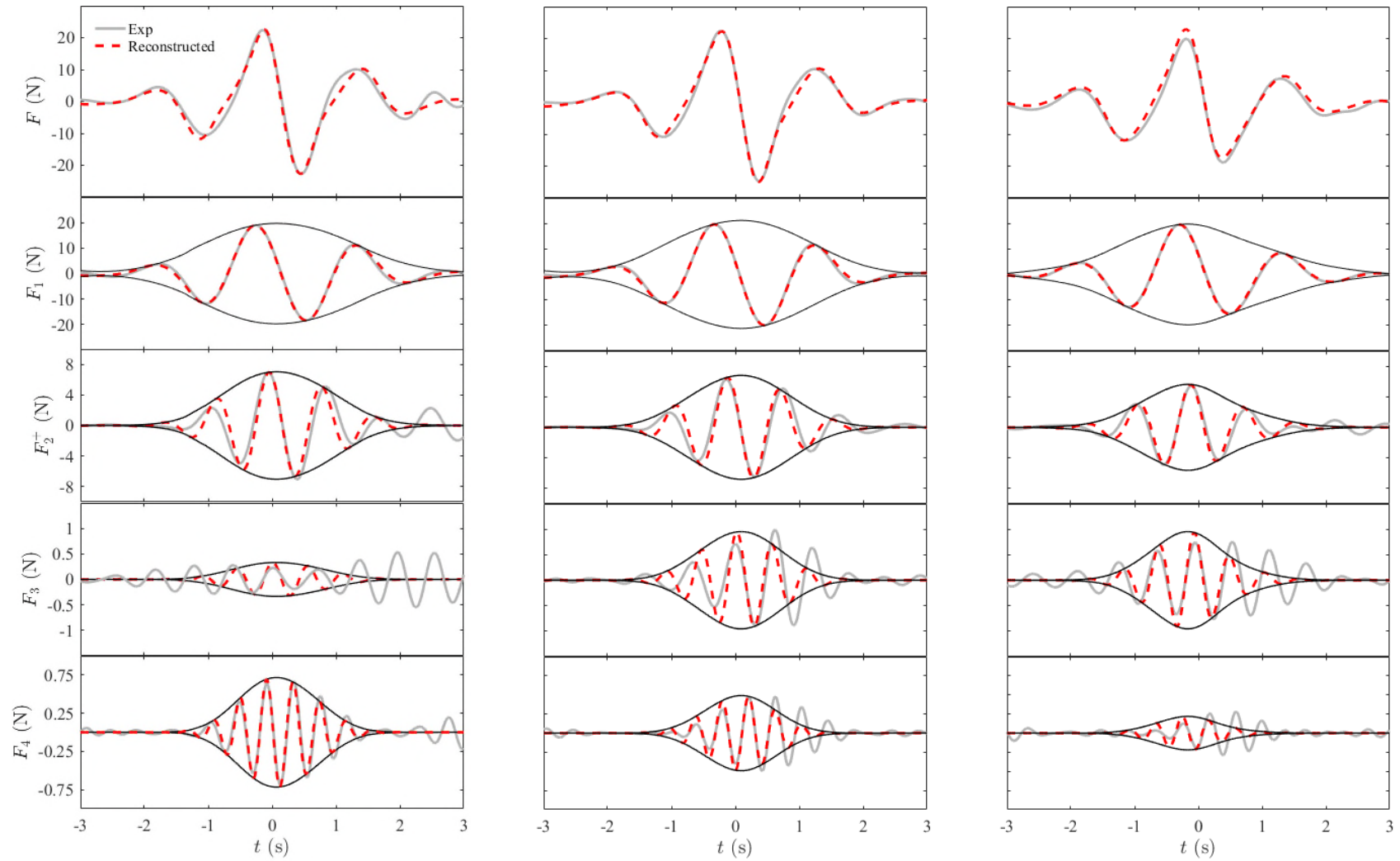


Figure 16 The reconstruction of horizontal wave loading on the smaller cylinder ( $D = 0.165$  m). From top to bottom: Total force, linear harmonic, 2<sup>nd</sup> order sum harmonic, 3<sup>rd</sup> order harmonic and 4<sup>th</sup> order harmonic. From left to right: following current; no current; adverse current.

## Appendix 1. Numerical schemes and solvers

The selected numerical schemes used to discretize the different terms in the governing equations, and the settings for the linear solvers and for the solution algorithm are summarized in Table A1. In OpenFOAM, they can be specified in the *fvSchemes* and *fvSolution* files, respectively.

