ASLIB

CRANFIELD RESEARCH PROJECT

FACTORS DETERMINING THE PERFORMANCE OF INDEXING SYSTEMS VOLUME 1. DESIGN

by

Cyril Cleverdon, Jack Mills and Michael Keen

Part 2. Appendices

An investigation supported by a grant to ASLIB by the National Science Foundation

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Cranfield 1966

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INTRODUCTION TO APPENDICES

The appendices which follow provide complete information concerning the collection of documents and the set of questions used in the investigation, and are included with the intention that it should be possible for anyone - if they should wish to do so - to repeat the test. The main elements in this are the list of 1,400 documents (Appendix 3C), the list of 279 questions (Appendix 3D), and the complete relevance assessments of all the documents against all the questions (Appendix 3G). Not included is the indexing for each of the documents, since this would have added some seven hundred pages to this volume. However, to anyone interested, copies of the indexing sheets could be made available, either the complete set or any of the subsets which are given in Appendix 3E.. Alternatively, as the indexing has been put on to magnetic tape, a copy of the tape could be supplied.

The later appendices are mainly concerned with various arrangements of the terms used in the indexing of the documents, and illustrate the problems discussed in Chapters 4 and 5 of this volume.

APPENDIX 3A

First letter and enclosures sent to authors of research papers

For some years, with the aid of grants from the National Science Foundation, Washington, work has been undertaken at Cranfield into the efficiency of various techniques for indexing scientific and technical information. This investigation is now moving into a new phase, for which we are seeking the assistance of a number of scientists and engineers, and this letter is addressed to you as the author of a recent research paper.

The basic problem of information retrieval lies in the relationship between the desired recall of relevant documents ("message") and the unwanted recall of non-relevant documents ("noise"). In this respect, we have been able to make rough measurements of the performance of certain existing systems, and in the future work, we shall attempt to refine these measurements and reach a point where systems can be designed to meet given performance and economic requirements.

Before an investigation of this nature can be done, it is necessary to have:

- (a) a collection of documents,
- (b) a number of questions to be used for searching,
- (c) an assessment of the documents in the collection regarding their relevance to each of the search questions.

Briefly, our hope is that we will, with your assistance, be able to meet these requirements in the following way. The collection of documents will consist of the references (apart from books or papers written prior to 1954) contained in approximately 400 recent papers which will be mainly in the field of aerodynamics. We ask you, as author of such a paper, to give us the original problem which

occasioned the research and also some other questions which arose in the course of the work and which might have been responsible for the inclusion of any particular papers in the list of references. We then ask that each reference shall be given an assessment rating for each question.

If you feel that you cannot spare the time to do this, I should be grateful if you would let me know, but I am, of course, hoping that you will be able to assist us and assure you that your help will be most welcome. The preceding paragraph is a general outline of what we wish you to do, but in the enclosed sheets we have given more details and have also included some examples for your guidance. The sheet we ask you to complete and return lists the references in your papers which would be included in the document collection.

As I have said earlier, we hope for and would greatly appreciate your co-operation in helping us to investigate this problem of information retrieval. If it is a subject in which you are interested, I should be pleased to send you a copy of a report that was recently issued on our work so far.

Yours sincerely,

Director, Aslib Cranfield Project.

INSTRUCTIONS FOR ENTERING QUESTIONS AND RELEVANCE ASSESSMENT

On the front of the attached sheet is entered the title and reference of a paper which you have published fairly recently. Would you please enter at (A) the basic problem, as closely as possible in the form of a search question, which was the reason for the research being undertaken. In addition would you please list any other questions (up to three in number) which, in the course of the work, you did put or might have put to an information service.

A number of the references which were given in your paper are listed on the reverse of the attached sheet, and it is assumed that they are all, in varying degrees, relevant to at least one of your questions. We would ask you to mark on this page in the appropriate column your assessment (1, 2, 3, 4 or 5) of the relevance of each reference in relation to each of the questions that you have listed.

This assessment of relevance should be based on the following definitions :-

- Mark as (1) references which are a complete answer to the question. Presumably this would only apply for supplementary questions, i.e. (B), (C) or (D) since if they applied to the main question, there would have been no necessity for the research to be done.
- Mark as (2) references of a high degree of relevance, the lack of which either would have made the research impracticable or would have resulted in a considerable amount of extra work.
- Mark as (3) references which were useful, either as general background to the work or as suggesting methods of tackling certain aspects of the work.
- Mark as (4) references of minimum interest, for example, those that have been included from a historical viewpoint.
- Mark as (5) references of no interest.

It is appreciated that it is not easy to do this objectively, for personal considerations tend to influence judgement. If you find that in certain cases you are uncertain as to which rating should be used, then we suggest that your doubts should be shown by giving a combined rating. An instance of this occurs in the example which we enclose of a sheet that has been completed by Dr. J. F. Clarke. In relation to his basic question A, he decided that references 1, 2, 3 and 5 were considered of no interest, and were therefore marked with a rating of '5'; reference 4 was assessed as falling between ratings '2' and '3' and this is indicated by his marking of '2-3'. For question B references 1 and 2 were rated at '1' while for question C, references 3 and 4 received this rating. Reference 4, as will be seen, was considered by Dr. Clarke to have varying degrees of relevance to questions A, B and C, but was not relevant to question D.

A.73

EXAMPLE

ASLIB CRANFIELD INVESTIGATION INTO PERFORMANCE OF INDEX LANGUAGES

Please complete and return to: C. W. Cleverdon, The College of Aeronautics, Cranfield, Bletchley, Bucks., England.

AUTHOR

J. F. Clarke

TITLE

Reaction-resisted Shock Fronts.

REFERENCE College of Aeronautics Report No. 150, May 1961.

BASIC QUESTION

(A) Has anyone investigated the role of chemical reactions in determining shock wore structure?

Copers on acoustic wave propagation in reacting gases.

Have other types of 'non-viscous' compression weres been investigated analytically or Ibserved experimentally

ls there any simple, but realistic, "model"
gos which can be used to expedite analysis?

LIST OF REFERENCES

	CONTRACTOR STATE CONTRA						
			ASSESSMENT				
Ref. No.		Question					
1.	Clarke, J. F. Flow of chemically	A	В	С	D		
1.	reacting gas mixtures. College of Aeronautics Report 117.	5		5	3		
2.	Clarke, J.F. Linearized flow of a dissociating gas. Jnl. Fluid Mech., 7, 1960, pp 577-595.	5	,	5	5		
3.	Griffith, W.C. and Kenny, A. Jnl. Fluid Mech. 3, 1957, p. 268.	5	5	4	5		
4.	Lighthill, M. J. Survey in Mechanics, Ed. by G.K. Batchelor and R.M. Davies, Cambridge University Press, 1956.	2-3	3	e de la companya de l	5		
5.	Lighthill, M.J. Dynamics of a dissociating gas. Jnl. Fluid Mech., 2, 1957, pp 1-32.	5	5	5	Comp.		
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APPENDIX 3B

Second letter and enclosures sent to authors of research papers

You may recollect that some months ago you were kind enough to help in our investigation into the efficiency of indexing systems. Since that time we have been engaged in indexing the documents in the collection and preparing for the test programme. In this we shall be using the questions which you, together with some two hundred other authors, compiled for us. You may remember that at that time we also asked you to make a relevance assessment against the questions you gave us, of some of the references included in your paper.

There have been external developments concerning our project during the past few months, and this is largely due to the co-operation which we have received from you and others whose assistance we requested. As you may possibly know, major efforts are being made to develop the use of computers for automatic translation and also for 'The latter work has been the automatic indexing of documents. proceeding at various levels but a number of groups have been concerned with what is usually known as associative indexing. Briefly, this involves having a computer analyse the key terms used in a collection of documents, and working out, usually by some statistical technique, the most useful associations of terms to be used in searching. some time the National Science Foundation have been interested in having a collection of documents that can be used as a common corpus by the various groups working in this field, so that a relative assessment can be made of their differing techniques. The collection which we have

compiled has been selected for such a common corpus, with the result that the work which you did for us will be used additionally by groups working at Cambridge and at a number of American Universities or Research Establishments

In addition, there is a further recently developed technique known as Citation Indexing. This depends on building up an index which allows one to find which papers have quoted an earlier paper. For instance, if one wished to find information on the area rule one might look up the early paper by G. Ward on this subject, and find, from the index, references to all the papers which had quoted it, on the assumption that these papers would have some relevance to the original subject. This is an assumption, the validity of which we are testing.

As you may have guessed, this preamble is a lead in to ask whether you would be willing to help us further in two ways. In relation to the questions which you provided, you gave a relevance assessment of some or all of the references in your original paper. It is these references which, with those from similar papers, make up our collection of 1,400 documents. The result is that there may be in the total collection additional papers which are relevant to your questions. This could be either because they appeared after your work was published, because you knew of them but did not list them, because you included them in your list of references but we omitted them from our list, or because they did not come to your attention. a number of post-graduate students analysing every document in the collection against each question, with the intention that we should have a matrix which will show the relevance of each document to each question. My instructions to the students were to list anything which might possibly be relevant, so my first request to you is to ask if you will check these assessments that have been done by the students. The papers concerned are listed on the attached sheet, together with the original questions, and I also include abstracts of the papers to help you decide.

The second matter concerns the questions which you prepared for us. Developments in testing indexing systems during the past few months have shown up, in a way not previously recognised, the importance of the actual search programmes. It is usually found that, within a question, certain aspects are more important than others, and this we would like you to indicate as far as possible by giving a "weighting" to each concept, as described in the attached sheets. Finally,

if you can suggest any related terms which you consider could be used as alternatives in searching, we would be grateful if you would give them.

I have set out these additional tasks in the order of their importance. Whilst we would, of course, be pleased if you could supply the answers to all these enquiries, most important is the relevance assessment, and I shall be very grateful if you can find time to deal with this matter.

Director
Aslib Cranfield Project

ASSESSMENT OF ADDITIONAL DOCUMENTS

The first attached sheet is a copy of the form which you originally completed by entering the questions that arose in the course of your research work. These questions were marked A, B, C, and D. The next sheet lists references to additional documents which we have located in the total collection and consider might be relevant to one or more of your questions. The four-figure file number for each document is followed by a letter or letters and these refer to your questions. It is only in relation to the particular question so indicated that we ask for your assessment of the relevance of the document. To assist you in this task we include abstracts of each paper listed.

The assessment of relevance should be, as before, based on the following definitions:

- Mark as (1) references which are a complete answer to the question.

 Presumably this would only apply for supplementary questions,
 i.e. (B), (C) or (D) since if they applied to the main question,
 there would have been no necessity for the research to be done.
- Mark as (2) references of a high degree of relevance, the lack of which either would have made the research impracticable or would have resulted in a considerable amount of extra work.
- Mark as (3) references which were useful, either as general background to the work or as suggesting methods of tackling certain aspects of the work.
- Mark as (4) references of minimum interest, for example, those that have been included from a historical viewpoint.
- Mark as (5) references of no interest.

WEIGHTING AND NEW TERMS

On the attached sheets are listed the terms as given in your questions. In the box by the side of each term we would ask you to indicate its relative importance by marking 1, 2, or 3. These figures will indicate:

- 1. A paper that did not cover this term would be of no use.
- 2. It is desirable that this term should be covered by the document.
- 3. This is a term which is not absolutely essential to the enquiry.

It may well be that you will rate each term as one; this would not be surprising but we would not wish to influence your decision in this respect.

Over the remainder of the sheet space is left for you to insert any alternative terms that you consider might be used in the search programme. In this case, however, it is more likely that you would wish to show alternative concepts rather than single terms, so the terms have been grouped into concepts. As examples of alternative terms or concepts, in the context of a question dealing with "pressure gradient", it might be reasonable to substitute "pressure distribution" or "pressure variation"; the concept "suddenly heated wall" might be changed to "rapidly cooled solid". Alternatively, the complete question might be rephrased, an example of this being where the original question was "has anyone investigated relaxation effects on gaseous heat transfer to a suddenly heated wall". Rephrased the question becomes "has anyone investigated conditions at the wall behind a plain reflected shock front in a real gas by theoretical analysis". If you would wish to rewrite your question in this way space is available on the sheet.

QUESTION 249 Is it possible to estimate the transonic drag-rise and buffet boundaries of aerofoils without recourse to experiment

Search Terms	Weight	Alternative Terms or Concepts
Transonic	4	Mixed subsonic/supersonic
Drag	2	
Rise	2	
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Buffet	3	Onset of shock induced boundary layer
Boundary	3	separation
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Aerofoil	1	Wing section
- Angel Michigan Andrew (Colon (1997)		
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Estimation	-	
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		- Parameters

Rephrased Question:

Is it possible to obtain a theoretical estimate of the transonic drag-rise and onset of shock-induced boundary layer separation Mach numbers for two-dimensional aerofoils

LIST OF ADDITIONAL DOCUMENTS OF POSSIBLE RELEVANCE

	LIST OF ADDITIONAL DOCOMENTS OF LODGINGS AND			•	
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1313	SPREITER, J.R. On alternative forms for the basic equations of transonic flow theory				
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APPENDIX 3C

LIST OF 1400 DOCUMENTS IN THE TEST COLLECTION

- 1001 ERENCKMAN, M. Experimental investigation of the aerodynamics of a wing in a slipstream.

 J. Ae. Scs. 25, 1958, 324.
- 1002 LI,T.Y. Simple shear flow past a flat plate in an incompressible fluid of small viscosity. J.Ae. Scs. 22, 1955, 651.
- 1003 GLAUERT, M.B. The boundary layer in simple shear flow past a flat plate. J. Ae. Scs. 24, 1957, 848.
- 1004 YEN, K.T. Approximate solutions of the incompressible laminar boundary layer equations for a plate in shear flow. J. Ae. Scs. 22, 1955, 728.
- 1005 WASSERMAN, B. One-dimensional transient heat conduction into a double-layer slab subjected to a linear heat input for a small time internal. J. Ae. Scs. 24, 1957, 924.
- 1006 CAMPBELL, W.F. One-dimensional transient heat flow in a multilayer slab. J. Ae. Sos. 25, 1958, 340.
- 1007 Van DRIEST, E.R. and McCAULEY, W.D. The effect of controlled three-dimensional roughness on boundary layer transition at supersonic speeds. J. Ae. Scs. 27, 1960, 261.
- 1008 KLEBANOFF, P.S. Measurements of the effect of two-dimensional and three-dimensional roughness elements on boundary layer transition. J. Ae. Scs. 22, 1955, 803.
- 1009 KORKEGI, R.H. Transition studies and skin friction measurements on an insulated flat plate at a Mach number of 5.8. J. Ae. Scs. 23, 1956, 97.
- 1010 CHAMBRE, P.L. and SCHAAF, S.A. The theory of the impact tube at low pressure. J. Ae. Scs. <u>15</u>, 1948, 735.
- 1011 NAPOLITANO, L. Similar solutions in compressible laminar free mixing problems.

 J. Ae. Scs. 23, 1956, 389.
- 1012 BISPLINGHOFF, R.L. Some structural and aerelastic considerations of high speed flight.

 J. Ae. Scs. 23, 1956, 289.
- 1013 TSIEN, H.S. Similarity laws for stressing heated wings. J. Ac. Scs. 20, 1953, 1.
- 1014 ASHLEY, H. and ZARTARIAN, G. Piston theory a new aerodynamic tool for the aeroelastician. J. Ae. Scs. 23, 1956, 1109.
- 1015 FUNG, Y.C. On two-dimensional panel flutter. J. Ac. Scs. 25, 1958, 145.
- 1016 MAGER, A. Transformation of the compressible turbulent boundary layer.

 J. Ae. Scs. 25, 1958, 305.
- 1017 TING, L. and LIBEY, P.A. Remarks on the eddy viscosity in compressible mixing flows. J. Ae. Scs. 27, 1960, 797.
- 1018 RHYMING, I.L. The flow field in the diffuser of a radial compressor.

 J. Ae. Scs. 27, 1960, 798.
- 1019 ZAKKAY, V. An investigation of the pressure distribution on conical bodies in hypersonic flows. J. Ae. Scs. 26, 1959, 457.
- 1020 LOVE, E.S. Generalised-Newtonian theory. J. Ae. Scs. 26, 1959, 314.
- 1021 MASLEN, S.H. On heat transfer in slip flow. J. Ae. Scs. 25, 1958, 400.
- 1022 OMAN, R.A. and SCHEUING, R.A. On slip-flow heat transfer to a flat plate. J. Ae. Scs. 26, 1959, 126.
- 1023 SEBAN, R.A. and BOND, R. Skin-friction and heat transfer characteristics of a laminar boundary layer on a cylinder in axial incompressible flow. J. Ae. Scs. 18, 1951, 671.
- 1024 FAY, J. A. and RIDDELL, F.R. Theory of stagnation point heat transfer in dissociated air. J. As. Scs. 25, 1958, 73.
- 1025 LEES, L. and KUBOTA, T. Inviscid hypersonic flow over blunt-nosed slender bodies.

 J. Ae. Scs. 24, 1957, 195.
- 1026 CHENG, H.K. and PALLONE, A.J. Inviscid leading-edge effect in hypersonic flow. J. Ae. Scs. 23, 1956, 700.
- 1027 COLE, J.D. Newtonian flow theory for slender bodies. J. Ae. Scs. 24, 1957, 448.
- 1028 FREEMAN, N.C. A note on the explosion solution of Sedov with application to the Newtonian theory of unsteady hypersonic flow. J. Ae. Scs. 27, 1960, 77.
- 1029 ISAKSON, G. A simple model study of transient temperature and thermal stress distribution due to aerodynamic heating. J. Ae. Scs. 24, 1957, 611.

- 1030 GERARD, G. and TRAMPOSCH, H. Photo-thermoelastic investigation of transient thermal stresses in a multiweb wing structure. J. Ae. Scs. 26, 1959, 783.
- 1031 HOFF, N.J. Thermal buckling of supersonic wing panels. J. Ae. Scs. 23, 1956, 1019.
- 1032 FRIEDRICH, H.R. and DORE, F.J. The dynamic motion of a missile descending through the atmosphere. J. Ae. Scs. 22, 1955, 628.
- atmosphere. J. Ae. Scs. 22, 1955, 628.

 1033 RESLER, E.J. and SEARS, W.R. The prospects for magneto-aerodynamics.

 J. Ae. Scs. 25, 1958, 235.
- 1034 KERREEROCK, J.P. and MARBLE, F.E. Constant-temperature magneto-gasdynamic channel flow. J. Ae. Scs. 27, 1960, 78.
- 1035 LI,T.Y. and GEIGER,R.E. Stagnation point of a blunt body in hypersonic flow. J. Ae. Scs. 24, 1957, 25.
- 1036 SERBIN, H. Supersonic flow around blunt bodies. J. Ae. Scs. 25, 1958, 58.
- 1037 FERRI, A. and LIBBY, P.A. A new technique for investigating heat transfer and surface phenomena under hypersonic flow conditions. J. Ae. Scs. 24, 1957, 464.
- 1038 SINNOTT, C.S. On the prediction of mixed subsonic/supersonic pressure distributions.

 J. Ae. Scs. 27, 1960, 767.
- 1039 SINNOTT, C.S. On the flow of a sonic stream past an airfoil surface. J.Ae.Scs. 26, 1959, 169.
- 1040 Van DRIEST, E.R. and BOISON, J.C. Experiments on boundary layer transition at super-'sonic speeds. J. Ae. Scs. 24, 1957, 885.
- 1041 MCRKOVIN, M.V. On transition experiments at moderate supersonic speeds. J. Ae. Scs. 24, 1957, 480.
- 1042 SCANLAN, R.H. and TRUMAN, J.C. The gyroscopic effect of a rigid rotating propeller on engine and wing vibration modes. J. Ae. Scs. 17, 1950, 653.
- 1043 POTTER, J.L. and WHITFIELD, J.D. The relation between wall temperature and the effect of roughness on boundary layer transition. J. Ae. Scs. 28, 1961, 663.
- 1044 BERTRAM, M.H. Tip-bluntness effects on cone pressures at M=6.85. J.Ae.Scs. 23, 1956,898.
- 1045 CHARVAT, A.F. An investigation of separated flows, Part II: Flow in the ca vity and heat transfer. J. Ae. Scs. 28, 1961, 513.
- 1046 KLEIN, B. Some comments on the inversion of certain large matrices.

 J. Ae. Scs. 28, 1961, 432.
- 1047 SAMSON, S.H. and BERGMANN, H.W. Analysis of low-aspect-ratio aircraft structures. J. Ae. Scs. 27, 1960, 679.
- 1048 WILLETT, J.E. Supersonic flow at the surface of a circular cone at angle of attack. J. Ac. Scs. 27, 1960, 907.
- 1049 CHAPMAN, D. and RUBESIN, M. Temperature and velocity profiles in the compressible laminar boundary layer with arbitrary distribution of surface temperature.

 J. Ae. Scs. 16, 1949, 547.
- 1050 Van DRIEST, E.R. Investigation of laminar boundary layer in compressible fluids using the Crocco method. NACA TN.2597, 1952.
- 1051 O'SULLIVAN, W.J. Theory of aircraft structural models subjected to aerodynamic heating and external loads. NACA TN.4115, 1957.
- 1052 MORGAN, H.G. Procedure for calculating flutter at high supersonic speed including camber deflections, and comparison with experimental results. NACA TN.4335, 1958.
- 1053 LARSON, H.K. and KEATING, S.J. Transition Reynolds numbers of separated flows at supersonic speeds. NASA TN. D349, 1960.
- MORDUCHOW, M. and CLARKE, J.H. Method for calculation of compressible laminar boundary layer characteristics in axial pressure gradient with zero heat transfer.

 NACA TN.2784, 1952.
- 1055 MORDUCHOW, M. and GRAPE, R.G. Separation, stability and other properties of compressible laminar boundary layer with pressure gradient and heat transfer. NACA TN. 3296, 1955.
- 1056 EHRET, D.M. An analysis of the applicability of the hypersonic similarity law to the study of the flow about bodies of revolution at zero angle of attack.

 NACA TN. 2250, 1950.

- 1057 ROSSOW, V.J. Applicability of the hypersonic similarity rule to pressure distributions which include the effects of rotation for bodies of revolution at zero angle of attack. NACA TN.2399, 1951.
- 1058 AMICK, J.L. Pressure measurements on sharp and blunt 5° and 15° half-angle cones at Mach number 3.86 and angles of attack to 100°. NASA TN.D753, 1961.
- 1059 BROWN, W.D. and DONOUGHE, P.L. Tables of exact laminar-boundary layer solutions when the wall is porous and fluid properties are variable. NACA TN.2479, 1951.
- 1060 BOBBITT, P.J. and MALVESTUTO, F.S. Estimation forces and moments due to rolling for several slender tail configurations at supersonic speeds. NACA TN. 2955, 1953.
- 1061 ROSSOW, V.J. On flow of electrically conducting fluids over a flat plate in the presence of a transverse magnetic field. NACA TN. 3971, 1957.
- 1062 COHEN, C.B. and RESHOTKO, E. Similar solutions for the compressible laminar boundary layer with heat transfer and pressure gradient. NACA TN. 3325, 1955.
- 1063 TALBOT, L. Hypersonic viscous flow over slender cones. NACA TN. 4327, 1958.
- 1064 MOORE, F.K. Unsteady oblique interaction of a shock wave with plane disturbances. NACA TN. 2879, 1953.
- 1065 RIENER, H.S. Convection of a pattern of vorticity through a shock wave. NACA TN. 2864, 1953.
- 1066 GRIFFITH, G.E. and MILTONBERGER, G.H. Some effects of joint conductivity on the temperature and thermal stresses in aerodynamically heated skin-stiffener combinations. NACA TN. 3609, 1956.
- 1067 TOBAK and ALLEN. Dynamic stability of vehicles traversing ascending or descending paths through the atmosphere. NACA TN.4275, 1958.
- 1068 LOVE, E.S. Some aspects of air-helium simulation and hypersonic approximations. NASA TN.D49, 1959.
- 1069 KAATTARI, G.E. Predicted shock envelopes about two types of vehicles at large angles of attack. NASA TN. D860, 1961.
- 1070 NITZBURG, G.E. and CRANDALL, S. A study of flow changes associated with airfoil section drag rise at supercritical speeds. NACA TN. 1813, 1949.
- 1071 MIRELS, H. Laminar boundary layer behind shock advancing into stationary fluid.
 NACA TN. 3401, 1955.
- 1072 MIRELS, H. Boundary layer behind shock or thin expansion wave moving into stationary fluid. NACA TN.3712, 1956.
- 1073 LEES, L. and LIN, C.C. Investigation of the stability of the laminar boundary layer in a compressible fluid. NACA TN.1115, 1946.
- 1074 GOODERUM, P.N. An experimental study of the turbulent boundary layer on a shock tube wall.
 NACA TN. 4243, 1958.
- 1075 HESS, N.W. Studies of structural failure due to acoustic loading. NACA TN. 4050, 1957.
- 1076 MULL, H.R. and ALGRANTI, J.S. Flight measurement of wall pressure fluctuations and boundary-layer turbulence. NASA TN. D280, 1960.
- 1077 EGGERS, A.J. A comparative analysis of the performance of long range hypervelocity vehicles. NACA TN.4046, 1957.
- 1078 REED, W.H. and BLAND, S.R. An analytical treatment of aircraft propeller precession instability. NASA TN.D659, 1961.
- 1079 JACK, J.R. Effects of extreme surface cooling on boundary layer transition. NACA TN.4094, 1957.
- 1080 BRASLOW, A.L. Effect of distributed three-dimensional roughness and surface cooling on boundary layer transition and lateral spread of turbulence at supersonic speeds.

 NASA TN. D53, 1959.
- 1081 OSTRACH, S. and THORNTON, P. Compressible laminar flow and heat transfer about a rotating isothermal disk. NACA TN.4320, 1958.
- 1062 ADAMS, E.W. Theoretical investigation of the ablation of a glass-type heat protection shield of varied material properties at the stagnation point of a re-entering IREM.

 NASA TN. D564, 1961.

- 1083 ANDERSON, K. and SINCHTEL, C.D. Discussion of solar proton events and manned space flights. NASA TN. D671, 1961.
- 1084 LIBBY, PA. and CRESCI, R.J. Experimental investigation of the downstream influence of stagnation point mass transfer. J. Ae. Scs. 28, 1961, 51.
- 1085 FELDMAN,S. On trails of axisymmetric hypersonic blunt bodies flying through the atmosphere. J. Ae. Scs. 28, 1961, 433.
- 1086 STRAND, T. Inviscid-incompressible flow theory of static peripheral jets in proximity to the ground. J. Ae. Scs. 1961, 27.
- 1087 LU, Pau-Chang. Free-convection magnetohydrodynamic flow past a porous flat plate.

 J. Ae. Scs. 28, 1961, 346.
- 1088 CRAMER, K.R. Magnetohydrodynamic free-convection pipe flow. J. Ae. Scs. 28, 1961, 736.
- 1089 CHARWAT, A.F. An investigation of separated flows, Part I: The pressure field. J. Ae. Scs. 28, 1961, 457.
- 1090 STONECYPHER, T.E. Periodic temperature distributions in a two-layer composite slab. J. Ae. Scs. 27, 1960, 152.
- 1091 CAMPEELL, W.F. Periodic temperature distribution in a two-layer composite slab. J. Ae. Scs. 27, 1960, 633.
- 1092 CHRICHLOW, W.J. and HAGGENMACHER, G.W. The analysis of redundant structures by the use of high-speed digital computers. J. Ae. Scs. 27, 1960, 595.
- 1093 Van DYKE, M.D. The supersonic blunt body problem Review and extensions. J. Ae. Scs. 25, 1958, 485.
- 1094 PROBSTEIN, R.F. and ELLIOTT, D. The transverse curvature effect in compressible axially symmetric laminar boundary layer flow. J. Ae.Scs. 28, 1956, 206.
- 1095 POHLE, F.V. and OLIVER, H. Temperature distribution and thermal stresses in a model of a supersonic wing. J. Ae. Scs. 21, 1954, 8.
- 1096 DRYDEN, H.L. Review of published data on the effect of roughness on transition from laminar to turbulent flow. J. Ae. Scs. 20, 1953, 477.
- 1097 CROCCO, L. and LEES, L. A mixing theory for the interaction between dissipative flows and nearly isentropic streams. J. Ae. Scs. 19, 1952, 649.
- 1098 MILLSAPS, K. and POHLHAUSEN, K. Heat transfer by laminar flow to a rotating plate.

 J. Ae. Scs. 19, 1952, 120.
- 1099 Von KARMAN, T. Fundamentals of statistical theory of turbulence. J.Ae.Scs. 4, 1937, 131.
- 1100 TAYLOR, E.S. and BROWNE, K.A. Vibration isolation of aircraft power plants.

 J. Ae. Scs. 6, 1938, 43.
- 1101 LEES, L. Laminar heat transfer over blunt nosed bodies at hypersonic flight speeds.

 Jet Prop. 26, 1956, 259.
- 1102 SOHEY, A.J. Advantages and limitations of models. J. R. Ae. S. 63, 1959, 646.
- 1103 SPALDING, D.B. Theory of mixing and chemical reaction in the opposed jet diffusion flame. A.R.S. Jnl. 31, 1961, 763.
- 1104 REEVES, B.L. Similar solutions of a free convection boundary layer equation for an electrically conducting fluid. A.R.S. Jnl. 31, 1961, 517.
- 1105 STEWARTSON, K. The asymptotic boundary layer on a circular cylinder in axial incompressible flow. Q.App.Math. 13, 1955, 113.
- 1106 CAMPHELL, I.J. The transverse potential flow past a body of revolution. Q.J.Mech.App.Math. 9, 1956, 140.
- 1107 TING, Lu. On the mixing of two parallel streams. J. Math. Phys. 38, 1959, 153.
- 1108 MacDONALD. Properties of the confluent hypergeometric function.
- 1109 OWEN, P.R. and ZIENKIEWICZ, H.K. The production of uniform shear flow in a wind tunnel. J. Fluid Mech. 2, 1957, 521.
- 1110 LIGHTHILL, M.J. Dynamics of a dissociating gas. J. Fluid Mech. 2, 1957, 1.
- 1111 LEIGH, D.C.F. The laminar boundary layer equation: a method of solution by means of an automatic computer. Proc. Cam. Phil. S. <u>51</u>, 1955, 320.

- 1112 SHERCLIFF, J.A. Steady motion of conducting fluids in pipes under transverse magnetic fields. Proc. Cam. Phil.S. 49, 1953, 136.
- 1113 SMITH, M.W. and LAMBERT, R.F. Acoustical signal detection in turbulent airflow. J. Acous. S. Am. 32, 1960, 858.
- 1114 DYER, I. Response of plates to a decaying and convecting random pressure field.

 J.Acous. S. Am. 31, 1959, 922.
- 1115 CONSTANTINESCU, V.N. On turbulent lubrication. Proc. Inst. Mech. E. 173, 1959, 881.
- 1116 JONES, E.E. The elliptic cylinder in a shear flow with hyperbolic velocity profile. Q.J.Mech.App.Math. 12, 1959, 191.
- 1117 MATHER, D.J. The motion of a viscous liquid past a paraboloid. Q.J.Mech.App.Math. 14, 1961, 423.
- 1118 TOMOTIKA, S. and HASIMOTO, Z. The transonic flow of a compressible fluid through an axially symmetrical nozzle. J.Math.Phys. 29, 1950, 105.
- 1119 VODICKA, V. Conduction of fluctuating heat flow in a wall consisting of many layers. App.Sc.Res. 5, 1955, 108.
- 1120 GRANVILLE, R.A. and BOXALL, G. Measurement of convective heat transfer by means of the Reynolds analogy. Brit. J. App. Phys. 11, 1960, 471.
- 1121 KORST, H. H. A theory for base pressures in transonic and supersonic flow. J. App. Mech. 23, 1956, 593.
- 1122 JACOBS, W.F. A simplified approximate method for the calculation of the pressures around conical bodies of arbitrary shape in supersonic and hypersonic flow.

 J.Aero.Scs. 28, 1961, 987.
- 1123 CRESCI, R.J. and LIBBY, P.A. The downstream influence of mass transfer at the nose of a slender cone. J.Aer. Scs. 29, 1962, 815.
- 1124 MORRIS, D.N. A summary of the supersonic pressure drag of bodies of revolution. J.Aero.Scs. 28, 1961, 563.
- 1125 PAPPAS, C.C. and OKUNO, A.F. Measurements of skin friction of the compressible turbulent boundary layer on a cone with foreign gas injection. J.Aero.Scs. 27, 1960, 321.
- 1126 CHARWAT, A.F. and YAKURA, J.K. An investigation of two-dimensional supersonic base pressures. J. Aero. Scs. 25, 1958, 122.
- 1127 CLIPPINGER, R.F. Supersonic axially symmetric nozzles. B. R. L. R794, 1951.
- 1128 LI,T-Y. Effects of free stream vorticity on the behaviour of a viscous boundary layer.

 J.Aero.Scs. 23, 1956, 1128.
- 1129 GERARD, J.H. An investigation of the noise produced by a subsonic air jet. J.Aero.Scs. 23, 1956, 855.
- 1130 CLAUSER, F. H. The behaviour of non-linear systems. J. Aero. Scs. 23, 1958, 411.
- 1131 PAI, S.I. Two-dimensional jet mixing of a compressible fluid. J.Aero.Scs. 16, 1949, 463.
- 1132 LIGHTHILL, M.J. Viscosity effects in sound waves of finite amplitude: in Survey in mechanics. Ed. by G.K. Batchelor and R.M. Davies. C.U.P. 1956.
- 1133 MURPHY, J.S. Some effects of surface curvature on laminar boundary layer flow. J. Aero. Scs. 20, 1953, 338.
- 1134 FERRI, A. and LIBBY, P.A. Note on the interaction between the boundary layer and the inviscid flow. J. Aero. Scs. 21, 1954, 130.
- 1135 DIACONIS, N.S. The calculation of wall shearing stress from heat transfer measurements in compressible flows. J.Aero.Scs. 21, 1954, A201.
- 1136 ROA, G.V.R. Recent developments in rocket nozzle configurations. A.R.S.Jnl. 31, 1961,1488.
- 1137 CURLE, N. The generation of sound by aerodynamic means. J.Roy. Ae. S. 65, 1961, 724.
- 1138 PEARSON, H. and McKENZIE, A.B. Wakes in axial compressors. J.Roy. Ae. S. 63, 1959, 415.
- 1139 McMILLAN, F.A. Viscous effects on pitot tubes at low speeds. J.Roy. Ae. S. 58, 1954, 570.
- 1140 PRESTON, J.H. The determination of turbulent skin friction by means of pitot tubes.

 J.Roy.Ae.S. 58, 1954, 109.
- 1141 HAMILTON, J.A. and HUFTON, P.A. Free-flight techniques for high speed aerodynamic research.
 J.Roy.As.S. 60, 1956, 151.

- 1142 VAN DRIEST, E.R. The problem of aerodynamic heating. Aero. Eng. Rev. 15, 1956.
- 1143 VERTREGT, M. Interplanetary orbits. J. Brit. Inter. S. 16, 1958, 326.
- 1144 MAYER, E. Heat flow in composite slabs. J. Am. R.S. 22, 1952, 150.
- 1145 YOUNG, A.D. Skin friction in the laminar boundary layer in compressible flow. Aero. Quart. 1, 1949, 137.
- 1146 FRAENKEL and PORTNOY. Supersonic flow past slender bodies with discontinuous profile slope. Aero.Quart. 6, 1955, 114.
- 1147 LORD, W.T. and BREBNER, G.G. Supersonic flow past slender pointed wings with "similar" cross sections at zero lift. Aero. Quart. 10, 1959, 79.
- 1148 LIGHTHILL, M.J. On displacement thickness. J. Fluid Mech. 4, 1958, 383.
- 1149 PROUDMAN, I. and PEARSON, J.R.A. Expansions at small Reynolds number for the flow past a sphere and a circular cylinder. J. Fluid Mech. 2, 1957, 237.
- 1150 MEKSYN, D. Integration of the boundary layer equations. Proc. Roy. S. A. 237, 1956, 543.
- 1151 PROUDMAN, J. The generation of noise by isotropic turbulence. Proc.Roy.S.A 214,1952,119.
- 1152 RAYLEIGH, Lord. On the flow of a compressible fluid past an obstacle. Phil.Mag.32,1916, 1.
- 1153 MILLIKAN, C.B. On the steady motion of viscous, incompressible fluids, with particular reference to a variation principle. Phil.Mag. 7, 1929, 641.
- 1154 HALL, A.A. and HISLOP, G.S. Velocity and temperature distributions in the turbulent wake behind a heated body of revolution. Proc. Cam. Phil.S. 34, 1938, 345.
- 1155 HOWARTH, L. On the solution of the laminar boundary layer equations. Proc. Roy.S. A, 164, 1938, 547.
- 1156 HAVELOCK, T.H. The effect of shallow water on wave resistance. Proc.Roy.S.A, 100,1922,499.
- 1157 LIGHTHILL, M.J. The hodographic transformation in transonic flow. Proc. Roy. S. A, 191, 1947, 323.
- 1158 HEISLER, M.P. Temperature charts for induction and constant temperature heating.
 A.S.M.E.Trans. 14, 1947, 227.
- 1159 DUSINEERRE, G.M. Numerical methods for transient heat flow. A.S.M.E.Trans. 12, 1945, 703.
- 1160 MIRELS, H. Approximate analytical solutions for hypersonic flow past slender power-law bodies. NASA R-15, 1959.
- 1161 Van DYKE, M.D. and GORDON, H.D. Supersonic flow past a family of blunt symmetric bodies. NASA R-1, 1959.
- 1162 LOW, G.L. Nearly circular transfer trajectories for descending satellites. NASA R-3, 1959.
- 1163 CHAPMAN, D.R. An analysis of the corridor and guidance requirements for supercircular entry planetary atmospheres. NASA R-55, 1959.
- 1164 CHAPMAN, D.R. An approximate analytical method for studying entry into planetary atmospheres. NASA R-11, 1959.
- 1165 SMITH, D.W. and WALKER, J.H. Skin friction measurements in incompressible flow. NASA R-26, 1959.
- 1166 CLARKE, J.F. Flow of chemically reacting gas mixtures. CoA R117, 1961.
- 1167 CLARKE, J.F. Linearized flow of a dissociating gas. J. Fluid Mech. 7, 1960, 577.
- 1168 CLARKE, J.F. Heat conduction through a gas with one inert internal model. CoA N102, 1960.
- 1169 CLARKE, J.F. On the sudden contact between a hot gas and a cold solid. CoA R124, 1962.
- 1170 MARY, H. The interaction of a reflected shock wave with the boundary layer in a shock tube. NACA TM.1418.
- 1171 SHERMAN, F.S. A low density wind tunnel study of shock wave structure and relaxation phenomena in gases. NACA TN.3298.
- 1172 CORTRIGHT, E.M. Some aerodynamic considerations of nozzle afterbody combination. Aero. Eng. Rev. <u>15</u>, 1956, 59.
- 1173 REID, J. and HASTINGS, R.C. The effect of a central jet on the base pressure of a cylindrical afterbody in a supersonic stream. ARC R & M.3224, 1962.

- 1174 BROMM, A.F. and O'DONNEL, R.M. Investigation at supersonic speeds of the effects of jet Mach number and divergence angle of the nozzle upon the pressure of the base annulus of a body of revolution. NACA RM L54I16, 1954.
- 1175 HOLDER, D.W., NORTH, R.J. and CHINNECK, A. Experiments with static tubes in a supersonic at stream. ARC R & M 2782, 1953.
- 1176 CRAVEN, A.H. Base pressure at subsonic speeds in the presence of a supersonic jet. CoA R129, 1960.
- 1177 JOHANNESEN, N.H. The mixing of free axially symmetric jets of Mach number 1.4. ARC R18,968, 1959.
- 1178 GRIFFITH, W.C. and KENNY, A. On full dispersed shock waves in carbon dioxids.

 J.Fluid Mach. 3, 1957, 286
- 1179 CHAPMAN, D. An analysis of base pressure at supersonic speeds and comparison with experiment. NACA TN.2137, 1950.
- 1180 TING, L. Boundary layer over a flat plate in presence of shear flow. Phys. Fluids, 13, 1960, 78.
- 1181 VODICKA, V. Some problems on heat conduction in stratiform bodies. J. Phys. Soc. Japan, 14, 1959, 216.
- 1182 Van DRIEST, E.R. and BLUMER, C.B. Effect of roughness on transition in supersonic flow.

 AGARD R255, 1960.
- 1183 HARRIS, E.L. and PATTERSON, G.N. Properties of impact pressure probes in free molecules flow. UTIA R52, 1958.
- 1184 MOLYNEUX, W.G. Scale models for thermo-aeroelastic research. RAE TN. Struct. 294, 1961.
- 1185 CHAPMAN, D.R. Some possibilities of using gas mixtures other than air in aerodynamic research. NACA R1259, 1956.
- 1186 GADD, G.E., HOLDER, D.W. and REGAN, J.D. Base pressure in supersonic flow. ARC CP271, 1996
- 1187 CHAPMAN, D.R., KUEHN, D.M. and LARSON, H. Investigation of separated flows in supersonic and subsonic streams with emphasis on the effects of transition. NACA R1356, 1958.
- 1188 CHAPMAN, D.R. An analysis of base pressure at supersonic velocities and comparison with experiment. NACA R1051, 1951.
- 1189 CHAPMAN, D.R., WIMEROW, W.R. and KESTER, R.H. Experimental investigation of base pressure on blunt-trailing-edge wings of supersonic velocities. NACA R1109.
- 1190 KANWAL, R. On magnetchydrodynamic shock waves. J. Math. Mech. 9, 1960, 681.
- 1191 HALL, N.G. A theory for the core of a leading edge vortex. RAE R.Aero. 2644, 1960.
- 1192 YASHURA, M. On the hypersonic viscous flow past slender bodies of revolution. J.Phys.Soc. Japan, 11, 1956, 878.
- 1193 EGGERS, A.J., SYVERTSON, C.A. and KRAUS, S. A study of inviscid flow about airfoils at high supersonic speeds. NACA R1123, 1953.
- 1194 ALLEN, H.J. General theory of airfoil sections having arbitrary shape or pressure distribution. NACA R833, 1945.
- 1195 TRAMPOSCH, H. and GERARD, G. Correlation of theoretical and photo-thermoelastic results on thermal stresses in idealized wing structure. J.App. Mech. 27, 1960.
- 1196 CAMPRELL, I.J. and LEWIS, R.G. Pressure distributions. Axially symmetric bodies in oblique flow. ARC CP213, 1955.
- 1197 SWIHART, J.W. and WHITCOMB, C.F. Pressure distributions on three bodies of revolution to determine the effect of Reynolds number up to and including the transonic speed colgan NACA RM L53H04, 1953.
- 1198 NICHOLS, M.R. and KEITH, A.L. Investigation of a systematic group of NACA 1 series cowlings with and without spinners. NACA R950, 1949.
- 1199 WIGHT, K.C. Measurement of two dimensional derivatives on a wing-aileron-tab system.

 Part I, ARC R & W 2934, 1955. Part II ARC R & M 3029, 1958.
- 1200 WATSON, J. Calculation of derivatives for a cropped delta wing with subsonic leading edges oscillating in a supersonic airstream. ARC R & M 3060, 1958.

- 201 Van DYKE, M.D. Supersonic flow past oscillating airfoils including nonlinear thickness effects. NACA R1183, 1954.
- 1202 WILLIAMS, J. Aircraft flutter. ARC R & M 2492, 1951.
- 1203 SINOTT, C.S. Calculated velocity distributions and force derivatives for a series of high-speed aerofoils. ARC R & M 3045.
- 1204 HUNTON, L.W. A study of the application of airfoil section data to the estimation of the high subsonic speed characteristics of swept wings. NACA RM A55023, 1955.
- 1205 WALKER, H.J. and MAILLARD, W.C. A correlation of airfoil section data with the aerodynamic loads measured on a 45° sweptback wing at subsonic Mach numbers. NACA RM A55008, 1955.
- 1206 WOODS, L.C. The applications of the polygon method to the calculation of the compressible subsonic flow round two-dimensional profiles. ARC CP115, 1953.
- 1207 SCHUBAUER, G.B. and SKRAMSTAD, H.K. Laminar boundary layer oscillations and transition on a flat plate. NACA R909, 1948.
- 1208 SATO, H. The Hall effect in the viscous flow of ionized gas between parallel plates under transverse magnetic field. J.Phys.Soc. Japan, 16, 1961, 1427.
- 1209 RIBNER, H.S. Boundary layer induced noise in the interior of aircraft. UTIA R37, 1956.
- 1210 RIBNER, H.S. Propeller in yaw. NACA R820, 1945.
- 1211 CHERNYI, G.G. Effect of slight blunting of leading edge of an immersed body on the flow around it at hypersonic speed. NASA TT F-35, 1960.
- 1212 STRATFORD, B.S. and SANSONE, G.E. Theory and tunnel tests of rotor blade for supersonic turbines. ARC R & M 3275, 1960.
- 1213 STRATFORD, B.S. and SANSONE, G.E. The performance of supersonic turbine nozzles.

 ARC R & M 3273, 1959.
- 1214 STANIFORTH, R. On the testing of supersonic compressor cascades. NGTE R212, 1957.
- JOHNSTON, I.H. and DRANSFIELD, D.C. The test performance of highly loaded turbine stages designed for high pressure ratio. NGTE R235, 1959.
- 1216 KANTROWITZ, A. The supersonic axial flow compressor. NACA R974, 1950.
- 1217 OSWATITSCH, K. and ROTHSTEIN, W. Flow pattern in a converging-diverging nozzle. NACA TM.1215.
- 1218 LAURENCE, J. Intensity, scale and spectra of turbulence in mixing region of free subsonic jet. NACA R1292, 1956.
- 1219 RIBNER, H.S. On the strength distribution of noise sources along a jet. UTIA R51, 1958.
- 1220 ALLCOCK, G.A., TANNER, P.L. and McLACHAN, K. A general purpose analogue correlator for the analysis of random noise signals. Univ. Southampton A.A. R205, 1962.
- 1221 LORD, W.T. A theoretical study of annular supersonic nozzles. ARC R & M 3227, 1961.
- 1222 MARSDEN, D.J., SIMPSON, R.W. and RAINBIRD, W.J. The flow over delta wings at low speeds with leading edge separation. CoA R114, 1957.
- 1223 GADD, G.E. A note on the theory of the Stanton tube. ARC R & M 3147, 1958.
- 1224 JONES, J.G. Quasi-cylindrical surfaces with prescribed loadings in the linearised theory of supersonic flow. RAE TN. Aero. 2769.
- 1225 JORGENSEN, L.H. Elliptic cones alone and with wings at supersonic speeds. NACA TN4045, 1957
- 1226 ROBINSON, A. Aerofoil theory of a flat delta wing at supersonic speeds. RAE R.Aero.2151, 1946.
- 1227 RANDALL, D.G. A technique for improving the predictions of linearised theory on the drag of straight edge wings. ARC CP394, 1957.
- 1228 I-DEE CHANG. Navier-Stokes solutions at large distances from a finite body.

 J.Math.Mech. 10, 1961, 811.
- 1229 OWEN, P.R. and ANDERSON, R.G. Interference between the wings and tail surfaces of a combination of slender body, cruciform wings and cruciform tail set at both incidence and yaw. RAE R.Aero.2471, 1952.
- 1230 OWEN, P.R. and MASKELL, E.C. Interference between the wings and tail plane of a slender wing-body tailplane combination. RAE R. Aero. 2441, 1951.

- 1231 Van DYKE, M.D. Practical calculation of second-order supersonic flow past non-lifting bodies of revolution. NACA TN.2744.
- 1232 EHRET, D.M. Accuracy of approximate methods for predicting pressure on pointed non-lifting bodies of revolution in supersonic flow. NACA TN.2764.
- 1233 FRAENKEL, L.E. The theoretical wave drag of some bodies of revolution. RAE R.Aero.2420.
- 1234 SYVERTSON, C.A. and DENIS, D.H. A second order shock-expansion method applicable to bodies of revolution near zero lift. NACA TN. 3527.
- 1235 EMINTON, E. On the minimisation and numerical evaluation of wave drag. RAE R. Aero. 2564.
- 1236 RUDIN, M. Criteria for thermodynamic equilibrium in gas flow. Phys. Fluids, 1, 1958.
- 1237 DIMMOCK, N.A. A compressor routine test code. NGTE R246, 1961.
- 1238 OWER, E. and JOHANSEN, F.C. On a determination of the pitot-static tube factor at low Reynolds numbers, with special reference to the measurement of low air speeds.

 ARC R & M 1437, 1931.
- 1239 KETTLE, D.J. Design and calibration at low speeds of a static tube and a pitot-static tube with semi-ellipsoidal nose shapes. RAE TN. Aero. 2247, 1953.
- 1240 CHAPMAN, D.R. A theoretical analysis of heat transfer in regions of separated flow. NACA TN.3729, 1956.
- 1241 NASH, J.F. Laminar mixing of a non-uniform stream with a fluid at rest. ARC 22245, 1960.
- 1242 KIRK, F.N. An approximate theory of base pressure in two dimensional flow at supersonic speeds. RAE TN. Aero. 2377, 1954.
- 1243 GOODERUM, P.B., WOOD, G.P. and BREVOORT, M. Investigation with an interferometer of the turbulent mixing of a free supersonic jet. NACA R963, 1950.
- 1244 PRESTON, J.H. and SWEETING, N.E. An improved smoke generator for use in the visualisation of airflow, particularly boundary layer flow at high Reynolds numbers.

 ARC R & M 2023, 1943.
- 1245 HUGGETT, D.J. The ground effect on the jet flap in two dimensions. ARC 19,713, 1957.
- 1246 FALKNER, V.M. The design of minimum drag tip fins. With an Appendix on the conformal transformation of a wing with a fin. ARC R & M 2279, 1945.
- 1247 WEHER, J. The calculation of the pressure distribution on thick wings of small aspect ratio at zero lift in subsonic flow. ARC R & M 2993.
- 1248 OHMAN, L. The application of Lighthill formula for numerical calculation of pressure distributions on bodies of revolution at supersonic speed and zero angle of attack. SAAB TN.45, 1960.
- 1249 WATSON, E.J. Formulae for the computation of the functions employed for calculating the velocity distribution about a given aerofoil. ARC R & M 2176.
- 1250 EMINTON, E. Pressure distributions at zero lift for delta wings with rhombic cross sections. ARC CP.525, 1960.
- 1251 NAYSMITH, A. A collection of longitudinal stability derivatives of wings at supersonic speeds. RAE TN.Aero.2423, 1956.
- 1252 O'HARA,F., SQUIRE,L.C. and HAINES,A.B. An investigation of interference effects on similar models of different size in various transonic tunnels in the U.K. RAE TN.Aero.2606, 1959.
- 1253 LILLEY, G.M. and SPILLMAN, J.J. On the ground level disturbance from large aircraft flying at supersonic speeds. CoA N103.
- 1254 CRAVEN, A.H. Boundary layers with suction and injection. A review of published work on skin friction. CoA R136.
- 1255 LILLEY, G.M. An approximate solution of the turbulent boundary layer equations in incompressible and compressible. CoA R134, 1960.
- 1256 STANEROOK, A. An experimental study of the glancing interaction between a shock wave and a turbulent boundary layer. ARC CP.555, 1960.
- 1257 PAI, S.I. On turbulentflow between parallel plates. J.App.Mech. 20, 1953, 109.
- 1258 CHOU, Y.T. and SAIBEI, E. The effect of turbulence on slider bearing lubrication. J. App. Mech. 25, 1959, 122.

- HILSENRATH, J. and EECKETT, C.W. Thermodynamic properties of Argon-free air (0.78847 N2, 0.2115302) to 15,000°K. NBS R3991, 1955.
- 1260 RUBESIN, M.W. and JOHNSON, H.A. A critical review of skin friction and heat transfer solutions of the laminar boundary layer of a flat plate. ASME Trans. 1949.
- 1261 YASHURA,M. Experiments on axi-symmetric boundary layers along a long cylinder in incompressible flow. Trans. Japan Soc.Ae.Sc. 2, 1959.
- 1262 TAYLOR,G.I. The formation of a blast wave by a very intense explosion. Proc. Roy. Soc. A, 201, 1950, 159.
- 1263 LIN,S.C. Cylindrical shock waves produced by instantaneous energy release.

 J.App.Phys. 25, 1954, 54.
- 1264 ECKHAUS, W. Asymptotic solution of the two dimensional oscillating aerofoil problem for high subsonic Mach numbers. Proc. 9th Int. Con. App. Mech. 1956.
- 1265 LAMBOURNE, N.C. Some instabilities arising from the interaction between shock waves and boundary layer. NPL Aero. 348, 1958.
- 1266 SMITH, A.N.C. and PIERCE, J. Exact solution of the Neumann problem. Calculation for non-circulatory plane and axially symmetric flows about or within arbitrary boundaries.
 3rd Nat. Con. App. Mech. 1958.
- 1267 GUPTA,S. Steady and transient free convection of an electrically conducting fluid from a vertical plate in the presence of a magnetic field. App. Sc. Res. 2, 1959.
- 1268 CRAMER, K.R. Several magnetohydrodynamic free-convection solutions. 5th Nat. Heat Transfer Con. 1962.
- MORI,Y. On a laminar free-convection flow and heat transfer of electrically conducting fluid on a vertical flat plate in the presence of a transverse magnetic field.

 Trans. Japan Soc. Ae. Sc. 2, 1959.
- 1270 MORI, Y. On combined free and forced convection laminar magnetohydrodynamic flow and heat transfer in channels with transverse magnetic field. Int. Devel. in Heat Transfer, 1961.
- 1271 SQUIRE, W. An experimental test of compressibility transformation for turbulent boundary layer. J. Ac. Sc. 29, 1962.
- 1272 STETSON, K.F. Boundary layer transition on blunt bodies with highly cooled boundary layers. J. Ae. Sc. 1960, 81.
- 1273 CAPIAUX, R. and KARCHMAR, L. Flow past slender blunt bodies a review and extension. Paper 61-201-1904, Nat. IAS-ARS Joint Meeting, 1961.
- 1274 ADAMS, E.W. Analysis of quartz and teflon shields for a particular re-entry mission. Heat Transfer and Fluid Mech. Inst. 1961, 222.
- 1275 WONG, T. and SLYE, R. The effect of lift on entry corridor depth and guidance requirements for the return lunar flight. NASA R-80, 1960.
- 1276 KEENAN, J.H. Reaction tests of turbine nozzles for supersonic velocities.

 ASME Trans. 1949, 773.
- 1277 HAUSER, C.H., PLOHR, H.W. and SONDER, G. Study of flow conditions and deflection angle at exit of two-dimensional cascade of turbine rotor blades at critical and supercritical pressure ratios. NACA RM E9K25, 1950.
- 1278 ROBINSON, A. On source and vortex distributions in the linearised theory of steady supersonic flow. Q.J.Mech.App.Math. 1948.
- 1279 BEANE, B.J. and RYNA, B.M. Supersonic drag calculations for a cylindrical shell wing of semicircular cross section combined with a central body of revolution.

 Douglas SM22627, 1956.
- 1280 STANBROOK, A. The surface oil flow technique as used in high speed wind tunnels in the United Kingdom. RAE TN. Aero. 2712, 1960.
- 1281 STOKER, J.J. Non-linear vibrations in mechanical and electrical systems. Chapt. 5a, Intersc. Publ. 1950.
- 1282 BAUGHMAN, L.E. and KOCHENDORFER, F.D. Jet effects on base pressure of conical afterbodies at Mach 1.91 and 3.12. NACA RE E57E06.
- 1283 KEMP, N.H., ROSE, P.H. and DETRA, R.W. Laminar heat transfer around blunt bodies in dissociated air. J.Ae.Sc. 26, 1959.

- 1284 BIOT, M.A. The divergence of supersonic wings including chordwise bending.
 J. Ac. Scs. 23, 1956.
- 1285 HEDGEPETH, J.M. On the flutter of panels at high Mach numbers. J.Ae.Scs. 23, 1956.
- 1286 MURPHY, C.H. Effect of roll on dynamic instability of symmetric missiles.

 J. Ae. Scs. 21, 1954.
- 1287 BIRD, J.D. Some theoretical low-speed loading characteristics of swept wings in roll and sideslip. NACA R969, 1950.
- 1288 SPREITER, J.R. and SACKS, A.H. The rolling up of the trailing vortex sheet and its effect on the downwash behind wings. J. Ae. Scs. 18, 1951.
- 1289 SPREITER, J.R. and SACKS, A.H. A theoretical study of the aerodynamics of slender cruciform-wing arrangements and their wakes. NACA R1296, 1957.
- 1290 BOLZ, R.E. Dynamic stability of a missile in rolling flight. J. Ae. Scs. 19, 1952.
- 1291 STALKER, R.J. Sweepback effects in the turbulent boundary-layer shock-wave interaction.
 J. Ae. Scs. 27, 1960.
- 1292 SMITH, A.M.O. Rapid laminar boundary layer calculations by piece-wise application of similar solutions. J. Ae. Scs. 23, 1956.
- 1293 WISNIEWSKI, R.J. and JACK, J.R. Recent studies on the effect of cooling on boundary layer transition at Mach 4. J. Ae. Scs. 28, 1961.
- 1294 CRESCI, R.J. and MACKENZIE, D.A. An investigation of laminar transitional and turbulent heat transfer on blunt-nosed bodies in hypersonic flow. J. As. Scs. 27, 1960.
- 1295 ECONOMOS, C. and LIBBY, P.A. A note on transitional heat transfer under hypersonic conditions. J. Ae. Scs. 29, 1962.
- 1296 LIGHTHILL, M.J. Notes on waves through gases at pressures small compared with the magnetic pressure, with applications to upper atmosphere aerodynamics.

 J. Fluid Mech. 9, 1960, 465.
- 1297 McCUNE, J.E. and RESLER, E.L. Compressibility effects in magneto-aerodynamic flows past thin bodies. J. Ae. Scs. 27, 1960.
- 1298 YEN, K.T. Incompressible wedge flows of an electrically conducting viscous fluid in the presence of a magnetic field. J. Ae. Scs. 27, 1960.
- 1299 MEKSYN, D. Magnetohydrodynamic flow past a semi-infinite plate. J. Ae. Scs. 29, 1962.
- 1300 REEVES, B.L. and KIPPENHAN, C.J. On a particular class of similar solutions of the equations of motion and energy of a viscous fluid. J. Ae. Scs. 29, 1962.
- 1301 RAHMAN, M.A. Approximate design of sharp cornered supersonic nozzles. J.Ae.Scs. 29, 1962.
- 1302 HANSEN, C.F. Approximations to the thermodynamic and transport properties of high temperature air. NASA R50, 1959.
- 1303 FENSTER, S.J. and NEYMAN, R.J. Effect of variable heat recombination on stagnation point heat transfer. J. Ae. Scs. 29, 1962.
- 1304 OGUCHI, H. First-order approach to a strong interaction problem in hypersonic flow over an insulated flat plate. Univ. Tokyo Aero.Res R330, 1958.
- 1305 LI, T.Y. and GROSS, J.F. Hypersonic strong viscous interaction on a flat plate with surface mass transfer. Heat Transfer and Fluid Mech. Inst. 1961, 146.
- 1306 MASLEN, S.H. Second approximation to laminar compressible boundary layer on flat plate in slip flow. NACA TN.2818, 1952.
- 1307 NAGAKURA, T. and NARUSE, H. An approximate solution of hypersonic laminar boundary layer equations and its application. J. Phys. Soc. Japan, 12, 1957, 1298.
- 1308 YASUHARA, M. On the hypersonic viscous flow past a flat plate with suction or injection.
 J. Phys. Soc. Japan, 12, 1957, 177.
- 1309 STEWARTSON, K. On the motion of a flat plate at high speed in a viscous compressible fluid, II, steady motion. J. Ae. Scs. 22, 1955, 303.
- 1310 LEES, L. and PROBSTEIN, R.F. Hypersonic viscous flow over a flat plate.
 Princeton Univ. Aero. Eng. R195, 1952.
- 1311 PEARCEY, H. H. A method for predicting the onset of buffeting and other separation effects from wind tunnel tests on rigid models. NPL Aero. 358, 1958.

- 1312 LADSON, C.L. Chordwise pressure distributions over several NACA 16 series airfoils at transonic Mach numbers up to 1.25. NASA Memo 6-1-59L, 1959.
- 1313 SPREITER, J.R. On alternative forms for the basic equations of transonic flow theory.

 J. Ae. Scs. 21, 1954.
- 1314 BRASLOW, A.L. and KNOX, E.C. Simplified method for determination of the critical height of distributed roughness particles for boundary layer transition at Mach numbers from 0 to 5. NACA TN.4363, 1958.
- 1315 HAINES, A.B., HOLDER, D.W. and PEARCEY, H.H. Scale effects at high subsonic and transonic speeds and methods for fixing transition in model experiments. ARC R & M3012, 1954.
- 1316 PEARCEY, H.H. The occurrence and development of boundary layer separations at high incidence and high speeds. ARC R & M 3109, 1955.
- 1317 FREEMAN, N.C. Dynamics of a dissociating gas, III. Non-equilibrium theory. J. Fluid Mech. 4, 1958.
- 1318 MASLEN, S.H. and MOECKEL, W.E. Inviscid hypersonic flow past blunt bodies. J. Ae. Scs. 24, 1957.
- 1319 MOORE, F.K. and GIBSON, W.E. Propagation of weak disturbances in a gas subject to relaxation effects. J. Ae. Scs. 27, 1960.
- 1320 LEIGH, D.C. Comment on improved numerical solution of the Blasius problem with three-point boundary conditions. J. Ae. Scs. 29, 1962.
- 1321 CHRISTIAN, W.J. Improved numerical solution of the Blasius problem with three-point boundary conditions. J. Ae. Scs. 28, 1961.
- 1322 TOBA, K. On the numerical solution of the Blasius problem with three-point boundary conditions. J. Ae. Sos. 29, 1962, 480.
- 1323 KEAP, N.H. Vorticity interaction at an axisymmetric stagnation point in a viscous incompressible fluid. J. Ae. Scs. 26, 1959.
- 1324 ROTT, N. and LENARD, M. Vorticity effect on the stagnation point flow of a viscous incompressible fluid. J. Ae. Scs. 26, 1959.
- 1325 LEVY,S. Heat transfer to constant property laminar boundary layer flows with power function free stream velocity and wall temperature variation. J.Ae.Scs. 19, 1952.
- 1326 SHEN, S.F. and SOLOMON, J.M. First-order slip effects on the compressible laminar boundary layer over a slender body of revolution in axial flow. J.Ae.Scs. 28, 1961.
- 1327 MOORE, F.K. On local flat plate similarity in the hypersonic boundary layer. J. Ae. Scs. 28, 1961.
- 1328 HERRIN; T.R. The boundary layer near the stagnation point in hypersonic flow past a sphere. J. Fluid Mech. 7, 1960, 257.
- 1329 PROBSTEIN, R.F. and KETP, N.H. Various aerodynamic characteristics in hypersonic rarefied gas flow. J. Ae. Scs. 27, 1960.
- 1330 EMA.ONS, H.W., CHANG, C.R. and WATSON, B.C. Taylor instability of finite surface waves. J. Fluid Bech. 7, 1960.
- 1331 BELLNAN, R. and PENNINGTON, R.H. Effects of surface tension and viscosity on Taylor instability. Q. App. Math. 12, 1954.
- 1332 CHENG, H.K. Similitude of hypersonic real-gas flows over slender bodies with blunted noses. J. Ae. Scs. <u>26</u>, 1959, 575.
- 1333 WHALEN, R.J. Boundary layer injection on a yawed infinite wing in hypersonic flow. J. Ae. Scs. <u>26</u>, 1959.
- 1334 LEES, L. Influence on the leading edge shock wave on the laminar boundary layer at hypersonic speeds. J. Ae. Scs. 23, 1956.
- 1335 LIEPMANN, H.W. The interaction between boundary layer and shock waves in transonic flow. J. Ae. Scs. 13, 1946.
- 1336 ROTT, N. and CRABTREE, L.F. Simplified laminar boundary layer calculations for bodies of revolution and for yawed wings. J. Ae. Scs. 19, 1962.
- 1337 SCOTT, C.J. and ANDERSON, G.E. Boundary layer transition with gas injection. J. Ae. Scs. 25, 1958.

- 1338 LEADON, B.M., SCOTT, C.J. and ANDERSON, G.E. Mass transfer cooling at Mach number 4.8. J. Ae. Scs. 25, 1958.
- 1339 BARTLE, E.R. and LEADON, B.H. Experimental evaluation of heat transfer with transpiration cooling in a turbulent boundary layer at M=3.2. J. Ae. Scs. 27, 1960.
- 1340 RADBILL, J.R. and KAYE, J. Analysis of effects of diffusion of a foreign gas into the laminar boundary layer of a supersonic flow of air in a tube. J.Ae.Scs. 26, 1959.
- 1341 FOELSCH, K. The analytical design of an axially symmetric Laval nozzle for a parallel and uniform jet. J. Ae. Scs. 16, 1949.
- 1342 SMITH, J.W. Effect of diffusion fields on the laminar boundary layer. J.Ze.Scs. 21, 1954.
- 1343 LEADON, B.M. and SCOTT, C.J. Transpiration cooling experiments in a turbulent boundary layer at E=3. J. Ae. Scs. 23, 1956.
- 1344 LEADON, B.M. Some experimental techniques in mass transfer cooling. Aero/Space Eng. 18, 1959.
- 1345 BARRY, F.W., SHAPIRO, A.H. and NEUMANN, B.P. The interaction of shock waves with boundary layer on a flat surface. J. Ae. Scs. 18, 1951, 229.
- 1346 COLES, D. Measurements of turbulent friction on a smooth flat plate in supersonic flow. J. Ae. Scs. 21, 1954, 433.
- 1347 HILL, F.K. Boundary layer measurements in hypersonic flow. J. Ae. Scs. 23, 1956, 35.
- 1348 Van DRIEST, E.R. Turbulent boundary layer in compressible fluids. J.Ae.Scs. 18, 1951, 145.
- 1349 KRAIR, R.F. and LIEERSTEIN, H. Numerical solution of the boundary layer equations without similarity assumptions. J. Ae. Scs. 26, 1959, 508.
- 1350 PAI,S.I. Laminar jet mixing of two compressible fluids with heat release. J. Ae. Scs. 23, 1956, 1012.
- 1351 MILLSAPS, K. and POHLHAUSEN, K. Thermal distributions in Jeffrey-Hamel flows between nonparallel walls. J. Ae. Ses. 20, 1953, 187.
- 1352 MURDUCHOW, M. On heat transfer over a sweat-cooled surface in laminar compressible flow with a pressure gradient. J. Ae. Scs. 19, 1952, 705.
- 1353 HOSHIZAKI, H. and SMITH, H.J. The effect of helium injection at an axially symmetric stagnation point. J. Ae. Scs. 26, 1959, 399.
- 1354 ZAKKAY, V. Laminar heat transfer and pressure measurements over blunt-nosed cones at large angles of attack. J. Ae. Scs. 25, 1958, 794.
- 1355 SCALA,S.M. The injection of air into the dissociated hypersonic laminar boundary layer.
 J. Ae. Scs. 25, 1958, 461.
- 1356 TAN, H.S. On optimum nose curves for missiles in the super-aerodynamic regime. J. Ae. Scs. 25, 1958, 56.
- 1357 CARTER, W.J. Optimum nose shapes for missiles in the super-aerodynamic region. J. Ac. Scs. 24, 1957, 527.
- 1358 MAGER, A. On the model of the free shock separation, turbulent boundary layer.
 J. Ae. Scs. 1956, 181.
- 1359 LEES, L. Note on the hypersonic similarity law for an unyawed cone. J. Ae. Scs. 18, 1951.
- 1360 GRIEMINTER, G., WILLIA S, E.P. and YOUNG, G. Lift on inclined bodies of revolution in hypersonic flow. J. Ae. Scs. 17, 1950.
- 1361 TOMOTIKA,S. and YOSINOBU,H. The flow of a viscous liquid past a flat plate at small Reynolds number. J. Math.Phys. 35, 1956.
- 1362 CHU, W.H. Three dimensional effect of flutter in a real fluid. J. Ac. Scs. 29, 374.
- 1363 CHU, W. H. and AHRAMSON, H. N. An alternative formulation of the problem of flutter in real fluids. J. Ae. Scs. 26, 1959.
- 1364 LIBBY, PA. and PALLONE, A. A method for analysing the insulating properties of the laminar compressible boundary layer. J. Ae. Scs. 21, 1954.
- 1365 LIBBY, P.A. The homogeneous boundary layer at an axisymmetric stagnation point with large rates of injection. J. Ae. Scs. 29, 1962.
- 1366 FOX, H. and LIBBY, P.A. Helium injection into the boundary layer at an axisymmetric stagnation point. J. Ae. Scs. 1962, 921.

- 1367 KALMAN, R.E. and BERTRAM, J.E. Control system and analysis and design via the second method of Lyapunov. ASME Trans. 82, 1960.
- 1368 BEST, D. Some problems of polar missile control. J.Roy.Ae.Soc., 64, 1960.
- 1369 TRAUGOTT,S. An approximate solution of the supersonic blunt body problem for prescribed arbitrary axisymmetric shapes. J. Ae. Scs. 27, 1960, 361.
- 1370 CASACCIO, A. Theoretical pressure distribution on a hemisphere cylinder combination. J. Ae. Scs. <u>26</u>, 1959, 63.
- 1371 FENNETT, F.D. Note on tip-bluntness effects in the supersonic and hypersonic regimes.

 J. Ae. Scs. 24, 1957, 314.
- 1372 OLIVER, R.E. An experimental investigation of flow about simple blunt bodies at a nominal Mach number of 5.8. J. Ae. Scs. 23, 1956, 177.
- 1373 EGHRS, A.J., SAVIM, R.C. and SYVERTSON, C.A. The generalized expansion method and its application to bodies travelling at high supersonic airspeeds. J.Ae.Scs. 22, 1955, 231.
- 1374 KELLY, H.J. An investigation of optimum zoom climb techniques. J. Ae.Scs. 26, 1959, 794.
- 1375 ILLINGWORTH, C.R. Steady flow in the laminar boundary layer of a gas. Proc. Roy. Soc. A, 199, 1949.
- 1376 VAN LE, N. Transformation between compressible and incompressible boundary layer equations. J. Ae. Scs. 20, 1953.
- 1377 CULICE, F.E. and HILL, J. A turbulent analog of the Stewartson-Illingworth Transformation. J. Ae. Scs. 25, 1958.
- 1378 ECKERT, E. Survey on heat transfer at high speeds. WADC TR54-70, 1954.
- 1379 FLAX, A.H. Reverse flow and variational theorems for lifting surfaces in nonstationary compressible flow. J. Ae. Scs. 20, 1953.
- 1380 DUGUNDI, J. Effect of quasi-steady air forces on incompressible bending-torsion flutter.

 J. Ae. Scs. 1958.
- 1381 GLAURT, M.B. and LIGHTHILL, M.J. The axisymmetric boundary layer on a long thin cylinder. Proc. Roy. Soc. A, 230, 1955, 188.
- 1382 KELLY, H.R. A note on the laminar boundary layer on a circular cylinder in axial incompressible flow. J. Ae. Scs. 21, 1954.
- 1383 MERSYN,D. Integration of the boundary layer equations for a plane in compressible flow with heat transfer. Proc. Roy. Soc. A, 231, 1955, 274.
- 1384 LAVELDER, R.E. and DEEP, R.A. Application of second-order shock-expansion theory to several types of bodies of revolution. J. Ae. Scs. 23, 1956, 1052.
- 1385 LILLLY, J.M. On a generalised porous-wall "Couette type" flow. J. Ae. Scs. 26,1959, 685.
- 1386 CRAUR, K.R. A generalised porous-wall "Couette type" flow. J. Ae. Scs. 26, 1959, 121.
- 1387 III. AL., R.L. Heat transfer for laminar flow in an annulus with porous wall. J. Ae. Scs. 26, 1959, 532.
- 1363 GLAUGHT, M.B. The pressure gradient induced by shear flow past a flat plate. J. Ae. Scs. 1962, 540.
- 1389 LI, T.Y. Simple shear flow past a flat plate in a compressible viscous fluid. J. Ae. Scs. 22, 1955, 724.
- 1390 JOMES, D. J. Some panel-flutter studies using piston theory. J. Ae. Scs. 25, 1958.
- 1691 HEDGERETH, J.M. Flutter of rectangular simply supported panels at high supersonic speeds.

 J. Ae. Scs. 24, 1957.
- 1392 JOGARAO, V. and LAKSHIKANTHAV, C. Natural frequencies of rectangular plates with edges elastically restrained against rotation. J. Ac. Scs. 1957.
- 1393 SAKURAI, T. The shear flow along a flat plate with uniform suction. J.Ae.Scs. 24, 1957.
- 1394 STUART, J.T. The viscous flow near a stagnation point when the external flow has uniform vorticity. J. Ae. Scs. <u>26</u>, 1959.
- 1395 BIOT, M.A. New methods in heat flow analysis with application to flight structures. J. Ae. Scs. 24, 1957.
- 1396 BIOT, L.A. Variational and Lagrangian thermodynamics of thermal convection-fundamental shortdomings of the heat transfer coefficient. J. Ae. Scs. 29, 1962.

- 1397 TURCOTTE, D.L. A sublayer for fluid injection into the incompressible turbulent boundary layer. J. Ae. Scs. 27, 1960.
- 1398 RANNIE, W.D. Heat transfer in turbulent shear flow. J. Ae. Scs. 23, 1956.
- 1399 JAEGER, J.C. Composite slabs. Q. App. Math. 8, 1950.
- 1400 BUDIANSKY, B. and CONNOR, R.W. Buckling stress of clamped rectangular plates in shear. NACA TN.1559, 1948.
- 1401 HALL, J.G., ESCHENROEDER, A.W. and MARRONE, P.V. Inviscid hypersonic airflows with coupled non-equilibrium processes. IAS Paper 62-67, 1962.
- 1402 De HOFFMAN, F. and TELLER, E. Magnetohydrodynamics shocks. Phys. Rev. 80, 1950, 693.
- 1403 HELGER.L. Magnetohydrodynamic shock waves. Astrophys. J. 117, 1953, 177.
- 1404 KUO, Y.A. Two dimensional transonic flow past airfoils. NACA TN. 2356.
- 1405 HILSENRATH, J., HECKETT, C.W. et al. Tables of thermal properties of gases.

 NBS Circular 564, 1955.
- 1406 MONAGHAN, R.J. On the behaviour of boundary layers at supersonic speeds. 5th Int. Astro. Conf. 1955.
- 1407 GERSHUNI, G.Z. and SHUKHOVITSKII, E.M. Stationary convection flow of an electrically conducting liquid between parallel plates in a magnetic field. Soviet Physics, 34, 1958.
- 1408 REGIRER, S.A. On convective motion of a conducting fluid between parallel vertical plates in a magnetic field. Soviet Physics, 37, 1960.
- 1409 CHOW, W.L. On the base pressure resulting from the interaction of a supersonic external stream with a sonic or subsonic jet. J. Ae. Scs. 1959, 176.
- 1410 GRAVALOS, F.G., EDELFLET, I.H. and EMMONS, H. The supersonic flow about a blunt body of revolution for gases at chemical equilibrium. 9th Int. Astro. Fed. 1958.
- 1411 HERBERLE, J.W., WOOD, G.P. and GOODERUM, P.B. Data on shape and location of detached shock waves in cones and sphere. NACA TN. 2000, 1950.
- 1412 BATDORF, S.B. and HOUBOLT, J.C. Critical combinations of shear and transverse direct stress for an infinitely long flat plate with edges elastically restrained against rotation.

 NACA R847, 1946.
- 1413 MATTING, F.W., CHAPMAN, D.R., NYHOLM, J.R. and THOMAS, A.G. Turbulent skin friction at high Mach numbers and Reynolds numbers in air and helium. NASA R82, 1960.
- 1414 Von KARMAN, T. The problem of resistance in compressible fluids. 5th Volta Cong. 1955,226.
- 1415 PEARCEY, H.H. The aerodynamic design of section shapes for swept wings. 2nd Int. Cong. Aero. Sc. 1960.
- 1416 PEARCEY, H. H. and STUART, C.M. Methods of boundary layer control for postponing and alleviating buffeting and other effects of shock-induced separation. SMF Fund Paper FF22, 1959.
- 1417 LIN, C.C. On the stability of two dimensional parallel flows. Pt.III Stability in a viscous fluid. Q. App. Math. 3, 1945, 277.
- 1418 MORKOVIN, M.V. Transition from laminar to turbulent shear flow. ASME Trans, 80, 1958, 1121.
- 1419 ROCKEY, K.C. The design of intermediate vertical stiffeners on web plates subjected to shear. Aero. Quart. 1956, 275.
- 1420 JASZLICS, I. and TRILLING, L. An experimental study of the flow field about swept and delta wings with sharp leading edges. J. Ac. Scs. 1959, 487.
- 1421 Van HISE, V. Analytic study of induced pressure on long bodies of revolution with varying nose bluntness at hypersonic speeds. NASA R78, 1960.
- 1422 LEISSA, A.W. and NIEDENFUHR, F.W. Bending of a square plate with two adjacent edges free and the others clamped or simply supported. A.I.A.A. Jnl. 1, 1963, 116.
- 1423 MACHELL, R.M. and O'ERYANT, W.T. An experimental investigation of the flow over bluntnosed cones at a Mach number of 5.8. Guggenheim Aero. Lab. Memo 32, 1956.
- 1424 HOLL, D.L. Cantilever plate with concentrated edge load. J. App. Mech. 1937, 8.
- 1425 MACNEAL, R.H. The solution of elastic plate problems by electrical analogies. J. App. Mech. 1951, 59.

- 1426 FERRI.A. Preliminary analysis of axial flow compressors having supersonic velocity at the entrance of the stator. NACA RM: L9306, 1949.
- 1427 DEICH, M.E. Flow of gas through turbine lattices. NACA TM. 1393, 1956.
- 1428 PORTNOY, J. The quasi-cylinder of specified thickness and shell loading in supersonic flow. Aero. Quart. 11, 1960, 387.
- 1429 POOLE, J.A. A description of the R.A.E. high speed supersonic tunnel. RAE TN.Aero. 2678, 1960
- 1430 ANDREWS, D.R. and EROWN, C.S. Calibration of the flow in the Mach 4 working section of the 4ft. x 3ft. high supersonic speed wind tunnel at RAE Bedford. RAE TN.Aero.2820, 1962.
- 1431 GRRENWOOD, G.H. Free-flight measurements of the zero-lift drag and base pressure on a wind tunnel interference model (M=0.8 1.5). RAE TN.Aero.2725, 1960.
- 1432 ADAMS, G.J. and DUGAN, D.W. Theoretical damping in roll and rolling moment due to differential wing incidence for slender cruciform wings and wing-body combinations.

 NACA R1088, 1952.
- ROJERS, A.W. Application of two dimensional vortex theory to the prediction of flow fields behind wings of wing-body combinations at subsonic and supersonic speeds.

 NACA TN. 3227, 1954.
- 1434 SPAHR, J.R. Contributions of the wing panels to the forces and moments of supersonic wingbody combinations at combined angles. NACA TN.4146, 1958.
- BECKWITH, I.E. and COHEN, N.B. Application of similar solutions to calculations of laminar heat transfer on bodies with yaw and large pressure gradients in high speed flow.

 NADA TR.D625, 1961.
- 1436 HOSHIZAKI, H. Heat transfer in planetary atmospheres at supersonic speeds. ARS Prep.2173-61, 1961.
- 1437 SCALA, S.M. and WARREN, W.R. Hypervelocity stagnation point heat transfer in dissociated air. ARS J. 1962.
- 1438 OFFERHARTZ, E., WISBLATT, H. and FLAG, R.F. Stagnation point heat transfer measurements at super satellite speeds. J. Roy. Aero. Soc., <u>66</u>, 1962.
- 1439 WOOD, J.P. and GOODERUE, P.B. A factor affecting transonic leading edge flow separation. NACA TN. 3804, 1956.
- 1440 LINUSEY, W.F. and LAMDRUM, E. Compilation of information on the transonic attachment of flows at the leading edge of airfoils. NACA TN.4204, 1958.
- 1441 HALFMAN, R.L., JOHNSON, H.C. and HALEY, S.M. Evaluation of high angle-of-attack aerodynamic derivative data and stall-flutter prediction techniques. NAJA TN. 2533, 1951.
- 1442 WOOLSTON, D.C. and CASTILE, G.E. Some effects of variations in several parameters including fluid density on the flutter speed of light uniform cantilever wings. MACA TM. 2558, 1951.
- 1443 PINKERTON, R.M. Calculated and measured pressure distribution over the midspan section of the NACA 4412 airfoil. NACA R563, 1936.
- 1444 ROTT.N. and GEORGE,M.B.T. An approach to the flutter problem in real fluids. Inst. Aero. Sos. Perp.509, 1955.
- 3445 REISSNER, E. On the application of Mathieu functions in the theory of subsonic compressible flow past oscillating airfoils. NACA TN. 1961.
- 1446 RAND,S. Wake of a satellite traversing the ionosphere. Phys. Fluids, 3, 1960.
- 1447 YOSHIRA, H. Motion of thin bodies in a highly rarefied plasma. Phys. Fluids, 4, 1961. .
- 1448 WYATT, P.J. Induction drag on a large negatively charged satellite moving in a magnetic-field-free ionosphere. J. Geophys. Rev. 65, 1960.
- 1449 DAVIS, A.H. and HARRIS, I. Interaction of a charged satellite with the ionosphere. NASA TN. D704, 1961.
- 1450 SAKAO,F. Some physical interpretations of magnetohydrodynamic duct flows. A.I.A.A. Jnl. 1, 1963, 915.
- 1451 PARKS, P.C. The methods of Lyapunov in the theory of automatic control. Control, 1961
- 1452 TSIEN, H.S. Symmetric Joukowsky airfoils in shear flow. Q. App. Math. 1, 1943, 130.

- 1453 VIDAL, R.J., CURTIS, J.T. and HILTON, J.H. The influence of two-dimensional stream shear on airfoil maximum lift. CAL Rep. A1-1190-A-7, 1961.
- 1454 NASH, W.A. Several approximate analyses of the bending of a rectangular cantilever plate by uniform normal pressure. J.App.Mech. 1952, 33.
- 1455 GLICK, H.S. Modified Cross-Lees mixing theory for supersonic separated and reattaching flows. GALCIT HYP. Res. Proj. Memo 53, 1960.
- 1456 VAGLIO-LAURIN, R. and TRELLA, M. A study of flow fields about some typical blunt-nosed slender bodies. PIBAL R.623, 1960.
- 1457 GOLDSTEIN,S. On laminar boundary-layer flow near a position of separation. Q. J. Mech. App. Mech. 1, 1948, 43.
- 1458 GORTLER, H. A new series for calculation of steady laminar boundary layer flows.

 J. Math. Mech. 6, 1957, 1.
- 1459 TANI, I. On the solution of the laminar boundary layer equations. J. Phys. Soc. Japan, 4, 1949, 149.
- 1460 STEWARTSON, K. Correlated incompressible and compressible boundary layers. Proc. Roy. Soc. A, 200, 1949, 84.
- 1461 CURLE, N. and SKAN, S.W. Approximate methods for predicting separation properties of laminar boundary layers. Aero. Quart. 8, 1957, 257.
- 1462 GERARD, G and GILBERT, A.C. Photo-thermoelasticity. J. App. Mech. 24, 1957.
- 1463 TRAMPOSCH, H. and GERARD, G. Physical properties of plastics for photo-thermoelastic investigation. J. App. Mech. 25, 1958.
- 1464 MICHAEL, W.H. Flow studies on flat plate delta wings at supersonic speeds. NACA TN.3472, 1955.
- 1465 WEEER, J. Slender delta wings with sharp edges at zero lift. RAE TN. Aero. 2508.
- 1466 McGREGOR, I. Development of the vapour screen method of flow visualization in the 3ft tunnel at RAE Bedford. R.A.E. Tech. Note Aero. 2709, 1960
- 1467 SPREITER, J.R. and ALKSNE, A.Y. Thin airfoil theory based on approximate solution of the transonic flow equations. NACA TN. 3970, 1957.
- 1468 HOSOWAKA, A. A refinement of the linearised transonic flow theory. J.Phys.Soc. Japan, 15, 1960.
- 1469 MAEDER, P.F. and THOMMEN, H.U. Linearised transonic flow about slender bodies at zero angle of attack. ASME Trans. J. App. Mech. 28, 1961.
- 1470 MIYAI, Y. Some notes for the small disturbance linear theory of the method of local linearisation of the flow over an airfoil at Mach number of unity. Proc. 10th Japan Nat. Cong. App. Mech. III-4, 1960, 207.
- 1472 LIEPMAN, H.W. and ROSHKO, A. Elements of gasdynamics. Waves in supersonic flow, p 85-88.
- 1473 TRUITT, R.W. Hypersonic aerodynamics. p 250 Freeman method. Ronald Press.
- 1474 CHAPMAN, D.R. Laminar mixing of a compressible fluid. NACA R958, 1950.
- 1475 LOCK, R.C. The velocity distribution in the laminar boundary layer between parallel streams. Q. J. Mech. App. Math. 4, 1951, 42.
- 1476 NAPOLITANO, I.G. The Blasius equation with three-point boundary conditions. Q. App. Math. 16, 1958, 397.
- 1477 POTTER, O.E. Laminar boundary layers at the interface of co-current parallel streams.
 Q. J. Mech. App. Math. 10, 1957, 302.
- 1478 EMMONS, H.W. and LEIGH, D.C. Tabulation of the Blasius function with blowing and suction.

 ARC CP. 1913, 1953.
- 1479 HARTREE, D.R. On an equation occurring in Falkner and Skan's approximate treatment of the equation of the boundary layer. Proc. Cam. Phil. S. 33, 1937, 223.
- 1480 SPALDING, D.B. Adiabatic wall temperature due to mass transfer cooling with a combustible gas. ARS J. 1959, 666.
- 1481 ECKERT, E., SCHNEIDER, P., HAYDAY, A. and LARSON, R. Mass transfer cooling of a laminar boundary layer by injection of a light weight foreign gas. Jet Prop. 1958, 34.

- 1482 LOVE, E.S. A re-examination of the use of the simple concepts for predicting the shape and location of detached shock waves. NACA TN.4170, 1957.
- 1483 AMBROSIO, A. and WORTMAN, A. Stagnation point shock detachment distance for flow around spheres and cylinder. ARS J. 32, 1962.
- 1484 VIDAL, R.J. The influence of two-dimensional stream shear for airfoil maximum lift. J. Ae. Scs. 29, 1962, 889,
- 1485 REID, W.P. Linear heat flow in a composite slab. J.Ae.Scs. 29, 1962.
- 1486 DUJUNDJI, J. Similarity laws for aerothermoelastic testing. J.Ae.Scs. 29, 1962, 935.
- 1487 ROM, J. Theory for supersonic two-dimensional, laminar, base-type flows using the Crocco-Lees mixing concepts. J.Ae.Scs. 29, 1962, 963.
- 1488 LEONARD, M. A reaction-rate parameter for gasdynamics of a chemically reacting gas mixture. J.Ae.Scs. 29. 1962, 995.
- 1489 MORDUCHOW, M. On calculation of the laminar separation point and results of certain flows. J.Ae.Scs. 29, 1962, 996.
- 1490 GUNDERSEN, R.M. Normal-shock relations in magnetohydrodynamics. J.Ae.Scs. 29, 1962, 997.
- 1491 BURTON, R.A. On the close relationship between turbulen plane-Quette and pressure flows.

 J. Ae. Scs. 29, 1962, 1004.
- 1492 KEENER, E.R. Prediction of Ogive-forebody pressures at angles of attack. J.Ae.Scs. 29, 1962.
- 1493 WILSON, R.E. Real-gas laminar boundary layer skin friction and heat transfer. J. Ae. Scs. 29, 1962, 640.
- 1494 YASHURA, M. Axisymmetric viscous flow plast very slender bodies of revolution. J. Ae. Scs. 29, 1962, 667.
- 1495 BORCHER, E.F. On similar solutions for strong blast waves and their application to steady hypersonic flow. J. Ae. Scs. 29, 1962, 694.
- 1496 ECKHAUS, W. A theory of transonic aileron buzz, neglecting viscous effects.
 J. Ae. Scs. 29, 1962, 712.
- 1497 TRAMFOSCH, H. Theoretical and experimental investigation of thermal stresses in hypersonic aircraft wing structures. J. Ae. Scs. 29, 1962.
- 1498 HESS, J.L. Calculation of potential flow about bodies of revolution having axes perpendicular to the free-stream direction. J. Ae. Scs. 29, 1962.
- 1499 GREENSITE, A.L. A closed-form solution for the oscillations of a vehicle entering a planetary atmosphere. J. Ae. Scs. 29, 1962, 745.
- 1500 CRAMER, K.R. Joule heating in magnetohydrodynamic free-convection flows. J.Ae.Scs., 29, 1962, 746.
- 1501 RIDYARD, H.W. Stagnation-point shock detachment of blunt bodies in supersonic flow. J. Ae. Scs. 29, 1962, 751.
- 1502 MAGER, A. On Squire's test of the compressibility transformation. J. Ae. Scs. 29, 1962,752
- 1503 SINNOTT, C.S. Theoretical prediction of the transonic characteristics of airfoils.

 J. Ae. Scs. 29, 1962, 275.
- 1504 OSTRACH,S. Stability of compressible boundary layers induced by a moving wave. J. Ae. Scs. 29, 1962, 289.
- 1505 LYONS, W.C. Transition measurements on cones in free flight ballistics range tests. J. Ae. Scs. 29, 1962, 352.
- 1506 BRANDMAIER, H.E. A note on Havelock's shallow-water wave-resistance curves.

 J. Ae. Scs. 29, 1962, 257.
- 1507 GOLDSTEIN, A.W. Energy equation approximations in fluid mechanics. J. Ac. Scs. 29, 1962, 358
- 1508 GREENBERG, R.A. A correlation of nose-bluntness induced pressures on cylindrical and conical after-bodies at hypersonic speeds. J. Ae. Scs. 29, 1962, 359.
- 1509 ADAMS, E.W. A graphical approximation for temperatures and sublimation rates at surfaces subjected to small net and large gross heat transfer rates. J. Ac. Scs. 29, 1962, 360.
- 1510 WEISS, D.C. Manoeuvring technique for changing the plane of circular orbits with minimum fuel expenditure. J. Ae. Scs. 29, 1962, 368.

- 1511 STRATFORD, B.S. Tunnel tests on a double cascade to determine the interaction between the rotor and the nozzles of a supersonic turbine. NGTE M359.
- 1512 MOORE, K.C. Quasi-cylindrical surfaces with prescribed thickness distributions. RAE TN. Aero. 2815, 1962.
- 1513 BRITTON, J.W. Pressure measurements at supersonic speeds on three uncambered conical wings of unit aspect ratio. RAE TN. Aero. 2821, 1962.
- 1514 SQUIRE, L.C. Pressure distributions and flow patterns on some conical shapes with sharp edges and symmetrical cross-sections at M=4.0. RAE TN.Aero.2823, 1962.
- 1515 NEUMARK, S. Self sustained oscillations of a system with non-linear damping of a particular type. RAE TN. Aero. 2839, 1962.
- 1516 GREENWOOD, G.H. Free-flight measurements of the dynamic longitudinal stability characteristics of a wind tunnel interference model (M=0.92 to 1.35). RAE TN.Aero.2798, 1961.
- 1517 CLARKE, J.F. Reaction-resisted shock fronts. CoA R. 150, 1961.
- 1518 CLARKE, J.F. Heat conduction through a polyatomic gas. CoA R.149, 1961.
- 1519 CRAVEN, A.H. Base pressure at supersonic speeds in the presence of a supersonic jet. Coa R.144, 1960.
- 1520 BENTON, E.R. Wing-tail interference as a cause of 'Magnus' effects on a finned missile.
 J. Ae. Scs. 29, 1962, 1358.
- 1521 HOSOKAWA, I. A note on application of transonic linearization to an airfoil with a round leading edge. J. Ae. Scs. 29, 1962, 1395.
- 1522 ZAKKAY, V. and CALLAHAN, C.J. Laminar, transitional and turbulent heat transfer to a conecylinder-flare body at Mach 8.O. J. Ae. Scs. 29, 1962, 1403.
- 1523 RAHMAN, M.A. Approximate determination of position of the sonic line for a blunt body in hypersonic flow. J. Ae. Scs. 29, 1962.
- 1524 ROZYCKI, R.C. and FENSTER, S.J. Stagnation point heat transfer in partially ionized air. J. Ae. Scs. 29, 1962.
- 1525 TIEN, C.L. On hypersonic viscous flow over an insulated flat plate with surface mass transfer. J. Ae. Scs. 29, 1962, 1024.
- 1526 MAREY, D.G. Leading edge attachment in transcnic flow with laminar or turbulent boundary layers. J. Ae. Scs. 29, 1962.
- 1527 MARTIN, E.D. Note on the three-point boundary layer problem for the Blasius equations.

 J. Ae. Scs. 29, 1962.
- 1528 SOLOMON, J.M. First-order slip effects on the laminar boundary layer over a slender body of revolution with zero pressure gradient. J. Ac. Scs. 29, 1962, 1136.
- 1529 GOUSE, S.W., BROWN, G.A. and KAYE, J. Some effects of injection of foreign gases in a decelerating laminar boundary layer in supersonic flow. J. Ae. Scs. 29, 1962, 1250.
- 1530 CHU, Wen-Hwa. An aerodynamic analysis for flutter in Oseen-type viscous flow. J. Ae. Scs. 29, 1962, 781.
- 1531 BIRD, G.A. The flow about a moving body in the upper ionosphere. J. Ae. Scs. 29, 1962.
- 1532 PARKS, P.C. Pitch-yaw stability of a missile oscillating in roll via the second method of Lyapunov. J. Ae. Scs. 29, 1962, 874.
- 1533 AMBROSIO, A. and WORTMAN, A. Stagnation-point shock-detachment distance for flow around spheres and cylinders in air. J. Ae. Scs. 29, 1962.
- 1534 INMAN, R.M. Consideration of energy separation for laminar slip flow in a circular tube.

 J. Ae. Scs. 29, 1962, 1014.
- 1535 DUNLAP,R. Shroud design for simulating hypersonic flow over the nose of a hemisphere. J. Ae. Scs. 29, 1962, 757.
- 1536 HIDALGO, H., TAYLOR, R.L. and KECK, J.C. Transition in the viscous wakes of blunt bodies at hypersonic speeds. J. Ae. Scs. 29, 1962.
- 1537 STOUDARD, F.J. Stagnation point viscous hypersonic flow. J. Ae. Scs. 29, 1962, 1138.
- 1538 BURGRAF, O.R. The conpressibility transformation and the turbulent boundary layer equations. J. Ae. Scs. 19, 1962.
- 1539 WEISS, D. Local heat transfer to a yawed, infite, circular cylinder in laminar compressible flow. J. Ae. Scs. 29, 1962.