The treatment of the first order time derivative terms ( $\partial/\partial t$ ) in the momentum equations is specified in the *ddt* scheme. Three transient schemes are widely used for engineering applications including *Euler*, *Backwards* and *CrankNicolson (CN)*. The *Euler* scheme corresponds to the first-order forward Euler scheme, while *Backwards* is a second-order implicit time discretization scheme in which the results from the current and two previous time steps are used. A blending factor is introduced in the *CrankNicolson (CN)* scheme to improve its stability and robustness; the blending factor of 1 corresponds to a pure *CN* scheme with a second-order accuracy, and 0 corresponds to pure *Euler*. The simulations with the *Euler* scheme are faster but may lead to a heavy diffusion of the air-water interface. The use of a *CN* scheme is recommended for waves with long propagation distances and times (Larsen et al., 2018).

One of major challenges in CFD calculations is the treatment of convective/advective terms in the governing equations. Different schemes are specified for different convective terms as they are fundamentally different. The standard finite volume discretization of Gaussian integration is implemented in OpenFOAM in which the integral over a control volume is converted to a surface integral using the Gauss theorem. Accordingly, the word “*Gauss*” is specified in the numerical schemes. The Gaussian integration requires the interpolation of the field variable from cell centres to face centres using for example central/linear or upwind differencing. The former is second-order accurate, but may cause oscillations (unboundedness) in the solution, while the latter is first order accurate, thus, is more diffusive. In lieu of this, various total variation diminishing (TVD) and normalized variable diagram (NVD) schemes that utilize combined upwind and linear differencing are implemented in OpenFOAM, including schemes of *limitedLinear* and *vanLeer*. The use of upwind differencing or linear upwind differencing for the momentum flux is preferable if the loads on the structure are of main concern, such as the cases in this study. A similar conclusion is presented in Larsen et al. (2018).

Generally, the linear schemes are used for calculating the gradients and the interpolation from cell centres to face centres although higher order accurate schemes are available. The *laplacian* scheme requires the specification of an interpolation scheme for e.g. the dynamic viscosity  $\mu$ , and a surface normal gradient scheme for e.g.  $\nabla \mathbf{u}$ . Again, linear schemes are often used with orthogonality corrections for surface normal gradients. For more detailed descriptions on various numerical schemes in OpenFOAM, the reader is referred to the OpenFOAM user’s guide (Greenshields, 2015) and programmer’s guide (Greenshields, 2015) as well as Larsen et al. (2018).

The iterative solvers, solution tolerances and algorithm settings for solving the discretised algebraic equations are specified in the *fvSolution* file. Various iterative solvers are implemented in OpenFOAM, including preconditioned (bi-) conjugate gradient solvers (PCG/PBiCG) and *smoothSolver* in which the specification of preconditioning of matrices (preconditioner) and smoother is required, respectively. The generalised geometric-algebraic multi-grid (GAMG) solver is also commonly used in which the initial guess of the accurate solution on the finer simulation mesh is obtained by mapping the quicker solutions on a coarser mesh to this finer mesh. Generally, the GAMG solver is quicker than the *smoothSolver*, whereas the latter may yield more accurate results. The use of PCG/PBiCG solver sits in between. Detailed descriptions refer to the OpenFOAM user’s and programmer’s guides (2015).

In this study, the compression velocity  $\mathbf{u}_\alpha$  in the equation (5) equals to the flow velocity at the interface by specifying *cAlpha* to be 1. A larger value of *cAlpha* leads to a sharper interface but also

the appearance of wiggles in the air-water interface which is found to be responsible for un-physical steepening of waves and over-estimations of wave celerity (Larsen et al., 2018). Whereas, the use of a smaller  $cAlpha$  reduces the wiggles but at the same time leads to a more significant smearing interface. Another two important controls over the  $\alpha$  equation are  $nAlphaCorr$  and  $nAlphaSubCycles$ ; the former specifies how many times the  $\alpha$  field should be solved within a time step, and the latter represents the number of sub-cycles for the  $\alpha$  equation within a given time step.

As aforementioned, the PISO algorithm is applied in this study, thus,  $nOuterCorrectors = 1$ , and the parameter  $nCorrectors$  is the number of pressure corrector iterations in the PISO loop and the  $momentumPredictor$  is a switch that controls solving of the momentum predictor. Each time step will be begun by solving the momentum equation rather than the pressure equation if the momentum predictor is turned on.