- 1540 FORBES DEWEY, C. Use of local similarity concepts in hypersonic viscous interaction problems. A.I.A.A. J. 1963, 20.
- 1541 INGER, G.R. Similitude of hypersonic flows over slender bodies in non-equilibrium dissociated gases. A.I.A.A. J. 1963, 46.
- 1542 LARDNER, T.J. Biot's variational principle in heat conduction. A.I.A.A. J. 1963, 196.
- 1543 DOYLE, M.D.C. The stacking of compressor stage characteristics to give an overall compressor performance map. Aero. Quart. 13, 1962.
- 1544 EAST, R.A. A theoretical and experimental study of oscillating wedge shaped aerofoils in hypersonic flow. A.A.S.U. R228, 1962.
- 1545 SELLS, C.C.L. Calculation of sideslip derivatives and pressure distribution in asymmetric flight conditions on a slender wing-fin configuration. RAE TN. Aero. 2805, 1962.
- 1546 WOODLEY, J.G. Measurements of aerodynamic heating on a 15° cone of graded wall thickness at a Mach number of 6.8. RAE IN. Aero. 2847, 1962.
- 1547 CATHERALL, D. Boundary layer characteristics of caret wings. RAE IN.Aero. 2835, 1962.
- 1548 KING-MELE, D.G., and COOK, G.E. The contraction of satellite orbits under the influence of air drag. Pt.IV With scale height dependent on altitude. RAE TN. Space 18, 1962.
- 1549 HARTNETT, J.P. and ECKERT, E.R.G. Experimental study of the velocity and temperature distribution in a high-velocity vortex-type flow. ASEE Trans. 70, 1957.
- 1550 SPARROW, E.M. and LIN, S.H. Laminar heat transfer in tubes under slip-flow conditions. ASES Paper 61-WA-165.
- 1551 LIVESLEY, R.K. and BIRCHALL, P.C. Analysis of a loaded cantilever plate by finite difference methods. RAE TN. FS26, 1956.
- 1552 WRAY, K.L. Chemical kinetics of high temperature air. Hypersonic flow research, p 181, Academic Press, New York, 1962.
- 1553 HIDALGO,H. Ablation of glassy materials around blunt bodies of revolution. ARS J. 30, 1960.
- 1554 DETRA, R.W. and KIDALGO, H. Generalized heat transfer formulas and graphs. ARS J. 1961.
- 1555 HIDALGO, H. Closing reply to comments on generalized heat transfer formulas and graphs for nose cone re-entry into the atmosphere. ARS J. 1962.
- 1556 FELDMAN,S. Numerical comparison between exact and approximate theories of hypersonic inviscid flow past slender blunt nosed bodies. ARS J. 30, 1960.
- 1557 CHENG, K.H. and CHANG, A.L. A numerical comparison between exact and approximate theories of hypersonic inviscid flow past slender blunt nosed bodies. ARS J. 31, 1961.
- 1558 SLATTERY, R.E. and CLAY, W.G. Experimental measurements of turbulent transition motion, statistics and gross radial growth behind hypervelocity object. Physics of Fluids, 5, 1962, 849.
- 1559 RESHOTKO, E. and COHEN, C.B. Heat transfer at the forward stagnation point of blunt bodies.
 NACA TN. 3513, 1955.
- 1560 RUBESIN, M.W. and INOUYE, M. A theoretical study of the effect of upstream transpiration-cooling on the heat transfer and skin friction characteristics of a compressible laminar boundary layer. NACA TN. 3969, 1957.
- 1561 GRAHAL, E.W. A geometric problem related to the optimum distribution of lift on a planar wing in supersonic flow. J. Ae. Scs. 1958, 771.
- 1562 HOWARTH, L. Concerning the effect of compressibility on laminar boundary layers and their separation. Proc. Roy. Soc. A, 194 1948.
- 1563 COLES, D. The law of the wake in the turbulent boundary layer. J. Fluid Mech. 1, 1956, 191.
- 1564 BECKWITH, I.E. and GALLAGHER, J.J. Local heat transfer and recovery temperature on a yawed cylinder at a Mach number of 4.15 and high Reynolds numbers. NASA Memo 2-27-59L, 1959.
- 1565 BECKWITH, I.E. Similar solutions for the compressible boundary layer on a yawed cylinder with transpiration cooling. NACA TN. 4345, 1958.
- 1566 GOODWIN.G., CREAGER, M.O. and WINKLER, B.L. Investigation of local heat transfer and pressure drag characteristics of a yawed circular cylinder at supersonic speeds. NACA RM.A55H31, 1956.

- 1567 PENLAND, J.A. Aerodynamic characteristics of a circular cylinder at Mach number of 6.86 and angles of attack up to 90°. NACA TN.3861, 1957.
- 1568 LI.T.Y. and NAGAMATSU, H.T. Shock wave effects on the laminar skin friction of an insulated flat plate at hypersonic speeds. J. As. Scs. 20, 1953, 345.
- 1569 KENDALL, J.M. An experimental investigation of leading edge shock wave boundary layer interaction at Mach 5.8. J. Ac. Sos. 24, 1957, 47.
- 1570 LEES, L. On the boundary layer equations in hypersonic flow and their approximate solutions. J. Ae. Scs. 20, 1953, 143.
- 1571 NAGANATSU, H.T., WEIL, H.A. and SHEET, R.E. Heat transfer to flat plate in high temperature rarefied ultra-high Mach mumber flow. ARS J. 32, 1962, 533.
- 1572 CHENG, H.K., HALL, J.G., GOLIAN, T.C. and HERTZEERG, A. Boundary layer displacement and leading edge bluntness effects in high temperature hypersonic flow.

 J. Ae. Sos. 28, 1961, 353.
- 1573 HAYER.W.D. and PROBSTEIN, R.F. Viscous hypersonic similitude. J. Ac. Scs. 26, 1959, 815.
- 1574 BLOOM, M.H. and STEIGER, M.H. Inviscid flow with nonequilibrium molecular dissociation for pressure distribution encountered in hypersonic flight. J. Ae. Scs. 27, 1960, 821.
- 1575 BRAY, K.N.C. Atomic recombination in a hypersonic wind tunnel nozzle. J.Fluid Mech. 6, 1959, 1.
- 1576 INGER, G.R. Viscous and inviscid stagnation flow in a dissociated hyper-velocity free stream. Proc. 1962 Heat Transfer and Fluid Mech. Inst.
- 1577 HAYES, W.D. On hypersonic similitude. Q. App. Math. 5, 1947, 105.
- 1578 GIBSON, W.E. Dissociation scaling for nonequilibrium blunt nose flows. ARS J. 32, 1962,285.
- 1579 BIOT, M.A. Further developments of new methods in heat flow analysis. J.Ae.Scs. 26, 1959.
- 1580 BIOT, M.A. New thermo-mechanical reciprocity relations with application to thermal stress analysis. J. Ae. Scs. 26, 1959.
- 1581 JOHNS, D. J. Approximate formulas for thermal stress analysis. J. Ae. Scs. 25, 1958.
- 1582 GOODMAN, T.R. and SHEA, J.J. The melting of finite slabs. J. App. Mech. 27, 1960, 16.
- 1583 SCHNEIDER, P.J. Radiation cooling of finite heat-conducting solids. J.Ae.Scs. 27, 1960.
- 1584 JAEGER, J.C. Conduction of heat in a solid with a power law of heat transfer at its surface. Proc. Cam. Phil. S. 46, 1950, 634.
- 1585 CHAMBRE, P.L. Nonlinear heat transfer problem. J. App. Phys. 30, 1959, 1683.
- 1586 YANG, K.T. and SZEWCZYK, A. An approximate treatment of unsteady heat conduction in semi-infinite solids with variable thermal properties. J. Heat Trans. ASME Trans. 81, 1959, 251.
- 1587 BIOT, N.A. and DAUGHADAY, H. Variational analysis of ablation. J. Ae. Scs. 29, 1962.
- 1588 BENSER, W.A. Aerodynamic design of axial flow compressors, Chapter XIII Compressor operation with one or more blade rows stalled. NACA RM E56BO3b, 1956.
- 1589 HUPPERT, M.C. and BENSER, W.A. Some stall and surge phenomena in axial flow compressors.

 J. As. Scs. 1953, 835.
- 1590 STONE, A. Effects of stage characteristics and matching on axial flow compressor performance. ASME Trans. 80, 1958, 1273.
- 1591 CSANADY, G.T. An approximate equation for the 'choke line' of a compressor.

 J. Ae. Scs. 1960, 637.
- 1592 HOWELL, A.R. Design of axial compressor. Proc. I. Mech. E. 153, 1945.
- 1593 MORGAN, H.G., RUNYAM, H.L. and HUCKELL, V. Theoretical considerations of flutter at high Mach number. J. Ae. Scs. 25, 1958.
- 1594 BRATT, J.B. Wind tunnel techniques for the measurements of oscillatory derivatives.

 ARC 22, 146, 1960.
- 1595 EAST, R. A. and PENNELEGION, L. The equilibrium piston technique for gun tunnel operation.

 ARC 22, 852, 1961.
- 1596 WITTRICK, W.H. The relation between torque and rotation of a crossed flexure pivot.

 Aero. Quart. 2, 1950.

- 1597 PUGH.P.G. and WOODGATE, L. Measurement of pitching moment derivatives for blunt-nose aerofoils oscillating in two-dimensional supersonic flow. ARC 23, 012, 1961.
- 1598 STALMACH, C.J. and COOKSEY, J.M. New test techniques for a hyper-velocity wind tunnel.

 Aerospace Eng. 21, 1962.
- 1599 SACKS, A.H. Aerodynamic forces, moments and stability derivatives for slender bodies of general cross section. NACA TN.3283, 1954.
- 1600 ROSS, A.J. The calculation of lateral stability derivatives of slender wings at incidence including fin effectiveness, and correlation with experiment. RAE R.Aero.2647, 1961.
- 1601 MANGLER, K.W. and SMITH, J. Calculation of the flow past slender delta wings with leading edge separation. RAE R. Aero. 2593, 1957.
- 1602 CRABTREE, L.F. and CRANE, J. The 7 x 7 in. hypersonic wind tunnel at RAE Farnborough,
 Part 1, Design, instrumentation and flow visualization techniques. RAE TN. Aero. 2716.
- 1603 CRANE, J. The 7 x 7 in. hypersonic wind tunnel at RAE Farnborough. Part II, Heater performance. RAE TN. Aero. 2717.
- 1604 CRANE, J. The 7 x 7 in. hypersonic wind tunnel at RAE Farnborough. Part III, Calibration of the flow of the working section. RAE TN. Aero. 2718.
- 1605 WOODLEY, J.G. Pressure measurements on a cone-cylinder-flare configuration at small incidence for M=6.8. RAE IN. Aero. 2739.
- 1606 MONAGHAN, R.J. Formulae and approximations for aerodynamic heating rates in high speed flight. RAE TN. Aero. 2407.
- 1607 CHANG, C.C. and LUNDGEN, T.S. Duct flow in magnetohydrodynamics. Z. Angew. Math. Phys., 12 100-114, 1961.
- 1608 LAUFER, J. Aerodynamic noise in supersonic wind tunnels. J. Ae. Scs. 28, 1961.
- 1609 PECKMAN, D.H. On three dimensional bodies of delta planform which can support plane attached shock waves. RAE IN. Aero. 2812.
- 1610 GERSTEN, K. Corner interference effects. AGARD R.299, 1959.
- 1611 MONAGHAN, R.J. An approximate solution of the compressible laminar boundary layer on a flat plate. RAE TN. Aero. 2025.
- 1612 SQUIRE, L.C. Pressure distributions and flow patterns at M=4. on some delta wings of inverted 'v' cross section. RAE TN. Aero. 2838, 1962.
- 1613 KING-HELE, D.G., COOK, G.E. and WALKER, D.M.C. The contraction of satellite orbits under the influence of air drag. Part I. With spherically symmetrical atmosphere. RAE TN. GW.533, 1959.
- 1614 KING-HELE, D.G., COOK, G.E. and WALKER, D.M.C. The contraction of satellite orbits under the influence of air drag. Part II. With oblate atmosphere. RAE TN, GW. 565, 1960.
- 1615 KING-HELE, D.G. The contraction of satellite orbits under the influence of air drag.

 Part III. High eccentricity orbits. RAE TN. Space 1, 1962.
- 1616 GROVES, G.V. Determination of upper atmosphere air density and scale height from satellite observations. Proc. Roy. Soc. A252, 1959, 16.
- 1617 GROVES, G.V. Determination of upper atmosphere air density profile from satellite observations. Proc. Roy. Soc. A252, 1959, 28.
- 1618 MICHIELSEN, H.F. Orbit decay and prediction of the motion of artificial satellites.

 Advances in Astro. Sc. 4, Plenum Press 1959, 255.
- 1619 KING-HELE, D.G. Density of the upper atmosphere from analysis of satellite orbits: further results. Nature, 184, 1959, 1267.
- 1620 PRIESTER, W., MARTIN, H.A. and KRAMP, K. Earth satellite observations and the upper atmosphere. Nature, 188, 1960, 200.
- 1621 GROVES,G.V. Latitude and diurnal variations of air densities from 190 to 280 km as derived from the orbits of Discoverer satellites. Proc. Roy. Soc. A263, 1961, 212.
- 1622 KING-HELE, D.G., and HUGHES, K.M. Scale height in the upper atmosphere, derived from changes in satellite orbits. RAE TN. Space 4, 1962.
- 1623 TEWFIK, O.E. On the coupling between heat and mass transfer. J. Ae. Scs. 28, 1962, 1009.
- 1624 STRAND, T. Cruise performance of channel-flow ground effect machines.

 J. Ae. Scs. 29, 1962, 702.

- 1625 WHALEN, R.J. Viscous and inviscid nonequilibrium gas flows. J. Ae. Scs. 29, 1962, 1222.
- 1626 TRAUGOTT, S.C. Some features of supersonic and hypersonic flow about blunted cones.

 J. Ae. Scs. 29, 1962, 389.
- 1627 RATTAYYA.J.V. Flutter analysis of circular panels. J. Ae. Scs. 20, 1962, 534.
- 1628 GOLLNICK, A.F. Thermal effects on a transpiration cooled hemisphere. J. As. Scs. 29, 1962, 583.
- 1629 MASLEN, S.H. Second-order effects in laminar boundary layers. A.I.A.A. J. 1963, 33.
- 1630 CHENG, H.K. and CHANG, A.L. Stagnation region in rarefied high Mach number flow. A.I.A.A. J. 1963, 231.
- 1631 BAGLEY, J.A. Low speed wind tunnel tests on a two dimensional aerofoil with split flap near the ground. ARC CP.568, 1961.
- 1632 BAGLEY, J. A. and JOYCE, G.M. Calculated lift distributions in incompressible flow on some sweptback wings. RAE TN. Aero. 2836, 1962.
- 1633 PORTNOY, H. An extension of the method of generalised conical flows for lifting wings in supersonic flow. RAE TN. Aero. 2849, 1962.
- 1634 GOETZ, R.C. Effects of leading edge bluntness on flutter characteristics of some squareplanform double-wedge airfoils at a Mach number of 15.4. NASA TN. D1487, 1962.
- 1635 WEINSTEIN, I. Heat transfer and pressure distributions on a hemisphere-cylinder and a bluff afterbody model in methane-air combustion products and in air.

 NASA TN.D1503, 1962.
- 1636 LEISS, A. Pressure distribution induced on a flat plate at a free-stream Mach number of 1.39 by rockets exhausting upstream and downstream. NASA TN.D1507, 1962.
- 1637 DRISCHLER, J. A. An integral equation relating the general time-dependent lift and downwash distributions on finite wings in subsonic flow. NASA TN. D1521, 1963.
- 1638 SPENCER, B. Longitudinal aerodynamic characteristics at low subsonic speeds of a highly swept wing utilizing nose deflection for control. NASA TN.D1482, 1962.
- 1639 TOBAK, M. Analytical study of the tumbling motions of vehicles entering planetary atmospheres. NASA TN.D1549, 1962.
- 1640 CLARKSON, B.L. The design of structures to resist jet noise fatigue. J. Roy. Aero. Soc., 66, 1962, 603.
- 1641 MORLEY, L.S.D. Reduction of the clamped plate to two membrane problems with an application to uniformly loaded sectors. RAE R.Struct.277, 1962.
- 1642 SOBEY, A.J. The buckling strength of a uniform circular cylinder loaded in axial compression. RAE R. Struct, 279, 1962.
- 1643 GAUKROGER, D.R. and CURRAN, J.K. An investigation of wing-aileron flutter using ground launched rocket models. RAE TN. Struct. 308, 1962.
- 1644 LEISSA, A.W. and NIEDENFUHR, F.W. A study of the cantilever square plate subjected to a uniform loading. J. Aero. Sc. 29, 1962.
- 1645 BARON, J.R. Thermodynamic coupling in boundary layers. ARS Space Flight Rep. to the Nation, 2206-61, 1961.
- 1646 TEWFIK, O.E., ECKERT, E.R.G. and SHIRTLIFFE, C.J. Thermal diffusion effects on energy transfer in a turbulent boundary layer with helium injection to be presented at the 1962 Heat Transfer and Fluid Mech. Inst. Seattle, 1962.
- 1647 HUANG, M.K. and CONWAY, D. Bending of a uniformly loaded rectangular plate with two adjacent edges and the others either simply supported or free. J. App. Mech. 1952, 451.
- 1648 CONWAY, H.D. The approximate analysis of certain boundary value problems. J. App. Mech. 1960, 275.
- 1649 CREWE, P.R. and EGGINGTON, W.J. The hovercraft a new concept in maritime transport. Trans. Roy. Inst. Naval Arch. 1960.
- 1650 STANTON-JONES, R. Some design problems of hovercraft. Inst. Aero. Sc. Paper 61-45, 1961.
- 1651 SEBAN, R.A., EMERY, A. and LEVY, A. Heat transfer to separated and reattached subsonic turbulent flows obtained downstream of a surface step. J. Ae. Scs. 1959, 809.

- 1652 BAGLEY, J.A. Pressure distribution on two dimensional wings near the ground. RAE R. Aero. 2625, 1960.
- 1653 LUNDGEN, T.S., ATABECK, B.H. and CHANG, C.C. Transient magnetohydrodynamic duct flow. Phys. Fluids 4, 1961, 1006.
- 1654 SAKURI, A. On the propagation and structure of the blast wave. Part 1. J. Phys. Soc. Japan, 2, 1954.
- 1655 BERTRAM, M.H. and HENDERSON, A. Effects of boundary layer displacement and leading edge bluntness on pressure distribution, skin friction and heat transfer of bodies at hypersonic speeds. NACA TN. 4301, 1958.
- 1656 BRAY, K.N.C. Departure from dissociation, equilibrium in a hypersonic nozzle.

 ARC 19,983, 1958.
- 1657 GIESE, J.H. and EERGDOLT, V.E. Interferometric studies of supersonic flows about truncated cones. J.App.Phys. 24, 1953, 1389.
- 1658 GOODMAN, L.E. and RATTAYYA, J.V. Review of panel flutter and effects of aerodynamic noise, Part I, Panel flutter. App. Mech. Rev. ASME 13, 1960.
- 1659 MURRAY, J.D. Nonuniform shear flow past cylinders. Q. J. Mech. App. Math. 10, 1957, 406.
- 1660 LIGHTHILL, M.J. The fundamental solution for small steady three dimensional disturbances to a two dimensional parallel shear flow. J. Fluid Mech. 3, 1957, 113.
- 1661 LIVINGOOD, J.N.B. and DONOUGHE, P. Summary of laminar boundary layer solutions for wedgetype flow over convection and transpiration cooled surfaces. NACA TN.3588, 1955.
- 1662 STINE, H.A. and WANLASS, K. Theoretical and experimental investigation of aerodynamic-heating and isothermal heat transfer parameters on a hemisphere nose with laminar boundary layer at supersonic Mach numbers. NACA TN.3344, 1954.
- 1663 KUO, Y.H. Viscous flow along a flat plate moving at high speeds. J.Ae.Scs. 23, 1956, 125.
- 1664 MURRAY, J.D. The boundary layer on a flat plate in a stream with uniform shear. J. Fluid Mech. 11, 1961, 309.
- 1665 SYCHEV, V.V. On the theory of hypersonic gas flow with a power law shock wave. J. App. Math. Mech. 24, 1960, 756.
- 1666 FERRI, A., ZAKKAY, V. and TING, L. Blunt body heat transfer at hypersonic speed and low Reynolds number. J. Ae. Scs. 28, 1961, 862.
- 1667 CHENG, H.K. Hypersonic shock layer theory of the stagnation region at low Reynolds number. Proc. 1961 Heat Trans. and Fluid Mech. Inst. Stanford Univ. Press, 1961, 161.
- 1668 FERRI, A. and ZAKKAY, V. Measurements of stagnation point heat transfer at low Reynolds number. J. Ae. Scs. 28, 1962, 847.
- 1669 HABERMAN, WL. Subsonic potential flow past a sphere inside a cylindrical duct. J. Ae. Scs. 1962, 356.
- 1670 FERRI, A., ZAKKAY, V. and TING, L. On blunt-body heat transfer at hypersonic speed and low Reynolds numbers. J. Ae. Scs. 29, 1962, 882.
- 1671 BREBNER, G.G. and BAGLEY, J.A. Pressure and boundary layer measurements on a two dimensional wing at low speed. ARC R & M.2886, 1962.
- 1672 PANKHURST, R.C. and HOLDER, D.W. Wind tunnel techniques. Chapt.8, Tunnel interference effects. Pitwan, 1962.
- 1673 WALLACE, R. Investigation of full scale split trailing edge wing flaps with various chords and hinge locations. NACA R.539, 1935.
- 1674 BAGLEY, J.A. and BEASLEY, J.A. The shapes and lift-dependent drags on some sweptback wings designed for M_O = 1.2. RAE R. Aero. 2620, 1959.
- 1675 GARNER, H.C. and WALSHE, D.E. Pressure distribution and surface flow on 5% and 9% (Mick wings with curved tip and 60° sweepback. ARC R & M.3244, 1959.
- 1676 MUCHEMANN, D. A simple method for calculating the span- and chordwise loading on straight and swept wings of any aspect ratio at subsonic speeds. RAE R. Aero. 2476, 1962.
- 1677 MULTHOPP, H. Methods for calculating the lift distribution of wings (subsonic lifting surface theory). RAE R. Aero. 2353, 1950.
- 1678 KUCHEMANN, D. and KETTLE, D.J. The effect of end plates on swept wings. RAE R. Aero. 2429, 1952.

- 1679 WEBER, J., BREENER, G.G. and KUCHEMANN, D. Low speed tests on 45° sweptback wings.
 RAE R. Aero. 2374, 1958.
- 1680 LOMAX, H. and HEASLETT, M.A. Generalized conical flow fields in supersonic wing theory.

 NACA TN. 2497, 1951.
- 1681 LOMAX, H., HASLETT, M.A. and FULLER, F.B. Integrals and integral equations in linearized wing theory. NACA R. 1054, 1951.
- 1682 LANCE.G.N. The lift of twisted and cambered wings in supersonic flow. Aero.Quart, 6, 1955.
- 1683 SMITH, J. H. B. and MANGLER, K.W. The use of conical camber to produce flow attachment at the leading edge of a delta wing and to minimize the lift-dependent drag at sonic and supersonic speeds. RAE R. Aero. 2584, 1957.
- 1684 HEUMAN, C. Tables of complete elliptic integrals. J. Maths. Phys., 20, 1941, 127.
- 1685 HANSON, P.W. Aerodynamic effects of some configuration variables on the aeroelastic characteristics of lifting surfaces at Mach numbers from 0.7 to 6.86.

 NASA TN. D984, 1961.
- 1686 MORGAN, H.G. and MILLER, R.W. Flutter tests of some simple models at a Mach number of 7.2 in helium flow. NASA Memo 4-8-59L, 1959.
- 1667 LIGHTHILL, M.J. Oscillating airfoils at high Mach number. J. Ae. Scs. 20, 1953, 402.
- 1688 WELLS, W.R. and ARMSTRONG, W.O. Tables of aerodynamic coefficients obtained from developed Newtonian expressions for complete and partial conic and spheric bodies at combined angles of attack and sideslip with some comparisons with hypersonic experimental data.

 NASA TR R127, 1962.
- 1689 CRAWFORD, D.H. and McCAULEY, W.D. Investigation of the laminar aerodynamics heat transfer characteristics of a hemisphere cylinder in the Langley 11-inch hypersonic tunnel at a Mach number of 6.8. NACA R.1323, 1957.
- 1690 CRAWFORD, D.H. Investigation of the flow over a spiked-nose hemisphere cylinder at a Mach number of 6.8. NASA TN.D118, 1959.
- 1691 LEYHE, E.W. and HOWELL, R.R. Calculation procedure for thermodynamic transport and flow properties of the combustion products of a hydro-carbon fuel mixture burned in air with results for ethylene-air and methane-air mixtures. NASA TN. D914, 1962.
- 1692 BRESSETTE, W.E. Investigation of the jet effects on a flat surface downstream of the exit of a simulated turbojet nacelle at a free-stream Mach number 2.02. NACA RM L54E05a, 1954.
- 1693 BRESSETTE, W.E. and LEISS, A. Investigation of jet effects on a flat surface downstream of the exit of a simulated turbojet nacelle at a free-stream Mach number of 1.39.
- 1694 IEISSE & BRESSETTE Pressure distribution induced in a flat plate by a supersonic and sonic jet at a free-stream Mach number of 1.80. NACA RM L56106, 1957
- 1695 BRESSETTE, W.E. Some experiments relating to the problem of simulation of hot jet engines in studies of jet effects on adjacent surfaces at a free stream Mach number of 1.80.

 NACA RM L56E07, 1956.
- 1696 FALANGAN, R.A. and JANOS, J.J. Pressure loads produced on a flat plate wing by rocket jets exhausting in a spanwise direction below the wing and perpendicular to a free-stream flow of Mach number 2.0. NASA TN.D893, 1961.
- 1697 BRESSETTE, W.E. and LEISS, A. Effects on adjacent surfaces from the firing of rocket jets.
 NACA RM L57D19a, 1957.
- 1698 JONES, R.T. The unsteady lift of a wing of finite aspect ratio. NACA R.681, 1940.
- 1699 DRISCHLER, J.A. Approximate indical lift functions for several wings of finite span in incompressible flow as obtained from oscillatory lift coefficients.

 NACA TN.3639, 1956.
- 1700 LOMAX et al. Two and three dimensional unsteady lift problems in high speed flight. NACA R.1077, 1952.
- 1701 MAZELSKY, B. Numerical determination of indical lift of a two dimensional sinking airfoil at subsonic Mach numbers from oscillatory lift coefficients with calculations for Mach number 0.7. NACA TN.2562, 1951.
- 1702 MAZELSKY, B. and DRISCHLER, J.A. Numerical determination of indical lift and moment functions for a two dimensional sinking and pitching airfoil at Mach numbers 0.5 and 0.6. NACA TN.2739, 1952.

- 1703 KUSSNER, H.G. General airfoil theory. NACA TM. 979, 1941.
- 1704 WATKINS, C.E., WOOLSTON, D.S. and CUNNINGHAM, H.J. A systematic kernel function procedure for determining aerodynamic forces on oscillating or steady finite wings at subsonic speeds. NASA TR.R48, 1959.
- 1705 WATKINS, C.E., RUNYAN, H.L. and WOOLSTON, D.S. On the kernel function of the integral equation relating the lift and downwash distributions of oscillating finite wings in subsonic flow. NACA R. 1234, 1955.
- 1706 GARRICK, I.E. On some reciprocal relations in the theory of nonstationary flows. NACA R.629, 1938.
- 1707 HANAWALT, A.J., BLESSING, A.H. and SCHMIDT, C.M. Thermal analysis of stagnation regions with emphasis on heat-sustaining nose shapes at hypersonic speeds.

 J. Ae. Scs. 1959, 257.
- 1708 LADSON, C.L. and JOHNSTON, P.J. Aerodynamic characteristics of two winged re-entry vehicles at supersonic and hypersonic speeds. NASA TM. X346, 1961.
- 1709 OLSTAD, W.B. Static longitudinal aerodynamic characteristics at transonic speeds and angles of attack up to 99 of a re-entry glider having folding wing-tip panels.

 NASA TM.X610, 1961.
- 1710 SMITH, M.O. and CLUTTER, D.W. The smallest height of roughness capable of affecting boundary layer transition. J. Ae. Scs. 1959, 229.
- 1711 SPENCER, B. An investigation at subsonic speeds of aerodynamic characteristics at angles of attack from -4° to 100° of a delta-wing re-entry configuration having folding wingtip panels. NASA TM.X288, 1960.
- 1712 SPENCER, B. and HAMMOND, A.D. Low-speed longitudinal aerodynamic characteristics associated with a series of low-aspect ratio wings having variations in leading-edge contour. NASA TN.D1374, 1962.
- 1713 MAYO, E.E. Static longitudinal stability characteristics of a blunted glider re-entry configuration having 79.5° sweepback and 45° dihedral at a Mach number of 6.2 and angles of attack up to 20°. NASA TM.X222, 1959.
- 1714 HERRICT, J.G. Blockage corrections for three-dimensional flow closed throat wind tunnels, with considerations of the effect of compressibility. NACA R.995, 1950.
- 1715 ALLEN, H.J. Notion of a ballistic missile angularly misaligned with the flight path upon entering the atmosphere and its effect upon aerodynamic heating, aerodynamic loads and miss distance. NACA TN.4048, 1957.
- 1716 SOMMER, S.C. and TOBAK, M. Study of the oscillatory motion of manned vehicles entering the earth's atmosphere. NASA Memo 3-2-59A, 1959.
- 1717 PETERSON, V.L. Motions of a short 10° blunted cone entering a Martian atmosphere at arbitrary angles of attack and arbitrary pitching rates. NASA TN.D1326.
- 1718 ROY, M. Means and examples of aeronautical research in France at ONERA.

 J. Ae. Scs. 26, 1959, 193.
- 1719 REMMLER, K.L. Tumbling bodies entering the atmosphere. ARS Jnl. 32, 1962, 92.
- 1720 MEAD, D.J. A note on the use of sandwich structures in severe acoustic environments. Univ. Southampton R. A.A.S.U.145, 1960.
- 1721 CALLAGHAN, E.E., HOVES, W.L. COLES, W.D. and MULL, R.H. Near noise field of a jet engine exhaust. NACA R.1338.
- 1722 CLARKSON, B.L. and FORD, R.D. Random excitation of a tailplane section by jet noise. Univ. Southampton R. A.A.S.U.171.
- 1723 POWELL, A. On the fatigue failure of structures due to vibrations excited by random pressure fields. J. Acoust. Soc. America, 30, 1958, 1130.
- 1724 SCHJELDRUP, H.C. Structural acoustic proof testing. Air Eng. 1959.
- 1725 CLARKSON, B.L. and FORD, R.D. The response of a typical aircraft structure to jet noise.
 J. Roy. Aero. Soc. 1962.
- 1726 MILES, J.W. On structural fatigue under random loading. J. Ae. Scs. 21, 1954, 753.
- 1727 HESS, R.W., HERR, R.W. and MAYES, W.H. A study of the acoustic fatigue characteristics of some flat and curved aluminium panels exposed to random and discrete noise.

 NASA TN.D1.

- 1728 LIN,Y.K. Free vibrations of continuous skin stringer panels. J.App.Mech. 1960.
- 1729 LIN, Y.K. Stresses in continuous skin stiffener panels under random loading.
 J. Ae. Scs. 1962.
- 1730 WEINSTEIN, A., ROCK, D. H. On the bending of a clamped plate. Q. App. Math. 2, 1944, 262.
- 1731 DIAZ,G.B., and GREENEERG,H.G. Upper and lower bounds for the solution of the first biharmonic boundary value problem. J. Math. Phys. 27, 1948, 193.
- 1732 SOUTHWELL, R.V. On the analogues relating flexure and extension of flat plates. Q. J. Mech. App. Math. 2, 1950, 257.
- 1733 CARRIER, G.F. The bending of a sectorial plate. J. App. Nech. 11, 1944, 134.
- 1734 HASSE, H.R. The bending of uniformly loaded clamped plate in the form of a circular sector. Q. J. Mech. App. Math. 2, 1950, 271.
- 1735 CONWAY, H.D. and HUANG, M.K. The bending of uniformly loaded sectorial plates with clamped edges. J. App. Mech. 19, 1962, 5.
- 1736 WOINOWSKY-KRIEGER, S. The bending of a wedge shaped plate. J. App. Mech. 20, 1953, 77.
- 1737 MANSFIELD, E.H. On the analysis of elastic plates of variable thickness. Q. J. Mech. App. Math. 15, 1962, 167.
- 1738 FRANK, W.L. Finding zero's of arbitrary functions. J. Assoc. Comput. Mach. 5, 1958, 154.
- 1739 Yon KARMAN, T. and TSIEN, H.S. The buckling of thin cylindrical shells under axial compression. J. Ae. Scs. 8, 1941, 303.
- 1740 LEGGETT, D.M.A. and JONES, R.P.N. The behaviour of a cylindrical shell under axial compression when the buckling load has been exceeded. ARC R & M.2190, 1942.
- 1741 MICHIELSEN, H.F. The behaviour of thin cylindrical shells after buckling under axial compression. J. Ae. Scs. 15, 1948, 738.
- 1742 KEMPNER, J. Post-buckling behaviour of axially compressed circular cylinder shells.

 J. Ae. Scs. 21, 1954, 329.
- 1743 THIELMANN, W.F. New developments in the non-linear theories of the buckling of thin cylindrical shells. Aeronautics and Astronautics, p 79, Pergamon Press.
- 1744 TSIEN, H.S. Lower buckling load in the non-linear buckling theory of thin shells. Q. App. Math. 5, 1947, 236.
- 1745 ROSENEROCK, H. H. An automatic method for finding the greatest or least value function. Computer Jnl. 1960.
- 1746 BROADBENT, E.G. Aeroelastic problems in connection with high speed flight. J. Roy. Aero. Soc. 1956.
- 1747 GAUKROCER, D.R. Body freedom flutter of ground launched rocket models at supersonic and high subsonic speeds. RAE R. Struct.237, 1957.
- 1748 MINHINNICK, I.T. Subsonic aerodynamic flutter derivatives for wings and control surfaces.

 RAE R. Struct.87, 1950.
- 1749 MOLYNEUX, W.G. and HALL, H. The aerodynamic effects of aspect ratio and sweepback on wing flutter. ARC R & M. 3011.
- 1750 HALL, I.M. Transonic flow in two dimensional and axially symmetrical nozzles.

 ARC 23,347, 1961.
- 1751 COWDREY, C.F. A note on the use of end plates to prevent three dimensional flow at the ends of bluff cylinders. NPL Aero. R. 1025, 1962.
- 1752 COOKE, J.C. Slender not-so-thin wing theory. RAE R. Aero. 2660, 1962.
- 1753 GRAVITZ,S.I., LAIDLAW,W.R., BRYCE,W.W. and COOPER,R.E. Development of a quasi-steady approach to flutter and correlation with kernel-function results. J.Ae.Scs. 29, 1962, 445.
- 1754 BOURNE, D.E. and WARDLE, S. Heat transfer through laminar boundary layers on semiinfinite cylinders of arbitrary cross section. J. Ac. Scs. 29, 1962, 460.
- 1755 THOMPSON, J.S. Oscillatory derivative measurements on sting-mounted wind tunnel models: method of test and results for pitch and yaw on a cambered ogen wing at Mach numbers up to 2.6. RAE R. Aero. 2668, 1962.

- 1756 SAMSON, C.H. Further comments on the inversion of large structural matrices.

 J. Ae. Scs. 29, 1962, 373.
- 1757 ROGERS, E.W.E., HALL, I.M. and BERRY, C.J. An investigation of the flow about a plane half-wing of cropped delta planform and 6% symmetrical section at stream Mach numbers between 0.8 and 1.41. ARC R & M.3286, 1960.
- 1758 CARLSON, H.W. The lower bound of attainable sonic-boom over-pressure and design methods of approaching this limit. NASA TN.D1494, 1962.
- 1759 COLTRANE, L.C. Stability investigation of a blunted cone and a blunted ogive with a flared cylinder afterbody at Mach numbers from 0.30 to 2.85. NASA TN. D1506, 1962.
- 1760 LEE, L.H.N. Inelastic buckling of initially imperfect cylindrical shells subject to axial compression. J. Ae. Scs. 29, 1962, 87.
- 1761 YAO, J.C. Buckling of sandwich under normal pressure. J. Ae. Scs. 29, 1962, 264.
- 1762 GATEWOOD, B.E. and GEHRING, R.W. Allowable axial loads and bending moments for inelastic structures under nonuniform temperature distribution. J. Ae. Scs. 29, 1962, 513.
- 1763 WEINGARTEN, V.I. Effects of internal pressure on the buckling of circular-cylindrical shells under bending.
- 1764 LINDHOLM, U.S., KANA, D.D. and ABRAMSON, H.N. Breathing vibrations of a circular shell with an internal liquid. J. Ae. Scs. 29, 1962, 1052.
- 1765 VAFAKOS, W.P. Clamped short cval cylindrical shells under hydrostatic pressure. J. Ae.Scs. 29, 1962, 1347.
- 1766 DIXON, S.C. Experimental investigation at Mach number of 3.0 of effects of thermal stress and buckling on flutter characteristics of flat single-bay panels of length-width ratio 0.96. NASA TN.D1485, 1962.
- 1767 HOWE, P.W.H. Mathematical techniques applying to the thermal fatigue behaviour of high temperature alloys. Aero. Quart. 13, 1962, 368.
- 1768 PHILLIPS, J. Formulae for use with the fatigue load meter in the assessment of wing fatigue life. RAE TN. Struct.279, 1960.
- 1769 JOHNS, D. J. Local circumferential buckling of thin circular cylindrical shells. NASA TN. D1510, 1962, 267.
- 1770 CAPLAN, C. The flow of a compressible fluid past a sphere. NACA TN.762, 1940.
- 1771 TAMADA, K. On the flow of a compressible fluid past a sphere. Proc. Physico-Math. Soc Japan, 21, 1939, 743.
- 1772 CLARK, E.L. and ORDWAY, D.E. An experimental study of jet flap compressor blades. J. Ae. Scs. 1959, 698.
- 1773 GREEN, L. and NALL, K.L. Experiments on porous-wall cooling and flow separation control in a supersonic nozzle. J. Ae. Scs. 1959, 689.
- 1774 SAUER, R. General characteristics of the flow through nozzles at near critical speeds. NACA TM.1147, 1944.
- 1775 TOMOTIKA, S. and TAMADA, K. Studies on two dimensional flows of compressible fluid. Q. App. Math. 7, 1950, 381.
- 1776 COWDREY, C.F. and LAWES, J.A. Force measurements on square and dodecagonal sectional cylinders at high Reynolds numbers. NPL. Aero. 351.
- 1777 LIGHTHILL, M.J. A technique for rendering approximate solutions to physical problems uniformly valid. Phil. Mag. 40, 1949, 1179.
- 1778 POWELL, E.O. An integral related to the radiation integrals. Phil. Mag. 34, 1943, 600.
- 1779 DRISCHLER, J. A. Calculation and compilation of the unsteady lift functions for a rigid wing subjected to sinusoidal gusts and to sinusoidal sinking oscillations.

 NACA TN. 3748, 1956.
- 1780 NELSON, W.H. and KRUMM, W.J. The transonic characteristics of 38 cambered rectangular wings of varying aspect ratio and thickness as determined by transonic-bump technique. NACA TN.3502, 1955.
- 1781 SEWALL, J.L., HERR, R.W. and WATKINS, C.E. Use of subsonic kernel function in an influence-coefficient method of aeroelastic analysis and some comparisons with experiment. NASA TN.D515, 1960.

- 1782 QUEIIJO, M.J. and RILEY, D.R. Calculated subsonic span loads and resulting stability derivatives of unswept and 45° sweptback tail surfaces in side-slip and steady roll. NACA TN. 3245, 1954.
- 1783 GRAY, W.L. A method for calculating the steady-state loading on an airplane with a wing of arbitrary planform and stiffness. NACA TN.3030, 1953.
- 1784 BOURNE, D.E. and DAVIES, D.R. Heat transfer through the laminar boundary layer on a circular cylinder in axial incompressible flow. Q. J. Mech. App. Math. 11, 1958, 52.
- 1785 COOKE, J.C. The flow of fluid along cylinders. Q. J. Mech. App. Math. 10, 1957, 312.
- 1786 BATCHELOR, G.K. The skin friction on infinite cylinders moving parallel to their length. Q. J. Mech. App. Math. 7, 1954, 179.
- 1787 HASIMOTO, H. Rayleigh's problem for a cylinder of arbitrary shape. J. Phys. Soc. Japan, 9, 1954, 611.
- 1788 VARLEY, E. An approximate boundary layer theory for semi-infinite cylinders of arbitrary cross-section. J. Fluid Mech. 3, 1958, 601.
- 1789 BOURNE, D.E., DAVIES, D.R. and WARDLE, S. A further note on the calculation of heat transfer through the axisymmetrical laminar boundary layer on a circular cylinder. Q. J. Mech. App. Math. 12, 1959, 257.
- 1790 BEAM, B.H. A wind tunnel test technique for measuring the dynamic rotary stability derivatives at subsonic and supersonic speeds. NACA R.1258, 1956.
- 1791 TAYLOR, C.R. Measurements at Mach numbers up to 2.8 of the longitudinal characteristics of one plane and three cambered slender 'Ogee' wingr. RAE R. Aero. 2658, 1961.
- 1792 SPENCE, A. and LEAN, D. Some low speed problems of high speed aircraft. J. Roy. Aero. Soc. 66, 1962, 211.
- 1793 HALL, I.M. and ROGERS, E.W.E. The flow pattern on a tapered sweptback wing at Mach numbers between 0.6 and 1.6. ARC R & M 3271, 1960.
- 1794 HALL, I.M. and ROGERS, E.W.E. Experiments with a tapered swept-back wing of Warren 12 planform at Mach numbers between 0.6 and 1.6. ARC R & M 3271, 1960.
- 1795 HALL, I.M. The operation of the NPL 18in x 14in. wind tunnel in the transonic speed range.

 ARC CP. 338, 1957.
- 1796 RCGERS, E.W.E. and HALL, I.M. An investigation at transonic speeds of the performance of various distributed roughness bands used to cause boundary layer transition near the leading edge of a cropped delta half-wing. ARC CP.481, 1959.
- 1797 ROGERS, E.W.E., TOWNSEND, J.E.G. and BERRY, C.J. A study of the effect of leading edge modifications on the flow over a 50° sweptback wing at transonic speeds.

 ARC R & M 3270, 1960.
- 1798 HOLDER, D.W., PEARCEY, H.H. and GADD, G.E. Interaction between shock waves and boundary layers. With a note on the effects of the interaction on the performance of supersonic intakes, by J. Seddon. ARC CP. 180, 1954.
- 1799 PEARCEY, H.H., SINNOTT, C.S. and OSBORNE, J. Some effects of wind tunnel interference observed in tests on two dimensional aerofoils at high subsonic and transonic speeds.

 NPL. Aero. 373, 1959.
- 1800 ROGERS, E.W.E. and HALL, I.M. Wall interference at transonic speeds on a hemisphere cylinder model. ARC CP.510, 1959.
- 1801 PAGE, W.A. Experimental study of the equivalence of transonic flow about slender conecylinders of circular and elliptic cross section. NACA TN.4233, 1958.
- 1802 WHITHAM, G.B. The behaviour of supersonic flow past a body of revolution far from the axis. Proc. Roy. Soc. A, 201, 1950, 89.
- 1803 WALKDEN, F. The shock pattern of a wing-body combination far from the flight path. Aero. Quart. 2, 1958, 164.
- 1804 MULLENS, M.E. A flight test investigation of the sonic boom. AFFTC-TN-56-20, Air Res. and Dev. Command, U.S.A.F., 1956.
- 1805 MAGLIERI, D.J., HUBBARD, H.H. and LANSING, D.L. Ground measurements of the shock wave noise from airplanes in level flight at Mach numbers to 1.4 and at altitudes to 45,000 feet. NASA TN.D48, 1959.
- 1806 LINA, L.J. and MAGLIERI, D.J. Ground measurements of airplane shock wave noise at Mach numbers to 2, and at altitudes of 60,000 feet. NASA TN.D235, 1960.

- MAGLIERI, D. J. and HUBBARD, H. H. Ground measurements of the shock wave noise from supersonic bomber airplanes in the altitude range from 30,000 to 50,000 feet.

 NASA TN. D880, 1961.
- 1808 CARLSON, H.W. An investigation of some aspects of the sonic boom by means of wind tunnel measurements of pressures about several bodies at a Mach number of 2.01.

 NASA TN.D161, 1959.
- 1809 CARLSON, H.W. An investigation of the influence of lift on sonic-boom intensity by means of wind tunnel measurements of the pressure fields of several wing-body combinations at a Mach number of 2.01. NASA TN.D881, 1961.
- 1810 MAGLIERI, D.J. and CARLSON, H.W. The shock wave noise problem of supersonic aircraft in steady flight. NASA Memo 3-4-59L, 1959.
- 1811 MORRIS, J. An investigation of lifting effects on the intensity of sonic booms. J. Roy. Aero. Soc. 64, 1960, 610.
- 1812 COLTRANE, L.C. Investigation of two bluff shapes in axial free flight over a Mach number range from 0.35 to 2.15. NACA RM L58A16, 1958.
- 1813 NELSON, R.L. The motion of rolling symmetrical missiles referred to a body-axis system.

 NACA TN. 3737, 1956.
- 1814 TOBAK, M. and WEHREND, W.R. Stability derivatives of cones at supersonic speeds.
 NACA TN.3788, 1956.
- 1815 FISHER, L.R. and DICAMILLO, J.R. Investigation of several blunt bodies to determine transonic aerodynamic characteristics including effects of spinning and of extendible afterbody flaps and some measurements of unsteady base pressures. NASA Memo 1-21-59L, 1959.
- 1816 LETKO, W. Experimental investigation at a Mach number of 3.11 of the lift, drag and pitching moment characteristics of a number of blunt low-fineness-ratio bodies.

 NASA Memo 1-18-159L, 1959.
- 1817 HANKELMAN, G.H. and WARNER, W.H. Loading paths and the incremental stress law. J. Math. and Phys. 33, 1954, 157.
- 1818 HODGE, P.G. and WHITE, G.N. A quantitative comparison of flow and deformation theories of plasticity. J. App. Mech. 17, 1950.
- 1819 MAGASARIAN,O.L. Stresses in the plastic range around a normally loaded circular hole in an infinite sheet. J. App. Mech. 27, 1960, 65.
- 1820 BATDORF, S.B. Theories of plastic buckling. J. Ae. Scs. 16, 1949, 405.
- 1821 SHANLEY, F.R. Inelastic column theory. J. Ae. Scs. 14, 1949, 261.
- 1822 DONNELL, L.H. and WAN, C.C. Effects of imperfections on buckling of thin cylinders and columns under axial compression. J. App. Mech. 1950.
- 1823 LEE, L.H.N. and ADES, C.S. Plastic torsional buckling strength of cylinders including the effects of imperfections. J. Ae. Scs. 24, 1957, 241.
- 1824 DRUCKER, D.C. and ONAT, E.T. On the concept of stability of inelastic systems. J.Ae.Scs., 21, 1954, 543.
- 1825 ONAT, E.T. and DRUCKER, D.C. Inelastic instability and incremental theories of plasticity.

 J. Ae. Scs. 20, 1953, 181.
- 1826 REISSNER, E. Small bending and stretching of sandwich type shells. NACA TN.1832, 1949.
- 1827 KAPLAN, A. and FUNG, Y.C. A nonlinear theory of bending and buckling of thin elastic shallow spherical shells. NACA TN. 3212, 1954.
- 1828 REISSNER, E. Stresses and small displacements of shallow spherical shells.

 J. Math. Phys. 25, 1946, 80.
- 1829 DONNEL, L.H. Stability of thin-walled tubes under torsion. NACA R.479, 1933, 12.
- 1830 REISS, E.L., GREENBERG, H.J. and KELLER, H.B. Nonlinear deflections of shallow spherical shells. J. Ae. Scs. 24, 1957, 533.
- 1831 REISS, E.L. Buckling of shallow shells under external pressure. J. App. Mech. 25, 1958,556.
- 1832 WEGSTEIN, J.H. Accelerating convergence of iteration processes. Comm. of Assoc. for Computing Mach. 1958.

- 1833 KLEIN, B. A simple method of matric structural analysis, Part IV, Non-linear problems. J. Ae. Scs. <u>26</u>, 1959.
- 1834 RIEDINGER, L.A. Limit design for economical missile structures. Space-Aero., 1958.
- 1835 GATEWOOD, B.E. The problem of strain accumulation under thermal cycling. J. Ae. Scs. 27, 1960.
- 1836 SPRAGUE, G.H. and HUANG, P.C. Analytical and experimental investigation of stress distributions in long plates subjected to longitudinal loads and transverse temperature gradients. WADC TR 55-350, 1955.
- 1837 PADLOG, J., HUFF, R.D. and HOLLOWAY, G.F. Inelastic behaviour of structures subjected to cyclic thermal and mechanical stressing conditions. WADC TR 60-271, 1960.
- 1838 DOW, M.B. and PETERSON, J.P. Bending and compression tests of pressurised ring-stiffened cylinders. NASA TN. D360, 1960.
- 1839 SUER, H.S., HARRIS, L.A., SKENE, W.T. and BENJAMIN, R.J. The bending stability of thinwalled unstiffened circular cylinders including the effects of internal pressure. J. Ae. Scs. 25, 1958.
- 1840 STEIN, M. and HEDGEPETH, J.M. Analysis of partly wrinkled membrane. NASA TN. D813, 1961.
- 1841 SEIDE, P. and WEINGARTEN, V.I. On the bending of circular cylindrical shells under pure bending. J. App. Mech. 28, 1961.
- 1842 MORLEY, L.S.D. An improvement on Donnell's approximation for thin-walled circular cylinders. Q. J. Mech. App. Math. 12, 1959.
- 1843 BATDORF, S.B. A simplified method of elastic stability analysis for thin cylindrical shells. I. Donnell's equation. NACA TN. 1341, 1947.
- 1844 ARNOLD, R.N. and WARBURTON, G.B. Flexural vibrations of the walls of thin cylindrical shells having freely supported ends. Proc. Roy. Soc. A, 197, 1949, 238.
- 1845 ARNOLD, R.N. and WARBURTON, G.B. The flexural vibrations of thin cylinders. J. Proc. I. Mech. E. 167, 1953, 62.
- 1846 FUNG, Y.C., SECHLER, E.E. and KAPLAN, A. On the vibration of thin cylindrical shells under internal pressure. J. Ae. Scs. 27, 1957, 650.
- 1847 GOTTENBERG, W.G. Experimental study of the vibrations of a circular cylindrical shell. J. Acoust. Soc. America, 32, 1960, 1002.
- 1848 BERRY, J.G. and REISSNER, E. The effect of an internal compressible fluid column on the breathing vibrations of a thin pressurised cylindrical shell. J. Ae. Scs. 25, 1958, 288.
- 1849 TOBIAS, S.A. Theory of imperfection for the vibrations of elastic bodies of revolution. Engineering, 172, 1951, 409.
- 1850 KEMPNER, J. Remarks on Donnell's equations. J. App. Mech. 22, 1955.
- 1851 KEMPNER, J. Energy expressions and differential equations for stress and displacement analysis of arbitrary cylindrical shells. J. Ship Res. 1958, 8.
- 1852 ROMANO, F. and KEMPNER, J. Stress and displacement analysis of a simply supported noncircular cylindrical shell under lateral pressure. PIBAL R.415, 1958.
- 1853 HOFF, N.J. The accuracy of Donnell's equations. J. App. Mech. 22, 1955.
- 1854 HOFF, N.J. Boundary-value problems of the thin-walled circular cylinder. J. App. Mech. 21, 1954.
- 1855 POHLE, F.V. and NARDO, S.W. Simplified formulas for boundary value problems of the thinwalled circular cylinder. J. App. Mech. 22, 1955.
- 1856 SYLVESTER, M.A. and BAKER, J.E. Some experimental studies of panel flutter at Mach 1.3. NACA TN.3914, 1957.
- 1857 SYLVESTER, M.A. Experimental studies of flutter of buckled rectangular panels at Mach numbers from 1.2 to 3.0 including effects of pressure differential and of panel width-length ratio. NASA TN.D833, 1961.
- 1858 DIXON, S.C., GRIFFITH, G.E. and EOHON, H.K. Experimental investigation at Mach numbers 3.0 of the effects of thermal stress and buckling on the flutter of four-bay aluminium alloy panels with length-width ratios of 10. NASA TN. D921, 1961.
- 1859 GUY, L.O. and SOHON, H.L. Flutter of aerodynamically heated aluminium-alloy and stainless steel panels with length-width ratio of 10 at Mach 3.0. NASA TN. D1353, 1962.

- 1860 HELDENFELS, R.S., ROSECRANS, R. and GRIFITH, G.E. Test of an aerodynamically heated multiweb wing structure (MW-1) in a free jet at Mach number 2. NACA RM L53E27, 1953.
- 1861 LEE, D.B. and FAGET, M.A. Charts adapted from Van Driest's turbulent flat-plate theory for determining values of turbulent aerodynamic friction and heat transfer coefficients.

 NACA TN. 3811, 1956.
- 1862 STEIN, M. The phenomenon of change in buckle pattern in elastic structures.

 NASA TR R39, 1959.
- 1863 STEIN, M. Loads and deformations of buckled rectangular plates. NASA TR R40, 1959.
- 1864 LEONARD, R.W. and HEDGEPETH, J.M. Status of flutter of flat and curved panels.
 NACA RM L57D24c, 1957.
- 1865 GLENNY, E. and TAYLOR, T.A. A study of the thermal fatigue behaviour of metals. The effect of test conditions on nickel-base high temperature alloys. J.Inst. Metals, 88, 1960, 449.
- 1866 GRAHAM, A. and WALLES, K.F.A. Regularities in creep and hot fatigue data. ARC CP.379, 1958.
- 1867 LANDAU, C.S. Low frequency fatigue of Nimonic 90. Low frequency fatigue a rheological approach. Engineer, 213, 1962, 911.
- 1868 STANWORTH, C.G. and PAINE, D.S.C. Design and operation of the N.G.T.E. Thermal shock analogue. ARC CP.557, 1960.
- 1869 HOWE, P.W.H. The calculation of transient temperature in turbine blades and tapered discs using Biot's variational method. ARC CP.617, 1961.
- 1870 FREUDENTHAL, A.M. Effect of rheological behaviour on thermal stresses. J.App.Phys. 25, 1954.
- 1871 WEERTMAN, J. Steady-state creep through dislocation climb. J.App. Phys. 28, 1957, 362.
- 1872 BIOT, M.A. Fundamentals of boundary layer heat transfer with streamwise temperature variations. J. Ae. Scs. 1962, 558.
- 1873 BIOT, M.A. Lagrangian thermodynamics of heat transfer in systems including fluid motion. J. Ae. Scs. 29, 1962, 568.
- 1874 DUNCAN, W.J. The use of models for the determination of critical flutter speeds.

 ARC R & M.1425, 1931.
- 1875 TEMPLETON, H. Models for aeroelastic investigation. ARC CP.255, 1955.
- 1876 LAMBOURNE, N.C. and SCRUTON, C. On flutter testing in high speed wind tunnels.

 ARC R & M.3054, 1956.
- 1877 MANSFIELD, E.H. The influence of aerodynamic heating on the flexural rigidity of a thin wing. ARC R & M.3115, 1957.
- 1878 MOLYNEUX, W.G. Experimental model techniques and equipment for flutter investigations. Air. Eng. 1958.
- 1879 TARGOFF, W.P. and WHITE, R.P. Flutter model testing at transonic speeds. Inst. Aero. Sc. Perp. 706, 1957.
- 1880 McCARTHY, J.F. and HALFMAN, R.L. The design and testing of supersonic flutter models. J. Ae. Scs. 1956.
- 1881 MINER, M.A. Cumulative damage in fatigue. J. App. Mech. 12, 1945, A159.
- 1882 BULLEN, N.I. The variation of gust frequency with gust velocity and altitude.

 ARC CP. 324, 1956.
- 1883 HEYWOOD, R.B. Correlated fatigue data for aircraft structural joints. ARC CP.227, 1955.
- 1884 CHILVER, A.H. The estimation of fatigue damage on structural elements. RAE TN.Struct.125, 1954.
- 1885 HOFF, N.J. Buckling of thin cylindrical shells under hoop stresses varying in axial direction. J. App. Mech. 24, 1957, 405.
- 1886 ZUK, W. Thermal buckling of clamped cylindrical shells. J. Ae. Scs. 24, 1957, 359.
- JOHNS, D.J. Buckling due to thermal stress of cylindrical shells subjected to axial temperature distributions. CoA R.147, 1961.
- 1888 ANDERSON, M.S. Combinations of temperature and axial compression required for buckling of a ring-stiffened cylinder. NASA TN.D1224, 1962.

- FATEORF, S.B. A simplified method of elastic stability analysis for thin cylindrical shells. NACA R.874, 1947.
- 1890 JOHNS, D.J. Comments on thermal buckling of clamped cylindrical shells. J. Ae. Scs. 26, 1959, 58.
- 1891 ALMROTH, B.O. and BRUCH, D.O. Buckling of a finite length cylindrical shell under a circumferential band of pressure. J. Ae. Scs. 28, 1961, 573.
- 1892 JONES, W.P. Research on unsteady flow. J. Ae. Scs. 29, 1962, 249.
- 1893 SALTER, C. A new design of pitot-static tube with a discussion of pitot-static tubes and their calibration factors. NPL. Aero. R.1013, 1962.
- 1894 SMITH, G.E. Flutter of a two dimensional simply supported buckled panel with elastic restraint against edge displacement. RAE R. Struct. 269, 1961.
- 1895 DAVIES, D.E. The airforces on the low aspect ratio rectangular wing oscillating in sonic flow. RAE TN. Struct. 311, 1962.
- 1896 SEAL, D.M. The calculation of loads on a supersonic weapon in the steady circling case. RAE TN. Struct. 275, 1960.
- 1897 KEMPNER, J. Some results on buckling and postbuckling of cylindrical shells. NASA TN.D1510, 1962, 173.
- 1898 SEIDE, P. A survey of buckling theory and experiment for circular conical shells of constant thickness. NASA TN.D1510, 1962, 401.
- 1899 MOORE, F.K. Aerodynamic effects on boundary layer unsteadiness. 6th A.A. Aero. Conf. 1957.
- 1900 LAMBOURNE, N.C. and BRYER, D.W. Some measurements in the vortex flow generated by a sharp leading edge having 65° sweep. ARC CP.477, 1959.
- 1901 BROWN, C.E. and MICHAEL, W.H. Long slender delta wings with leading edge separation. NACA TN. 3430.
- 1902 LAMBOURNE, N.C. Some current and proposed investigations into the flow for slender delta and other wings in unsteady motion. ARC 21,844, 1960.
- 1903 ECKHAUS, W. Two dimensional transonic unsteady flow with shock waves. M.I.T. Fluid Dynamics Res. Group R.59-3, 1959.
- 1904 KETTLE, D.J. Calibration of the standard pitot-static head used in the RAE low speed wind tunnels. RAE Tech. Memo 222, 1951.
- 1905 MERRIAM, K.G. and SPAULDING, E.R. Comparative tests of pitot-static tubes. NACA TN.546.
- 1906 FOLSON, R.G. Review of the pitot tube. ASME Trans. 70, 1956.
- 1907 ROUSE, H. and McNOWN, J.S. Cavitation and pressure distribution on various load forms. Studies in Eng. Bulletin 32, Inst. Hydr. Res. State Univ. of Iowa, 1948.
- 1908 CRANDAL, S. Random vibration. App. Mech. Rev. 12, 1959, 739.
- 1909 CLARKSON, B.L. The effect of jet noise on aircraft structures. Aero. Quart. 10, 1959, 103.
- 1910 AYRE, R.S. and JACOBSON, L.S. Natural frequencies of continuous beams of uniform span length. J. App. Mech. 17, 1950, 391.
- 1911 CLARKSON, B.L. and FORD, R.D. Experimental study of the random vibrations of an aircraft structure excited by jet noise. U.S.A.A. R.128, Univ. of Southampton, 1960.
- 1912 HAMA, P.R., RECESSO, J.V. and CHRISTIAENS, J. The axisymmetric free-convection temperature field along a vertical thin cylinder. J. Ae. Scs. 1959, 335.
- 1913 MILES, J.W. Vibrations of beams on many supports. Fice. A.S.C.E., 82, 1956.
- 1914 ISAACS, R.P. Transtability flutter of supersonic aircraft panels. Rand Corp. R. P48750.
- 1915 HAYES, W. A buckled plate in a supersonic stream. N.A.A. Inc. R.AL1029.
- 1916 LANDAHL, M.T. The flow around oscillating low aspect ratio wings at transonic speeds. KTH Aero. TN.40, 1954.
- 1917 MANGER, K.W. A method of calculating the short period longitudinal stability derivatives of a wing in linearised unsteady compressible flow. RAE R. Aero. 2468.
- 1948 Miles, J.W. On the low aspect ratio oscillating rectangular wing in supersonic flow.

 Aero. Quart. 4, 1953, 231.

- 1919 LANDAHL, M.T. Theoretical studies of unsteady transonic flow. Part III, The oscillating low aspect ratio rectangular wing. Aero. Res. Inst. of Sweden (FFA) R.79, 1958.
- 1920 STEWARTSON, K. Supersonic flow over an inclined wing of zero aspect ratio. Proc. Cam. Phil. Soc. 46, 1950, 307.
- 1921 ADAMS, M.C. and SEARS, W.R. Slender body theory review and extension. J.Ae.Scs. 29, 1953.
- 1922 LIGHTHILL, M.J. Supersonic flow past slender bodies of revolution, slope of whose median section is discontinuous. Q. J. Mech. App. Math. 1948, 90.
- 1923 NIELSEN, J.N. and KAATTARI, G.E. Methods for estimating lift interference of wing-body combinations at supersonic speeds. NACA TIL. 2950.
- 1924 NIELSEN, J.N., KAATTARI, G.E. and ANASTASIO, R.F. A method for calculating the lift and centre of pressure of wing-body-tail combinations at subsonic, transonic speeds.

 NACA TIL. 3959.
- 1925 HENDERSON, A. and BERTRAM, M.H. Factors affecting loads at hypersonic speeds. NACA TIL. 5576.
- 1926 KEMPNER, J. Post buckling behaviour of circular cylinderical shells under hydrostatic pressure. J. Ae. Scs. 24, 1957, 253.
- 1927 MELLO, J.F. Investigation of normal force distributions and wake vortex characteristics of bodies of revolution at supersonic speeds. J. Ae. Scs. 1959, 155.
- 1928 DONNELL, L.H. A new theory for the buckling of thin cylinders under axial compression and bending. ASME Trans. 56, 1934, 795.
- 1929 MARGUERRE, K. Stability of the cylindrical shell of variable curvature. NACA TM.1302,1951.
- 1930 DILL, E.H. General theory of large deflections of thin shells with special applications to conical shells. NASA TN.D826, 1961.
- 1931 KEMPNER, J. Stability equations for conical shells. J. Ae. Scs. 25, 1958, 137.
- 1932 LACKMAN, L. and PENZIEN, J. Buckling of circular cones under axial compression. J. App. Mech. 27, 1960, 458.
- 1933 COLEMAN, W.S. The characteristics of roughness from insects as observed for two-dimensional, incompressible flow past airfoils. J. Ae. Scs. 1959, 264.
- 1934 MUSHTARI, K.M. and SACHENKOV, A.V. Stability of cylindrical and conical shells of circular cross section, with simultaneous action of axial compression and external normal pressure. NACA TM.1433, 1958.
- 1935 RADKOWSKI, P.P. Buckling of thin single and multi-layer conical and cylindrical shells with rotationally symmetrical stresses. Proc. 3rd U.S. Nat. Cong. App. Mech. 1958, 443.
- 1936 SEIDE, P. A Donnell-type theory for asymmetrical bending and buckling of thin conical shells. J. App. Mech. 24, 1957, 547.
- 1937 SEIDE, P. On the buckling of truncated conical shells in torsion. J. App. Mech. 28, 1962.
- 1938 SEIDE, P. Calculations for the stability of thin conical frustums subjected to external uniform hydrostatic pressure and axial loads. J. Ae. Scs. 29, 1962, 951.
- 1939 BERLOT, R.R. Some explicit solutions for constant-temp. magnetogas-dynamic channel flow. J. Ae. Scs. 29, 1962, 748.
- 1940 NASH, J.F. The effect of an initial boundary layer on the development of a turbulent free shear layer. NPL. Aero. R1019, 1962.
- 1941 WILLIAMS, J.C. Viscous compressible and incompressible flow in slender channels. A.I.A.A. J. 1963, 186.
- 1942 WALKER, R.E., STONE, A.R. and SHANDOR, M. Secondary gas injection in a conical rocket nozzle. AIAA Jnl. 1, 1963, 334.
- 1943 DENISON, M.R. and BAUM, E. Compressible free shear layer with finite initial thickness. AIAA Jnl. 1, 1963, 342.
- 1944 WELLS, W.R. and McLELLAN, C.H. One dimensional heat conduction through the skin of a vehicle upon entering a planetary atmosphere at constant velocity and entry angle. NASA TN.D1476, 1962.
- 1945 STOCKMAN, N.O. and KRAMER, J.L. Method for design of pump impellers using a high speed digital computer. NASA TN.D1562, 1963.
- 1946 ROMEO, D.J. and STERRETT, J.R. Exploratory investigation of the effect of a forward facing jet on the bow shock of a blunt body in a Mach number 6 free stream. NASA TN.D1605, 1963

- 1947 FOHRMAN, M.J. Static aerodynamic characteristics of a short blunt 10° semi-vertex angle cone at a Mach number of 15 in helium. NASA TN.D1648, 1963.
- 1948 BOHON, H.L. Panel flutter tests on full scale X-15 lower vertical stabilizer at Mach number of 3.0. NASA TN.D1385, 1962.
- 1949 JORGENSEN, L. H. and BAUM, G. M. Charts for equilibrium flow properties of air in hyper-velocity nozzles. NASA TN. D1333, 1962.
- 1950 JAHSMAN, W.E. and FIELD, F.A. Comparison of theoretical and experimental creep buckling times of initially straight, centrally loaded columns. J. Ae. Scs. 29, 1962, 431.
- 1951 STOWELL, E.Z. and THEIN WAH. A unified theory of creep buckling under normal loads. J. Ae. Scs. 29, 1962, 658.
- 1952 SERPICO, J.C. Study of creep collapse of a long circular cylindrical shell under various distributed force systems. J. Ae. Scs. 29, 1962, 1316.
- 1953 ARMENAKAS, A.E. Vibrations of infinitely long cylindrical shells under initial stress. AIAA Jnl. 1, 1963, 100.
- 1954 JOHNS, R.H., MORGAN, W.C. and SPERA, D.A. Analysis of stress at several junctions in pressurized shells. AIAA Jnl. 1, 1963, 455.
- 1955 McCOMB, H.G., ZENDER, G.W. and MIKULAS, M.M. The membrane approach to bending instability of pressurized cylindrical shells. NASA TN.D1510, 1962, 229.
- 1956 HARRIS, L.A. and BAKER, E.G. Elastic stability of simply supported corrugated core sandwich cylinders. NASA TN.D1510, 1962, 331.
- 1957 NEWMAN, M. and REISS, E.L. Axisymmetric snap buckling of conical shells. NASA TN.D1510, 1962, 451.
- 1958 BERNER, F. and CAMAC, M. Air scooping vehicle. Avco-Everett Res. Lab. R.76, 1959.
- 1959 LARSON, H.K. Heat transfer in separated flows. J. Ae. Scs. 1959, 731.
- 1960 LIEPHAN, H.W. and LAUFER, J. Investigation of free turbulent mixing. NACA TN. 1257, 1947.
- 1961 KORST, H.H. and PAGE, R.H. Compressible two dimensional jet mixing at constant pressure.
 Univ. of Illinois Dept. Mech. Eng. ME-TN-392-1, 1954.
- 1962 LIGHTHILL, M.J. Contributions to the theory of heat transfer through a laminar boundary layer. Proc. Roy. Soc. A, 202, 1950, 359.
- 1963 AGRAWAL, H.C. A variational principle for convection of heat II. J. Math. Mech. 9, 1960.
- 1964 RIVAS, M.A. and SHAPIRO, A.H. On the theory of discharge coefficients for round entrance flowmeters and venturis. ASME Trans. 78, 1956, 489.
- 1965 SIMMONS, F.S. Analytical determination of discharge coefficients for flow nozzles. NACA TN.3449, 1955.
- 1966 MASLEN, S.H. On fully developed channel flows; some solutions and limitations, and effects of compressibility, variable properties, and body forces. NASA TR. R34, 1959.
- 1967 MARTIN, D.E. A study of laminar compressible viscous pipe flow accelerated by a body force, with application to magnetogas-dynamics. NASA TN. D855, 1961.
- 1968 SUTTON, G.P. Rocket propulsion systems for interplanetary flight. J.Ae.Scs. 26, 1959.
- 1969 LIEPMAN, H.P. On the use of side-jets as control devices. ARS Jnl. 29, 1959, 453.
- 1970 JANOS, J.J. Loads induced on a flat plate wing by an air jet exhausting perpendicularly through the wing and normal to a free-stream flow of Mach number 2.0.

 NASA TN. D649, 1961.
- 1971 CUBRISON, R.W., ANDERSON, B.H. and WARD, J.J. Surface pressure distributions with a sonic jet normal to adjacent flat surfaces at Mach 2.92 to 6.4. NASA TN. D580, 1961.
- 1972 ROMEO, D. J. and STERRETT, J.R. Aerodynamic interaction effects ahead of a sonic jet exhausting perpendicularly from a flat plate into a Mach number 6 free stream.

 NASA TN. D743, 1961.
- 1973 VINSON, P.W., AMICK, J.L. and LIEPMAN, H.P. Interaction effects produced by jet exhausting laterally near base of ogive-cylinder model in supersonic main stream.

 NASA Nemo 12-5-58W, 1959.
- 1974 WU,J.M., CHAPKIS,R.L. and MAGER,A. An approximate analysis of thrust vector control by fluid injection. ARS Jnl. 1677-1685, 1961.

- 1975 EGGERS, A.J. One dimensional flows of an imperfect diatomic gas. NACA R.959, 1950.
- 1976 LEES, L. and HROMAS, L. Turbulent diffusion in the wake of a blunt nosed body at hypersonic speeds. J. Ae. Scs. 29, 1962, 976.
- 1977 GOLDSTEIN,S. Concerning some solutions of the boundary layer equations in hydrodynamics. Proc. Cam. Phil. Soc. <u>26</u>, 1930, 18.
- 1978 DEWEY, C.F., SCHLESINGER, S.I. and SAHSKIN, L. Temperature profiles in a finite solid with moving boundary. J. Ae. Scs. 27, 1960.
- 1979 LEHNERT, R. and SCHERMERHORN, V.L. Correlation of base pressure and wake structure of sharp and blunt-nose cones with Reynolds number based on boundary layer momentum thickness.

 J. Ae. Scs. 26, 1959.
- 1980 HILLS, P.R. A method of computing the transient temperature of thick walls from arbitrary variation of adiabatic wall temperature and heat transfer coefficient.

 NACA R. 1372. 1958.
- 1981 EERGLES, A.E. and KAYE, J. Solutions to the heat conduction equation with timedependent boundary conditions. J. Ae. Scs. 28, 1961, 251.
- 1982 SUTTON,G.W. The temperature history in a thick skin subjected to laminar heating during entry into the atmosphere. Jet. Prop. 28, 1959, 40.
- 1983 DETRA, R.W., KEMP, N.H. and RIDDELL, F.R. Addendum to 'Heat transfer to satellite vehicles re-entering the atmosphere'. Jet. Prop. 27, 1957, 1256.
- 1984 HAMRICK, J.T. Method of analysis for compressible flow through mixed-flow centrifugal impellers of arbitrary design. NACA R. 1082, 1952.
- 1985 SMITH, K.J. and HAMRICK, J.T. A rapid approximate method for the design of hub shroud profiles of centrifugal impellers of given blades shapes. NACA TN. 3399, 1955.
- 1986 OSBORN, W.M. Design and test of mixed-flow impellers. VIII Comparison of experimental results for three impellers with shroud redesigned by rapid approximate method.

 NACA RM E56LO7, 1957.
- 1987 WU,C. A general theory of three dimensional flow in subsonic and supersonic turbomachines of axial-radial and mixed-flow types. NACA TN.2604, 1952.
- 1988 KRAMER, J. J. Non-viscous flow through a pump impeller on a blade-to-blade surface of revolution. NASA TN. D1108, 1962.
- 1989 KRAMER, J.J. Incompressible nonviscous flow through a pump rotor with splitter vanes.
 NASA TN.D1186, 1962.
- 1990 STANITZ, J.D. and PRIAN, V.D. A rapid approximate method for determining velocity distribution on impeller blades of centrifugal compressors. NACA TN. 2421, 1951.
- 1991 LOPATOFF, M. Wing-flow study of pressure drag reduction at transonic speed of projecting a jet of air from the nose of a prolate spheroid of fineness ratio 6.

 NACA RM L51E09, 1951.
- 1992 LOVE, E.S. The effects of a small jet of air exhausting from the nose of a body of revolution in supersonic flow. NACA RM L52I19a, 1952.
- 1993 HAYMAN, L.O. and McDEARMON, R.W. Jet effects on cylindrical afterbodies housing sonic and supersonic nozzles which exhaust against a supersonic stream at angles of attack from 90° to 180°. NASA TN. D1016, 1962.
- 1994 CHARCZENKO, N. and HENNESSY, K.W. Investigation of a retrocket exhausting from the nose of a blunt body into a supersonic free stream. NASA TN.D751, 1961.
- 1995 PETERSON, V.L. and McKENZIE, R.L. Effects of simulated retrockets on the aerodynamic characteristics of a body of revolution at Mach numbers from 0.25 to 1.90.

 NASA TN.D1300.
- 1996 STERRETT, J.R. and EMERY, J.C. Extension of boundary layer separation criteria to a M=6.5 utilizing flat plates with forward-facing steps. NASA TN.D618, 1960.
- 1997 LOVE, E.S. Experimental and theoretical studies of axisymmetric free jets. NASA TR.R6, 1959.
- 1998 AMES Research Staff. Equations, tables and charts for compressible flow. NACA R.1135, 1953.

- 1999 TREON,S.L. Static aerodynamic characteristics of short blunt cones with various nose and base cone angles at Mach numbers of 0.6 to 5.5 and angles of attack to 180°.

 NASA TN.D1327, 1962.
- 2000 INTRIERI, P.F. Free-flight measurements of the static and dynamic stability and drag of a 10° blunted cone at Mach numbers 3.5 and 8.5. NASA TN. D1299, 1962.
- 2001 WEHREND, W.R. Wind tunnel investigation of the static and dynamic stability characteristics of a 10° semivertex angle blunted cone. NASA TN. D1202, 1962.
- 2002 BOGDONOFF, S.M. and VAS, I.E. Preliminary investigations of spiked bodies at hypersonic speeds. J. Ae. Scs. 1959, 65.
- 2003 ERICKSON, W.D. Real gas correction factors for hypersonic flow parameters in helium. NASA TN. D462, 1960.
- 2004 HENDERSON, A. and SWALLEY, F.E. Effects of air contamination in a helium tunnel.
 NASA TN.D406, 1960.
- 2005 MARGOLIS, K. Theoretical evaluation of the pressures, forces and moments at hypersonic speeds acting on arbitrary bodies of revolution undergoing separate and combined angle of attack and pitching motions. NASA TN. D652, 1961.
- 2006 INOUYE, M. and LOMAX, H. Comparison of experimental and numerical results for the flow of a perfect gas about blunt-nosed bodies. NASA TN. D.1426, 1962.
- 2007 MUELLER, J.N. Equations tables and figures for use in the analysis of helium flow at supersonic and hypersonic speeds. NASA TN. D4063, 1957.
- 2008 KORDES, E.E. Flutter research on skin panels. NASA TN. D451, 1960.
- 2009 ERICKSON, W.D. and CREEKMORE, H.S. A study of equilibrium real-gas effects in hypersonic air nozzles, including charts of thermodynamic properties for equilibrium air.

 NASA TN. D231, 1960.
- 2010 YOSHIKAWA, K.K. and KATZEN, E.D. Charts for air-flow properties in equilibrium and frozen flows in hypervelocity nozzles. NASA TN. D693, 1961.
- 2011 MOECKEL, W.E. and WESTON, K.C. Composition and thermodynamic properties of air in chemical equilibrium. NACA TN.4265, 1958.
- 2012 SHANLEY, F.R. "Principles of creep buckling" Weight-strength analysis of aircraft structures. Part III. p 343, McGraw-Hill, New York, 1952.
- 2013 GERARD, G. A creep buckling hypothesis. J. Ae. Scs. 23, 1962, 879.
- 2014 RABOTNOV, G.N. and SHESTERIKOV, S.A. Creep stability of columns and plates. J. Mech. Phys. Solids, 6, 1957, 27.
- 2015 MARIN, J. Creep deflections in columns. J. App. Phys. 18, 1947.
- 2016 CARLSON, R.L. Time-dependent tangent modulus applied to column creep buckling.

 J. App. Mech. 23, 1956, 390.
- 2017 GERARD, G. Note on creep buckling of columns. J. Ae. Scs. 19, 1962, 714.
- 2018 MATHAUSER, E.E. and BROOKS, W.A. An investigation of the creep lifetime of 75 S-T6 Aluminium alloy columns. NACA TN. 3204, 1954.
- 2019 CARLSON, R.L. and MANNING, G.K. Investigation of compressive creep properties of aluminium columns at elevated temperature. Part 2, Stability problems. WADC TR.52-251, 1954.
- 2020 PATEL, S.A., KEMPNER, J.M., ERICKSON, B. and MOCASSERY, A.H. Correlation of creep-buckling tests with theory. NACA RM C.20, 1956.
- 2021 WAH, T. and GREGORY, R.K. Creep collapse of long cylindrical shells under high temperature and external pressures. J. Ae. Scs. 28, 1961.
- 2022 ERICKSON, B., PATEL, S.A., FRENCH, F.W., LEDERMAN, S. and HOFF, N.J. Experimental investigation of creep bending and buckling of thin circular cylindrical shells. NACA RM 57E17, 1957.
- 2023 HOFF, N.J. Creep buckling. Aero. Quart. 1956.
- 2024 KEMONER, J. Creep buckling and buckling of linearly viscoelastic columns. NACA TN.3136.
- 2025 LIBOVE, C. Creep buckling of columns. J. Ae. Scs. 19, 1952, 459.

- 2026 PATEL, S.A. Buckling of columns in the presence of creep. Aero. Quart. 1956.
- 2027 STOWELL, E.Z. A phenomenological relation between stress, strain rate and temperature for metals at elevated temperature. NACA TN.4000, 1957.
- 2028 STOWELL, E.Z. A phenomenological theory for the transient creep of metals at elevated temperature. NACA TN.4396, 1958.
- 2029 SANDERS, J.L., McCOMB, H.G. and SCHLECHTE, F.R. A variational theorem for creep with applications to plates and columns. NACA R. 1342, 1958.
- 2030 PEARSON, C.E. General theory of elastic stability. Q. App. Math. 14, 1956, 133.
- 2031 BORESI, A.P. A refinement of the theory of buckling of rings under uniform pressure.

 J. App. Mech., 22, 1955, 85.
- 2032 BODNER, S.R. On the conservativeness of various distributed force systems. J. Ae. Scs. 25, 1958.
- 2033 TIMOSHENKO,S. Theory of elastic stability, p 222, The design of tubes under uniform external pressure on the basis of assumed inaccuracies. McGraw, 1936.
- 2034 HOFF, N.J., JAHSMAN, W.H. and NACHBAR, W. A study of creep collapse of a long circular cylindrical shell under uniform external pressure. J. Ae. Scs. 26, 1959.
- 2035 STEVENS, G.W.H. The stability of a compressed elastic ring. Q. J. Mech. App. Math. 5, 1955, 221.
- 2036 REISSNER, E. On transverse vibrations of thin, shallow elastic shells. Q. App. Math., 13, 1955, 169.
- 2037 HERMANN, G. and ARMENAKAS, A.E. Shear buckling of bars. J. App. Mech. 27, 1960, 455.
- 2038 JOHNS, R.H. and ORANGE, T.W. Theoretical elastic stress distributions arising from discontinuities and edge loads in several shell-type structures. NASA TR.R103, 1961.
- 2039 MORGAN, W.C. and BIZON, P.T. Experimental investigation of stress distributions near abrupt change in wall thickness in thin-walled pressurized cylinders.

 NASA TN. D1200, 1962.
- 2040 BLOOM, M.H. and PALLONE, A. Shroud tests of pressure and heat transfer over short afterbodies with separated wakes. J. Ae. Scs. 1959, 626.
- 2041 TAYLOR, C.E. and WENK, E. Analysis of stress in the conical elements of shell structures.

 DTMB R. 981, 1956.
- 2042 FLUGGE, W. Stress problems in pressurized cabins. NACA TN.2612, 1952.
- 2043 RADKOWSKI, P.P., DAVIS, R.M. and BOLDUC, M.R. A numerical analysis of the equations of thin shells of revolution. ARS Prep. 1580-60, 1960.
- 2044 REISSNER, E. On the theory of thin elastic shells. Reissner Anniv. 1949, 231.
- 2045 ZENDER, G.W. The bending strength of pressurized cylinders. J. Ae. Scs. 29, 1962, 362.
- 2046 BATDORF, S.B. Critical stress of thin-walled cylinders in torsion. NACA TN. 1344, 1947.
- 2047 DUNCAN, W.J. Galerkin's method in mechanics and differential equations. ARC R & M 1798, 1937.
- 2048 STEIN, M. and MAYERS, J. A small deflection theory for curved sandwich plates. NACA R. 1008, 1951.
- 2049 LIBOVE, C. and HUBKA, R.E. Elastic constants for corrugated core sandwich plates. NACA TN. 2289, 1951.
- 2050 STEIN, M. and MAYERS, J. Compressive buckling of simply supported curved plates and cylinders of sandwich construction. NACA TN.2601, 1952.
- 2051 HARRIS, L.A. The stability of thin-walled unstiffened circular cylinders under axial compression including the effects of internal pressure. J. Ae. Scs. 24, 1957, 587.
- 2052 NASH, W.A. Recent advances in the buckling of thin shells. App. Mech. Rev. 13, 1960, 161.
- 2053 KELLER, H.B. and REISS, E.L. Spherical cap snapping. J. Ae. Scs. 26, 1959, 643.
- 2054 KELLER, H.B. and REISS, E.L. Iterative solutions for the non-linear bending of circular plates. Comm. on Pure and App. Math. 11, 1958, 273.
- 2055 KELLER, H.B. and REISS, E.L. Non-linear bending and buckling of circular plates. Proc. 3rd U.S. Nat. Cong. App. Mech. 1958, 375.

- 2056 NEWMAN, M. and FORRAY, M. Axisymmetric large deflections of circular plates subjected to thermal and mechanical load. Proc. Aerospace Forum I Session, S.M.F. Paper FF-30, 1962, 56.
- 2057 ALMEN, J.O. and LASZLO, A. The uniform section disk spring. ASME Trans. 58, 1936, 305.
- 2058 WEMPNER, G.A. The conical disk spring. Proc. 3rd U.S. Nat. Cong. App. Mech. 1958, 473.
- 2059 SCHMIDT, R. and WEMPNER, G.A. The nonlinear conical spring. ASME Trans. 81E, 1959, 681.
- 2060 KELLER, H.B., KELLER, J.B. and REISS, E.L. Buckled states of circular plates. Q. App. Math. 20, 1962, 55.
- 2061 VASILU, J. Turbulent mixing of a rocket exhaust jet with a supersonic stream including chemical reactions. J. Ae. Scs. 29, 1962, 19.
- 2062 WILBY, P.G. An experimental and theoretical investigation of second-order wing-body interference at high Mach number. FFA R.91, Stockholm, 1962.
- 2063 RUSSELL, D.B. On obtaining solutions to the Navier-Stokes equations with high speed digital computers. ARC 23,797, 1962.
- 2064 WINSTON, M.M. Propeller slipstream effects as determined from wing pressure distribution on a large-scale six-propeller VTOL model at static thrust. NASA TN.D1509, 1962.
- 2065 WINTERS, C.W., WITTE, W.G., RASHIS, B. and HOPKO, H.N. A free-flight investigation of ablation of a blunt body to a Mach number of 13.1. NASA TN.D1500, 1962.
- 2066 HANSON, P.W. and DOGGETT, R.V. Wind tunnel measurements of aerodynamic damping derivatives of a launch vehicle vibrating in free-free bending modes at Mach numbers from 0.70 to 2.87 and comparisons with theory. NASA TN.D1391, 1962.
- 2067 GERARD,G. Plastic stability theory of geometrically orthotropic plates and cylindrical shells. J. Ae. Scs. 29, 1962, 956.
- 2068 HERRMANN,G. Instability analysis of cylindrical shells under hydrostatic pressure.
 NASA TN.D1510, 1962, 239.
- 2069 CUNNINGHAM, J.H. and JACOBSON, M.J. Design and testing of honeycomb sandwich cylinders under axial compression. NASA TN.D1510, 1962.
- 2070 SINGER, J. Buckling of orthotropic and stiffened conical shells. NASA TN. D1510, 1962,463.
- 2071 MESCALL, J. Stability of thin torispherical shells under uniform internal pressure.
 NASA TN.D1510, 1962, 671.
- 2072 MARBLE, F. and ADAMSON, T.C. Ignition and combustion in a laminar mixing zone. Jet Prop. 24, 1954, 85.
- 2073 CRANK, J. and NICOLSON, P. A practical method for numerical evaluation of solutions of partial differential equations of the heat-conduction type. Proc. Cam. Phil. Soc., 43, 1947, 50.
- 2074 LANDAHL, M., DROUGGE, G. and BEANE, B.J. Theoretical and experimental investigation of second-order supersonic wing-body interference. J. Ae. Scs. <u>27</u>, 1960, 694.
- 2075 WILBY, P.G. An experimental and theoretical investigation of second-order supersonic wing-body interference, for a non-lifting body with wings at incidence.

 Aero. Res. Inst. of Sweden, FFA R.87, 1960.
- 2076 BERTRAM, M.H. An approximate method for determining the displacement effects and viscous drag of laminar boundary layers in two dimensional hypersonic flow. NACA TN.2773,1952.
- 2077 TRIMPI,R.L. and JONES,R.A. A method of solution with tabulated results for the attached oblique shock wave system for surfaces at various angles of attack, sweep, and dihedral in an equilibrium real gas including the atmosphere. NASA TR.R63, 1960.
- 2078 APPELT, C.J. The steady flow of a viscous fluid past a circular cylinder at Reynolds numbers 40 and 44. ARC R & M.3175.
- 2079 NICLEY, W.G. Finite difference formulae for the square lattices. Q.J.Mech. App. Math, 1, p35.
- 2080 JENSON, V.G. Viscous flow round a sphere at low Reynolds numbers. Proc. Roy. Soc. A, 249, 1959, 346.
- 2081 KAWAGUTI, M. Numerical solution of the Navier-Stokes equations for the flow around a circular cylinder at Reynolds number 40. J. Phys. Soc. Japan, 8, 1953, 747.

- 2082 LESTER, W.G.S. The flow past a pitot tube at low Reynolds numbers. Pt.1, The numerical solution of the Navier-Stokes equations for steady viscous axisymmetric flow. Pt. 2, The effects of viscosity and orifice size on a pitot tube at low Reynolds numbers. ARC 22,070.
- 2083 THOM, A. An investigation of fluid flow in two dimensions. ARC R & M 1194, 1928.
- 2084 THOM, A. The flow past circular cylinders at low speeds. Proc. Roy. Soc. A, 141, 1933,651.
- 2085 THOM, A. and APELT, C.J. Note on the convergence of numerical solutions of the Navier-Stokes equations. ARC R & M 3061, 1956.
- 2086 WOODS, L.C. A note on the numerical solution of fourth order differential equations.

 Aero. Quart. 5, 1954, 176.
- 2087 FRANKEL, S.P. Convergence rates of iterative treatments of partial differential equations.

 Math. Tables Aids Comp. 4, 1950, 65.
- 2088 YOUNG, D. Iterative treatments for solving partial difference equations of elliptic type. Trans. Amer. Math. Soc. <u>76</u>, 1954. 92.
- 2089 KIRBY, R.H. Aerodynamic characteristics of propeller-driven VTOL aircraft. NASA TN.D730, 1961.
- 2090 CURRIE, M.M. and DUNSBY, J.A. Pressure distribution and force measurements on a VTOL tilting wing-propeller model. Pt.II, Analysis of results. Rep. LR-284, Nat. Res. Council of Canada, 1960.
- 2091 HUSTON, R.J. and WINSTON, M.M. Data from a static thrust investigation of a large scale general research VTOL-STOL model in ground effect. NASA TN. D397, 1960.
- 2092 ROBINSON, R.G. and HERRNSTEIN, W.H. Wing-nacelle-propeller interference for wings of various spans. Force and pressure distribution tests. NACA R. 569, 1936.
- 2093 SPREEMANN, K.P. Induced interference effects on jet and buried-fan VTOL configurations in transition. NASA TN. D731, 1961.
- 2094 KUHN, R.E. Investigation of the effects of ground proximity and propeller position on the effectiveness of a wing with large chord slotted flaps in redirecting propeller slipstream downward for vertical take-off. NACA TN. 3629, 1956.
- 2095 KUHN, R.E. and DRAPER, J.W. Investigation of effectiveness of large-chord slotted flaps in deflecting propeller slipstreams downward for vertical take-off and low-speed flight. NACA TN.3364, 1955.
- 2096 RASHIS, B., WITTE, W.G. and HOPKO, R.N. Qualitative measurements of the effective heats of ablation of several materials in supersonic air jets at stagnation temperature up to 11,000° F. NACA RM. L58E22, 1958.
- 2097 BOND, A.C., RASHIS, B. and LEVIN, L. Experimental ablation cooling. NACA RM.L58E15a, 1958.
- 2098 RASHIS, B. and WALTON, T.E. An experimental investigation of ablating material at low and high enthalpy potentials. NASA TM.X-263, 1960.
- 2099 ROBERTS, L. A theoretical study of stagnation point ablation. NASA TR.R9, 1959.
- 2100 RASHIS, B. and HOPKO, R.N. An analytical investigation of ablation. NASA TM.X-300, 1960.
- 2101 WINTERS, C.W. and BRACALENTE, E.M. A sensor for obtaining ablation rates. NASA TN.D800, 1961.
- 2102 SWANGON, A.G. A five-stage solid fuel sounding rocket system. NASA Memo 3-6-59L, 1959.
- 2103 The Rocket Panel: pressures, densities and temperatures in the upper atmosphere. Phys. Rev. <u>88</u>, 1952, 1027.
- 2104 STONEY, W.E. Aerodynamic heating of blunt nose shapes at Mach numbers up to 14. NACA RM.L58E05a, 1958.
- 2105 FULLER, F.B. Numerical solutions for supersonic flow of an ideal gas around blunt two-dimensional bodies. NASA TN.D791, 1961.
- 2106 McCONNELL, D.G. Free-flight observation of a separated turbulent flow including heat transfer up to Mach 8.5. NASA TN.D278, 1961.
- P107 TENDELAND, T. The flow field over blunted flat plates and its effect on turbulent boundary growth and heat transfer at a Mach number of 4.7. NASA TN.D689, 1961.
- 2108 VanDYKE, M.D. A study of second-order supersonic flow theory. NACA R.1071, 1952.

- 2109 YANG, K-T. Unsteady laminar compressible boundary layers on an infinite plate with suction or injection. J. Ae. Scs. 1959, 653.
- 2110 STONE, AH. On supersonic flow past a slightly yawing cone. J. Math. Phys. 27, 1948, 67.
- 2111 GARRICK, I.E. Some research on high speed flutter. 3rd A.A. Aero. Conf. 1951, 419.
- 2112 BOND, R. and PACKARD, B.B. Unsteady aerodynamic forces on a slender body of revolution in supersonic flow. NASA TN. D859, 1961.
- 2113 OLSSON, C.O. and ORLIK-RUCKEMAN, K. An electronic apparatus for automatic recording of the logarithmic decrement and frequency for oscillations in the audio and subaudio frequency range. Aero. Res. Inst. Sweden, R.52, 1954.
- 2114 COE, C.F. Steady and fluctuating pressures at transonic speeds on two space-vehicle payload shapes. NASA TM.X503, 1961.
- 2115 RAINEY, A.G. Measurements of aerodynamic forces for various mean angles of attack on an airfoil oscillating in pitch and on two finite-span wings oscillating in bending with emphasis on damping in the stall. NACA R.1305, 1957.
- 2116 BECKER, H. General instability of stiffened cylinders. NACA TN. 4237, 1958.
- 2117 HESS, T.E. Stability of orthotropic cylindrical shells under combined loading. ARS J. 31, 1961.
- 2118 BECKER, H. and GERARD, G. Elastic stability of orthotropic shells. J. Ae. Scs. 29, 1962.
- 2119 GERARD, G. Plastic stability theory of thin shells. J. Ae. Scs. 24, 1957.
- 2120 STOWELL, E.Z. A unified theory of plastic buckling of columns and plates. NACA R.898, 1948.
- 2121 GERARD,G. Compressive and torsional buckling of thin-wall cylinders in the yield region. NACA TN.3726, 1956.
- 2122 GERARD, G. On the role of initial imperfections in plastic buckling of cylinders under axial compression. J. Ae. Scs. 29, 1962.
- 2123 LOO,T.T. An extension of Donnell's equations for a circular cylindrical shell. J. Ae. Scs. 1957, 390.
- 2124 STRAND, T. Design of missile bodies for minimum drag at very high speeds thickness ratio, lift, and center of pressure given. J. Ae. Scs. 1959, 568.
- 2125 WINDENBURG, D.F. and TRILLING, C. Collapse by instability of thin cylindrical shells under external pressure. ASIE Trans. 56, 1934, 819.
- 2126 CUNNINGHA, J.H. An engineer's conceptual approach to the buckling of cylindrical shell (axial loading). IAS Paper 62-106, 1962.
- 2127 HOFF, N.H. and MAUTNER, S.F. The buckling of sandwich type panels. J. Ae. Scs. 1945, 285.
- 2128 YUSEFF, S. Face wrinkling and core strength in sandwich construction. J. Ae. Scs. 1960.
- 2129 BODNER, S.R. General instability of a ring stiffened circular cylindrical shellunder hydrostatic pressure. J. App. Mech. <u>24</u>, 1957, 269.
- 2130 BECKER, H. Handbook of structural stability. Pt.VI. Strength of stiffened curved plates and shells. NACA TN.3786, 1958.
- 2131 SINGER, J. The effect of axial constraint on the instability of thin conical shells under external pressure. J. App. Mech. 29, 1962, 212.
- 2132 GALLETLY, D.D. General instability of ring stiffened cylindrical shells subject to external hydrostatic pressure a comparison of theory and experiment.