Numerical schemes			
Terms in equations	Representation in OpenFOAM	Discretization schemes	Description
Time derivatives	ddt	Euler	First order forward Euler scheme
Gradients	grad	Gauss linear	--
Divergence (momentum flux)	div(rho*phi, U)	Gauss linearUpwind, grad(U)	Second order, upwind-biased, specification of velocity gradient is required.
Divergence (mass flux)	div(phi, alpha)	Gauss vanLeer	Total variation diminishing (TVD)
Divergence	div (phib, alpha)	Gauss linear	--
Laplacian	laplacian	Gauss linear corrected	Interpolation and snGrad schemes are required.
Interpolation	interpolation	linear	--
Surface normal gradient	snGrad	corrected	Linear with orthogonality correction
Iterative solvers			
Equations	Variable field	Solvers	Left to right are: solver, preconditioner/smoother, tolerance, relative tolerance
Pressure $p^*$	pcorr/p_rgh/ p_rghFinal	PCG, DIG, 1e-5, 0	
Velocity $U$	U	smoothSolvers, symGaussSeidel, 1e-06, 0	
VOF function $\alpha$	alpha.water	smoothSolvers, symGaussSeidel, 1e-08, 0	
Algorithm controls			
Artificial compression term $u_\alpha$	cAlpha	1	$u_\alpha = u$ in which $u$ is the flow velocity at the interface
PISO loop	momentumPredictor	no	Loop starts by solving the pressure equation
PIMPLE loop	nOuterCorrectors	1	PISO is used, otherwise, PIMPLE is used.
PISO loop	nCorrectors	3	pressure corrector iterations
Loop over the $\alpha$ equation	nAlphaCorr	2	$\alpha$ corrector iterations
	nAlphaSubCycles	1	Number of sub-cycles

Table A1 The selected numerical schemes and iterative solvers.



## Appendix 2. Flow kinematics calculated using no- and free-slip boundary conditions

Figure A1 shows flow kinematics computed at and below the peak of the main wave crest at focus in the absence of cylinders using the Direct Method. The bottoms of the wave flumes are modelled by both no- and free-slip walls. The results of the Coupling Method are not shown here for brevity. It can be found that there is essentially no difference in terms of flow kinematics outside the boundary layer and thus the forces on the cylinder do not depend on the bed or side wall conditions.

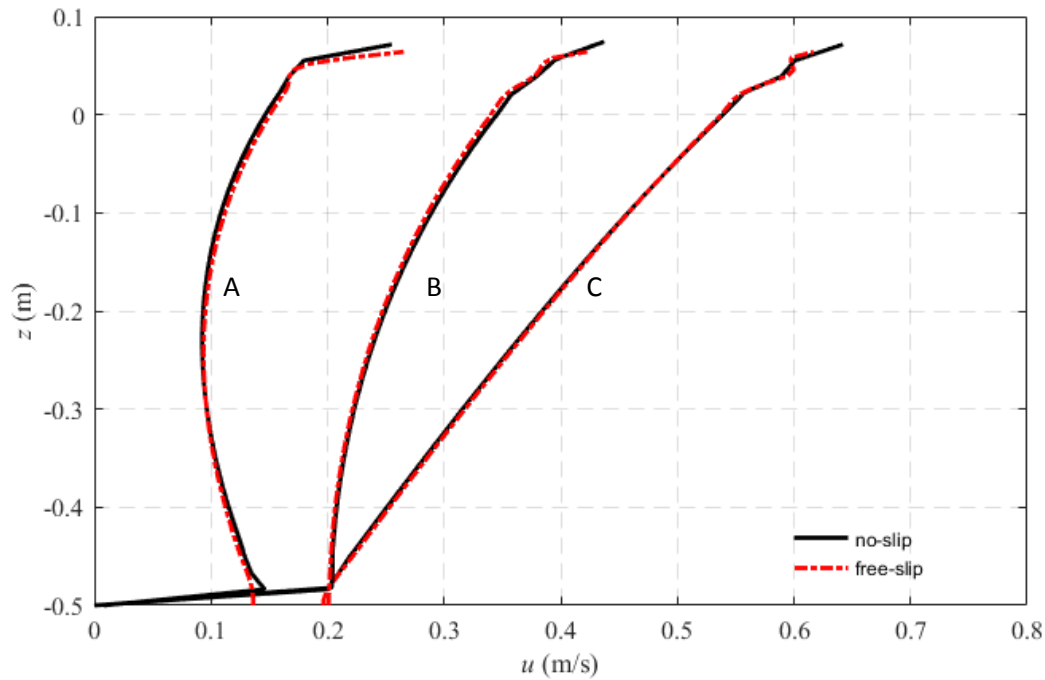


Figure A1 Velocity profiles under the wave crest for focussed wave groups for all three cases considered calculated using the Direct Method with both no- and free-slip boundary conditions. A -- Adverse current; B -- no current; C -- following current.

### Appendix 3. Detailed model comparisons and coefficients used for the reconstruction

Table A2 summarized the results for maximum surface elevation and peak forces, and all differences in Table A2 are calculated with respect to the experimental data and they are only used for a qualitative model comparison. Integral spectra errors are reported in Table A3 and used as an approach demonstrating model accuracy in depth. Table A4 summarized the coefficients used for reconstructing the higher order harmonics from the linear components alone, as shown in Figures 15-16.