 J. App. Mech. 25, 1958, 259.
- 2133 GALLETLY, G.D. and REYNOLDS, T.E. A simple extension of Southwell's method for determining the elastic general instability pressure of ring stiffened cylinders subject to external hydrostatic pressure. Proc. Soc. Exp. Stress Analysis, 13, 1956, 141.
- 2134 GALLETLY, G.D. Torispherical shells a caution to designers. J. Eng. for Industry, 1959, 51.
- · 2155 DRUCKER, D.C. and SHEELD, R.T. Limit analysis of symmetrically loaded thin shells of revolution. J. App. Mech. <u>26</u>, 1959, 61.
 - 2136 SHIELD, R.T. and DRUCKER, D.C. Design of thin walled torispherical and toriconical pressure vessel heads. J. App. Mech. 28, 1961, 292.

- 2137 CLARK, R.A. On the theory of thin elastic toroidal shells. J. Math. Phys. 29, 1950, 146.
- 2138 CLARK, R.A. Asymptotic solutions of toroidal shell problems. Quart. App. Math. 16, 1958, 47.
- 2139 POND, H.L. The effect of entrance velocity on the flow of a rarefied gas, through a tube.

 J. Ae. Scs. 29, 1962, 917.
- 2140 CHAUDHURY, P.R. Shock-standoff distance for spherical bodies at high Mach numbers. J. Ae. Scs. 29, 1962, 745.
- 2141 ZEISBERG, S.L. The wake behind an oscillating vehicle. J. Ae. Scs. 29, 1962, 1344.
- 2142 TIRUMALESA, D. and SATYANARAYANA, B. Effect of wall divergence on sonic flows in solid wall tunnels. J. Roy. Soc. 66, 1962, 125.
- 2143 CUNNINGHAM, B.E. and KRAUS, S. A one-foot hypervelocity shock tunnel in which high-enthalpy real gas flows can be generated with flow times of about 180 milliseconds.

 NASA TN.D1428, 1962.
- 2144 NEWSOM, W.A. and TOSTI, L.P. Slipstream flow around several tilt-wing VTOL aircraft models operating near the ground. NASA TN.D1382, 1962.
- 2145 BRUSH, D.O. and ALEROTH, B.O. Buckling of core-stabilised cylinders under axisymmetric external loads. J. Ae. Scs. 1962, 1164.
- 2146 ANDERSON, M.S. Thermal buckling of cylinders. NASA TN. D1510, 1962, 255.
- 2147 STALDER, J.R. and JUKOFF, D. Heat transfer to bodies travelling at high speed in the upper atmosphere. NACA R.944, 1948.
- 2148 DeMARCUS, W.C. and HOPPER, E.H. Knudsen flow through a circular capillary. J. Chem. Phys. 23, 1955, 1344.
- 2149 STOJANOVIC, D. Similar temperature boundary layers. J. Ae. Scs. 1959, 571.
- 2150 McISAAC, J., POND, H. and STERGIS, G. Preliminary results of density measurements from an Air Force satellite. Air Force, Cam. Lab., 63-24, 1963.
- 2151 KENDALL, J.M. Experiments on supersonic blunt-body flows. J.P.L. Prog. R. 20-372, 1959.
- 2152 LIN, C.C. On periodically oscillating wakes in the Oseen Approximation. Studies in Maths. and Mechs., presented to R. Von Mises, p 170, Academic Press, 1954.
- 2153 SPREITER, J.R., SMITH, D.W. and HYETT, B.J. A study of simulation of flow with free stream Mach number one in a choked wind tunnel. NASA TR.R73, 1960.
- 2154 BERNDT, S.B. On the influence of wall boundary layers in closed transonic test section. FFA R.71, Stockholm, 1957.
- 2155 PETERSOHN, E.G.M. Some experimental investigations on the influence of wall boundary layers upon wind tunnel measurements at high subsonic speeds. FFA R.44, Stockholm, 1952.
- 2156 JONES, J.J. Experimental investigation of attenuation of strong shock waves in a shock tube with hydrogen and helium as driver gases. NASA TN.4072, 1957.
- 2157 NAGAMATSU, H.T. Hypersonic shock tunnel. ARS J. 29, 1959, 332.
- 2158 WITTLIFF, C.E. The tailor-interfaced hypersonic shock tunnel. J. Ae. Scs. 26, 1959, 219.
- 2159 CUNNINGHAM, B.E. and KRAUS, S. Experimental investigation of the effect of yaw on rates of heat transfer to transverse circular cylinders in a 6500-foot-pre-second hypersonic air stream. NACA RM.A58E19, 1958.
- 2160 GROSS, R.A. and OPPENHEIM, A.K. Recent advances in gaseous detonation. ARS J. 29, 1959, 173
- 2161 ROSE, P.H. and STARK, W.I. Stagnation point heat transfer measurements in dissociated air. J. Ae. Sos. 26, 1958, 86.
- 2162 TOSTI, L.P. Force-test investigation of the stability and control characteristics of a 1/8 scale model of a tilt-wing vertical take-off and landing airplane.

 NASA TN. D44, 1960.
- 2163 NEWSON, W.A. and TOSTI, L.P. Force-test investigation of the stability and control characteristics of a $\frac{1}{4}$ scale model of a tilt-wing vertical take-off and landing aircraft. NASA Memo 11-3-58L, 1959.
- 2164 NEWSON, W.A. Effect of ground proximity on the aerodynamic characteristics of a fourengined vertical take-off and landing transport airplane model with tilting wing and propellers. NACA TN.4124, 1957.

- 2165 O'BRYAN, T.C. An investigation of the effect of downwash from a VTOL aircraft and a helicopter in the ground environment. NASA TN. D977, 1961.
- 2166 KUHN, R.E. An investigation to determine conditions under which downwash from VTOL aircraft will start surface erosion from various types of terrain. NASA TN. D56. 1959.
- 2167 O'BRYAN, T.C. An experimental study of the effect of downwash from a twin-propeller VTOL aircraft on several types of ground surfaces. NASA TN. D1239, 1962.
- 2168 FEGG, R.J. Damage incurred on a tilt-wing multipropeller VTOL/STOL aircraft operating over a level, gravel-covered surface. NASA TN.D535, 1960.
- 2169 REEDER, J.P. Handling qualities experience with several VTOL research aircraft. NASA TN.D735, 1961.
- 2170 WARD, J.F. Structural loads surveys on two tilt-wing VTOL configurations. NASA TN. D729, 1961.
- 2171 SEIDE, P. The stability under axial compression and lateral pressure of circular cylindrical shells with a soft elastic core. J. Ae. Scs. 1962.
- 2172 GOREE, W.S. and NASH, W.A. Elastic stability of circular cylindrical shells stabilized by a soft elastic core. Exp. Mech. 2, 1962.
- 2173 WEINGARTEN, V.I. The buckling of cylindrical shells under longitudinal varying loads. ASLE Paper 61-WA-153, 1961.
- 2174 LANGHAAR, H.L. General theory of buckling. App. Mech. Rev. 11, 1958.
- 2175 BIJLAARD, P.P. Stresses from local loadings in cylindrical pressure vessels. ASNE Trans. 77, 1953.
- 2176 PETERSON, J.P. Bending tests of ring-stiffened circular cylinders. NACA TN. 3735, 1956.
- 2177 PRIDE, R.A. Effects of rapid heating on strength of airframe components. NACA TN. 4051, 1957.
- 2178 ANDERSON, M.S. and CARD, M.F. Buckling of ring-stiffened cylinders under a pure bending moment and a nonuniform temperature distribution. NASA TN.D1513, 1962.
- 2179 SWIGART, R.J. A theory of asymmetric hypersonic blunt-body flows. AIAA Jnl. 1963, 1034.
- 2180 LIBBY, P.A. and SCHETZ, J.A. Approximate analysis of the slot injection of a gas in laminar flow. AIAA Jnl. 1963, 1056.
- 2181 CHU, C.K. and LYNN, Y.M. Steady magnetohydrodynamic flow past a non-conducting wedge. AIAA Jnl. 1963, 1062.
- 2182 JOSSON, R.L. An approximate solution for laminar boundary layer flow. AIAA Jnl. 1963, 1088.
- 2183 LYKOUDIS, P.S. Laminar hypersonic trail in the expansion-conduction region. AIAA Jnl. 1963, 772.
- 2184 STEIGER, M.H. and BLOOM, M.H. Three dimensional effects in viscous wakes. AIAA Jnl. 1963,776
- 2185 CULICK, F.E.C. An integral method for calculating heat and mass transfer in laminar boundary layers. AIAA Jnl. 1963, 783.
- 2186 MESSITER, A.F. Lift of slender delta wings according to Newtonian theory. AIAA Jnl. 1963,795
- 2187 ARENS, M. and SPIEGLER, E. Shock-induced boundary layer separation in overexpanded conical exhaust nozzles. AIAA Jnl. 1963, 578.
- 2188 GOEBEL, T.P. MARTIN, J.J. and BOYD, J.A. Factors affecting lift-drag ratios at Mach numbers from 5 to 20. AIAA Jnl. 1963, 640.
- 2189 CAPIAUX, R. and WASHINGTON, M. Nonequilibrium flow past a wedge. AIAA Jnl. 1963, 1, 650.
- 2190 ZIERING, S. Flow of a gas near a solid surface. AIAA Jnl. 1, 1963, 661.
- 2191 HICKMAN, R.S. and GIEDT, W.H. Heat transfer to a hemisphere-cylinder at low Reynolds numbers. AIAA Jnl. 1, 1963, 665.
- 2192 YEN, Shee-Mang and THYSON, N.A. An integral method for calculation of supersonic laminar boundary layer with heat transfer on yawed cone. AIAA Jnl. 1, 1963, 672.
- 2193 EHRICH, F.F. Some exact solutions for cavitating curvilinear bodies. AIAA Jnl. 1963, 675.

- 2194 CUMBERBATCH, E., SARASON, L. and WEITZNER, H. Magnetohydrodynamic flow past a thin airfoil.
 AIAA Jnl. 1, 1963, 679.
- 2195 DOSANJH, D.S. and SHEERLAN, W.J. Experiments with two-dimensional, transversely impinging jets. AIAA Jnl. 1963, 1, 329.
- 2196 MURPHY, C.H. and DICKINSON, E.R. Growth of the turbulent wake behind a supersonic sphere.
 AIAA Jnl. 1, 1963, 339.
- 2197 YATES, J.E. and ZEYDEL, E.F.E. Unsteady aerodynamic forces on slender supersonic aircraft with flexible wings and bodies. AIAA Jnl. 1, 1963, 350.
- 2198 OGUCHI, H. The blunt-leading-edge problem in hypersonic flow. AIAA Jnl. 1, 1963, 361.
- 2199 FREEDMAN, S.I., RADBILL, J.R. and KAYE, J. Theoretical investigations of a supersonic laminar boundary layer with foreign-gas injection. AIAA Jnl. 1, 1963.
- 2200 TIEN, C.L. and GEE, C. Hypersonic viscous flow over a sweat-cooled flat plate.
 AIAA Jnl. 1963, 159.
- 2201 MIELE, A.A. A study of slender shapes of minimum drag using the Newton-Busemann pressure coefficient law. AIAA Jnl. 1, 1963, 168.
- 2202 WALLACE, J. and CLARKE, J.H. Uniformly valid second-order solution for supersonic flow over cruciform surfaces. AIAA Jnl. 1, 1963, 179.
- 2203 GUNDERSEN, R.M. The propagation of a nonuniform magnetohydrodynamic shock wave into a moving monatomic fluid. J. Ae. Scs. 1962, 1421.
- 2204 BLOXSON, D.E. and RHODES, B.V. Experimental effect of bluntness and gas rarefaction on drag coefficients and stagnation heat transfer on axisymmetric shapes in hypersonic flow. J. Ae. Scs. 1962.
- 2205 DUNLAP, R. and KUETHE, A.M. Effects of cooling on boundary layer transition on a hemisphere in simulated hypersonic flow. J. Ae. Scs. 1962, 1454.
- 2206 CUMBERBATCH, E. Magnetohydrodynamic Mach cones. J. Ae. Scs. 1962, 1476.
- 2207 MAGER, A. Supersonic airfoil performance with small heat addition. J. Ae. Scs. 1959, 99.
- 2208 EHLERS, F.E. and SHOEMAKER, E.M. A linearized analysis of the forces exerted on a rigid wing by a shock wave. J. Ae. Scs. 1959, 75.
- 2209 VIDAL, R.J. Aerodynamic processes in the downwash-impingement problem. J. Ae. Scs. 1962.
- 2210 RYHMING, I.L. On slender airfoil theory for nonequilibrium flow. J.Ae.Scs. 1962, 1076.
- VAN DRIEST, E.R. and BLUMER, C.B. Boundary layer transition at supersonic speeds-three-dimensional roughness effects (spheres). J. Aerospace Sc. 1962, 909-16.
- 2212 GODDARD, F.E. Effect of uniformly distributed roughness on turbulent skin-friction drag at supersonic speeds. J. Ae. Scs. 1959, 1.
- 2213 WITTCLIFF, C.E. and WILSON, M.R. Heat transfer to slender cones in hypersonic flow, including effects of yaw and nose bluntness. J. Ae. Scs. 1962, 761.
- 2214 ROBERTSON, J.M. and CLARK, M.E. The drag of elongated bodies over a wide Reynolds number range. J. Ae. Scs. 1962, 842.
- 2215 ROTT, N. and LENARD, M. The effect of slip particularly for highly cooled walls. J. Ae. Scs. 1962, 591.
- 2216 VASILU, J. Pressure distribution in regions of step-induced turbulent separation. J. Ae. Scs. 1962, 596.
- 2217 LEVINSKY, E.S. Application of inequality constraints to variational problems of lifting re-entry. J. Ae. Scs. 1962, 400.
- 2218 GEIGER, R.E. Experimental lift and drag of a series of glide configurations at Mach numbers 12.6 and 17.5. J. Ae. Scs. 1962, 410.
- 2219 ERYSON, A.E. Determination of lift or drag programs to minimize re-entry heating.

 J. Ae. Scs. 1962, 420.
- 2220 TANI, I. and KOMODA, H. Boundary layer transition in the presence of streamwise vortices.

 J. Ae. Scs. 1962, 440.
- 2221 TANI, I. Steady flow of conducting fluids in channels under transverse magnetic fields, with consideration of Hall effect. J. Ae. Scs. 1962, 297.

- 2222 HAINS, F.D. and HOLER, Y.A. Axisymmetric magnetohydrodynamic channel flow. J. Ae. Scs. 1962, 143.
- 2223 STRAND, T. Inviscid-incompressible-flow theory of static two-dimensional solid jets, in proximity to the ground. J. Ae. Scs. 1962, 170.
- 2224 VAGLIO-LAURIN, R. On the PLK method and the supersonic blunt-body problem.
 J. Ae. Scs. 1962, 185.
- 2225 McLAFFERTY, G.H. and BARBER, R.E. The effect of adverse pressure gradients on the characteristics of turbulent boundary layers in supersonic streams.

 J. Ae. Scs. 1962, 29, 1.
- 2226 FAULDERS, C.R. Heat transfer in the laminar boundary layer with ablation of vapor of arbitrary molecular weight. J. Ae. Scs. 1962, 76.
- 2227 NALEID, J.F. and THOMPSON, M.J. Pressure-gradient effects on the Preston tube in supersonic flow. J. Ae. Scs. 1961, 940.
- 2228 BROWER, W.B. Leading-edge separation of laminar boundary layers in supersonic flow. J. Ae. Scs. 1961, 957.
- 2229 BOGDONOFF, S.M. and VAS, R.E. The effect of sweep angle on hypersonic flow over blunt wings. J. Ae. Scs. 1961.
- 2230 NAGAMATSU, H.T., WORKMAN, J.B. and SHEER, R.E. Hypersonic nozzle expansion of air with atom recombination present. J. Ae. Scs. 1961, 833.
- 2231 CHAPKIS,R.L. Hypersonic flow over an elliptic cone: theory and experiment. J. Ae. Scs. 1961, 844.
- 2232 EHRICH, F.F. The curtain jet. J. Ae. Scs. 1961, 855.
- 2233 RHYMING, I.L. Supersonic shear flow past an airfoil between two parallel walls. J. Ae. Scs. 1961, 861.
- 2234 HOLT, M. Direct calculation of pressure distribution on blunt hypersonic nose shapes with sharp corners. J. Ae. Scs. 1961, 872.
- 2235 YEN, K.T. and TOBA, K. A theory of the two dimensional laminar boundary layer over a curved surface. J. Ae. Scs. 1961, 877.
- 2236 BUSH, W.B. The stagnation point boundary layer in the presence of an applied magnetic field. J. Ae. Scs. 1961, 610.
- 2237 NESS, N. Foreign-gas injection into a compressible turbulent boundary layer on a flat plate. J. Ae. Scs. 1961, 645.
- 2238 LYKOUDIS, P.S. The Newtonian approximation in magnetic hypersonic stagnation-point flow. J. Ae. Scs. 1961.
- 2239 CHEN, C.F. and CLARKE, J.H. Body under lifting wing. J. Ae. Scs. 1961, 547.
- 2240 PALLONE, A. Nonsimilar solutions of the compressible laminar boundary layer equations with applications to the upstream-transpiration cooling problem.
 J. Ae. Scs. 1961, 449.
- 2241 DENISON, M.R. The turbulent boundary layer on chemically active ablating surfaces. J. Ae. Scs. 1961.
- 2242 SHEN, S.F. Some considerations on the laminar stability of time-dependent basic flows. J. Ae. Scs. 1961, 397.
- 2243 RYHMING, I.L. and YOLER, Y.A. Supersonic boom of wing-body configurations.
 J. Ae. Scs. 1961, 313.
- 2244 CHENG, Sin-I. On the aerodynamic noise of a turbulent jet. J. Ae. Scs. 1961, 321.
- 2245 SEDNEY, R. Some aspects of nonequilibrium flows. J. Ae. Scs. 1961, 189.
- 2246 KILAKOWSKI,L.J. and HASKELL,R.N. Solution of subsonic nonplanar lifting surface problems by means of high-speed digital computers. J. Ae. Scs. 1961.
- 2247 RYHMING, I.L. The supersonic boom of a projectile related to drag and volume.

 J. Ae. Scs. 1961.
- 2248 WALDMAN, G.D. and PROBSTEIN, R.F. An analytic extension of the shock-expansion method. J. Ae. Scs. 1961, 119.
- 2249 YOSHIHARA, H. Plasma flow over a thin charged conductor. J. Ae. Scs. 1961, 141.

- 2250 BLOOM, M.H. and RUBIN, S. High-speed viscous corner flow. J. Ae. Scs. 1961, 145.
- 2251 STEWARTSON, K. Viscous flow past a quarter infinite plate. J. Ae. Scs. 1961, 1.
- 2252 DORRANCE, W.H. On the approach to chemical and vibrational equilibrium behind a strong normal shock wave. J. Ae. Scs. 161, 43.
- 2253 WU, Ching-Sheng. Hypersonic viscous flow near the stagnation point in the presence of magnetic field. J. Ae. Scs. 1960, 882.
- 2254 ESCHENROEDER, A.Q. Combustion in the boundary layer on a porous surface. J. Ae. Scs. 1960, 901.
- 2255 HUNZIKER, R.R. The flow about a charged body moving in the lower atmosphere.

 J. Ae. Scs. 1960.
- 2256 FUNG, Y.C. Fluctuating lift and drag acting on a cylinder in a flow at supercritical Reynolds numbers. J. Ae. Scs. 1960, 801.
- 2257 DAIBER, J.W. An optical boundary-layer probe. J. Ae. Scs. 1960.
- 2258 TEWFIK, O.K. and GIEDT, W.H. Heat transfer, recovery factor and pressure distributions around a circular cylinder normal to a supersonic rarefied air stream.

 J. Ae. Scs. 1960, 721.
- 2259 REVELL, J.D. Second-order theory for unsteady supersonic flow past slender pointed bodies of revolution. J. Ae. Scs. 1960, 730.
- 2260 ROTT, N. and ROSENZWEIG, M.L. On the response of the laminar boundary layer to small fluctuations of the free-stream velocity. J. Ae. Scs. 1960, 741.
- 2261 URAM, E.M. A method of calculating velocity distribution for turbulent boundary layers in adverse pressure distributions. J. Ae. Scs. 1960, 659.
- 2262 MARTELLUCCI, A. An extension of the linearized characteristics method for calculating the supersonic flow around elliptic cones. J. Ae. Scs. 1960, 667.
- 2263 ROSE, P.H., PROBSTEIN, R.F. and ADAMS, M.C. Turbulent heat transfer through a highly cooled, partially dissociated boundary layer. J. Ae. Scs. 1958, 751.
- 2264 HARTUNIAN, R.A., RUSSO, A.L. and MARRONE, P.V. Boundary layer transition and heat transfer in shock tubes. J. Ae. Scs. 1960, 587.
- 2265 YEN, K.T. On the thrust hypothesis for the jet flap including jet-mixing effects.
 J. Ae. Scs. 1960, 607.
- 2266 STRAND, T. Minimum wing wave drag with volume constraint. J. Ae. Scs. 1960, 615.
- 2267 KOGAN, A. On supersonic flow past thick airfoils. J. Ae. Scs. 1960, 504.
- 2268 TURCOTTE, D.L. Stable combustion of a high-velocity gas in a heated boundary layer. J. Ae. Scs. 1960.
- 2269 GROSS, R.A. and CHINITZ, W. A study of supersonic combustion. J. Ae. Scs. 1960, 517.
- 2270 FRAISER, H.R. Supersonic inlet dynamics. J. Ae. Scs. 1960, 429.
- 2271 ORDWAY, D.E. and HALE, R.W. Theory of supersonic propeller aerodynamics. J. Ae. Scs. 1960, 437.
- 2272 RODDEN, W.P. and REVELL, J.D. Oscillatory aerodynamic coefficients for a unified supersonic hypersonic strip theory. J. Ae. Scs. 1960, 451.
- 2273 TAO, L.N. Magnetohydrodynamic effects on the formation of Couette flow. J. Ae. Scs. 1960, 334.
- 2274 NAGAMATSU, H.T., GEIGER, R.E. and SHEER, R.E. Real gas effects in flow over blunt bodies at hypersonic speeds. J. Ae. Scs. 1960, 241.
- 2275 SPARROW, E.M. and GREGG, J.L. Flow about an unsteadily rotating disc. J. Ae. Scs. 1960, 252.
- 2276 EHLERS, F.E. and SHOEMAKER, E.M. A three-dimensional linearized analysis of the forces exerted on a rigid wing by a shock wave. J. Ae. Scs. 1960, 257.
- 2277 SCHAFFER, A. A study of vortex cancellation. J. Ac. Scs. 1960, 193.
- 2278 LOCHTENBERG, B.H. Transition in a separated laminar boundary layer. J.Ae.Scs.1960, 92.
- 2279 SCALA, S.M. Sublimation in a hypersonic environment. J. Ae. Scs. 1960, 1.
- 2280 GINZEL, I. and MULTHOPP, H. Wings with minimum drag due to lift in supersonic flow. J. Ae. Scs. 1960, 13.

- 2281 VAGLIO-LAURIN, R. Turbulent heat transfer on blunt-nosed bodies in two-dimensional and general three-dimensional hypersonic flow. J. Ae. Scs. 1960, 27.
- 2282 BUSH, W.B. Compressible flat-plate boundary-layer flow with an applied magnetic field. J. Ae. Scs. 1960, 49.
- 2283 DUNWOODY, N.T. On shearing flow between porous coaxial cylinders. J. Ae. Scs. 29, 1962, 494.
- 2284 ROGERS, R.H. The transition to tubulence in a boundary layer on a blunt cone in supersonic flow. J. Ae. Scs. 29, 1962, 501.
- 2285 PECKHAM, D.H. Experiments at hypersonic speeds on circular cones at incidence. RAE TN. Aero. 2863, 1963.
- 2286 BERNSTEIN,L. Equilibrium real-gas performance charts for a hypersonic shock-tube wind-tunnel employing nitrogen. ARC CP.633, 1961.
- 2287 GREGORY, N. and LOVE, E.M. Progress report on an experiment on the effect of surface flexibility on the stability of laminar flow. ARC CP.602, 1961.
- 2288 BROADWELL, J.E. Analysis of the fluid mechanics of secondary injection for thrust vector control. AIAA Jnl. 1963, 1067.
- 2289 MURROW, H.N., PRATT, K.G. and DRISCHLER, J.A. Numerical technique to lifting surface theory for calculation of unsteady aerodynamic forces due to continuous sinusoidal gusts on several wing planforms at subsonic speeds. NASA TN. D1501, 1963.
- 2290 YATES, E.C., LAND, N.S. and FOUGHNER, J.T. Measured and calculated subsonic and transonic flutter characteristics of a 45° sweptback wing planform in air and in Freon-12 in the Langley transonic dynamics tunnel. NASA TN.D1616, 1963.
- 2291 LEVY, L.L. Atmosphere entries with spacecraft lift-drag ratios modulated to limit decelerations. NASA TN. D1427, 1962.
- 2292 HINSON, W.F. and FALANGA, R.A. Effect of jet pluming on the static stability of conecylinder-flare configurations at a Mach number of 9.65. NASA TN.D1352, 1962.
- 2293 HEDGEPETH, J.M. Design of stiffened cylinders in axial compression. NASA TN. D1510. 1962.
- 2294 LANSING, W., JONES, I.W. and RATNER, P. Non-linear shallow shell analysis by the matrix force method. NASA TN.D1510, 1962, 753.
- 2295 LI, T.Y. Recent advances in nonequilibrium dissociating gasdynamics. ARS Jnl. 1961.
- 2296 ESCHENROEDER, A.Q., BOYER, D. and HALL, J.G. Non-equilibrium expansions of air with coupled chemical reactions. Phys. Fluids, 1962.
- 2297 ESCHENROEDER, A.Q. Ionization nonequilibrium in expanding flows. ARS Jnl. 1962, 32.
- 2298 HAMMERLING, P., TEARE, J.D. and KIVEL, B. Theory of radiation from luminous shock waves in nitrogen. Phys. Fluids, 1959, 2, 422.
- 2299 SCALA, S.M. Hypersonic viscous shock layer. ARS Jnl. 1959, 29, 520.
- 2300 MOECKEL, W.E. Some effects of bluntness on boundary layer transition and heat transfer at supersonic speeds. NACA R.1312, 1957.
- 2301 MANGLER, K.W. Compressible boundary layers on bodies of revolution. ARC R.9740, 1946.
- 2302 ROGERS, R.H. The development of the boundary layer in supersonic shear flow. RAE TN.Aero.2738, 1961.
- 2303 TAYLOR, G.I. AND MACCOLL, J.W. Air pressure on a cone moving at high speeds. Proc. Roy. Soc. A, 139, 1933, 278.
- 2304 GIRAUD, J.P. Newtonian flow over a surface. Colston Symposium on Hypersonic Flow, Bristol, 1959.
- 2305 PECKHAN, D.H. A proposed programme of wind tunnel tests at hypersonic speeds to investigate the lifting properties of geometrically slender shapes. RAE TN.Aero.2730.
- 2306 HOLT, M. and BLACKIE, J. Experiments on circular cones at yaw in supersonic flow. J. Ae. Scs. 23, 1956, 931.
- 2307 CONTI,R.J. Laminar heat-transfer and pressure measurements at a Mach number of 6 on sharp and blunt 15 half-angle cones at angles of attack up to 90 NASA TN.D962, 1961.
- 2308 ROBERTS, R.C. and RILEY, J.D. A guide to the use of the M.I.T. Cone Tables. J. Ae. Scs. 21, 1954, 336.

- 2309 CHENG, H.K. Hypersonic flows past a yawed circular cone and other pointed bodies.

 J. Fluid Mech. 12, 1962, 169.
- 2310 CRABTREE, L.F. Survey of inviscid hypersonic flow theory for geometrically slender shapes. RAE TN. Aero. 2695, 1960.
- 2311 BERTRAM, M.H. and BLACKSTOCK, T.A. Some simple solutions to the problem of predicting boundary layer self-induced pressures. NASA TN. D798, 1961.
- 2312 BERNSTEIN, L. Tabulated solutions of the equilibrium gas properties behind the incidents and reflected normal shock-wave in a shock-tube. I. Nitrogen, II. Oxygen.

 ARC CP.626, 1961.
- 2313 HOLDER, D.W. and SCHULTZ, D.L. On the flow in a reflected shock tunnel. ARC R & M 3265, 1960.
- 2314 RESLER, E.L., LIN, S.C. and KANTROWITZ, A. Production of high temperature gases in shock tubes. J. App. Phys., 23, 1852, 1390.
- 2315 WOODS, N.A. Performance estimates for the RAE 6in. high-pressure shock tube. ARC 20,440, 1958.
- 2316 CLOUSTON, J.G., GAYDON, A.G. and HURLE, I.R. Temperature measurements of shock-waves by spectrum-line reversal. II, A double beam method. Proc. Roy. Soc. A252, 1959, 143.
- 2317 ANDERSON,G. Shock-tube testing time. J. Ae. Scs. 26, 1959, 184.
- 2318 STOLLERY, J.L. Stagnation temperature measurements in a hypersonic gun tunnel using the Sodium-line reversal method. ARC 22,854, 1961.
- 2319 NAGAMATSU, H.T., GEIGER, R.E. and SHEET, R.E. Real gas effects in flow over blunt bodies at hypersonic speeds. ARC 21,083, 1959.
- 2320 HANCOCK,G. Divergence of plate airfoils of low aspect ratio at supersonic speeds. J. Ae. Scs. 1959, 495.
- 2321 BENJAMIN, T.B. Effects of a flexible boundary on hydrodynamic stability. J. Fluid Mech. 9, 1960, 513.
- 2322 NONWEILER, T.R.F. Qualitative solutions of the stability equation for a boundary layer in contact with various forms of flexible surface. ARC 22,670, 1961.
- 2323 CUMMING, R.W., GREGORY, B. and WALKER, W.S. An investigation of the use of an auxiliary slot to re-establish laminar flow on low drag aerofoils. ARC R & M.2742, 1950.
- 2324 GREGORY, N. and WALKER, W.S. The effect on transition of isolated surface excrescences in the boundary layer. Pt. 1, ARC R & M 2779, 1950.
- 2325 GREGORY, N. and WALKER, W.S. Experiments on the use of suction through perforated strips for maintaining laminar flow. Transition and drag measurements. ARC R & M 3083, 1955.
- 2326 NEWTON, J.F. and SPAID, F.W. Interaction of secondary injectants and rocket exhaust for thrust vector control. ARS Jnl. 32, 1962, 1203.
- 2327 SAKURAI, A. On the propagation and structure of the blast wave. J. Phys. Soc. Japan, 8, 1953, 662.
- 2328 WILLMARTH, W.W. The production of aerodynamic forces by heat addition on external surfaces of aircraft. Rand Corp. RM. 2078, 1957.
- 2329 SEARS, W.R. Some aspects of non-stationary airfoil theory and its practical application.
 J. Ae. Scs. 8, 1951, 104.
- 2330 GARRICK, I.E. On some Fourier transforms in the theory of non-stationary flows. Proc. 5th Int. Cong. App. Mech. 1938, J. Wiley, 590.
- 2331 BENNETT, F.V. and PRATT, K.G. Calculated responses of a large sweptwing airplane to continuous turbulence with flight-test comparisons. NASA TR R.69, 1960.
- 2332 DIEDERICH, F.W. and ZLOTNICK, M. Calculated spanwise lift distributions, influence functions and influence coefficients for unswept wings in subsonic flow.

 NACA R.1228, 1955.
- 2333 JONES, W.P. Aerodynamic forces on wings in non-uniform motion. ARC R & M.2117, 1945.
- 2334 DIEDERICH, F.W. and ZLOTNICK, M. Calculated spanwise lift distributions and aerodynamic influence coefficients for swept wings in subsonic flow. NACA TN. 3476, 1955.
- 2335 HUBER, P.W. Use of Freon-12 as a fluid for aerodynamic testing. NACA TN. 1024, 1946.

- 2336 VonDOENHOFF. Studies of the use of Freon-12 as a wind tunnel testing medium. NACA TN.3000, 1953.
- 2337 BARMBY, J.G. Study of effects of sweep on the flutter of cantilever wings. NACA R.1014, 1951.
- 2338 JONES, G.W. and UNANGST, J.R. Investigation to determine effects of center-of-gravity location on transonic flutter characteristics of a 45° sweptback wing.

 NACA RM. L55K30, 1956.
- 2339 YATES, E.C. Calculation of flutter characteristics for finite-span swept or unswept wings at subsonic and supersonic speeds by a modified strip analysis.

 NACA RM. L57L10, 1958.
- 2340 LAND, N.S. and ABBOTT, F.T. Method of controlling stiffness properties of a solid-construction model wing. NACA TN.3423, 1955.
- 2341 JONES, G.W. and DuBOSE, H.C. Investigation of wing flutter at transonic speeds for six systematically varied plan forms. NACA RM. L53G10a, 1953.
- 2342 FALKNER, V.M. The calculation of aerodynamic loading on surfaces of any shape. ARC R & M.1910, 1943.
- 2343 COHEN, D. Formulas for the supersonic loading, lift and drag of flat swept back wings with leading edges behind the Mach lines. NACA R. 1050, 1951.
- 2344 KATZEN, E.D. and LEVY, L.L. Atmospheric entries with vehicle lift-drag ratio modulated to limit deceleration and rate of deceleration vehicles with maximum lift-drag ratio of 0.5. NASA TN.D1145, 1961.
- 2345 LEES, L. The use of aerodynamic lift during entry into the earth's atmosphere. Space Tech. Lab. R. GM-TR-0165-00519, 1958.
- 2346 GRANT, F.C. Modulated entry. NASA TN. D452, 1960.
- 2347 LUIDENS, R.W. Approximate analysis of atmospheric entry corridors and angles. NASA TN. D590, 1961.
- 2348 YOSHIKAWA, K.K. and WICK, B.H. Radiative heat transfer during atmosphere entry at parabolic velocity. NASA TN.D1074, 1961.
- 2349 FETTERMAN, D.E. Effects of simulated rocket jet exhaust on stability and control of a research type airplane configuration at a Mach number of 6.86. NASA TM.X127, 1959.
- 2350 SALMI, R.J. Effects of jet billowing on stability of missile-type bodies at Mach 3.85.
 NASA TN.D284, 1960.
- 2351 FALANGA,R.A. Exploratory tests of the effects of jet plumes on the flow over conecylinder flare bodies. NASA TN.D1000, 1962.
- 2352 LOVE, E.S. Aerodynamic investigation of a parabolic body of revolution at Mach number of 1.92 and some effects of an annular supersonic jet exhausting from the base.

 NACA TN.3709, 1956.
- 2353 McLELLAN, C.H. Investigation of a two-step nozzle in the Langley 11in. hypersonic tunnel. NACA TN.2171, 1950.
- 2354 McLELLAN, C.H. Investigation of the flow through a single stage two dimensional nozzle in the Langley 11in. hypersonic tunnel. NACA TN.2223, 1950.
- 2355 BERTRAM, M.H. Boundary layer displacement effects in air at Mach numbers of 6.8 and 9.6.
 NASA TR.R22, 1959.
- 2356 SEIFF, A. Secondary flow fields embedded in hypersonic shock layers. NASA TN. D1304, 1962.
- 2357 SEIDE, P. and STEIN, M. Compressive buckling of simply supplorted plates with longitudinal stiffeners. NACA TN.1825, 1949.
- 2358 BUDIANSKY, B. and SEIDE, P. Compressive buckling of simply supported plates with transverse stiffeners. NACA TN.1557, 1948.
- 2359 PETERSON, J.P. and DOW, M.B. Compression tests on circular cylinders stiffened longitudinally by closely spaced Z-section stringers. NASA Memo. 2-12-59L, 1959.
- 2360 SHANLEY, F.R. Simplified analysis of general instability of stiffened shells in pure bending. J. Ae. Scs. 16, 1949, 590.
- 2361 TURREL, J. L. Large deflections of structures subjected to heating and external loads. J. Ae. Scs. 1960.

- 2362 LANSING, W. Non-linear analysis of heated, cambered, wings by the matrix force method. IAS Paper 62-107, 1962.
- 2363 ASHWELL, D.G. A characteristic type of instability in the large deflections of elastic plates. Proc. Roy. Soc. A, 214, 1952, 98.
- 2364 GADD, G.E., HOLDER, D.W. and REGAN, J.D. An experimental investigation of the interaction between shock waves and boundary layers. Proc. Roy. Soc. A, 226, 1954, 227.
- 2365 THWAITES, B. Approximate calculation of the laminar boundary layer. Aero. Quart. 1, 1949-1950.
- 2366 COHEN, C.B. and RESHOTKO, E. The compressible laminar boundary layer with heat transfer and arbitrary pressure gradient. NACA R. 1294, 1956.
- 2367 GADD, G.E. A theoreticalinvestigation of the effects of Mach number, Reynolds number, wall temperature and surface curvature on laminar separation in supersonic flow.

 ARC 18,494, 1956.
- 2368 STEIGER, M.H. and BLOOM, H.H. Three dimensional viscous wakes. J. Fluid Mech. 14, 1962, 233.
- 2369 LAMB, H. Hydrodynamics. p608 Steady motion treated by Oseen's method and others.

 Dover Publ. New York, 1945, 608.
- 2370 LEWIS, J.A. and CARRIER, G.F. Some remarks on the flat plate boundary layer. Q. App. Math. 7, 1949, 228.
- 2371 STEIGER, M.H. and BLOOM, M.H. Axisymmetric free mixing with swirl. PIBAL R.628.
- 2372 KIEINSTEIN,G. On axially symmetric, turbulent, compressible mixing in the presence of initial boundary layer. AIAA Jnl. 1, 1963, 945.
- 2373 TAN, H.S. Nose drag in free-molecule flow and its minimization. J.Ae.Scs. 1959, 360.
- 2374 LIBEY, P.A. Theoretical analysis of turbulent mixing of reactive gases with application to supersonic combustion of hydrogen. ARS Jnl. 32, 1962, 388.
- 2375 KLEINSTEIN,G. An approximate solution for the axisymmetric jet of a laminar compressible fluid. PIBAL R.648, 1961.
- 2376 MASTERS, J.I. Some applications in physics of the F-function. J.Chem. Phys. 23, 1955, 1865.
- 2377 VAGLIO-LAURIN, R. and FERRI, A. Theoretical investigation of the flow field about blunt nosed bodies in supersonic flight. J. Ae. Scs. 25, 1958, 761.
- 2378 BURKE, A.F. and CURTIS, J.T. Blunt-cone pressure distributions at hypersonic Mach numbers. J. Ae. Scs. 29, 1962, 237.
- 2379 ALLEN, H.J. Hypersonic flight and the re-entry problem. J. Ae. Scs. 25, 1958, 217.
- 2380 BROWN, C.E. and McLEAN, F.E. The problem of obtaining high lift-drag ratios at supersonic speeds. J. Ae. Scs. 26, 1959, 298.
- 2381 CARROS, R.J. Effect of Mach number on boundary layer transition in the presence of pressure rise and surface roughness on an ogive-cylinder body with cold wall conditions.

 NACA RM A56B15, 1956.
- 2382 HARTREE, D.R. The solution of the equations of the laminar boundary layer for Schubauer's observed pressure distribution for an elliptic cylinder. ARC R & M. 2427, 1949.
- 2383 VonKARMAN, T. and MILLIKAN, C.B. On the theory of laminar boundary layer involving separation. NACA R.504, 1934.
- 2384 MILLIKAN, C.B. A theoretical calculation of the laminar boundary layer around an elliptic cylinder and its comparison with experiment. J. Ae. Scs. 2, 1936, 91.
- 2385 SCHUBAUER, G.B. Air flow in a separating laminar boundary layer. NACA R. 527, 1935.
- 2386 MORDUCHOW, M. Analysis and calculation by integral methods of laminar compressible boundary layer with heat transfer and with and without pressure gradient.

 NAUA R.1245, 1955.
- 2387 LEGGETT, D.M.A. The buckling of a square panel under shear, when one pair of opposite edges is clamped and the other pair is simply supported. ARC R & M.1991, 1941.
- 2388 GHLS,S. A process for the step-by-step integration of differential equations in an automatic digital computing machine. Proc. Cam. Phil. Soc. 47, 1951, 96.
- 2389 GARABEDIAN, P.R. Numerical construction of detached shock waves. J. Math. Phys. 36, 1957, 192.

- 2390 GARABEDIAN, P.R. and LIEBERSTEN, H.M. On the numerical calculation of detached bow shock waves in hypersonic flow. J. Ae. Scs. 25, 1958, 109.
- 2391 PROBSTEIN, R.F. Shock wave and flow field development in hypersonic re-entry. ARS Jnl. 31, 1961, 185.
- 2392 HOPKINS, H.G. The solution of small displacement, stability or vibration problems concerning a flat rectangular panel when the edges areeither clamped or simply supported. ARC R & M.2234, 1945.
- 2393 SIRULKIN, M.J. Heat transfer near the forward stagnation point of a body of revolution. J. Ae. Sos. 19, 1952, 570.
- 2394 NEICE, S.E., RUTKOWSKI, R.W. and CHANN, K.K. Stagnation point heat transfer measurements in hypersonic low density flow. J. Ae. Scs. 27, 1960, 387.
- 2395 WILSON, M.R. and WITTLIFF, C.E. Low density stagnation point heat transfer measurements in the hypersonic shock tunnel. ARS Jnl. 32, 1962, 275.
- 2396 COOK, I.T. and ROCKEY, K.C. Shear buckling of clamped and simply-supported infinitely long plates reinforced by transverse stiffeners. Aero. Quart. 13, 1962, 41.
- 2397 STEIN, M. and FRALICH, R.W. Critical shear stress of an infinitely long simply supported plate with transverse stiffeners. NACA TN. 1851, 1949.
- 2398 WAY,S. Stability of rectangular plates under shear and bending forces. J. App. Mech. 2, 1936, A131.
- 2399 WANG, T.K. Buckling of transverse stiffened plates under shear. J. App. Mech. 3, 1947, A269.
- 2400 KLEEMAN, P.W. The buckling shear stress of simply-supported infinitely long plates with transverse stiffeners. ARC R & M. 2971, 1953.

APPENDIX 3D

List of 279 questions used in the test

ĄΙ	what similarity laws must be obeyed when constructing aeroelastic models of heated high speed aircraft.
Q2	What are the structural and aeroelastic problems associated with flight of high speed aircraft.
Q3	How can one describe the aerodynamic forces and heating rates acting on a high speed aircraft.
Q4	What problems of heat conduction in composite slabs have been solved so far.
Q8	Can a criterion be developed to show empirically the validity of flow solutions for chemically reacting gas mixtures based on the simplifying assumption of instantaneous local chemical equilibrium.
Q9	What chemical kinetic system is applicable to hypersonic aerodynamic problems.
Q10	What theoretical and experimental guides do we have as to turbulent Couette flow behaviour.
Q12	Is it possible to relate the available pressure distributions for an ogive forebody at zero angle of attack to the lower surface pressures of an equivalent ogive forebody at angle of attack.
Q13	What methods - exact or approximate - are presently available for predicting body pressures at angle of attack.
Q14	What would be the effect of a slight gas rarefaction (i.e. velocity and temperature discontinuities at walls) on energy separation in tubes and ducts.
Q15	Papers on internal 'slip flow' heat transfer studies.
Q16	Can the reference enthalpy method be used to predict real-gas laminar-boundary layer skin friction and heat transfer.
Q17	Has the reference enthalpy method been compared with exact calculations for the perfect-gas case over a wide range of free-stream Mach numbers.
Q18	Are real-gas transport properties for air available over a wide range of enthalpies and densities.
Q22	Is it possible to find an analytical, similar solution of the strong blast wave problem in the Newtonian approximation.
Q23	How can the aerodynamic performance of channel flow ground effect machines be calculated.
Q26	What is the basic mechanism of the transonic aileron buzz.
Q27	Papers on shock-sound wave interaction.
⊋2 9	Material properties of photoelastic materials.

Q31	Can the transverse potential flow about a body of revolution be calculated efficiently by an electronic computer.
Q32	Can the three-dimensional problem of a transverse potential flow about a body of revolution be reduced to a two-dimensional problem.
Q33	Are experimental pressure distributions on bodies of revolution at angle of attack available.
Q34	Does there exist a good basic treatment of the dynamics of re-entry, combining consideration of realistic effects with relative simplicity of results.
Q35	Has anyone formally determined the influence of Joule heating, produced by the induced current, in magnetohydrodynamic free convection flows under general conditions.
Q39	Why does the compressibility transformation fail to correlate the high speed data for helium and air.
Q40	Did anyone else discover that the turbulent skin friction is not over sensitive to the nature of the variation of the viscosity with temperature.
Q41	What progress has been made in research on unsteady aerodynamics.
Q49	What are the factors which influence the time required to invert large structural matrices.
Q50	Does a practical flow follow the theoretical concepts for the interaction between adjacent blade rows of a supersonic cascade.
Q51	What is a single approximate formula for the displacement thickness of a laminar boundary layer in compressible flow on a flat plate.
Q52	How is the design of ring or part ring wings by linear theory affected by thickness.
Q53	What application has the linear theory design of curved wings.
Q54	What is the effect of cross sectional shape on the flow over simple delta wings with sharp leading edges.
Q55	Papers on flow visualization on slender conical wings.
Q56	What size of end plate can be safely used to simulate two-dimensional flow conditions over a bluff cylindrical body of finite aspect ratio.
Q57	To find an approximate correction for thickness in slender thin-wing theory
Q58	How do interference-free longitudinal stability measurements (made using free-flight models) compare with similar measurements made in a low-blockage wind tunnel.
Q59	Have wind tunnel interference effects been investigated on a systematic

basis.

Q61	Are there any papers dealing with acoustic wave propagation in reacting gases.
Q62	Has anyone investigated relaxation effects on gaseous heat transfer to a suddenly heated wall.
Q63	Are there any papers which treat heat conduction problems involving gas-phase-excited and surface-excited internal energy modes.
Q64	What is the effect on the base pressure of a body from which a jet issues of varying (a) jet design Mach number, (b) free stream Reynolds number, (c) jet radius/base radius.
Q66	Are there any theoretical methods for predicting base pressure.
Q67	Does transition in the hypersonic wake depend on body geometry and size.
Q68	How can one detect transition phenomena in boundary layers.
Q69	How can one detect transition phenomena in hypersonic wakes.
Q70	What work has been done on estimating forces and moments on cruciform wing-body configurations which are rolling.
Q71	Has anyone investigated and developed a simple model for the vortex wake behind a cruciform wing.
Q72	What is a criterion that the transonic flow around an airfoil with a round leading edge be validly analyzed by the linearized transonic flow theory.
Q74	Can the transonic flow around an arbitrary smooth thin airfoil be analysed in a simple approximate way.
Q79	What are the details of the rigorous kinetic theory of gases. (Chapman-Enskog Theory).
Q80	Has anyone investigated the effect of surface mass transfer on hypersonic viscous interactions.
Q81	What is the combined effect of surface heat and mass transfer on hypersonic flow.
Q82	What are the existing solutions for hypersonic viscous interactions over an insulated flat plate.
Q83	What controls leading-edge attachment at transonic speeds.
Q84	Can the three-point boundary-value problem for the Blasius equation be integrated numerically, using suitable transformations, without iteration on the boundary conditions.
Q85	What are the effects of small amounts of gas rarifaction on the charac-

teristics of the boundary layers on slender bodies of revolution.

Q86	What is the available information pertaining to boundary layers on very slender bodies of revolution in continuum flow (the "transverse curvature" effect).
Q87	What is the available information pertaining to the effect of slight rarifaction on boundary layer flows (the "slip" effect).
Q9 2	Is there a theory for prediction of aerodynamic force and movement on an oscillating thin airfoil in a real fluid (medium: water and air, angle of attack: small or large.)
Q93	What investigations have been made of the flow field about a body moving through a rarefied, partially ionized gas in the presence of a magnetic field.
Q94	How is the heat transfer downstream of the mass transfer region effected by mass transfer at the nose of a blunted cone.
Q95	To what extent can the available information for incompressible boundary layers be applied to problems involving compressible boundary layers.
Q97	To what extent can readily available steady-state aerodynamic data be utilized to predict lifting-surface flutter characteristics.
Q98	What are the significant steady and non-steady flow characteristics which affect the flutter mechanism.
Q99	Is it possible to determine rates of forced convective heat transfer from heated cylinders of non-circular cross-section, (the fluid flow being along the generators).
Q100	How much is known about boundary layer flows along non-circular cylinders.
Q1 01	Is there any simple, but practical, method for numerical integration of the mixing problem (i.e. the Blasius problem with three-point boundary conditions.)
Q102	Does there exist a closed-form expression for the local heat transfer around a yawed cylinder.
Q103	How far around a cylinder and under what conditions of flow, if any, is the velocity just outside of the boundary layer a linear function of the distance around the cylinder.
Q104	Where can I find pressure data on surfaces of swept cylinders.
Q105	Can't the static deflection shapes be used in predicting flutter in place of vibrational shapes? If so, can we provide a justification by means of an example.
Q106	Does the boundary layer on a flat plate in a shear flow induce a pressure gradient.

Can the procedure of matching inner and outer solutions for a viscous

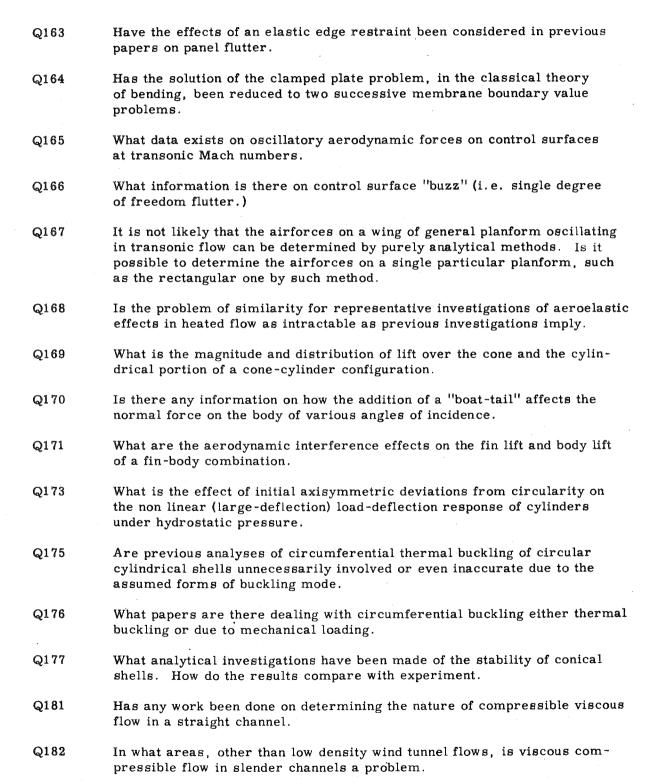
flow problem be applied when the main stream is a shear flow.

Q107

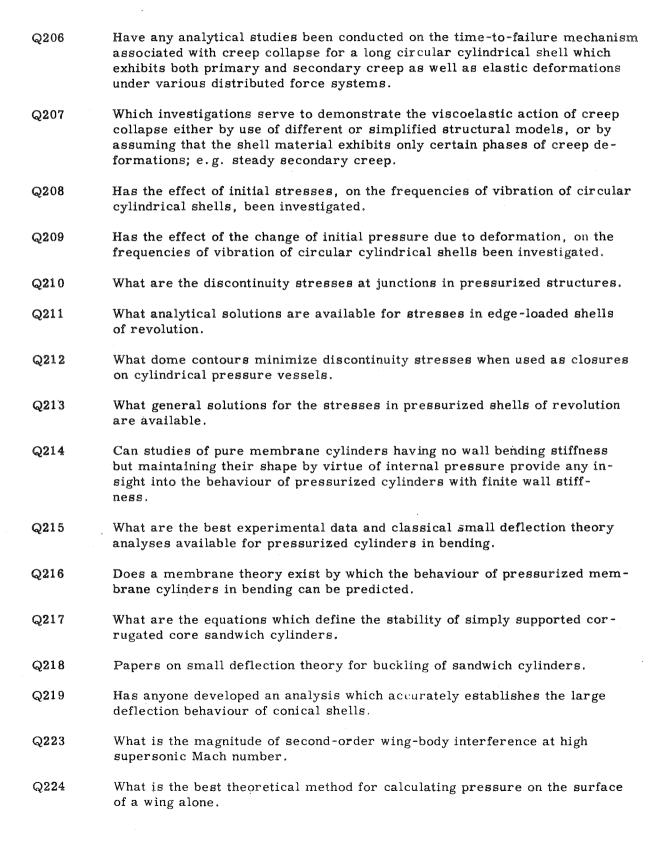
Q108	Can series expansions be found for the boundary layer on a flat plate in a shear flow.
Q109	What possible techniques are available for computing the injection distribution corresponding to an isothermal transpiration cooled hemisphere.
Q110	What is known regarding asymptotic solutions to the exact boundary layer equations.
Q111	Previous solutions to the boundary layer similarity equations.
Q112	Experimental results on hypersonic viscous interaction.
Q113	What has been done about viscous interactions in relatively low Reynolds number flows, particularly at high Mach numbers.
Q114	What role does the effect of chemical reaction (particularly when out of equilibrium) play in the similitude laws governing hypersonic flows over slender aerodynamic bodies.
Q115	Are there any experimental or exact numerical solution results around that might indicate the significance of nonequilibrium - dissociation effects on small disturbance flow fields.
Q116	How significant is the possible pressure of a dissociated free stream with respect to the realization of hypersonic simulation in high enthalpy wind tunnels.
Q117	Is it possible to use Biot's variational principle for general one dimensional and two dimensional heat conduction problems beyond those treated by Biot.
Q118	Do the discrepancies among current analyses of the vorticity effect on stagnation-point heat transfer result primarily from the differences in the viscosity - temperature law assumed.
Q119	How far can one trust the linear viscosity - temperature solution assumed in some of the analyses of hypersonic shock layer at low Reynolds number.
Q120	How close is the comparison of the shock layer theory with existing experiments in the low Reynolds number (merged-layer) regime.
Q121	Has anyone explained the kink in the surge line of a multi-stage axial compressor.
Q122	Have any aerodynamic derivatives been measured at hypersonic Mach numbers and comparison been made with theoretical work.
Q123	Are methods of measuring aerodynamic derivatives available which could be adopted for use in short running time facilities.
Q124	Can the irrotational-flow pressure distribution, and certain characteristic "sideslip" parameters, be found in closed form in general motion (not symmetrical about vertical plane.)

Q125	lift on an aerofoil with flaps.
Q126	What are wind-tunnel corrections for a two-dimensional aerofoil mounted off-centre in a tunnel.
Q127	What is the pressure distribution or load on a split flap on an aerofoil.
Q128	How do Kuchemann's and Multhopp's methods for calculating lift distributions on swept wings in subsonic flow compare with each other and with experiment.
Q129	What experimental measurements exist of span wise and chord wise loadings on swept wings at low subsonic speeds and small incidence.
Q130	What is the present state of the theory of quasi-conical flows.
Q131	References on the methods available for accurately estimating aerodynamic heat transfer to conical bodies for both laminar and turbulent flow.
Q132	What parameters can seriously influence natural transition from laminar to turbulent flow on a model in a wind tunnel.
Q133	Can a satisfactory experimental technique be developed for measuring oscillatory derivatives on slender sting-mounted models in supersonic wind tunnels.
Q134	What are the values of the stability derivatives in pitch and yaw for cambered slender ogee wings.
Q135	What effect has the boundary layer in modifying the basic inviscid flow behind the shock, neglecting effects of leading edge and corner.
Q136	How does a satellite orbit contract under the action of air drag in an atmosphere in which the scale height varies with altitude.
Q137	How is the flow at transonic speeds about a delta wing different from that on a closely-related tapered sweptback wing.
Q138	Recent data on shock-induced boundary-layer separation.
Q139	What interference effects are likely at transonic speeds.
Q140	Given complete freedom in the design of an airplane, what procedure would be used in order to minimize sonic boom intensity, and is there a limit to the degree of minimizing that can be accomplished.
Q141	Can methane-air combustion product be used as a hypersonic test medium and predict, within experimental accuracies, the results obtained in air.
Q142	What is the theoretical heat transfer rate at the stagnation point of a blunt body.

Q143	What is the theoretical heat transfer distribution around a hemisphere.
Q145	Has anyone investigated the unsteady lift distributions on finite wings in subsonic flow.
Q146	What information is available for dynamic response of airplanes to gusts or blasts in the subsonic regime.
Q147	Will forward or apex located controls be effective at low subsonic speeds and how do they compare with conventional trailing-edge flaps.
Q148	Given that an uncontrolled vehicle will tumble as it enters an atmosphere, is it possible to predict when and how it will stop tumbling and its subsequent motion.
Q149	What are the effects of initial imperfections on the elastic buckling of cylindrical shells under axial compression.
Q150	Why does the incremental theory and the deformation theory of plastic stress-strain relationship differ greatly when applied to stability problems.
Q152	Basic dynamic characteristics of structures continuous over many spans.
Q153	Is the information on the buckling of sandwich sphere available.
Q154	Can the load deformation characteristics of a beam be obtained with the material being inelastic and a non uniform temperature being present.
Q155	What is the effect of an internal liquid column on the breathing vibrations of a cylindrical shell.
Q156	Experimental techniques in shell vibration.
Q157	In summarizing theoretical and experimental work on the behaviour of a typical aircraft structure in a noise environment is it possible to develop a design procedure.
Q158	What data is there on the fatique of structures under acoustic loading.
Q159	What are the effects of thermal stress and buckling on the flutter characteristics of elastically restrained nearly square panels.
Q160	What procedures are available for calculating skin temperatures of panels subjected to aerodynamic heating.
Q161	Can increasing the edge loading of a plate beyond the critical value for buckling change the buckling mode.
Q162	Can thermal fatique results, obtained experimentally in engines or laboratory tests, be related to other experimental results, such as creep, mechanical fatigue, plastic deformation.



Q183	Jet interference with supersonic flows - experimental papers.
Q184	Thrust vector control by fluid injection - papers.
Q186	Is there experimental data on base pressure and wake angle for conical bodies in laminar hypersonic flow.
Q187	Is it possible to obtain a reasonably simple analytical solution to the heat equation for an exponential (in time) heat input.
Q189	Has anyone programmed a pump design method for a high-speed digital computer.
Q190	Has anyone derived simplified pump design equation from the fundamental three-dimensional equations for incompressible nonviscous flow.
Q191	Has anyone obtained meridional plane or blade-to-blade plane solutions of pump flow.
Q193	What are the effects of forward facing jets due to pressure ratio, exit mach number, ratio of specific heats, angle of attack, and exit to body diameter ratio.
Q194	What is the static force and moment characteristics and the shape of the bow shock wave on a short blunt 10° semivertex angle cone with a flat base, and also one with a conical afterbody having a semivertex angle of 50° , in helium at M = 15.
Q195	Effects of helium contamination on static pressures on surface of bodies and shock wave shapes in helium at $M=15$.
Q196	What are the flutter characteristics of the exposed skin panels of the X-15 vertical stabilizer when subjected to aerodynamic heating.
Q198	What information is available concerning the thermodynamic and transport properties of air for temperature to about 10,000°K and pressures to about 1,000 atmospheres.
Q200	What agreement is found between theoretically predicted instability times and experimentally measured collapse times for compressed columns in creep.
Q201	Theoretical studies of creep buckling.
Q202	Experimental studies of creep buckling.
Q203	Is it possible to correlate the results on the creep buckling of widely different structures within the framework of a single theory.
Q204	What are the experimental results for the creep buckling of columns.
Q205	What are the results for the creep buckling of round tubes under external pressure.



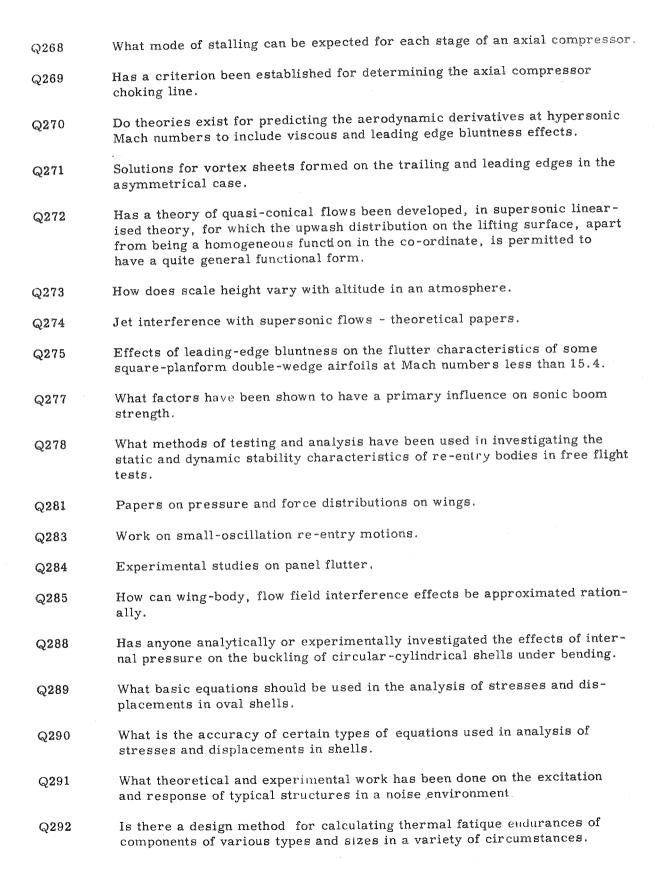
Q 22 5	How can the effect of the boundary-layer on wing pressure be calculated, and what is its magnitude.
Q226	How should the Navier-Stokes difference equations be solved.
Q227	Which iterative method for solving linear elliptic difference equations is most rapidly convergent.
Q228	Is there any available information on the aerodynamic loading distribution on wings immersed in high-energy, non-uniform slipstream of the type common to V/STOL aircraft from large scale experimental investigation.
Q229	Items on V/STOL aircraft operating in ground effect.
Q230	Technical report on measurement of ablation during flight.
Q231	What qualitative & quantitative material is available on ablation materials research.
Q 232	Have flow fields been calculated for blunt-nosed bodies and compared with experiment for a wide range of free stream conditions and body shapes.
Q233	What are the available properties of high-temperature air.
Q234	What is the magnitude of aerodynamic damping in flexible vibration modes of a slender body of revolution characteristic of launch vehicles.
Q 2 35	Is there any published literature on plastic buckling of orthotropic plates or shells.
Q 2 36	Papers on plastic buckling of isotropic shells and plates.
Q237	How to stabilize thin faces of sandwich construction to carry maximum loads, i.e. stress faces up to yield point of material.
Q 2 38	Has the buckling of orthotropic and stiffened conical shells under external pressure and other loads been investigated.
Q241	Compressive circumferential stresses in a torispherical shell reveal the possibility of buckling under internal pressure. Has anyone investigated for which ranges of shell parameters these stresses are sufficiently large to cause elastic buckling.
Q245	Is there an integral method to give a single and sufficiently accurate method of calculating the laminar separate point for various incompressible and compressible boundary layers with zero heat transfer.
Q246	What accurate or "exact" solutions of the laminar separation point for various incompressible and compressible boundary layers with zero heat transfer are available.
Q 24 7	Can the hypersonic similarity results be applied to the technique of predicting surface pressures of an ogive forebody at angle of attack.

Is it possible to estimate the transonic drag-rise and buffet boundaries

of aerofoils without recourse to experiment.

Q**24**9





Q293	Will an analysis of panel flutter based on arbitrarily assumed modes of deformation prove satisfactory, and if so, what is the minimum number of modes that need be considered.
Q294	What is the criterion for true panel flutter, as opposed to small amplitude vibration arising from acoustic disturbances.
Q295	Papers dealing with uniformly loaded sectors.
Q296	General methods of solving clamped plate problems.
Q297	How can the analytical solution of the buckling strength of a uniform circular cylinder loaded in axial compression be refined so as to lower the buckling load.
Q298	In the problem of the buckling strength of uniform circular cylinders loaded in axial compression, does the linear solution help with improving the non-linear one.
Q299	The problem of similarity for representative investigation of aeroelastic effects in a flow with the absence of heating effects.
Q300	How is fatigue damage estimated using the normal long-hand method.
Q301	Is there any information available on the difference in the effects of various edge conditions on the buckling of cylindrical shells.
Q302	What approximate analytical or experimental investigations have been performed relating to the non-linear bending and buckling of conical shells.
Q303	Have non-linear large deflection analyses been conducted for shell shapes other than conical.
Q304	Are asymptotic methods sufficiently accurate in the determination of pre-buckling stresses in torispherical shells, or must we resort to numerical methods.
Q305	What is the effect of an entrance tube on the measurement of pressure and density by a gage placed in a rarefied gas flow.
Q306	What are the <u>nonequilibrium</u> chemical constituents in the <u>viscous</u> shock layer ahead of a blunt re-entry vehicle.
Q310	How is shearing flow between two coaxial cylinders affected by suction or blowing at the cylinder walls.
Q314	How accurate are existing analytical theories in estimating pressure distributions on cones at incidence, at hypersonic speeds.
Q315	Are simple empirical methods of any use for estimating pressure distribution in cones.

Q316	Do Alecone effects seriously modify by easing distributions.
Q317	Has anyone investigated theoretically whether surface flexibility can stabilize a laminar boundary layer.
Q318	What is the side force induced on a supersonic nozzle wall when a liquid or gas is ejected from the wall.
Q319	What is the effect of the shape of finite airplane wings on the lift and moment responses to sinisoidal gusts.
Q320	What has been published on the subject of non-steady aerodynamic forces acting on airplane wings due to gusts and/or turbulence.
Q321	How do subsonic and transonic flutter data measured in the new Langley transonic dynamics tunnel compare with similar data obtained in other facilities.
Q322	How do subsonic and transonic flutter data measured in Freon - 12 compare with corresponding data obtained in air.
Q323	How do large changes in new mass ratio quantitatively affect wing-flutter boundaries.
Q327	What is the effect of the shape of the drag polar of a lifting spacecraft on the amount of reduction in maximum deceleration obtainable by continuously varying the aerodynamic coefficients during re-entry.
Q328	What are the differences in range and aerodynamic heating during re-entry which may be associated with the use of different drag polars.
Q330	What are the effects of a highly underexpanded rocket jet exhaust on vehicle static and dynamic stability at hypersonic speeds and high altitudes.
Q331	What are the physical significance and characteristics of separated laminar and turbulent boundary layer flows.
Q332	Has anyone analytically investigated the stabilizing influence of soft elastic cores on the buckling strength of cylindrical shells subjected to non-uniform external pressure.
Q333	What papers are available on the buckling of empty cylindrical shells under non-uniform pressure.
Q335	What effect do thermal stresses have on the compressive buckling strength of ring-stiffened cylinders.
Q336	What is the effect on cylinder buckling of a circumferential stress system that varies in the axial direction.
Q338	Can non-linear shallow shall analysis be reduced to an engineering technique

by use of the matrix.

Q339	of the nose region flow field for a sphere in hypersonic flow.
Q 340	What investigations have been made of the wave system created by a static pressure distribution over a liquid surface.
Q347	Has anyone investigated the effect of shock generated vorticity on heat transfer to a blunt body.
Q348	What is the heat transfer to a blunt body in the absence of vorticity.
Q349	What are the general effects on flow fields when the Reynolds number is small.
Q35 2	Find a calculation procedure applicable to all incompressible laminar boundary layer flow problems having good accuracy and reasonable computation time.
Q353	Papers applicable to this problem (calculation procedures for laminar incompressible flow with arbitrary pressure gradient).
Q354	What stresses and displacements are found in square plates having two adjacent edges free and the others either clamped or simply supported when they are loaded either by a uniformly distributed load or a concentrated force at the free corner.
Q355	Has anyone investigated the shear buckling of stiffened plates.
Q356	Papers on shear buckling of unstiffened rectangular plates under shear.
Q357	What theoretical methods are available for calculating the pressure distribution and the flow over symmetrical conical bodies with sharp leading edges.
Q358	In what manner does the surface pressure on a supersonic or hypersonic blunted cone approach its asymptotic value.
Q360	In practice, how close to reality are the assumptions that the flow in a hypersonic shock tube using nitrogen is non-viscous and in thermodynamic equilibrium.
Q365	What design factors can be used to control lift-drag ratios at Mach numbers above 5.

APPENDIX 3E
SUBSETS OF DOCUMENTS AND QUESTIONS

In the course of the tests, various subsets were used of the 279 questions (as given in Appendix 3D) and of the 1400 documents (as given in Appendix 3C). The main subsets, which have been referred to in the text of this volume, are listed.

QUESTIONS

SUBSET 1.	35 Questions,	each having	seven starting	terms.
2	82	142	177	285
9	87	145	181	292
34	95	157	189	293
40	113	160	205	294
49	122	165	211	299
67	131	170	219	315
81	132	171	261	338
SUBSET 2,	42 Questions,	all dealing	with aerodynam	ilos
SUBSET 2,	42 Questions,	all dealing	with aerodynam	i cs 268
79	130	167	225	268
79 100	130 132	167 170	225 226	268 269
79 100 116	130 132 136	167 170 181	225 226 227	268 269 272
79 100 116 118	130 132 136 137	167 170 181 182	225 226 227 230	268 269 272 273
79 100 116 1'18 119	130 132 136 137 141	167 170 181 182 189	225 226 227 230 250	268 269 272 273 274 317 323
79 100 116 118 119 121	130 132 136 137 141 145	167 170 181 182 189	225 226 227 230 250 261	268 269 272 273 274 317

QUESTIONS

SU	BSET	3.

1	8 1	138	205	275
2	82	139	206	277
4.	83	140	208	283
8	84	141	209	284
9	85	142	210	285
10	86	143	211	288
12	87	145	212	2 9 1
1 3	93	148	213	292
15	94	149	214	293
18	95	150	215	294
22	9 7	152	216	295
23	98	153	217	296
26	99	154	218	297
27	100	155	219	298
29	101	156	223	299
31	102	157	226	300
32	103	158	227	301
33	104	160	230	303
34	105	161	231	304
35	106	163	232	306
39	107	164	233	314
40	108	165	234	315
41	109	167	241	316
49	110	168	245	317
50	111	169	246	321
51	112	170	247	323
52	113	171	250	327
53	114	173	251	331
54	116	175	252	332
55	118	176	253	333
56	119	177	254	335
57	120	181	255	336
58	121	182	257	338
59	122	183	259	339
61	123	184	261	340
62	126	187	264	347
66	128	189	265	348
67	130	190	266	349
68	131	196	267	352 353
69 	132	200	268	353
71	133	201	269	35 5
72	135	202	272	356 360
74	136	203	273	360 365
79	137	204	274	365
80				

DOCUMENTS

SUBSET	1.	200 docum	ments, al in Quest	l of whi ion Subs	ch are r et 2.	elevant	to at leas	t one
1302	1476	1618	1695	1779	1964	1994	2111	
1311	1509	1619	1696	1782	1965	1995	2150	
1316	1569	1620	1697	1783	1966	1997	2153	
1317	1572	1621	1698	1785	1967	2001	2154	
1320	1574	1622	1699	1786	1968	2002	2155	
1321	1575	1655	1700	1787	1970	2061	2157	
1322	1576	1656	1701	1788	1971	2074	2187	
1324	1578	1666	1702	1792	1972	2075	2274	
1335	1588	1667	1703	1793	1973	2076	2313	
1351	158 9	1670	1704	1794	1974	2077	2316	
1360	1590	1671	1705	1795	1978	2078	2317	
1367	1591	1672	1706	1796	1980	2080	2318	
1378	1592	1675	1707	1797	1981	2081	2319	
1383	1594	1676	1708	1798	1982	2082	2321	
1399	1596	1677	1709	1799	1983	2083	2322	
1406	1597	1680	1710	1800	1984	2084	2338	
1409	1598	1681	1711	1836	1985	2085	2339	
1415	1605	1682	1712	1874	1986	2087	2340	
1416	1606	1683	1713	1879	1987	2088	2341	
1420	1608	1684	1717	1880	1988	2099	2342	
1436	1613	1687	1719	1916	1989	2100	2364	
1437	1614	1688	1728	1919	1990	2101	2367	
1443	1615	1691	1729	1920	1991	2102	2379	
1451	1616	1692	1748	1921	1992	2103	2391	
1467	1617	1693	1772	1963	1993	2104	1694	
SUBSET	2							
	£.0	Consists	of Subse	t i, plu	is the fo	llowing	150 docume	nus
1014	1225	Consists 1309	of Subse	t 1, plu 1562	ns the fo 1686	llowing 1811	150 docume	nus
1014								nes
1016 1021	1225	1309 1310	1439	1562	1686	1811	2213	nts
1016 1021 102 2	1225 1231	1309 1310 1314 1315	1439 1440	1562 1566	1686 1714	1811 1922	2213 2243	nts
1016 1021 102 2 1023	1225 1231 1234 1252 1253	1309 1310 1314 1315 1326	1439 1440 1446	1562 1566 1567	1686 1714 1746	1811 1922 1923	2213 2243 2247	nts
1016 1021 102 2	1225 1231 1234 1252 1253 1255	1309 1310 1314 1315 1326 1329	1439 2440 1446 1447 1448 1449	1562 1566 1567 1570	1686 1714 1746 1749	1811 1922 1923 1924	2213 2243 2247 2252	nts
1016 1021 1022 1023 1025 1033	1225 1231 1234 1252 1253 1255 1265	1309 1310 1314 1315 1326 1329 1332	1439 1440 1446 1447 1448	1562 1566 1567 1570 1571	1686 1714 1746 1749 1775	1811 1922 1923 1924 1925	221 3 2243 2247 2252 2280	nts
1016 1021 1022 1023 1025 1033 1062	1225 1231 1234 1252 1253 1255	1309 1310 1314 1315 1326 1329 1332	1439 2440 1446 1447 1448 1449	1562 1566 1567 1570 1571 1573	1686 1714 1746 1749 1775 1780	1811 1922 1923 1924 1925 1927	2213 2243 2247 2252 2280 2304	nts
1016 1021 1022 1023 1025 1033 1062 1068	1225 1231 1234 1252 1253 1255 1265	1309 1310 1314 1315 1326 1329 1332 1375	1439 2440 1446 1447 1448 1449	1562 1566 1567 1570 1571 1573 1577	1686 1714 1746 1749 1775 1780 1784	1811 1922 1923 1924 1925 1927 1969	2213 2243 2247 2252 2280 2304 2305	nts
1016 1021 1022 1023 1025 1033 1062	1225 1231 1234 1252 1253 1255 1265 1271 1274 1283	1309 1310 1314 1315 1326 1329 1332 1375 1376 1377	1439 2440 1446 1447 1448 1449 1460	1562 1566 1567 1570 1571 1573 1577 1609 1612 1625	1686 1714 1746 1749 1775 1780 1784 1789	1811 1922 1923 1924 1925 1927 1969 2062	2213 2243 2247 2252 2280 2304 2305 2309	nts
1016 1021 1022 1023 1025 1033 1062 1068	1225 1231 1234 1252 1253 1255 1265 1271	1309 1310 1314 1315 1326 1329 1332 1375 1376 1377	1439 2440 1446 1447 1448 1449 1460 1464 1466	1562 1566 1567 1570 1571 1573 1577 1609 1612	1686 1714 1746 1749 1775 1780 1784 1789	1811 1922 1923 1924 1925 1927 1969 2062 2096	2213 2243 2247 2252 2280 2304 2305 2309 2326	nts
1016 1021 1022 1023 1025 1033 1062 1068 1082 1094 1101	1225 1231 1234 1252 1253 1255 1265 1271 1274 1283	1309 1310 1314 1315 1326 1329 1332 1375 1376 1377	1439 \$440 1446 1447 1448 1449 1460 1464 1466 1468	1562 1566 1567 1570 1571 1573 1577 1609 1612 1625	1686 1714 1746 1749 1775 1780 1784 1789 1790	1811 1922 1923 1924 1925 1927 1969 2062 2096 2097	2213 2243 2247 2252 2280 2304 2305 2309 2326 2344	nts
1016 1021 1022 1023 1025 1033 1062 1068 1082 1094 1101 1160	1225 1231 1234 1252 1253 1255 1265 1271 1274 1283 1296 1297	1309 1310 1314 1315 1326 1329 1332 1375 1376 1377 1379 1380 1381	1439 2440 1446 1447 1448 1449 1460 1464 1466 1468	1562 1566 1567 1570 1571 1573 1577 1609 1612 1625 1629	1686 1714 1746 1749 1775 1780 1784 1789 1790 1801 1802	1811 1922 1923 1924 1925 1927 1969 2062 2096 2097 1098	2213 2243 2247 2252 2280 2304 2305 2309 2326 2344 2345	nts
1016 1021 1022 1023 1025 1033 1062 1068 1082 1094 1101 1160 1161	1225 1231 1234 1252 1253 1255 1265 1271 1274 1283 1296 1297 1298	1309 1310 1314 1315 1326 1329 1332 1375 1376 1377 1379 1380 1381 1382	1439 \$440 1446 1447 1448 1449 1460 1464 1466 1468 1469 1470	1562 1566 1567 1570 1571 1573 1577 1609 1612 1625 1629 1643	1686 1714 1746 1749 1775 1780 1784 1789 1790 1801 1802 1803	1811 1922 1923 1924 1925 1927 1969 2062 2096 2097 1098 2108	2213 2243 2247 2252 2280 2304 2305 2309 2326 2344 2345 2346	nts
1016 1021 1022 1023 1025 1033 1062 1068 1082 1094 1101 1160	1225 1231 1234 1252 1253 1255 1265 1271 1274 1283 1296 1297	1309 1310 1314 1315 1326 1329 1332 1375 1376 1377 1379 1380 1381 1382	1439 \$440 1446 1447 1448 1449 1460 1464 1466 1468 1469 1470 1475	1562 1566 1567 1570 1571 1573 1577 1609 1612 1625 1629 1643 1659	1686 1714 1746 1749 1775 1780 1784 1789 1790 1801 1802 1803 1804	1811 1922 1923 1924 1925 1927 1969 2062 2096 2097 1098 2108	2213 2243 2247 2252 2280 2304 2305 2309 2326 2344 2345 2346 2347	nts
1016 1021 1022 1023 1025 1033 1062 1068 1082 1094 1101 1160 1161	1225 1231 1234 1252 1253 1255 1265 1271 1274 1283 1296 1297 1298	1309 1310 1314 1315 1326 1329 1332 1375 1376 1377 1379 1380 1381 1382 1390	1439 2440 1446 1447 1448 1449 1460 1464 1466 1468 1469 1470 1475 1478	1562 1566 1567 1570 1571 1573 1577 1609 1612 1625 1629 1643 1659 1660	1686 1714 1746 1749 1775 1780 1784 1789 1790 1801 1802 1803 1804 1805	1811 1922 1923 1924 1925 1927 1969 2062 2096 2097 1098 2108 2110 2112	2213 2243 2247 2252 2280 2304 2305 2309 2326 2344 2345 2345 2346 2347	nts
1016 1021 1022 1023 1025 1033 1062 1068 1082 1094 1101 1160 1161 1163	1225 1231 1234 1252 1253 1255 1265 1271 1274 1283 1296 1297 1298 1299	1309 1310 1314 1315 1326 1329 1332 1375 1376 1377 1379 1380 1381 1382 1390 1391	1439 2440 1446 1447 1448 1449 1460 1464 1466 1468 1469 1470 1475 1478 1492	1562 1566 1567 1570 1571 1573 1577 1609 1612 1625 1629 1643 1659 1660 1661	1686 1714 1746 1749 1775 1780 1784 1789 1790 1801 1802 1803 1804 1805	1811 1922 1923 1924 1925 1927 1969 2062 2096 2097 1098 2108 2110 2112 2114	2213 2243 2247 2252 2280 2304 2305 2309 2326 2344 2345 2345 2346 2347 2355	nts
1016 1021 1022 1023 1025 1033 1062 1068 1082 1094 1101 1160 1161 1163 1164	1225 1231 1234 1252 1253 1255 1265 1271 1274 1283 1296 1297 1298 1299	1309 1310 1314 1315 1326 1329 1332 1375 1376 1377 1379 1380 1381 1382 1390 1391	1439 \$440 1446 1447 1448 1449 1460 1464 1466 1469 1470 1475 1478 1492 1502	1562 1566 1567 1570 1571 1573 1577 1609 1612 1625 1629 1643 1659 1660 1661 1662	1686 1714 1746 1749 1775 1780 1784 1789 1790 1801 1802 1803 1804 1805 1806 1807	1811 1922 1923 1924 1925 1927 1969 2062 2096 2097 1098 2108 2110 2112 2114 2115	2213 2243 2247 2252 2280 2304 2305 2309 2326 2344 2345 2346 2347 2355 2377	nts

APPENDIX 3F

AUTHORS OF BASE DOCUMENTS AND QUESTION COMPILERS

The authors of the base documents are listed according to the country in which they were working at the time the paper was published. Against each author is shown the document number (as given in Appendix 3C) of his paper, and the number or numbers (as given in Appendix 3D) of the questions which he prepared and which were used in the test. Although all the authors prepared search questions, it will be noted that in some cases, none of the questions was used. There were various reasons for this; these are discussed in Chapter 3 of this volume.

AUSTRALIA	Document	Questions
G.A. Bird	1531	92, 263
FRANCE		
W. Eckhaus	1496	26, 27
GERMANY		
E.W. Adams	1509	253
INDIA		
D. Tirumalesa	2142	
ISRAEL		
J. Rom	1487	
J. Singer	2070	238
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I. Hosokawa	1521	72, 74
F. Sakao	1450	
M: Yasuhara	1494	
SWITZERLAND		
E.F. Brocher	1495	22

UNITED KINGDOM	Document	Que <u>stio</u> ns
T A Deviler	1631	125, 126, 127
J.A. Bagley	1632	128, 129
I Downstoin	2286	360
L. Bernstein	1754	99,100
D.E. Bourne	1513	54,55
J. W. Britton	1547	135
D. Catherall	1517	61
J.F. Clarke	1518	62, 63
D. I. Claubson	1640	157,158,291
B.L. Clarkson I.T. Cook	2396	355,356
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J.C. Coole	1751	56
C.F. Cowdrey	1519	64, 66
A.H. Craven	1895	167
D. E. Davies	1543	121, 268, 269
M.D.C. Doyle		310
N.T. Dunwoody	2283	122, 123, 270
R.A. East	1544	165, 166
D.R. Gaukroger	1643	58, 59, 2 59
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N. Gregory	2287	317
B. Glauert	1388	106,107, 108 254, 255
I.A. Hall	1750	
P.W.H. Howe	1767	162, 292
D.J. Johns	1769	175, 176, 301
W.P. Jones	1892	41
D.G. King-Hele	1548	136, 273
D.G. Mabey	1526	83, 260
W.G. Molyneux	1184	168, 299
K.C. Moore	1512	52, 53
L.S.D. Morley	1641	164, 2 95, 2 96
J.F. Nash	1940	
S. Neumark	1515	256
P.C. Parks	153 2	264
P.H. Peckham	22 85	314, 315, 316
J. Phillips	1768	300
H. Portnoy	1633	130, 272
R.H. Rogers	2284	
E.W.E. Rogers	1757	137, 138, 139
D.B. Russell	2063	226, 227
C. Salter	1893	257
D. M. Seal	1896	169, 170, 171
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C.S. Sinnott	1503	249, 250
G.E. Smith	1894	163, 293, 2 94
A.J. Sobey	1642	297, 2 98
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B.S. Stratford	1511	50, 51
J.S. Thompson	1755	133, 134
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M.S. Anderson	2146	335, 336
A.E. Armenakas	1953	208, 209
R.R. Berlot	1939	70 71
E.R. Benton	1520	70, 71
M. A. Biot	1872	100 904
H.L. Bohon	1948	196, 284 340
H.E. Brandmeier	1506 2288	318
J.E. Broadwell D.O. Brush	2145	332, 333
O.R. Burggraf	1538	95
R.A. Burton	1491	10
H. W. Carlson	1758	140, 277
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P.R. Choudhury	2140	306
Wen-Hwa Chu	1530	92
L.C. Coltrane	1759	278
K.R. Cramer	1500	35
R.J. Cresci	1123	94
B.E. Cunningham	2143	
J.H. Cunningham	2069	237
C.F. Dewey	1540	110, 111, 112
M.R. Denison	1943	186
S.C. Dixon	1766	159, 160, 161
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J. Dugundji	1486	1, 2, 3
R. Dunlap	1535	339
M.J. Fohrman	1947	194, 195
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G. Gerard	2188	285, 365
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A.F. Gollnick	1628	109
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A.L. Greensite	1499	34
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J.M. Hedgepeth	2293	21 29 22
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W.F. Hinson	2292	330, 331
G.R. Inger	1541	114, 115, 116
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M. Inouye	2006	233, 232
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E.R. Keener	1492	12, 13, 247
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G. Kleinstein	2372	
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W. Lansing	2294	338
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A. Leiss	1636	1) ~ A
A.W. Leissa	1422	354
M. Lenard	488	8, 9
L.L. Levy	2291	327, 328
Y.K. Lin	1729	15 2 155, 156
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E.D. Martin	1527	84, 261
W.C. Lyons	1505	4.4.0
S.H. Maslen	1629	113
H.G. McComb	1955	214, 215, 216
J. Mescall	2071	241, 304
M. Morduchow	1489	245, 246
H.N. Murrow	2289	319, 320
M. Newman	1957	219, 302, 303
W.A. Newsom	2144	0.51 0.50
S. Ostrach	1504	251, 252
H.L. Pond	2139	305
M.A. Rahman	1523	1.05
J.V. Rattayya	1627	105
W.P. Reid	1485	4
H. W. Ridyard	1501	100
D.J. Romeo	1946	193
R.C. Rozycki	1524	79
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J.C. Serpico	1952	206, 207
J.M. Solomon	1528	85, 86, 87
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T. Strand	1624	2 3
R.J. Swigart	2179	
O.E. Tewfik	1623	
C.L. Tien	1525	80, 81, 82
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M. Tobak	1639	148, 2 83
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J. Vasiliu	2061	
R.J. Vidal	1484	
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V. Zakkay	1522	
S.L. Zeiberg	2141	
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APPENDIX 3G

QUESTIONS - RELEVANT DOCUMENTS SETS

This Appendix lists the relevant documents for the 279 questions which are given in Appendix 3D. The document numbers refer to the documents as listed in Appendix 3C.

The questions are divided into the two groups of Basic Questions and Supplementary Questions. The difference in these groups of questions is discussed in Chapter 3; briefly the Basic Question is considered to be representative of the main problem which was the cause of the research being undertaken, while a Supplementary Question is a question which arose in the course of the research.

The Base Document refers to the original paper which was written by the compiler of the question, and as such it would always be expected to have a high relevance to the question. However, in the tests, the Base Document is eliminated in the search or searches to which it applies.

The relevant documents are divided into the four levels of relevance, as given on page 21. The papers are then listed in two groups, the first of which gives the papers which appeared in the list of references of the Base Document. The second group is marked with an asterisk against the document number. These refer to the additional papers which were located by students or by bibliographic coupling, and which were subsequently submitted to the authors for their decisions concerning relevance.

BASIC QUESTIONS

Question No.	Base Document	Relevance 1	Relevance 2	Relevance 3	Relevance
1	1486		1184, 1029*, 1031*, 1057*, 1378*, 1859*, 1875*	1012, 1051, 1102, 1185, 1030*, 1037*, 1056*, 1066*, 1095*, 1497*, 1858*, 1876*, 1879*, 1880*	1013, 1014, 1015 1052, 1142, 1195* 1462*
8	1488			1236, 1166*	
10	1491		1099	1115, 1257, 1258	
12	1492		1020	1056, 1057, 1058	1019
14	1534		1021, 1549	1550	
16	1493			1378*	1294*
22	1495			1027, 1028, 1262, 1263	1160, 1020*, 1654*
23	1624		1086, 1194, 1650, 1652		1649
26	1496		1064, 1265		1065, 1311*
31	1498		1266	1106, 1196	
34	1499		1032, 1067, 1164*	1639*, 1715*, 1716* 171 7 *	1719*, 2379*
35	1500			1087, 1088, 1104, 1267, 1268, 1269, 1270, 1407, 1408	
39	1502	1271	1016, 1413	1414	
41	1892	1900, 1902	1200, 1201, 1601 1901, 1544*, 1597* 1634*, 1687*, 1698* 1700*, 1704*, 1705* 2109*, 2112*, 2141* 2197*, 2256*, 2259* 2272*, 2289*	1899, 1903, 1593* 1749*, 1917*, 1919* 2290*	1199, 1594*, 2333*
49	1756		1046, 1047	1092	
50	1511		1213	1212, 1214, 1215, 1216, 1276, 1277, 1426, 1427	
52	1512		1224	1278, 1428	
54	1513		1250*, 1514*, 1609* 1612*	1225, 1793*	1464, 1465, 1466
56	1751				1776
57	1752		1465	1249, 1777, 1778	1247, 1250
58	1516			1252, 1431	1141
62	1518		1169	1168	
64	1519			1173, 1174, 1179, 1282	1172, 1176
67	1536		1024, 1283, 1552, 1553, 1554, 1555, 1556	12 72	1557, 1558
72			1468	1467, 1469, 1470	1775*
80	1525		1305, 1570*	1308, 1481*, 133 8 * 2185*, 1629*, 1663* 1798*, 1572*	2226*, 2355*

Question No.	Base Document	Relevance 1	Relevance 2	Relevance 3	Relevance 4
83	1526	143	9	1311, 1316, 1440, 1797*, 1798*	1187, 1314, 1315 1794*, 1265*
84	1527			1320	1478
85	1528	132	6, 1629*		1021, 1094, 1022*, 1306*
92	1530	·	۰	1361, 1441, 1442, 1443, 1445, 1748*	1362, 1363, 1444 1200*,1201*,1593* 1687*,1753*,1919* 2005*
93	1531			1296, 1033*	1446, 1447, 1448 1449, 1208*,1297* 1298*,1299*
94	1123	108	4	1024, 1101, 1294, 1364, 1365, 1560	1628*, 1661*
95	1538	101 156	6, 1375, 1460, 2	1378, 1255*, 1271* 1502*	1376, 1377
97	1753	138	io .	1749*, 2339*	1014, 1705, 1779 1780, 1379,1643* 1686*
99	1754	178	8	1785, 1786, 1787	1023, 1381, 1382 1784, 1789
101	1322	147	' 5	1476*, 1527*	1320*, 1478*
102	1539	156	5, 1566	1564, 2159*	1661*
106	1388		02, 1003, 1128, 54*, 1629*	1180, 1323, 1324, 1393, 1394, 1659, 1004*, 1384*, 2302*	1418*
113	1629	102	21, 1324, 1630*	1094, 1664,1 3 04*, 1570*, 1494*	1128, 1323, 1663 1305*,1309*,1571* 1655*,1388*,1062*
114	1541	133 140	32, 1572, 1578,)1*, 1625*	1573, 1574, 1421*, 1160*, 2213*, 2304* 1068*, 1571*, 2252*	1025, 1317, 1577 2198*,1655*,2355*
117	1542		95, 1579, 1580, 97, 1873*		
118	1630	166	57	1324, 1378, 1666, 1670	
121	1543	158	38	1589, 1590	
122	1544			1597, 1598	1688*,1708*,1713*
124	1545	159	99		1060, 1782*
125	1631			1652,1245*,2094* 2095	
128	1632	167	76, 1677	1678, 1679, 1205*	
133	17 55			1594, 1790	
13 5	1547		06, 1611	1562	1050, 1236, 1609 1612, 1406*
136	1 548	16	13, 1614, 1615	1616, 1617	1618
137	1757	179	93, 1794	1797	1795, 1796,1420*
140	1758	. 180	02, 1803	1808, 1809, 1811, 1253*, 2247*	1804, 1805, 1806 1807, 1810,2243*
141	1635			1691	

Question	Base Document	Relevance 1	Relevance 2	Relevance 3	Relevance 4
145	1637			1703, 1704, 1705, 2289*, 1779*	1698, 1699, 1700 1701, 1702, 1706 1792*, 2339*
147	1638		1713		1708, 1709, 1711 1712
148	1639		1717, 1719	2001*, 2379*	
149	1760		1821, 1822, 1824, 2122*	1820, 182 3, 1825, 2051*, 2121*	
153	1761			1826, 1828	
154	1762			1833, 1834, 1835, 1836, 1837	
155	1764			1848	1844, 1845, 1846 1847
157	1640		1725, 1728	1729, 1911*	1720, 1075*,1909*
159	1766			1391,1864, 1012*,1015*	1856, 1857, 1858 1859,1746*,1877*
162	1767		1865, 1867	1395, 1866 , 1868, 1869, 1870	
163	1894		1015, 1391, 1864*	1914, 1392*, 1627*, 1658*	
164	1641			1731	1730
167	1895	1919	1916	1920, 1921	
168	1184		1051, 1185	1878	1874
175	1769		1829, 1887, 1890, 2146*	1885, 1886, 1888, 1891	1889
177	1898		1932, 1936, 1937, 1938	1957*, 2 131 *	1931, 1934, 1935
183	1942		1969, 1970, 1971, 1972, 1174*, 1997*	1187, 1973, 1173*, 1176*, 1409*, 1946* 1992*, 1994*, 1995*	1974, 1177*
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189	1945			1985, 1990	
194	1947		1069, 2003, 2004, 2005, 2006, 2007	1999, 2000, 2001	1717
196	1948		1766*	1858,1859,2008, 1012*	
200	1950		2014, 2020	2013, 2016, 2 <mark>017,</mark> 2018, 2019	2012
203	1951		2027	2028	
206	1952		2029	2034, 1951*, 2021*	
208	1953	1846, 20 36			
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214	1955		1840, 2045	1763, 1838, 1839, 1841	
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226	2063			2078, 2083, 2084, 2085	2080, 2081, 2082
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235	2067			1829, 2116, 2117, 2118, 2119, 2120, 2121, 1743*, 2130*	
237	2069		1822, 2127, 2128	1742, 2050, 2121, 2126, 1826*, 1743*	2051*
238	2070		2131	1936, 2129	2117, 2118
241				2134, 2137, 2138	2135, 2136
245	1489		1054, 2386*	1562*, 1336*	1055, 1996
249	1503		1038, 1039	1206, 1313*	1203, 1204, 1205
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319	2289		1704, 2331,	2330, 2332, 2333, 2334, 2341*	1779, 2329
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332	2145		2171	1891, 2172, 2173, 2174	2030*
335	2146		1888, 1889, 2178	1885, 1887, 1886* 1890*, 1769*, 1891* 2173*, 1843*	1841, 2176, 2177
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339	1535				1037, 1035*
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347	2191	1666, 1667, 2395	2258, 2394	1668*, 1670*, 2204* 2300*, 1037*, 1559* 1630*, 2107*, 2213*	2391
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354	1422		1644	1647, 1648	1424, 1425, 1454 1551
355	2396			1400, 1419, 2387, 2397, 2400	1412, 2392, 2398 2399
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3	1486		1012, 1014, 1142, 1184, 1606*	1051, 1052, 1102, 2147*, 1658*	1015, 1185,1095*
4	1485			1005, 1006, 1090, 1091, 1119, 1144, 1181, 1399	
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130	1633		1680, 1681, 1682, 1683		
131	1546		1606, 1406*	1608, 1142*, 1294* 1962*, 2192*, 2213*	1354*, 1522*, 1873*
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156	1764		1847	1846, 1849	1844, 1845
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182	1941		1967, 1968	1964, 1965	
184	1942		1974, 2326*	1187, 1969, 1970 1971, 1972, 1973	
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193	1946			1991, 1992, 1993 1994, 1995, 1997	
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209	1953		2031, 2032, 2035		2037
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253	1509		1274	1082	
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255	1750	1118,1157 1774,1775			
259	1516	1252,1431			1141
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268	1543		1588, 1589	1592, 1772*	1590
269	1543	1591		1590	1588, 1589
270	1544		1201, 1572, 1577 1593	1025*, 2355*, 1068*	1569*,2311*,1540* 1332*
271	1545		1229*	1191, 1600, 1601 1433*, 1782*	
273	1548		1622	1619, 1620, 1621	1617,1616,2150*
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275	1634	1685		1686	
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281	2064	2090,2091 2092		1453, 1443*, 1675* 1676*, 1677*, 1780* 1783*, 1794*, 1484*	
283	1639			1032, 1067, 1715 1716, 1499*,2379*	1717
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285	2188			2309, 2378, 1923* 1924*, 2062*, 2074* 2075*, 1225*, 2280* 2305*, 2304*	1161, 1421, 2377 2379, 2380
289	1765	1851		1829, 1852	
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292	1767		1395, 1866, 1869	1865, 1868, 1870 1872, 1873	1867
293	1894	1015,1391	1285*	1390*, 1864*	
294	1894	1914,1915	1285*, 1857*, 1858*	1015,1391,1856* 1948*,1864*,1658*	1390*, 2008*
295	1641	1733,1734 1735,1736			
296	1641			1730, 1422*, 1648* 2392*	1731, 1733, 1734 1735, 1736
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301	1769		1887, 1888, 1889	1843*	
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320	2289		1779, 2331	2329 ,2197*,1637* 1699*	1698, 2330
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331	2292		2351, 2356	2350, 1187*, 1487* 2383*,2385*	1045*,1055*,1 316 * 2106*,2228*
333	2145		1891, 2172	2173, 2174, 1763*	1769*,2030*,20 31 * 2033*,2052*,2068*
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353	2182		1155, 2383, 2385	2382, 1062*, 1292* 1458*, 1459*, 1461* 2365*, 2366*, 1111* 1150*, 1479*	1241, 2370, 2384 2386
356	2396			1400, 2387, 2392 2398	
357	1514	1019,1122 1227			
358	1626	1025,1093 1161,1370 1372	1044, 1371, 1657 1423*, 1482*	1369, 1160*, 1332* 1572*	1318*
360	2286			1656, 2313, 2317 2318, 2319, 2157* 2274*	2316

APPENDIX 5.1

LIST OF INDEX TERMS

This Appendix contains the full list of the single terms used in the indexing of the complete set of 1400 documents. The terms are in alphabetical order; before each term is given a code which relates to the schedules of single terms to be found in Appendix 5.3.