Table A2 Comparisons between the two models in terms of wave crests/troughs and peak forces

Cylinders	Cases		Exp.	Direct Method		Coupling Method	
	Parameters	Current (Heading)		Num.	Differences (%)	Num.	Differences (%)
$D = 0.25$ m	Wave crest (m)	Following	0.144	0.135	-6	0.135	-6
		No current	0.118	0.108	-8	0.114	-3
		Adverse	0.107	0.107	0	0.093	-13
	Wave trough (m)	Following	-0.077	-0.065	-16	-0.070	-9
		No current	-0.064	-0.055	-14	-0.061	-5
		Adverse	-0.042	-0.046	10	-0.043	2
	Positive peak forces (N)	Following	54.17	53.07	-2	52.67	-3
		No current	48.10	48.10	0	49.70	3
		Adverse	46.90	43.89	-6	45.46	-3
Negative peak forces (N)	Following	-49.04	-44.75	-9	-48.90	0	
	No current	-51.25	-48.07	-6	-53.63	5	
	Adverse	-46.03	-48.33	5	-44.94	-2	
$D = 0.165$ m	Wave Crest (m)	Following	0.100	0.120	20	0.120	20
		No current	0.105	0.096	-9	0.105	0
		Adverse	0.090	0.085	-6	0.091	1
	Wave trough (m)	Following	-0.053	-0.049	-8	-0.055	4
		No current	-0.059	-0.054	-8	-0.051	-14
		Adverse	-0.040	-0.055	38	-0.047	18
	Positive peak forces (N)	Following	22.51	23.84	6	22.51	0
		No current	22.51	20.93	-7	21.69	-4
		Adverse	22.00	18.28	-17	18.28	-17
	Negative peak forces (N)	Following	-24.53	-18.67	-24	-20.17	-18
		No current	-25.71	-19.66	-24	-22.64	-12
		Adverse	-23.04	-20.9	-9	-19.90	-14

Table A3 Root-mean-square errors for various harmonics in spectra space

Cases			Direct Method				Coupling Method			
Cylinders	Parameters	Current (Heading)	2 <sup>nd</sup> order dif.	Linearized	2 <sup>nd</sup> order sum	3 <sup>rd</sup> order	2 <sup>nd</sup> order dif.	linearized	2 <sup>nd</sup> order sum	3 <sup>rd</sup> order
$D = 0.25$ m	Free surface elevations ( $\times 10^{-2}$ m)	Following	0.46	0.84	0.52	0.18	0.21	0.82	0.50	0.16
		No current	0.23	0.68	0.31	0.12	0.12	0.68	0.34	0.13
		Adverse	0.23	1.05	0.19	0.11	0.22	0.94	0.17	0.11
	Forces (N)	Following	0.56	0.96	0.76	0.27	0.31	0.88	0.28	0.11
		No current	0.48	0.70	0.43	0.12	0.88	0.97	0.82	0.11
		Adverse	0.19	0.58	0.51	0.073	0.89	0.85	0.29	0.09
$D = 0.165$ m	Free surface elevations ( $\times 10^{-2}$ m)	Following	0.42	0.17	0.15	0.06	0.15	0.17	0.09	0.06
		No current	0.13	0.12	0.08	0.04	0.05	0.10	0.03	0.03
		Adverse	0.05	0.14	0.07	0.04	0.2	0.11	0.04	0.02
	Forces (N)	Following	1.03	0.55	0.35	0.11	0.71	0.35	0.61	0.15
		No current	0.07	0.58	0.23	0.068	0.09	0.33	0.22	0.08
		Adverse	0.62	0.32	0.25	0.051	0.31	0.44	0.26	0.051

Table A4 Coefficients for reconstructing the higher order harmonics for all three flow conditions

$kR$	$kh$		Order	Coefficients		
				Following	No current	Adverse
0.242 (larger cylinder)	0.97	Amplitude ( $S_{FFn}$ )	2	3.03	2.58	2.06
			3	0.34	0.27	0.18
			4	0.98	0.54	0.23
		Phase (deg.) ( $\alpha_{FFn}, \beta_{FFn}$ )*	2	97	94	73
			3	49	305	148
			4	145	123	68
0.160 (smaller cylinder)	0.97	Amplitude ( $S_{FFn}$ )	2	2.01	1.89	1.51
			3	0.12	0.34	0.32
			4	0.33	0.22	0.09
		Phase (deg.) ( $\alpha_{FFn}, \beta_{FFn}$ )*	2	99	99	81
			3	183	245	190
			4	165	128	48

\* Phase =  $\arctan(\beta_{FFn} / \alpha_{FFn})$

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