Following each term is given the number of documents in which the term was used for indexing.

It will be noted that several terms do not have a code equivalent; such terms are proper names for which no reasonable place was available in the schedules.

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                                                                     V11
                                                                            Spectrography (1)
M27
       Shallow (20)
                                                                     R111b Spectrum (8)
M27a
       Shallowness (1)
                                                                     T85b
                                                                            Specular (3)
                                  M81
                                        Skewed (1)
F13
       Shank (1)
                                                                            Speed (168)
                                                                     P89
       Shanley (1)
                                  1140
                                         Skin (164)
                                  N60
                                                                            Sphere (44)
                                        Skip (4)
       Shape (238)
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                                  N61
                                        Skipping (1)
                                                                     K52
                                                                            Spherical (34)
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       Shaped (10)
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                                        Stiffened (36)
                                  D100
                                        Stiffener (20)
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                                                                            Supersonically (1)
      Spheroid (1)
K54
                                                                     069
                                                                            Supply (2)
E13
      Spike (2)
                                  D99
                                        Stiffening (4)
      Spiked (1)
                                        Stiffness (50)
                                                                     U69
                                                                            Supplying (1)
                                  R77
K47
                                        Still (1)
                                                                     D75
                                                                            Support (17)
N98
      Spin (3)
                                  X97
                                                                     E40
                                                                            Supported (50)
D36
      Spinner (3)
                                  G13
                                        Sting (9)
                                        Stochastic (1)
                                                                     V29e
                                                                            Supporting (5)
      Spinning (1)
                                  V119
Q7
                                                                            Suppression (1)
                                                                     U75
                                  W12
                                        Stoichiometric (2)
K51
      Spiral (4)
                                                                     H37
                                                                            Surface (300)
                                        Stokes (14)
077
       Spiralling (1)
      Split (4)
                                  C60
                                                                     E46
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                                        STOL (4)
D20
                                                                     P29
                                                                            Surge (5)
       Splitter (1)
                                        Stone (1)
F22
                                                                     J4
                                                                            Surrounding (3)
                                        Stopping (2)
       Spoiler (2)
D26
                                  U67
                                        Storage (4)
                                                                     Y57
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J15
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                                  U68
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                                                                            Sustained (2)
T56
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                                                                     U49a
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T55
                                                                            Sutherland (1)
                                  L58
                                        Straightness (3)
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                                                                     Q62
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                                  R28
                                        Strain (.64)
                                                                            Swallowing (1)
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                                                                            Sweat (4)
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                                  R64b
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                                                                            Sweep (15)
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                                        Stratford (2)
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                                        Stratiform (1)
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                                        Streamline (40)
P79
                                                                            Sweptback (28)
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                                  F56
                                        Streamtube (6)
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       Stabilized (4)
                                        Streamwise (15)
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                                                                     T82
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       Stabilizing (2)
                                                                            Symmetric (54)
                                        Stress (196)
                                                                     K31
X103
       Stable (8)
                                  R1
                                                                     K31
                                                                            Symmetrical (30)
       Stacking (3)
                                  E51
                                        Stressed (2)
F17
                                                                            Symmetrically (1)
                                                                     K32
F16
       Stage (13)
                                  E2
                                        Stressing (1)
                                                                           Symmetry (5)
                                  K99
                                        Stretched (1)
                                                                     K30a
M98
       Stagger (1)
                                                                     U92a
                                                                           Synthesis (1)
                                  R33
                                        Stretching (5)
M99
       Staggered (1)
                                                                     Y41
                                                                            System (29)
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                                  Q41
                                        Striking (2)
                                                                     M72a
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       Stagnant (2)
                                  D68
                                        Stringer (14)
X94
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                                  V24
                                        Strioscopy (1)
P10
A64
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                                  D94
                                        Strip (18)
                                                                     L52a
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                                                                     D25
                                                                            Tab (5)
                                  V12
P14
       Stall (17)
                                        Stroboscopic (1)
                                                                     Y55a
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                                  X14
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                                                                     Y55b
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                                                                           Tabulation (1)
                                        Strouhal (2)
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                                  WT
                                        Structural (28)
                                                                     D7
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                                        Structure (61)
                                                                     C57
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D77
                                  D40
                                        Strut (5)
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                                                                            Tailored (1)
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                                  D103
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Standoff (20)
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                                                                            Tailoring (1)
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                                                                            Tailplane (9)
                                        St. Venant (1)
                                                                     D9
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                                                                     N65
                                                                            Take-off (3)
                                  L65
       Stanton (7)
                                        Subarc (1)
 U28
       Start (2)
                                  R111
                                        Sub audio (1)
                                                                     J27a
                                                                           Tangency (1)
                                                                     Y16
                                                                            Tangent (18)
 U27
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                                  04
                                        Subcritical (1)
                                                                     K19
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                                  S94b
                                        Subject (3)
                                                                     F33
                                                                            Tank (7)
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                                  W87
                                        Subjected (3)
                                                                     V29g
                                        Sublayer (6)
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                                                                            Taper (10)
                                  S33
                                        Sublimation (4)
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       Stationary (25)
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 X95
                                                                           Tapped (1)
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                                                                     W101a
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                                        Subsonic (106)
                                                                            Tapping (1)
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                                        Sucking (1)
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                                                                            Taylor (12)
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       Steam (1)
                                        Suction (27)
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                                        Sudden (7)
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                                        Sun (1)
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                                  214
                                        Superaerodynamic (3)
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                                                                            Telemetering (1)
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                                  V115
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                                       Trajectory (25)
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                                 V100b Transcendant (1)
                                                                   J83b
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                                                                          Ultimate (2)
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                                 V91a
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                                 W60
                                       Transferred (1)
                                                                   J97b
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V71
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                                                                   P48
                                                                          Unbanked (1)
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F82
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                                                                   M116
                                                                          Unbounded (2)
B17
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                                                                          Unbuckled (2)
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                                       Transient (41)
       Theodorsen (3)
V84
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                                       Transit (1)
                                                                   L40b
                                                                          Uncambered (4)
      Theorem (10)
                                 042
                                       Transition (104)
                                                                   W32a
                                                                          Unchanged (1)
V29ha Theoretical (152)
                                                                   S71
                                                                          Uncooled (1)
                                 020
                                       Transitional (10)
V29h
      Theory (406)
      Thermal (86)
                                 P96
                                        Translation (4)
                                                                   Y33
                                                                          Uncoupled (5)
S24
                                 X35
                                                                   J43
                                                                          Under (4)
S26
      Thermally (5)
                                       Translational (3)
P31
                                 P97
                                        Translatory (2)
                                                                   034
                                                                          Underexpanded (1)
      Thermo-aeroelastic (1)
T13
      Thermo-chemical (1)
                                 T73
                                       Transmission (5)
                                                                   034a
                                                                          Underexpansion (2)
                                 06
                                                                   S94a
G29
                                        Transonic (72)
                                                                          Undergoing (1)
      Thermocouple (3)
                                 R124
                                       Transparent (1)
                                                                   N117
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S24
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                                       Transpiration (4)
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                                       Transversality (1)
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                                       Transversely (1)
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      Thickened (1)
047
                                        Trapezoidal (1)
                                                                   X84
                                                                          Unevenness (1)
      Thickening (2)
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                                 P88
                                       Travelling (4)
                                                                   F37
                                                                          Unfired (2)
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                                        Traversal (1)
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      Thin (140)
                                 N67
                                                                          Unflanged (1)
                                                                   R26
H86
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                                 U100
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                                                                          Unfolding (1)
                                 N68
                                        Traversing (3)
                                                                   S72
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M61
                                 U62a
                                       Treatment (4)
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                                        Trefftz (1)
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                                        Triangular (23)
                                                                   X74
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H79
                                 V110
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                                                                   X74a
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      Throat (20)
F78
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                                 P45
                                        Trim (8)
                                                                   X74b
                                                                          Uniformly (17)
                                 P46a
                                       Trimmed (2)
                                                                   R97
                                                                          Unimodal (1)
K18
      Through (5)
N48
      Thrust (45)
                                 P46
                                       Trimming (2)
                                                                   N'59
                                                                          Unit (11)
                                                                   M59a
                                 052
                                       Trip (5)
                                                                          Unity (5)
D21
      Thwaites (3)
                                                                   V59c
                                                                          Univac (3)
P42
      Tilt (9)
                                 045
                                        Tripped (1)
                                 052a
                                       Tripping (1)
                                                                   E33a
                                                                          Unloaded (1)
E31a
      Tilting (1)
N1
      Time (72)
                                 Y24b
                                       True (1)
                                                                   R26a
                                                                          Unloading (1)
N1
      Timewise (1)
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                                        Truncated (13)
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                                 D105
                                       Truss (3)
                                                                   N63
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A63
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                                        Tsien (1)
                                                                   D61
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                                 E7
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                                        Tubular (3)
                                                                   038
                                                                          Unstalled (1)
A51
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                                 K43
                                                                          Unstalling (1)
      TNT (1)
                                 V28
                                        Tuft (1)
                                                                   P17
A28
N127
      Tolerance (1)
                                 N92
                                        Tumbling (4)
                                                                   X77
                                                                          Unsteadily (1)
                                        Tungsten (16)
                                                                   X78
                                                                          Unsteadiness (3)
                                 A55
      Tollmien (5)
H16
                                 F84
                                        Tunnel (296)
                                                                   X76
                                                                          Unsteady (41)
      Top (2)
                                 E80
                                                                   E39
                                                                          Unstiffened (10)
                                       Turbine (14)
      Torda (2)
                                 E61
                                                                   Y33a
K48
      Toriconical (2)
                                        Turbojet (8)
                                                                          Unsupported (1)
                                                                   L10
                                                                          Unswept (12)
                                 E74
                                        Turbomachine (1)
K59
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      Toroidal (5)
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                                                                          Unsymmetrical (2)
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                                        Turbulent (132)
                                                                          Untapered (4)
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                                                                   L82b
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R14
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                                 T83
                                        Turning (8)
                                                                   W74
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                                 M39a
                                       Twin (1)
                                                                   K12
E20
      Torus (4)
M53
      Total (59)
                                 P57
                                        Twist (10)
                                                                   P19
                                                                          Upflow (1)
                                 W93
K7
                                       Twisted (5)
                                                                   P18a
                                                                          Upload (1)
      Towards (1)
Y60
                                 R15
      Traces (1)
                                        Twisting (7)
                                                                   J40
                                                                          Upper (28)
U99
      Tracing (1)
                                 1160
                                       Two (14)
                                                                   F18
                                                                          Uprating (1)
                                 м6
                                                                          Upstream (21)
R32
                                        Two-dimensional (168)
                                                                   K21
      Tractions (1)
P26
     .Trail (4)
                                 M6a
                                       Two-dimensions (1)
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                                                                          Upwash (9)
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       U.R. (1)
                                   E48a
                                         Walled (11)
U82
       Use (1)
                                         Walles (1)
                                         Ward (1)
L54
       V (3)
                                   E94
                                         Warhead (1)
B76
       Vacuum (5)
                                   P58
                                         Warp (1)
Y24a
       Valid (3)
                                         Warpage (1) Warped (1)
                                  R15a
       Value (39)
V73
                                   W105
Y99a
       Valued (1)
                                         Warren (1)
F43
                                  W68a
       Valve (1)
                                         Washed (1)
       Vandriest (1)
                                  B10
                                         Water (17)
       Van Dyke (4)
                                  U51b
                                         Waterproofing (1)
F21
       Vane (9)
                                  090
                                        Wave (244)
F67
       Vaneless (1)
                                  R111e Waveform (1)
C80
       Vanguard (1)
                                  R111d Wavelength (5)
Q20
       Vanishing (5)
                                  L82c
                                        Waviness (1)
       Vanishingly (1)
M47c
                                        Wavy (2)
Weak (11)
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B74
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                                  X14b
S34
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                                  U53b
                                        Weakening (1)
X86
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                                  E92
                                         Weapon (2)
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                                  D73
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V41
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                                         Wedge (63)
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C32
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                                        Weighting (1)
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                                        Weinstein (1)
Q57b
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                                        Weissinger (4)
F72
       Venturis (1)
                                  J28a
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C8
       Venus (3)
                                  Z21
                                        Wetting (1)
บ 105
       Verification (2)
                                  Q8
                                        Whirl (2)
H17
       Vertex (1)
                                        Whitham (3)
M75
       Vertical (34)
                                  H2
                                        Whole (1)
       Vertol (2)
C65
                                 M19
                                        Wide (4)
M516
      Very (4)
                                  M19a
                                        Widely (1)
       Vessels (9)
F35
                                 L100c Width (19)
       Vibrating (4)
R92
                                  C19
                                        Wind (266)
R91
       Vibration (49)
                                 D65
                                        Window (4)
X37
       Vibrational (10)
                                 K24
                                        Windward (4)
X38
       Vibrationally (3)
                                 D4
                                        Wing (252)
X36
       Vibratory (5)
                                 C39
                                        Winged (4)
       Vicinity (3)
Virtual (2)
J29
                                 C40
                                        Wingless (1)
Q30
                                 D90
                                        Wire (12)
016
       Viscid (2)
                                 X115a With (1)
R71
      Viscoelastic (3)
                                 I66a
                                        Within (1)
R70
      Viscoelasticity (1)
                                 Y38
                                        Without (1)
R80
      Viscoplastic (1)
                                 B1
                                        Wood (4)
P2
      Viscosity (44)
                                 Q29
                                        Work (5)
016
      Viscous (116)
                                 Y84
                                        Working (14)
N126
      Visibility (1)
                                 D91
                                        Woven (1)
N125
      Vision (1)
                                 L82a
                                        Wrinkled (1)
V9a
      Visual (1)
                                 R61
                                        Wrinkling (6)
V10
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      Volterra (1)
                                 C49
                                        X (1)
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                                        Xenon (1)
L103a Volumetric (1)
074
      Vortex (59)
                                 N85
                                       Yaw (32)
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      Vortical (2)
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                                       Yawed (17)
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      Vorticity (47)
                                 N86
                                       Yawing (13)
N54
      Voyage (1)
                                 Y48c
                                       Yield (10)
C61
      VTOL (20)
                                 U53c
                                       Yielding (1)
                                       Young (3)
L55
      W (1)
                                 £26
                                       Yo-Yo (1)
      Wagner (2)
H80a
      Waists (1)
                                       Z (2)
                                L55a
P22
      Wake (60)
                                M57a
                                       Zero (73)
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                                 0.7
                                       Zinc (2)
      Walkden (1)
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H42
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APPENDIX 5.2

SINGLE-TERM DICTIONARIES

This Appendix lists the 723 search starting terms which occurred in the set of 361 questions. These are shown, in the first column, in the form used by the questioner, and following each starting term are the terms in the three groups of Synonyms, Word-Endings and Quasi-Synonyms, these being denoted by the numbers (1), (2) and (3) respectively. These dictionaries are discussed on pages 66 to 68.

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ABLATING
                         (2) Ablated, Ablation (3) Ablation, Melting
 ABLATION
                         (2) Ablated, Ablating (3) Ablating, Melting
 ABOVE
                         (3) Higher
 ABSENCE
                         (3) Without
 ABSORPTION
                         (2) Absorbed
 ACCURACY
                         (2) Accurate (3) Accurate
 ACCURATE
                         (2) Accuracy (3) Exact, Perfect, Accuracy, True
 ACOUSTIC
                         (1) Sound, (2) Acoustically, (3) Sonic
 ACTION
                         (3) Behaviour, Process
 ADJACENT
                         (3) Near, Close, Proximity
 AERODYNAMIC
                         (2) Aerodynamically, Aerodynamics
 AERODYNAMICS
                        (2) Aerodynamic, Aerodynamically
 AEROELASTIC
                         (2) Aeroelasticity
 AEROFOIL
                         (3) Profile, Lifting x Element, Lifting x Surface
 AEROPLANE
                        (1) Aircraft (3) Air x Vehicle, Flight x Vehicle, Aerodynamic x Body
 AEROTHERMODYNAMIC
 AFTERBODY
 AFTER BURNING
                        (2) Afterburner
                        (3) Forward, Upstream
 AILERON
 AIR
                        (3) Atmosphere, Atmospheric
AIRCRAFT
                        (1) Aeroplane, (3) Air x Vehicle, Flight x Vehicle, Aerodynamic x Body
ALTITUDE
                        (3) Height
AMOUNT
                        (1) Quantity (3) Magnitude, Size, Total, Dimension
ANALYSIS
                        (2) Analyser, Analytic, Analytical (3) Analytical x Investigation
ANALYTICAL
                        (2) Analyser, Analysis, Analytic (3) Analysis
ANGLE
                        (2) Angled, Angular, Angularly
ANTISYMMETRIC
                        (1) Antisymmetric, Asymmetric, Asymmetrical, Nonsymmetric (2) Antisymmetric
ANTISYMMETRICAL
                        (1) Antisymmetric, Asymmetric, Asymmetrical, Non-symmetric
                        (2) Antisymmetric
APEX
                        (1) Vertex, (3) Top, Head, Crest, Crown, Cap, Semiapex, Semivertex
APPLICATION
                        (2) Applicability, Applied (3) Use
APPLIED
                        (2) Applicability, Application
APPROPR TATE
                        (3) Optimum, Ideal, Perfect
APPROXIMATE
                        (2) Approximating, Approximation (3) Generalized, Approximation, Simple,
                        Simplified, Simplification, Idealized, Near x Perfect, Imperfect
APPROXIMATION
                        (2) Approximate, Approximating (3) Simplification
ARBITRARILY
                        (2) Arbitrary
ARBITRARY
                        (2) Arbitrarily
AROUND
                        (3) Circumferential, Adjacent, Peripheral, Bounding, Ambient, Surrounding
ASPECT
ASSUMED
                        (1) Given, (2) Assuming, Assumption (3) Prescribed, Imposed
ASSUMPTION
                        (2) Assumed, Assuming (3) Theory, Theoretical, Hypothesis, Prediction,
                        Principle, Law, Rule
ASYMMETRICAL
                        (1) Antisymmetric, Antisymmetrical, Asymmetric, Non-symmetric,
                        (2) Asymmetric
ASYMPTOTIC
                        (2) Asymptote, Asymptotically
ATMOSPHERE
                        (2) Atmospheric, (3) Air
ATTACHMENT
                        (2) Attached, (3) Reattachment, Reattaching, Reattached
ATTACK
                       (3) Incidence
AXIAL
                       (2) Axially, Axis (3) Parallel
AXISYMMETRIC
                       (1) Axisymmetrical (2) Axisymmetrical, (3) Axial x Symmetric, Axial x
                       Symmetrical, Axially x Symmetric, Axially x Symmetrical
BASE
                       (2) Based, Basic (3) Bottom, Undersurface, End
BASTC
                       (1) Fundamental (2) Base, Based
BEAM
BEHAVIOUR
                       (3) Process, Action
BEHIND
                       (3) Back, Rearward
BENDING
                       (1) Flexure (2) Bend (3) Deflection
BETWEEN
                       (3) Mid
BTOT
BLADE
                       (1) Vane (2) Blading
BLASIUS
BLAST
                       (3) Explosion, Detonation, Gust
BLOCKAGE
                       (2) Block, Blocking (3) Choking, Stagnation
BLOWING
                       (2) Blower, Blown
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BLUFF

(3) Blunt

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BLUNT
                        (1) Blunted (2) Blunted, Blunting, Bluntness (3) Blunted, Bluff, Rounded
BLUNTED
                        (1) Blunt (2) Blunt, Blunting, Bluntness (3) Rounded, Bluff
BLUNTNESS
                        (2) Blunt, Blunted, Blunting (3) Blunt, Blunted, Blunting, Bluff, Round,
                        Rounded, Roundness
BOATTAIL
                        (2) Boattailed, Boattailing
RODA
                        (3) Surface, Structure, Fuselage
BOOM
BOUNDARY
                        (2) Bound, Bounded, Bounding (3) Threshold, Bound, Limit, Limiting, Onset.
                       Start, Starting, Appearance, Beginning
ROW
                        (2) Bowing (3) Ahead, Front, Forward, Forebody
                        (3) Pulse
(2) Buckle, Buckled
BREATHING
BUCKLING
                        (2) Buffeting (3) Buffeting, Oscillation
BUFFET
BUZZ
                        (3) Single x Degree x Freedom x Flutter
CALCULATED
                        (2) Calculating, Calculation, Calculus (3) Estimate, Estimation.
                       Prediction, Assessment
CALCULATING
                       (2) Calculated, Calculation, Calculus (3) Computation, Determination.
                       Solution, Quantitative x Measurement, Formula, Estimate, Estimation.
                       Prediction, Assessment
CALCIII.ATION
                       (1) Computation (2) Calculated, Calculating, Calculus (3) Determination,
                       Solution, Quantitative x Measurement, Formula, Estimate, Estimation,
                       Prediction, Assessment
CALIBRATION
                       (2) Calibrated
CAMBERED
                       (2) Camber (3) Camber
CAPACITY
                       (2) Capacitance
CARRYING
                       (1) Supporting (3) Loaded, Exposed, Subjected
CASCADE
                       (3) Lattice
                       (1) Middle (2) Central, Centrally, Centred (3) Central
CESSATION
                       (1) Stopping (3) Completion, Terminated, Braking, Arrest
CHANGE
                       (1) Alteration (2) Changing (3) Transition
CHANNEL
                       (1) Duct (3) Passage, Ducting
CHAPMAN
CHARACTERISTIC
                       (1) Property (3) Quality, Nature
CHARGED
                       (1) Ionized (2) Charge, Charging (3) Dielectric
CHEMICAL
                       (2) Chemically
CHEMICALLY
                       (2) Chemical
                       (2) Choke, Choking
                       (2) Choke, Choked
(2) Chord, (3) Chord
CHOKING
CHORDWISE
CIRCULAR
                       (2) Circularity, Circling (3) Round
CIRCULARITY
                       (2) Circular, Circling (3) Roundness
CIRCUMFERENTIAL
                       (1) Peripheral
CLAMPED
                       (2) Clamping
CLASSICAL
                       (1) Conventional
CLOSED
                       (2) Closure
CLOSURE
                       (1) Seal (2) Closed
COAXIAL
                       (3) Concentric
COEFFICIENT
                       (3) Factor
COLLAPSE
                       (2) Collapsing
COLUMN
COMBINATION
                       (2) Combined (3) System
COMBUSTION
                       (1) Burning (2) Combustible
COMPARISON
COMPONENT
                       (3) Element, Part
COMPOSITE
                       (2) Composed, Composition (3) Complex
                       (2) Compressibility, Compressible, Compressing, Compression, Compressive,
COMPRESSED
                       Compressor, Compressively
COMPRESSIBILITY
                       (2) Compressed, Compressible, Compressing, Compression, Compressive,
                       Compressor, Compressively
COMPRESSIBLE
                       (2) Compressed, Compressibility, Compressing, Compression, Compressive,
                       Compressor, Compressively (3) Subsonic, Transonic, Supersonic, Hypersonic
COMPRESSION
                       (2) Compressed, Compressibility, Compressible, Compressing, Compressive,
                       Compressor, Compressively (3) External x Pressure, External x Load
COMPRESSIVE
                       (2) Compressed, Compressibility, Compressible, Compressing, Compression,
                       Compressor, Compressively
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(1) Blower, Turbomachine (2) Compressed, Compressibility, Compressible, COMPRESSOR Compressing, Compression, Compressive, Compressively (3) Impeller, Pump COMPUTATION (1) Calculation (2) Computational, Compute, Computer (3) Derivation, Mathematical x Technique, Determination COMPUTER (2) Computation, Computational, Compute CONCENTRATED (2) Concentration (1) State (3) Problem, Concept CONDITION CONDUCTION (2) Conducting, Conductive, Conductivity, Conductor (3) Heat x Flow (2) Conducting, Conduction, Conductive, Conductor CONDUCTIVITY CONE (2) Coned, Conical, Conically (3) Conical x Body CONFIGURATION (3) Arrangement, System, Shape, Contour, Spatial x Distribution (2) Cone, Coned, Conically (3) Quasi-Conical, Cone x Shaped CONICAL CONSTRUCTION (3) Surface, Structure, Body CONTAMINATION (3) Foreign x Gas x Injection CONTINUOUS (2) Continuation, Continuity, Continuously, Continuum (3) Steady, Quasi-Steady CONTINUOUSLY (2) Continuation, Continuity, Continuous, Continuum (3) Constant, Steady, Continuous CONTINUUM (2) Continuation, Continuity, Continuous, Continuously (3) Steady x State CONTOUR (2) Contoured (3) Configuration, Profile, Geometry, Form, Shape CONTRACTION (2) Contracting (3) Reduction CONTROL (2) Controlled, Controlling (3) Guidance, Reduction, Limitation, Decreasing, Reduced, Decrement, Fall, Limiting, Minimisation, Minimising CONVECTION (2) Convected, Convecting, Convective CONVECTIVE (2) Convected, Convecting, Convection (3) Convection, Non-Radiating CONVERGENT (2) Convergence, Converging (2) Cool, Coolant, Cooler, Cooling (3) Insulated (2) Cool, Coolant, Cooled, Cooler (3) Insulation, Heat x Protection, COOLED COOLING Heat x Dissipation COORDINATE COER (3) Kernal, Centre, Middle COUNER (2) Cornered (2) Corrective, Corrector (3) Modified, Revised, Evaluation, Assessment, CORRECTION CORRELATION (1) Relation, Relationship (2) Correlated, Correlator, Correlogram (3) Similarity, Similitude, Analogy, Analogue, Coupling. CORRESPONDING (3) Similar CORRIDOR (1) Passage (3) Path CORRUGATED (2) Corrugation COUETTE COUPLED (2) Couple, Coupling (2) Creeping (3) Plastic x Flow CRITERION (3) Parameter, Value, Factor, Requirement, Principle CRITICAL (3) Maximum CROCCO CROSS (2) Crossed, Crossing CRUCIFORM (3) Crossing, Crossed, Intersecting, Cross CURRENT CURVATURE (2) Curve, Curved (3) Arc, Subarc, Curve, Curved CURVED (2) Curvature, Curve (3) Arc, Round CYLINDER (2) Cylindrical (3) Drum, Cylindrical x Body, Cylindrical x Shell CYLINDRICAL (2) Cylinder (3) Tubular, Quasi-Cylindrical, Quasi-Cylinder DAMAGE DAMPING (2) Damped (3) Reduction, Decrease, Decreasing, Decrement, Reduced DATA (3) Result, Output, Statistics, Valve DECELERATION (2) Decelerating, (3) Decelerating, Braking, Speed x Reduction, Motion x Arrest DEFLECTION (1) Displacement (2) Deflect, Deflected, Deflecting (3) Deviation DEFORMATION (1) Strain (2) Deformable, Deformed, Deforming (3) Failure DEGREE DELTA (1) Triangular (3) Isosceles, Caret DENSITY (2) Dense DERIVATION (2) Derivative (3) Transformation DERIVATIVE (2) Derivation DESIGN

(2) Detached (3) Separation

DETACHMENT

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DETECTION
                        (1) Finding (3) Determination, Deduction, Prediction, Observation,
                        Investigation
DETERMINATION
                        (3) Estimation, Prediction, Finding
                        (2) Developable, Developed (3) Formation, Generation, Production, Growth
DEVELOPMENT
DEVIATION
                        (3) Variation, Fluctuation, Change, Deflection
DIAMETER
                        (1) Differential (2) Different, Differential, Differentially (3) Differing
DIFFERENCE
                        (2) Difference, Differential, Differentially, Differing (3) Variable,
DIFFERENT
                        Variation, Varying, Alteration, Change, Deviation
DIFFERENTIAL.
                        (1) Difference (2) Difference, Different, Differentially (3) Differing
DIFFUSION
                        (2) Diffuse, Diffuser, Diffusivity (3) Joining, Merging, Merger, Merged,
                        Exchange, Impinging, Impingement
DIGITAL
DIMENSION
                        (2) Dimensional, Dimensionless
                       (2) Direct, Directed, Directional (3) Inclination(2) Discontinuous (3) Irregularity
DIRECTION
DISCONTINUITY
                        (1) Deflection (2) Displaced
DISPLACEMENT
                       (2) Dissociating, Dissociation
DISSOCIATED
                        (2) Dissociated, Dissociating (3) Breakdown, Decomposition
DISSOCIATION
DISTANCE
                        (2) Distant
DISTRIBUTED
                        (1) Spread (2) Distribution
                        (2) Distributed (3) Field, Gradient, Differential, Fluctuation, Variation,
DISTRIBUTION
                        Divergence, Profile, Slope, Load x Input
DISTURBANCE
                        (1) Perturbation (2) Disturbed, Disturbing (3) Instability, Unsteadiness,
                        Nonequilibrium, Alteration, Change, Interference
                        (2) Divergent, Diverging (3) Expansion
DIVERGENCE
DOME
                        (1) Hemisphere
DOUBLE
DOWNSTREAM
                        (3 Behind
DRAG
                       (3) Resistance
DRIVEN
                        (2) Drive, Driver, Driving (3) Forced
DHCT
                        (1) Channel (2) Ducted, Ducting (3) Passage, Ducting
                        (2) During (3) Period, Time
DURATION
DURING
                        (2) Duration
DYN AM IC
                       (2) Dynamically
EARTH
EDGE
                        (2) Edged, Edgewise (3) Lip, Surface, End, Boundary
                        (2) Effective, Effectiveness (3) Influence, Phenomena, Action
EFFECT
ELASTIC
                        (1) Flexible (2) Elastically, Elasticity
                        (1) Flexibly (2) Elastic, Elasticity
ELASTICALLY
ELECTRON
                        (2) Electronic
                        (2) Electron
ELECTRONIC
ELEMENT
                        (1) Oval (2) Elliptical, Ellipticity
ELLIPTIC
                       (3) Experimental, Semi-Empirical, Measured
EMPIRICAL
                        (2) Ended (3) Tip, Edge, Boundary
END
END PLATE
ENDURANCE
                        (3) Strength, Load x Carrying x Capacity, Fatigue x Resistance
ENERGY
ENGINE
                       (1) Motor (3) Power x Plant, Propulsion x System, Power Plant
ENSKOG
ENTHALPY
                        (3) Heat x Content
ENTRANCE
                        (1) Inlet (2) Entering, Entry (3) Orifice, Intake
ENTRY
                        (2) Entering, Entrance (3) Re-entry, Re-entering, Entering, Entrance
ENVIRONMENT
                        (2) Environmental (3) Surrounding, Condition
EQUATION
                        (3) Expression, Operator
                       (2) Equilibration, (3) Stability, Quasi-Equilibrium
EQUILIBRIUM
                        (2) Estimation (3) Prediction, Assessment
ESTIMATE
ESTIMATION
                        (2) Estimate (3) Prediction, Assessment
EXACT
                        (3) Accurate
EXCITATION
                        (2) Excited, Exciting (3) Exciting, Generation, Production
EXCITED
                        (2) Excitation, Exciting (3) Generated
EXHAUST
                        (1) Discharge, Efflux, Emission (2) Exhausting (3) Outflow, Issuing,
                       Leaving, Plume, Exhausting
EXTT
                        (1) Outlet (2) Exiting
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(2) Expanded, Expanding

(2) Experimental (3) Measuring, Measurement, Test, Investigation, Testing

EXPANSION

EXPER D. ENT

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(2) Experiment (3) Empirical, Semi-empirical, Measured
EXPERIMENTAL
                        (2) Exponential
EXPONENT
EXPONENTIAL
                        (2) Exponent
EXPOSED
                        (3) Bare, Open, Loaded, Subjected
                        (2) Expressible (3) Equation, Operator
EXPRESSION
                        (1) Outer (2) Externally (3) Outboard, Peripheral
EXTERNAL
                        (2) Faced, Facing (3) Surface
FACE
FACING
                        (1) Opposite (2) Face, Faced
                        (3) Constant, Variable, Criterion, Value, Function, Parameter
FACTOR
                        (2) Failure (3) Defect
FAILING
                        (2) Failing (3) Instability
FAILURE
                        (1) Quick, Rapid
FAST
FATIGUE
                        (1) Region (3) Area, Domain, Regime
FIELD
                        (2) Finned (3) Stabilizer
FIN
FINDING
                        (1) Detection (3) Estimation, Determination, Prediction
FINITE
                        (2) Flapping
FLAP
                        (2) Flattening (3) Plane, Planar
FLAT
                        (1) Elasticity (2) Flexible, Flexibly, Flexural, Flexure
FLEXIBILITY
                        (1) Elastic (2) Flexibility, Flexibly, Flexural, Flexure (3) Flexural (2) Flying (3) Flying, Fall, Falling
FLEXIBLE
FLIGHT
                        (1) Flux, Stream (2) Flowing (3) Motion, Lovement, Moving, Flowing
FLOW
                        (2) Fluidized (2) Fluttering
FLUID
FLUTTER
                        (1) Flow, Stream (3) Transfer
FLUX
                        (2) Flight (3) Flight, Fall, Falling
FLYING
                        (2) Forced, Forcible, Forcing (3) Load, Stress, Pressure, Loading
FORCE
                        (2) Force, Forcible, Forcing
FORCED
                        (1) Forepart, Foreplane (3) Nose, Cap, Bow, Head, Front, Leading x Edge
FORE BODY
                        (2) Formation, Formed (3) Shape
FORM
                        (2) Formulation (3) Solution, Theory, Principle
FORMULA
FYNRWARD
                        (3) Leading, Ahead
                        (1) Independent, Uncoupled (2) Freedom, Freely (3) Without, Negative,
FREE
                        Negatively, Ideal, Independence
                        (1) Independence (2) Free, Freely
FREEDOM
FREON
FREQUENCY
FRICTION
                        (2) Frictional (3) Resistance
FUNCTION
GAS
                        (2) Gaseous
GASEOUS
                        (2) Gas
GAUGE
                        (1) Meter (3) Measurement
                        (2) Generalized
GENERAL
GENERATED
                        (1) Formed (2) Generating, Generation, Generator (3) Excited
                        (2) Generated, Generating, Generation
GENERATORS
                        (2) Geometric, Geometrical, Geometrically (3) Shape, Profile, Contour,
GEOMETRY
                        Geometrical, Form, Proportion, Configuration
GIVEN
                        (1) Assumed (3) Prescribed, Imposed
                        (1) Slope (2) Graded (3) Load, Input, Distribution, Field, Differential,
GRADIENT
                        Fluctuation, Exchange, Variation, Divergence, Profile
GRAPHICAL
                        (3) Diagram
GROUND
                        (3) Surface
                        (3) Spread, Increase, Increased, Increasing, Increment, Rise, Jump,
GROWTH
                        Amplification, Spreading, Expansion
                        (2) Guide, Guided (3) Control
GUIDANCE
                        (3) Air x Current, Air x Blast
GUST
                        (2) Heated, Heating, Heater, Hot (3) Thermal
HEAT
                        (1) Hot (2) Heat, Heating, Heater, Hot (3) High x Temperature, Elevated x
HEATED
                        Temperature
HEATING
                        (2) Heat, Heated, Heater, Hot (3) Heat x Input, Heat x Addition,
                        Temperature x Rise
HEIGHT
                        (2) High, Higher, Highly (3) Altitude
HELIUM
HEMISPHERE
                        (1) Dome (2) Hemispheric, Hemispherical, Hemispherically
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(2) Higher, Highly, Height (3) Large

HIGH

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(2) High, Higher, Height (3) Fully, Entirely
HIGHLY
HOLOGENOUS
HOT
                        (1) Heated, (2) Heat, Heated, Heating, Heater (3) High x Temperature,
                        Elevated x Temperature
                        (3) Stagnant, Stationary x Fluid, Static x Fluid, Dead x Air, Still x Air
HYDROSTATIC
HYPERSONIC
                        (3) Hypervelocity
                        (3) Hypersonic x Flight, Hypersonic x Flow, Hypersonic x Speed, Ultra-
HYPERVELOCITY
                        High x Mach
IMPERFECTION
                        (1) Defect (2) Imperfect
                        (2) Incident (3) Attack
INCIDENCE
                        (3) Subsonic, Subcritical
INCOMPRESSIBLE
                        (2) Increased, Increasing (3) Increased, Increasing, Increment, Rise,
INCREASE
                        Spread, Spreading, Growth, Jump, Amplification, Expansion (1) Additional (2) Increment (3) Increasing
INCREMENTAL.
INDUCED
                        (2) Inductance, Induction (3) Generated, Caused
                        (1) Rigid, Unelastic
INELASTIC
INFINITE
                        (2) Infinitely, Infinitesimal, Infinitesimally (3) Unbounded
INFLUENCE
                        (2) Effect, Phenomena, Action
TNITTAL.
                        (1) First (2) Initialed, Initially, Initiation (3) Starting
INJECTION
                        (2) Injectant, Injected, Injector (3) Addition
                        (1) Internal (3) Interior
TNNFR
INPUT
INSTABILITY
                        (3) Failure
INSTANTANEOUS
                        (3) Sudden
INSTRUMENT
                        (2) Instrumentation (3) Device, Machine, Mechanism, Equipment, Apparatus
INSULATED
                        (1) Adiabatic, Nonconductive (2) Insulating, Insulation (3) Unheated,
                        Recovery x Temperature x Distribution, Zero x Heat x Transfer
INTEGRAL
                        (2) Integrated, Integrating, Integration, Integrative
INTEGRATION
                        (2) Integral, Integrated, Integrating, Integrative, (3) Integral x Method
INTENSITY
                        (2) Intense (3) Strength, Level, Volume
INTERACTION
                        (2) Interacting (3) Interference, Interferential, Disturbance, Perturbation
INTERBLADE
INTERFERENCE
                        (2) Interferential (3) Interaction, Interacting, Disturbance
                        (1) Inner (2) Internally, (3) Interior
INTERNAL
INTERVAL
                        (3) Gap
                        (2) Inverse, Inverted
INVERSION
INVESTIGATION
                        (1) Research, Study
INVISCID
                        (1) Non-viscous (3) Viscosity x Zero
                        (1) Charged (2) Ion, Ionization, Ionosphere (1) Non-rotating (3) Potential
IONIZED
IRROTATIONAL
ISENTROPIC
                        (3) Constant x Entropy
ISOTHERMAL
                        (3) Constant x Temperature
ISOTROPIC
ISSUING
                        (1) Exhausting, Leaving (3) Discharge, Emission, Efflux, Outflow
ITERATION
                        (2) Iterative (3) Iterative x Process, Finite x Difference x Method,
                        Finite x Difference x Procedure, Finite x Difference x Solution
ITERATIVE
                        (1) Step-by-step (2) Iteration (3) Finite x Difference
TET.
                        (3) Stream
JOULE
JUNCTION
                        (3) Joint, Intersection
KELVIN
KINETIC
                        (3) Dynamic, Motion
KINK
                        (2) Kinking (3) Deviation, Disturbance, Perturbation, Displacement,
                        Deflection, Divergence
LABORATORY
                        (2) Laminarization (3) Smooth
LAMINAR
LANGLEY
LARGE
                        (3) High, Extensive, Large x Order, Large x Size
LAUNCH
                        (2) Launched (3) Takeoff, Propelled
LAW
                        (1) Rule, Principle (3) Hypothesis, Assumption
LAYER
                        (2) Layered (3) Flow, Stream, Region, Regime
LEADING
                        (2) Lead (3) Front
LEES
LENGTH
                        (2) Lengthwise, Long (3) Span, Extent
LIFT
                        (2) Lifting
LIGHTWEIGHT
                        (1) Light
LINE
                        (2) Linear, Linearization, Linearized, Linearly, Liners (3) Point, Limit,
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Boundary

(2) Lift (3) Inclined

LIFTING

LINEAR (1) One-Dimensional (2) Line, Linearization, Linearized, Linearly LINEARIZED (2) Line, Linear, Linearization, Linearly, Liners (3) One-Dimensional. Quasi-One-Dimensional LIQUID (2) Liquefaction (3) Fluid LOAD (1) Loading, Stress (2) Loaded (3) Airload, Force, Pressure LOADED (2) Load, Loading (3) Exposed, Subjected, Supporting, Carrying LOADING (1) Load, Stress (2) Loaded, Load (3) Airload, Force, Pressure LOCAL (2) Locally, Located, Location (3) Ambient, Surrounding, Area, Proximity, Vicinity LOCATED (2) Local, Locally, Location (3) Location, Position LOCATION (1) Position, Point, Station (2) Local, Locally, Located LONG (2) Length, Lengthwise (3) Elongated, Extended (1) Lengthwise, Spanwise (2) Longitudinally (2) Losing (3) Transfer, Emission, Removal LONGITUDINAL LOSS LOW (1) Reduced, Small (2) Lower, Lowest LOWER (2) Low, Lowest (3) Base, Bottom, Under LUNAR LYAPUNOV MACH MACHINE (3) Vehicle MAGNETTO MAGNETOGASDYNAMIC (1) Magnetoaerodynamic, Electrogasdynamic, (3) Magnetohydrodynamic, Magnetofluidmechanic, Hydromagnetic, Conducting x Magnetic x Fluid. Conducting x Magnetic x Flow MAGNETOHYDRODYNAMIC (1) Hydromagnetic, Nagnetofluidmechanic (3) Nagnetogasdynamic, Nagnetoaerodynamic, Electrogasdynamic, Conducting x Magnetic, x Fluid, Conducting x Magnetic x Flow MAGNITUDE (1) Size (3) Amount, Quantity, Total Dimension MAIN (1) Major, Primary (3) First MAINTENANCE MASS (3) Inertia, Weight MATCHING (3) Integrating, Integration, Correlation MATERIAL (3) Medium MATHEMATICAL (3) Computational, Algebraic MATRIX (2) Latric MAXIMUM (2) Maxima (3) Complete, Extreme, Peak, Total MAXWELL (2) Maxwellian MEASURED (2) Measurement, Measuring (3) Measurement, Experiment, Experimental. Determination, Exact, Test, Testing MEASURING (2) Measured, Measurement (3) Measurement, Determination, Experiment, Test. Testing MEASUREMENT (2) Measured, Measuring (3) Measuring, Determination, Experiment, Test, Testing MECHANICAL. (2) Mechanics, Mechanism MECHANISM (2) Mechanical, Mechanics (3) Action, Characteristic, Nature, Problem, Structure MEDIUM: (3) Environment, Surrounding, Region, Atmosphere, Material MEMBRANE (3) Skin, Surface, Shell MERCED (1) Combined (2) Nerger, Merging (3) Lixed, Integrated, Interacting MERIDIONAL (2) Meridian (3) Central, Centre, Centred, Midplane, Median, Axis, Bisector, Fidpoint METHANE (1) Technique (3) Procedure, Treatment(2) Minimizing, Finimum (3) Minimizing, Reduction, Limitation, Limiting, METHOD MINIMISATION Removal MISSILE (1) Projectile (3) Rocket, Warhead MIXING (2) Fixed, Mixture (3) Diffusion, Joining, Merged, Merging, Merger, Exchange, Impinging, Impingement MIXTURE (2) Mixed, Mixing MODE (2) Modal (3) Type, Form, Shape, Method(3) Specimen, Idealized x Structure, Experiment, Test, Analysis, Theory, MODEL Hypothesis, Solution MODERATE (3) Low, Small, Reduced(2) Mol, Molecule MOLECULAR MOMENT (2) Momentary, Momentum (3) Aerodynamic x Influence x Coefficient MOTION (3) Moving, Movement, Travelling, Reaction, Reacting, Non-stationary

(1) Supported (2) Mounting, (3) Support, Suspension, Supporting, Resting,

MOUNTED

Mounting

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MOVING
                        (1) Non-Stationary, Travelling (2) Movable, Movement
MULTI
                        (1) Multiple
MULTISTAGE
                        (3) Multi, Multiple
NATURAL
                        (2) Nature (3) Self x Induced
NATURE
                        (2) Natural (3) Property, Characteristic, Quality
NAVTER
NEARLY
                        (2) Near, Nearest (3) Near, Quasi, Semi
NEUTRAL
NEWTONIAN
                        (2) Newton
NITROGEN
NOTSE
                        (3) Pressure x Fluctuation, Random x Pressure x Load, Acoustic x Disturbance
NON-CIRCULAR
                        (2) Non-Circulatory
NON-EQUILIBRIUM
                        (3) Frozen, Freezing, Instability, Unsteadiness
NON-LINEAR
                        (2) Non-Linearity
                        (1) Unsteady (3) Quasi-Unsteady, Variable, Varying, Fluctuating, Fluctuation,
NON-STEADY
                        Variation, Discontinuous, Unsteadiness, Unsteadily, Time x Dependent,
                        Alternating, Sinsusoidal
NON-UNIFORM
                        (1) Varying (2) Non-Uniformity, Non-Uniformly (3) Non-Linear, Variable.
                        Differential
NON-VISCOUS
                        (1) Inviscid (3) Viscosity x Zero
NON-ZERO
                        (3) Non-Negative, Positive
                        (2) Nosed (3) Cap, Head, Front, Forebody, Bow, Forepart, Foreplane,
NOSE
                        Leading x Edge
NOSED
                        (1) Headed, Capped (2) Nose
                        (3) Duct, Tube, Pipe, Venturi, Orifice, Wind x Tunnel x Contraction
NOZZLE
                        (2) Normally (3) Vertical, Perpendicular, Average, Mean
NORM AT.
NUMERICAL
                        (2) Number
OFF
OGEE
                        (2) Ogival, Ogive
OGIVE
                        (2) Ogee, Ogival (3) Gothic
ONE-DIMENSIONAL
                        (1) Linear (3) Linearized, Quasi-Onedimensional
ONSET
                        (3) Beginning, Appearance, Start
                        (2) Operation, Operational, Operator (3) Operation, Performance
OPERATING
                        (3) Visual(2) Optimal, Optimization (3) Ideal, Perfect, Appropriate
OPTICAL
OPTIMUM
                        (1) Trajectory, Path (2) Orbital, Orbiting
ORBIT
ORDER
                        (2) Ordered
ORTHOTROPIC
OSCILLATING
                        (1) Vibrating (2) Oscillation, Oscillator, Oscillatory (3) Vibrational,
                        Oscillatory
OSCILLATION
                        (1) Vibration (2) Oscillating, Oscillator, Oscillatory
OSCILLATORY
                        (1) Vibratory (2) Oscillating, Oscillation, Oscillator, (3) Wavy, Sinsusoidal,
OUTER
                        (1) External (2) Outermost (3) Outboard, Peripheral, Outlet, Output,
                        Outward
                        (3) External, Peripheral
OUTSIDE
                        (1) Elliptic
OVAL.
PANEL
                        (3) Plate
                        (3) Constant, Variable, Criterion, Value, Factor, Function (2) Partial, Partially, Partly (3) Component, Element
PARAMETER
PART
PARTIALLY
                        (1) Partly (2) Partial, Part (3) Part, Semi
PARTICLE
PERFECT
                        (2) Perfectly (3) Simple, Ideal, Idealized, Simplified, Approximate,
                        Model, Hypothetical, Pure
PERFORMANCE
                        (3) Operation, Operating
                        (2) Period, Periodically, Periodicity
PERIODIC
PERIPHERAL
                        (1) Circumferential
PHASE
                        (3) State
PHENOMENA
                        (2) Phenomenological (3) Effect, Characteristic
PHOTOELASTIC
PITCH
                        (2) Pitching
PITOT
PLANE
                        (2) Planar (3) Direction, Axis
PLANFORM
                        (3) Shape
PLASTIC
                        (2) Plasticity (3) Soft
PLATE -
                        (2) Plating (3) Panel
```

(1) Location, Position, Station (2) Pointed, (3) Area

(3) Curve

POINT

POLAR

```
(1) Location, Point, Station (2) Positioning
POSITION
                        (3 Irrotational, Non-Rotating
POTENTIAL
                        (3) Real, Working, Imperfect, Non-Perfect
PRACTICAL
PREBUCKLING
                        (2) Prebuckled
                        (3) Estimate, Estimation, Assessment, Determination, Calculation, Deduction (3) Ambient, Environment, Environmental, Surrounding, Proximity, Vicinity
PREDICTION
PRESENCE
                        (2) Pressurisation, Pressurized (3) Load, Stress, Force, Loading
PRESSURE
                        (2) Pressure, Pressurization
PRESSURIZED
                        (1) Main, Major
PRIM ARY
                        (1) Law, Rule (3) Theory, Theoretical, Hypothesis, Prediction, Assumption
PRINCIPLE
                        (3) Concept, Condition
PROBLEM
                        (3) Method, Technique, Treatment
PROCEDURE
                        (2) Production (3) Result
PRODUCT
                        (1) Generation (2) Product (3) Propagation
PRODUCTION
                        (3) Distribution, Field, Gradient, Differential, Fluctuation, Variation,
PROFILE
                        Divergence, Slope, Load x Input
                        (2) Programmed, Programming (3) Algorithm
PROGRAMME
                        (2) Propagating (3) Generation, Production, Originating, Source,
PROPAGATION
                        Development, Spread, Radiation, Emission, Excitation
                        (2) Propellant, Propelled(1) Characteristic (3) Quality, Nature
PROPELLER
PROPERTY
                        (3) Near, Close, Adjacent
PROXIMITY
                        (3) Blower, Compressor
PUMP
                        (3) Simple, Simplified, Ideal, Idealized, Approximate, Perfect, Model,
PURE
                        Hypothetical
                        (2) Quantity
QUANTITATIVE
QUASI-CONICAL
                        (2) Radial, Radially
RADIUS
                        (3) Distance
RANGE
RAREFACTION
                        (2) Rarefied (3) Free x Molecular x Flow, Rarefied x Fas x Flow, Semi-
                        Rarefied x Gas x Flow, High x Altitude x Flow, Low x Density x Flow,
                        Wall x Discontinuity x Velocity, Wall x Discontinuity x Temperature
                        (2) Rarefaction (3) Semi-Rarefied, Free x Molecule, Free x Molecular
RAREFIED
                        (3) Speed, Velocity
RATE
RATIO
                        (2) Reaction, Reactive, Reactor (3) Reactive, Active, Action,
REACTING
                        Activation, Activity
                        (2) Reacting, Reactive, Reactor (3) Energy, Force, Action, Behaviour,
REACTION
                        Kinetic, Response
                        (3) Non-Perfect, Imperfect, True
REAL
RECTANGULAR
                        (2) Rectangle
                        (2) Reduced (3) Reduced, Limitation, Limiting, Decreasing, Decrement,
REDUCTION
                        Fall, Control, Minimisation, Minimising
                        (2) Re-entering, Re-Entrant (3) Re-Entering, Entry, Entering, Entrance
RE-ENTRY
                        (3) Characteristic
REFERENCE
                        (3) Improvement
REFINEMENT
                        (2) Reflected, Reflecting (3) Deflection
REFLECTION
REGIME
                        (1) Domain
                        (1) Field (3) Area, Ambient, Regime, Local, Locally, Environment,
REGION
                        Presence, Surrounding
REINFORCED
                        (3) Stiffened
                        (1) Relationship, Correlation (2) Relative, Relatively, Relativistic,
RELATION
                        (3) Coupling
                        (1) Relation, Correlation (2) Relative, Relatively, Relativistic, Coupling
RELATIONSHIP
RELAXATION
                        (2) Relaxational, Relaxing
REQUIRED
                        (2) Requirement
                        (1) Investigation, Study
RESEARCH
                        (3) Storage x Vessel
RESERVOIR
RESPONSE
                        (3) Reaction
                        (2) Restraining, Restraint (3) Retarded, Controlled
RESTRAINED
RESTRAINT
                        (2) Restrained, Restraining (3) Constraint
                        (2) Resultant, Resulting (3) Data, Output, Statistics, Value
RESULT
                        (3) Descent, Descending, Deceleration, Approach, Homing
RETURN
REVOLUTION
                        (1) Rotation
REYNOLDS
```

(3) Annular, Round, Rounded

RING

```
RISE
                          (3) Increase, Increment
 ROCKET
                          (3) Missile, Projectile
 ROLLING
                          (2) Roll, Rolled
ROUGHNESS
                          (2) Rough (3) Irregularity
 ROUND
                         (2) Rounded, Rounding, Roundness (3) Rounded, Blunt, Convex, Semi-
                         circular, Circular, Circularity, Curved, Curvature
ROW
                         (3) Series
RUNNING
                          (3) Operation
 SANDWICH
                         (1) Multilayer, Laminate, Stratiform, (3) Lamination
 SATELLITE
 SCALE
                         (2) Scaling (3) Size, Dimension, Area
 SECOND
                         (2) Secondary (3) Two
SECTION
                         (2) Sectional, Sector, Sectorial (3) Segment, Plane, Sector
SECTOR
                         (2) Section, Sectional, Sectorial (3) Section, Plane, Segment
SELF
SEMIVERTEX
                         (3) Top, Head, Crest, Apex, Crown, Cap, Semi-Vertex, Vertex
SENSITIVITY
                         (2) Sensor
                         (2) Separating, Separation (3) Free, Detached
SEPARATED
SEPARATION
                         (2) Separated, Separating (3) Detachment
SERIES
                         (3) Row, Line, Bank
SHALLOW
                         (2) Shallowness
SHAPE
                         (2) Shaped (3) Configuration, Profile, Contour, Geometry, Form
SHARP
                         (2) Sharpness (3) Intermediate
SHEAR
                         (2) Sheared, Shearing (3) Shearing, Longitudinal x Velocity x Gradient
SHEARING
                         (2) Shear, Sheared (3) Shear, Longitudinal x Velocity x Gradient
SHEET
                         (3) Layer, Path, Core, Line, Trail, Field, Street
SHELL
SHOCK
                         (2) Shocking (3) Blast
SHORT
                         (2) Shortening
SHROUD
                         (2) Shrouded
SIDE
                         (2) Sided, Sidedness
SIDESLIP
                         (3) Cross x Load
SIMILAR
                         (2) Similarity, Similitude
                         (2) Similar, Similitude (3) Similitude, Correlation, Analogy, Analogue (2) Similar, Similarity (3) Similarity, Correlation, Analogue, Analogy
SIMILARITY
SIMILITUDE
                         (1) Simplified, Simplification (2) Simply, Simplified (3) Idealised,
SIMPLE
                         Approximate, Perfect, Model, Hypothetical, Ideal, Pure
                         (1) Simple (2) Simplification, Simply, Simple (3) Idealised, Approximate, Perfect, Model, Hypothetical, Ideal, Pure
SIMPLIFIED
                         (2) Simple, Simplification, Simplified
SIMULATION
                         (2) Simulated, Simulating, Simulator (3) Analogy
SINGLE
                         (2) Singly, Singular, Singularity (3) One
                         (2) Sinsusoidally, Sine
SINUSOIDAL
SIZE
                         (1) Magnitude (3) Dimension
SKIN
                         (3) Membrane, Surface, Shell
SLAB
SLENDER
                        (1) Thin (2) Slenderness (3) Narrow
SLIPSTREAM
                         (3) Downwash, Wake, Vortex, Trail, Afterflow
SMALL
                         (1) Low, Reduced (3) Infinitesimal, Infinitesimally, Vanishing, Minimum,
                        Slightly, Negligible
SMOOTH
                         (2) Smoothing, Smoothness (3) Polished
SOFT
                        (3) Plastic
SOLID
SOLUTION
                        (3) Result, Equation, Procedure, Expression
SONIC
                        (2) Sound, Sounding (3) Sound x Speed, Sound x Velocity
SOUND
                        (1) Acoustic (2) Sounding, Sonic
SPACECRAFT
                        (1) Spaceship (3) Space x Vehicle
SPAN
                        (2) Spanning, Spanwise (3) Length, Extent
SPANWISE
                        (1) Longitudinal, Lengthwise (2) Span, Spanning, (3) Longitudinally
SPECIES
                        (2) Specific
SPECIFIC
                        (2) Species (3) Exact, Particular
SPEED
                        (1) Velocity (3) Flow, Flight, Stream, Mach
SPHERE
                        (2) Spherical, Spherically, Spheroid (3) Spheroid, Spherical x Body
SPHERICAL
                        (2) Sphere, Spherically, Spheroid
SPLIT
                        (2) Splitter
```

SQUARE

```
(2) Stabilization, Stabilized, Stabilizer, Stabilizing, Stable
STABILITY
                        (3) Equilibrium, Quasi-Equilibrium
                        (2) Stability, Stabilized, Stabilizer, Stabilizing, Stable (3) Stability.
STABILIZATION
                        Stabilizing, Control, Equilibrium
                        (2) Stability, Stabilization, Stabilized, Stabilizing, Stable (3) Fin
STABILIZER
                        (2) Stability, Stabilization, Stabilized, Stabilizer, Stable
STABILIZING.
                        (3) Stability, Stabilization, Control, Equilibrium(2) Stability, Stabilization, Stabilized, Stabilizer, Stabilizing
STARLE
                        (3) Equilibrium, Steady
                        (3) Blading, Blade x Row, Blade x Set
STAGE
                        (2) Stagnant (3) Stationary x Flow, Zero x Velocity
STAGNATION
                        (2) Stall, Stalled (3) Flow x Reversal
STALLING
                        (3) Normal
STANDARD
                        (3) Detachment
STANDOFF
                        (1) Condition (3) Problem, Concept
STATE
STATIC
                        (2) Statically
                        (2) Statistical (3) Result, Data, Output, Value
STATISTICS
                        (3) Quasi-Steady, Continuous, Stable, Invariant, Time x Independent
STEADY
                        (2) Stiffener, Stiffening, Stiffness (3) Reinforced
STIFFENED
STIFFENER
                        (2) Stiffened, Stiffening, Stiffness (3) Stiffening
                        (2) Stiffened, Stiffener, Stiffening (3) Rigidity
STIFFNESS
STING
STOKES
                        (3) Vtol
STOT.
                        (2) Straightness (3) Rectilinear
STRAIGHT
                        (1) Deformation
STRAIN
                        (1) Flow, Flux
STREAM
                        (2) Strong (3) Intensity, Endurance, Load x Carrying x Capacity,
STRENGTH
                        Fatigue x Resistance
                        (1) Load, Loading (2) Stressed, Stressing (3) Strain, Concentrated x Force
STRESS
                        (2) Strength (3) Intense, Severe
STRONG
                        (2) Structure
STRUCTURAL
                        (2) Structural (3) System, Nature, Construction, State, Body, Surface
STRUCTURE
                        (1) Investigation, Research
STUDY
SUBJECTED
                        (2) Subject (3) Bare, Open, Loaded, Exposed
                        (3) Subcritical, Incompressible
SUBSONIC
SUCTION
                        (3) Rapid, Instantaneous, Fast, Quick
SUDDEN
                        (2) Supersonically (3) Supercritical, High x Speed
SUPERSONIC
SHPPORTED
                        (1) Mounted (2) Support, Supporting
                        (2) Surfaced (3) Face, Structure, Body
SURFACE
                        (3) Instability, Unsteadiness, Pressure x Rise
SURGE
                        (2) Sustaining (3) Supported
SUSTAINED
                        (2) Sweep (3) Sweptback
SWEPT
                        (2) Sweepback (3) Swept
SWEPTBACK
                        (t) Symmetrical (2) Symmetrical, Symmetrically, Symmetry (3) Regular,
SYMMETRIC
                        Uniform. Steady
                        (1) Symmetric (2) Symmetrically, Symmetry, (3) Regular x Shape
SYMMETRICAL
                        (2) Systematic
SYSTEM
TABLE
                        (3) Chart, Tabulation
                        (1) Pointed (2) Taper
TAPERED
                        (1) Method (3) Process, Operation, Procedure
TECHNIQUE
TEMPERATURE
                        (2) Testing (3) Experiment, Measurement
PEST
THEORETICAL
                        (2) Theory (3) Perfect, Ideal, Model
                        (2) Theoretical (3) Theoretical, Hypothesis, Prediction, Assumption,
THEORY
                        Principle, Law, Rule
                        (1) Thermodynamic (2) Thermally, (3) Heat
THERMAL
THERMOCHEMICAL
THERMODYNAMIC
                        (1) Thermal (3) Heat
                        (1) Nonslender (2) Thickened, Thickening, Thickness (3) Broad, Not x
THICK
                        Slender, Not x Thin
                        (2) Thick, Thickened, Thickening
THICKNESS
                        (1) Slender (3) Narrow
THIN
THREE
```

(2) Three-Dimensionality

(3) Contraction, Contracting x Section, Convergent x Duct

THREE-DIMENSIONAL

THROAT

```
(3) Boost, Resultant x Jet x Force
THRUST
                         (2) Tilting
TITT
                         (2) Timewise (3) Period, Duration
TIME
TORISPHERICAL
                         (2) Trail (3) Rear, Aft
(2) Transferred (3) Flow, Flux, Exchange, Transport
TRAILING
                         (2) Transform, Transformed (3) Transform, Transformed, Conversion
TRANSFER
TRANSFORMATION
                         (3) Discontinuous, Random
TRANSIENT
                         (2) Transit, Transitional (3) Change, Breakdown x Laminar x Flow (2) Transit, Transition (3) Changing, Breakdown x Laminar x Flow
TRANSITION
TRANSITIONAL
                         (3) High x Subsonic
TRANSONIC
TRANSPIRATION
                         (3) Sweat
                          (3) Distribution, Carrying
TRANSPORT
                          (1) Lateral (2) Transversality, Transversely, (3) Chord, Chordwise
TRANSVERSE
                          (3) Method, Technique, Procedure
TREATMENT
                          (1) Delta (2) Triangle (3) Isosceles, Caret
                          (3) Accurate, Exact, Real, Perfect, Accuracy
(1) Pipe, Tubing (2) Tubular, (3) Head, Probe, Hollow x Cylinder,
TRIANGULAR
TRUE
TUBE
                          Hollow x Body, Tubular x Body
                          (3) Falling, Plunging, Sinking, Diving
TUMBLING
                          (3) Tube
TUNNEL
                          (2) Turbulent (3) Mixing, Eddy, Velocity x Fluctuation
TURBULENCE
                          (2) Turbulence
THE BULENT
                          (2) Two-Dimension (3) Two-Dimension
 TWO-DIMENSIONAL
                          (2) Underexpansion (3) Expanded
 UNDEREXPANDED
                          (2) Uniformity, Uniformly (3) Regular
 UNIFORM
                          (2) Uniform, Uniformity
                          (1) Non-Steady (2) Unsteadily, Unsteadiness, (3) Alternating, Sinusoidal, Time x Dependent, Quasi-Unsteady, Variable, Varying, Fluctuating,
 UNIFORMLY
 UNSTEADY
                          Fluctuation, Variation, Discontinuous, Unsteadily, Unsteadiness
                          (3) Flexible, Elastic, Stiffener x Zero, Stiffness x Zero
 UNSTIFFENED
                          (1) Upflow (3) Upload
 UPWASH
                          (3) Value, Applicability, Application
 USE
                          (2) Valued (3) Data, Output, Result, Statistics
 VALUE
                          (2) Variation, Variational, Varying (3) Variation, Varying, Different,
 VARIABLE
                          Alteration, Change, Deviation
                          (2) Variable, Variational, Varying (3) Variable, Varying, Different,
 VARIATION
                          Differing, Change, Changing, Alteration, Fluctuation, Fluctuating,
                          Deviation
                           (2) Variable, Variation, Varying
 VARIATIONAL
                          (1) Non-Uniform (2) Variable, Variation, Variational
 VARYING
                           (3) Direction
 VECTOR
                           (3) Machine, Ship
 VEHICLE
                           (1) Speed (3) Flow, Flight, Stream, Mach
 VELOCITY
                           (2) Vertex (3) Perpendicular, Normal
 VERTICAL
                           (3) Highly
 VERY
 VESSEL
                           (1) Oscillation (2) Vibrating, Vibrational, Vibrationally, Vibratory
 VIBRATION
                           (2) Vibrating, Vibration, Vibrationally, Vibratory (3) Vibrating,
 VIBRATIONAL
                           Oscillating, Oscillatory
                           (2 Viscoelasticity
 VISCOELASTIC
                           (2) Viscous, Viscid (3) Internal x Friction (1) Viscid (2) Viscosity
  VISCOSITY
  VISCOUS
                           (2) Visual (3) Flow x Survey, Visual x Flow x Investigation
  VISUALISATION
                           (1) Eddy (2) Vortical, Vorticity (3) Swirl, Spiral, Spiralling x Fluid,
  VORTEX
                           Vorticity x Layer, Rotational x Flow, Rotational x Fluid
                           (2) Vortex, Vortical (3) Rotation, Spiral, Swirl
  VORTICITY
                           (3) Stol
  TOL
                           (1) Afterflow, (3) Trail
  WAKE
                           (2) Walled (3) Boundary, Bound, Limit
  WALL
                           (2) Wall (3) Bounded, Enclosed
  WALLED
  WATER
                           (2) Waviness, Wavy
  WAVE
                           (3) Triangular, Ramp
  WEDGE
                            (2) Weighting
  WEIGHT
```

(2) Wide, Widely (3) Thickness

WIDTH

WIND WING WITHOUT

YAW YAWED YIELD ZERO

(2) Windward

(2) Winged
(3) Absence
(2) Yawed, Yawing
(2) Yaw, Yawing
(2) Yielding

(3) Negative, Negligible, Without, Absence

APPENDIX 5.3

SINGLE TERM HIERARCHIES

The schedules list single terms only, and include all the terms used in indexing the 1,400 documents, with the exception of the proper names.

The single terms are in natural uncontrolled language, and different verbal forms of a word often appear in different places.

Notation is purely ordinal and in no way reflects the relations between the terms represented.

Plus signs indicate synonyms.

Superordinate, subordinate and coordinate relations are shown by indentation. These hierarchies represent an attempt to interpret hierarchical linkage as an obligatory recall device, that is, an original vocabulary is reduced by condensing species into containing genera, and no option is left in searching. Three stages of reduction are shown in the columns on the right of each page. Further information on these hierarchies is given on pages 69-73

Summary of main sections

```
Particles
A4
A15
      Materials; fuels, metals, non-metals, aggregates.
B68
        State and form of matter
C1
      Astronomical and geophysical phenomena
C30
      Constructs and structures
C32
        Vehicle: aircraft, missiles
D2
          Parts: aerofoils, control surfaces.
D40
        Structural systems, parts and elements
E4
        By shape: bodies
E56
        Machines, engines, auxiliary systems
F1
        Machine components
F80
      Apparatus, equipment: wind tunnels, instruments
Spatial relations; portion, position, direction, shape, dimension, arrangement
H1
      Temporal relations: duration, periodicity, rate.
N1
NLO
      Flying operations, processes, properties.
0a
      Aerodynamics
01
        Flow; types, by speed etc.
064
        Flow elements and phenomena, streamlines, vortices, jets, waves.
0100
        Flow attributes: viscosity, forces, loads, stagnation, wake,
P30
        Aeroelasticity
P42
        Aerodynamic reference parameters: attitude, planform, section etc.
P70
      Mechanics
P84
          Kinematics
Q32
          Force
R1
          Behaviour of deformable bodies
R91
            Vibration, wave phenomena
S23
              Thermodynamic behaviour and properties
S77
      Physical-chemical processes
T14
      General processes, activities, operations, change (of composition, state,
                              location, direction, dimension, etc.)
U59
        Technical operations and processes
U85
        Research
V1
          Experiment
          Mathematical operations
V30
Wa
      Common properties
X30
        Properties of processes (e.g. incipient, continuous, erratio)
X108
      General relations
Y40
      Abstract concepts, form.
```

		First Reduction	Second Reduction	Third Reduction
-	Particles	1	1/8a	1/8a
5	Electron + Beta	3/4		
} ;	Proton Atom	5/6		
-	Isotope	5/6		
,	Ion	7		
}	Molecule	8/8a		
a	Mol			
b	(Structure) Atomic	0) ///	0. /4.	
ic	Molecular	8b/14	8b/14	8b/14
d	Bimolecular			
)	Homonuclear			
a	Nuclear			
.0 1	Monatomic Diatomic			
2	Polyatomic			
3	Polymer			
4	Polycrystalline			
5	Matter + Material	15	15	15/19a
.6	(By use) Pigment	16	16 + 19	09/25
.0 .7	Lacquer	17/18	17/18	+ 27/35
.8	Phosphorescent	1,7,10	.,, .	
9	Ink	19		
9a	Injectant	19a	19a	
0 1	Fuel Methane	20/24	20/24	20/26
2	Ethylene			
3	Hydrocarbon			
4	Methanol			
6	Propellant	26	26	
?7 :8	Explosive TNT	27/28	27/28	
0	Coolant .	30	30	
1.	Lubricant	31	31	
2	Refractory	32	32	
2 a	Oxidizer	32a	32a	
4	(By origin) Electrodeposit	71/75	7./75	
5	Electroformed	34/35	34/35	
	(By constitution)	38/39a	38/39a	38/56
3	Metal		+48/56	+66/68
9	Alloy			
9 a	Bimetallic			
0	Aluminium	40/43	40/43	
L	Duralumin			
2 3	Alcoa Alclad			
) -	Nickel	44/47	44/47	
5	Inconel	44/41	44/41	
6	Nimonic			
7	Rex			
8 -	Brass	48		
19 10	Copper Zinc	49 50		
1	Titanium	50 5 1		
2	Lead	52		
3	Magnesium	53		
4	Molybdenum	54		
5 6	Tungsten Chromium	55 56		
7	Ferrous	5 7	57/65	57/65
8	Iron	58	211 42 .	J1, 4J
88a	Gamma	58a		
59	Steel	59/65		
	Mild Dec ar bu rize d			
0				
0	Columbium			
0 1 1a 2				
0 1 1a 2 3	Columbium Sicromo 2 Timken			
0 1 51a 52 54	Columbium Sicromo 2 Timken Stainless			
50 51 51 62 53 54	Columbium Sicromo 2 Timken Stainless SAE	64	66160	
0 1 51a 52 54	Columbium Sicromo 2 Timken Stainless	66 67	66/68	

		First Reduction	Second Reduction	Third Reduction
	[Metal] [Sodium]			
A70 A71	(Non≈metal) Carbon Graphite	70 71	70/71	70/75
A72 A73	Silica Fused Quartz	72/75	72/75	
A74 A75	Foamed		•	
A76 A77 A78	(Gas) Hydrogen Helium Argon	76 77/80	76 + 81/84 + 87 77/80	76/87
A79 A80 A81 A82 A83	Krypton Xenon Iodine Nitrogen Nitric (oxide)	81 82/84		
A84 A85 A86	Nitrons (oxide) Oxygen Freon	85 86	8 5 86	
B1 B2 B3 B4	(Natural materials) Wood Balsa Plywood Spruce	1/5	1/5	1/66
в5 в6 в6а	Pine Oil Rubber	6 6a	6/6b	
866 88 89	Paper Glycerine Hexachlorethane	6b 8 9	8/9	8/23
B9a B10 B11	Inorganic (compounds) Water Steam	9a 10 11	9a + 18/23 10/11	
B12 B13	Air Seeded	12/13	12/13	
B14 B14a B15 B16 B17 B18 B19 B20 B21 B23	Salt Carbonate Chloride Iodide Tetrachloride Amonium Monoxide Dioxide Silicate Fluorochemical	14/17 18 19 20 21 23	14/17	
B24 B25 B26	Ceramic Glass Glassy	2 4 2 5/2 6	24 25/26	24/28
B27 B28	Fibre Fibrous	27/28	27/28	
B30 B31 B32 B33	(Plastics) Nylon Melamine Cellulose Acetate	30 31 32/33	30/33+37/38 +46/50+55/59	30/59
в34 в35 в36	Resin Epoxy Polyester	34 35 36	34/36 +39/45&	
B37 B38 B39 B40 B41	(Polyester (plastics)) Mylar Paraplex Phenolic (resins) Bakelite Formica Asbestos	37 38 39/45a		
B42 B43 B44 B45 B45a	Raybestos Refrasil Astrolite Thermold J.			
в46 в47 в48	(Methyl methacrylates) Perspex Plexiglass Lucite	46/48		
B49 B50	Teflon Polystyrene	49 50		
B51 B52	(Reinforced plastics) Fibreglass Cimcinnati-Testing-Laborat	51/54 ory-Material	51/54	

	[Resin] [Reinforced plastics]	First Reduction	Second Reduction	Third Reduction
	[Cincinnati-Testing-Laborator			
953 854	Papreg Compre <i>g</i>			
B55	Castolite	55 56		
B56 B57	Haveg=Rocketon Hysol			
B58	Lexam	57 58		
B59	Lockfoam	59		
	(Ground: aggregates by particle s	ize)		
в60	Sod	60/67	60/67	60/67
в63 в64	Sand Gravel			
B65	Dirt			
в66	Dust			
в67	Debris			
- 60	(State and form of matter)			
в68 в69a	State + Condition Phase	68/68 a	68/68a	68/87
в69	Solid	69/69a	69/69a	
B69a	Crystal			
В70 В71	Non solid Fluid	70 71/72	70 71/72	
B7la	Liquid	10/12	1 1/ 12	
В72 В73	Moisture Gas + Gaseous	77/71	97/94	
B74	Vapour	73/74	73/74	
B75	Rarified	75	75/76	
В75 а В76	Semi⊸rarified Vacuum	75a 76		
B77	Film	77	7 7	
B78	Bubble	78	78	
B81 B82	Foam Suspension	81/83	81/83	
в83	Smoke			
B85 B87	Grain Plasma	85 87	85 8 7	
Do (I I COM CL	87	87	
Cl	(Astronomical and Geophysical phenor			
C2	Astrophysics Space	1/2	1/4	1/4
C3	Interplanetary	3		
C4 C5	Luni=solar Sun + Solar	4	F	
c6	Planet+Planetary	5 6/11	5 6/13	5 6/13
a 9	(Non-terrestrial)			
c8 c9	Venus Mars + Martian			
C10	Jupiter			
Cll Cl2	Saturn Meteor	40/47		
C13	Meteorite + Meteroid	12/13		
C14	Earth	14	14	14/15
C15 C16	Moon + Lunar Atmosphere+ Atmospheric	15 16/17	15 16/18+21/:	23a 16/23a
C17	Equational	10, 11	10/ 10/21/	L) a 10/ L) a
c18	Ionosphere	18		
	(Atmospheric phenomena)			
C19 C2 0	Wind Gust	19/20	19/20	
C21	Cloud	21/23a		
C22	Snow	- 1, - 2, -		
C23 C23a	Fog Ice			
C24	Ground	24		
C25	Sea	25		
	(Constructs and Structures)			
C30	Bridge Chimney	30	30/31	30/31
031 032	Vehicle	31 32	32	32
C33 C34	Ship + Marine + Maritime	33	33	33
034 035	Hovercraft Saunders-Roe SRN1	34/35	34/35	34/35
c36	Aircraft + Airplane	36/36a	36/36a	36/41+53/58
036a	Acronautical (vehicle)			- 01 -
038	Non-lifting	38	38	
C39	Winged	39	39/40	
C40	Wingless	40		
C43.	Finned	41	41	
C42	Civil	40/::	10/::	10/53
U44Z	Civil	42/44a	42/44a	42/52

		First Reduction	Second Reduction	Third Reduction
-1-	[Vehicle] [Aircraft] [Civil]			
C43 C44	Airliner Caracelle (Military)			
C45 C46 C47 C48 C49 C50	Bomber Fighter F-100 Boulton-Paul (Delta PIII) X-15 B ¹ 7A			
051 052	Avro (707A) F8U3			
053 054 055	(Winged) Monoplane Parasol Fairchild 22	53/55	53/56	
c56	Biplane	56		
C57 C58	Tailless Canard	57 58	57/58	
C59	(Aircraft) Helicopter	59	59	59
060 061 062 063 064 065	STOL VTOL Curtiss=Wright Bell Doak Vertol	60 61/66	60 61/66	60/66
066 066a 067 068	Hiller Glider Spacecraft + Spaceship Manned	66a 67+69/7 1 68	66a 67+69/72 68	66a 67/89
C69 C70 C71	Centaur Able X-15			
072 073 0714 075 076 077 078 079	Capsule Launch (vehicle) Satellite Fuelling Midas Atlas Explorer Sputnik Vanguard	72 73 74/81	73 74/81	
031 082 083	Discoverer Missile + Projectile Ballistic	82/83	82/89	
084 085 086 087 083 089	Guided Homing TREM ICEM Titan Probe	84/85 86 87 88 89		
D2 D3 D4 D5	(Parts of aircraft) Airframe Airfoil Wing Winglike	2 3 4/6	2 3 + 11 4/6	2+32/38 3/11
D6 D7 D3 D9	Nonweiler Tail Slabtail Tailpl a ne	7/9	7/9	
D1.1.	Foreplane (Control surfaces) Aileron	12	12	12/13a
D15 D16	Rudder Elevator	1 5 16/17	15 16/17	
D1.7 D1.3 D1.9 D20 D21 D21a	Geared Elevon Flap Split Thwaites Programmed (flap)	18 19/21a	18+22 19/21a	

		First Reduction	Second Reduction	Third Reduction
	[Programmed (flap)]	110440 11011	Ticker of the time	ACCASE OF TOTAL
D22	Horn	22		
D23 D24	Slat Slot	23/24	23/24	
D25 D26	Tab Spoiler	25 26/27	25 26/27	
D27	Bang-bang			
D28 D29	Stabilizer Fin	28+30/31 29	28/31a	
D30	Recoverable			
D31 D31a	Non-recoverable Mass balanced	31a		
D32	Fuselage	32/33	32/33+38	
D33	Cabin	7.4	7.4	
D34 D35	Nacelle Cowling	34 35/36	34 35/36	
D36	Spinner	-		
D37	Shield	37 38	37	
D38	Store (Structural systems, parts and elemer			
	(System)			
D¼1	Structure Built-up	40+57a 41	40/42 +45/46	40/57b
D42	Reinforced	42	+54/57b	
D43	Stratiform + Laminate + Sandwich	43/44	43/44	
D43a D43b	Lamination +Multilayer Layered			
D43e	Multilayered			
D44	Multiply	A.E.		
D45 D47	Honeycomb Cellular	45 47/48	47/48	
D43	Multicell			
D50 D51	Multiweb MW÷l	50/51	50/51	
D52	Bay	52/53	52/53	
D53	Multibay	5.4		
D54 D55	Multirib Multispar	54 55		
D56	Monocoque	56		
D57	Monolithic	57 5 7 -		
D57a D57b	Boxtype 1 Geodetic	57a 57b		
	(By core condition)			
D58 D5 9	Hollow Filled	58/61	58/61	58/61
D60	Pressurized			
D6 1	Unpressurized			
D63	(Structural Parts) Deck	63	63	63/74a
D64	Doors	64/65a	64/66	0)/ / 40
D65	Windows			
D65a D66.	Pane Roof	66		
D63	Stringer	68/70a	68/73	
D69	Bulkhead			
D70 D70a	Frame Framework			
D71.	Spar	71/72		
D71 a D72	Shaft Rib			
D73	Web	73		
Dγħ	Fairing	74	74	
D7/4a	Flange 9	74a	74a	
D7/H5	(Structural elements) Filler	74b	745/77	746/82
D75	Support	75/ 7 7	+80/82	140/02
D76	Mounting			
DY7 DY8	Stand Attachment	78	78	
D30	Ramp	80	, ~	
PGS DGI	Step Loop	81 82		
D'33	Sheet	83	83	83/88
D/3/4	Plate			
DSha DSS	En dplate Panel	84/84a 85/86	84/84a 85/87	
086	Handley-Page	-	- 2/ - 1	
D37 D38	Slab Shell	87 88	88	
U A)	ORGIL	oo	00	

		First	Second	Third
		Reduction	Reduction	Reduction
	[Shell]			
DS9	Bar	89/93	89/96	89/96
D9 0	Wire			
D91	Woven			
D92	Rod			
D93	Coupon Strip	94/96		
D94 D95	Ribbon	7 " 7 -		
D96	Band			
D97	Ноор	9 7	97/100	97/100
D99	Stiffening (ring)	99 100/100a		
D100	Stiffener Backing	1007 1004		
D100a D101	Beam	101/102	101/102	101/105
D102	Cantilever			
D103	Strut	103/104	103/104	
D104	Column	105	105	
D105	Truss			
	(Structures characterized mainly by sha	pe)		
El.	Bodies	1/4+23/26	1/4+23/26	1/4+22/26
E3	Block			
E4	Box			
77.6	(Body of revolution) Cylinder	6/6a+9	6/6a+10	6/21
E6 E6a	Quasi-cylinder	u, u,		
E7	Tube + Pipe + Tubing	7/8	7/8	
E9	Drum	40		
E10	Disc	10	11/13	
Ell	Cone	11/13	11/1)	
E12 E13	Frustum Spike			
E14	Sphere	14/15	14/16	
E15	Ball			
E16	Hemisphere + Dome	16	17/18	
E17	Ellipsoid	17/18	117.10	
E18	Semi-ellipsoid	19	19	
E19 E20	Ogive Torus	20	2Ó	
E21	Paraboloid	21	21	
E22	Boattail	22	22	
E23	Maikapar			
E24	Thor-Able			
E25 E26	Sears—Haack Yo-Yo			
1320	(Structures by edge, support and jur	oction)		
E27	Clamped	27	27/29	27/42
E27a	Jointed	270/29		
E23	Pinned			
E29	Pinjointed	70	30/31	
E30	Hinged	30 31	JU/ J1	
E31 E32	Tilting Flanged	32/32a	32/32a	
E32a	Unflanged	2		
E33	Free-free	33/34b	33/34a	
E33a	Unloaded			
E34	Loose			
E34a	Floating Adjustable			
E34b E35	Fixed	35	35/38	
E36	Rigid + Inelastic + Unelastic	36/38	+42	
E36a	Inexorable			
E36b	Rigidly			
E36c	Binding Prescribed			
E36d E36e	Non-elastic			
E37	Semi-rigid			
E38	Stiffened	**	70	
E39	Unstiffened	39 40/41	39 40/41	
E40	Supported + Mounted	40/41	40/41	
E40a E40b	Hooped Fremed			
E41	Cantilevered			
E42	Clamshell	42		
est es	(Structures by various other princi	ples) 43/44a	43/44u	43/51
E43 E44	Segmented Sectorial	7// 779	.2/ 1	
E44a	Lobed	,		
E45	Faced	45/46b	45/46a	
E46	Surfaced			

		First	Second	Third Reduction
,	[Surfaced]	Reduction	Reduction	Reduction
E46a	Sided			
E46b	Cornered	46 -		
E46c E47	Edged Covered	46c 47/49	47/49	
E48	Shielded		.,, .,	
E48a	Walled			
E49 E51	Ducted Stressed	51	51	
.E56	Machine + Mechanism* + Plant*	5 6/57	56/59	56/73a
E57	Mechanism Plant	58/59		
E58 E59	Power Plant	J01 JJ		
E60	Engine + Motor	60	60	
E61	(Gas turbine)	61/62	61/62	
E65	Turbojet Propulsive (jet)	0., 02	J., J.	
	(Direct combustion)	(7/160	67/60.77-	
E63 E64	Rocket Midcourse	63/69	63/69+73a	
E65	Retrorocket			
E66	Sounding (rocket)			
e68 e69	Sail (i.e. solar sail) Magnetoplasma			
E70	Reactor	70/73	70/73	
E71	Fission			
E72	Fusion Nuclear			
E73 E73a	Booster 2	73a		
	(Auxiliary systems)	m . lm/	9 196.90	74/90
E74	Compressor + Blower + Turbomachine Jumo	74/76	74/76+79	74/80
E75 E76	Multistage			
E77	Fan	77	77	
E79	Pump Turbine	79 80	80	
E80 E82	Exchanger (i.e. Heat exchanger)	82	82/84	82/88+96
E83	Cooler	83/84		
E84 E85	Aftercooler Generator	85	85	
E86	Convertor	86	86	
E88	Gyroscope	88	88	
=0.0	(Control systems)	89/91	89/91	89/91
E89 E90	Servo Relay	0),).	-,,,,	-2.2
E91	Autopilot		20101	00/04
E92	Weapon Warhead	93/94	92/94	92/94
E94 E96	Oscillator	96		
	[Machine components] (Fixed)			
Fl	Baffle	1/3	1/3+5/6	1/7
Fla	Foil			
F2	Screen Grid			
F3 F4	Cascade	4	4	
F5	Diaphragm	5		
F6 F7	Filter Stator	7	7	
F ((Reciprocating)			
F8	Piston	8	8	8/13a
F10	(Rotating) Rotor	10/13a	10/13a	
F1.2	Propeller	, ., ., .,		
F12a	Multipropeller			
F13 F13a	Shank Shaft			
rija	(Parts of compressor)			
F15	Impeller	15 16/18	15 16/18	15/22a
F16 F17	Stage Stacking	10/10	10/ 10	
F18	Uprating			
F19	Hub	19	19	
F20 F21	Overhang (Section) Blade + Blading + Vane	20 21/22a	20 21/22a	
F2la	Guide	on ty to the CO		
F22	Splitter			
F22a	Multiblade			
mat.	Outprogrammen recognist contract and the second sec	24	24+27/30	24/30b
F24 F26	Gear Decoupling	26	26	
F27	Bearing	27/29		

		First Reduction	Second Reduction	Third Reduction
	[Bearing]			
F28	Journal			
F29 F30	Slider Pivot	30	30	
F30a	Crank	30a/30b	30a/30b	
F30b _	Flywheel (Container)			
F32	Cover	32/32b	32/39a	32/39a
F32a	Shroud			
F32b F33	Sheath Tank	33/37		
F34	Receiver			
F35 F36	Vessel Reservoir			
F37	Unfired	/		
F38	Chamber	38/ 3 9a		
F39 F39a	Plenum Can			
F40	Seal + Closure	40/42	40/42	40/45
F41 F42	Plug Cap			
F43	Valve	43	43/43a	
F44	Spring	44/45	44/45	
F45 F46	Belleville Joint	46/47	46/53	46/53
F46a	Bond			
F47 F48	Fit Rivet	48/49b		
F49	Bolt	4-7-12-		
F49a	Pin			
F 49b	Screw Flexible			
F50	Hinge Bellows (Joint)	50 51/52		
F51 F52	Omega (Expansion joint)			
F53	Knuckle	53		
F54	Passage + Corridor	54	54	54/72
F 55	Capillary	55 57/60	5 5 5 7/ 62	
F57 F58	Entrance + Inlet Opening	51700	71/02	
F59	Orifice			
F60 F61	Intake Scoop	61/62		
F62	Sugar	ŕ		
F63	Outlet + Exit	63 64/65	63 64/65	
F64 F65	Channel + Duct + Ducting Interblade	04/0)		
F66	Diffuser	66/67	66/67	
F67 F63	Vaneless Nozzle	68/7 1a	68/71a	
F69	Overexpanding			
F70	Laval			
F71 F72	Contoured Venturi	7 2	72	
F73	Injector	73	73	73/79
F74 F75	Ejector Afterburner	74 75	74 75	
F76	Muffler + Silencer	76/77	76/77	
F77	Wideangle Throttle	78/79	78/79	
F78 F79	Accelerator	10/17	19/15	
F80	Apparatus + Equipment + Facility* +	- Device* 80/81	80/81	80/81+83/83 b
F80a F30b	Facility Device			
F81	Laboratory		22	00.04/04
F32 F33	Test + Testing (equipment) Model	82 83/83b	82 83/83ъ	82+84/91
F33a	Shrouded	0), 0),	- 5,7 - 52	
F33b	Non-separating (model)	84+89/91	84/91	
F84 F85	(Wind) Tunnel Blowdown	85	04/ 51	
F37	Tailored	87		
F83	Gun			
F89	NPL			
F90	Ame s			
F91.	ARA			
G2 G4	(Parts and components of wind Settling (chamber) Driver (gas) + Driving (gas)	tunnel) 2 4	2/7a+12/12a	1/15

		First Reduction	Second Reduction	Third Reduction
	[Driver (gas) + Driving (gas)]	4 5/6		
д5 с б	Heater Drier	2/ 6		
G7	Evacuator	7 7a		
G7 a	Barrel (Wall)	•		
G 8	Sidewall Liner	8/9	8/9	
G9 G 12	Cell	12/12a		
G12a	Dashpots			
-	(Model support)			
G13	Šting	13/15	13/15	
G15 G17	Jack Instrument + Instrumentation*	17/18+22/22&	17/18+22/22a	17/23
G17a	Instrumentation	+41a+52/53	+41a+52/53	+41a
G18 G20	Accelerometer Radiator	20	20	+52/53
G21	Ribbon (Vibrating)	21	21	
022 022a	Indicator Sensor			
G23	Corrector	23	23 24	24/34a
G24 G25	Measuring Gauge + Meter	24 25/29a	25/34a	+41
G26	Extensometer			
G27 G28	Manometer Piezometer			
G29	Thermocouple			
G29a G30	Pickup Flowmeter	30/32a		
G30a	Pitot (tubes)			
G31 G32	Pitometer Anenometer			
G32a	Potentiometer			
G33 G34	Dampometer Theodolite	34/34a		
G34a	Kinetheodolite	75	35/40	35/40
G35 G36	Optical Photographic	35 36/40	<i>)))</i> / 40	<i>))/ 40</i>
G36a	Photograph Camera			
G37 G38	Photomultiplier			
G39 G40	Television (Motion) Picture			
G41.	Interferometer	41	41	
G41a	Pendulum [Electric, Electronic equipment]	41a		
G42a	Circuitry	42a + 51	42/46	42/51a
G43 G44	Antenna Rada r	43 44	+51/51a	
G45	Microphone	45		
G46 G47	Cyclotron Cathode (ray tube)	46 47		
G48	Oscilloscope	48/49		
G49 G 51	Oscillograph Conductor			
G5la	Radiotrajectograph			
G52 G52a	Transducer Pickup			
G 53	Multiplier			
	(Spatial relations of parts)		. •	
H1 H2	Portion Whole	1/2a	1/2a	1/10
H2a	All			
Н3 Н4	Part Component	3/6	3/6+8/10	
H 5	Element			
H5a H5b	Sample Specimen			
H5c	Piece			
н6 н8	Factor Fraction	8/10		
Н9	Half			
H10 H11	Quarter Section	11/12	11/12	11/14
H12	Slice			
H13 H14	Sector Segment	13 14	13/14	
,	(By Position)			
	(Vertical plane)			
н16 н17	Top Apex + Vortex	16+19/22 17/18a	16/22	16/26a
н18	Semi apex	1// 108		

		First Reduction	Second Reduction	Third Reduction
H18a H19 H20 H21 H22 H23 H24 H25 H26 H26a	Semi apex Semi vertex Head Crest Crown Peak Bottom Base Bed Foundation Foot	23+25/26 a 24	23+25/26 a 24	
H27 H28 H29 H30 H31 H32 H33 H34	(Longitudinal plane) Front Forepart + Forebody + Foreplane Bow Nose Back Afterbody Rear Aft	27 28 29 30 31+33/34 32	27/29 30 31/34	27/35 a
н35 н35а	(Lateral plane) Side Sidedress	35/35a	35/35&	
H36 H37 H38 H39 H40	(Outer) Face Surface Undersurface Interface Skin	36 37/38 39 40	36 + 39 37/38 40/41a	36/42
H41 H41a H42 H43 H44	Membrane Envelope Wall Layer Sublayer (Inner)	41/41a 42 43 44	42 43/42	43/47
н45 н46 н47	Interior Core Kernel	45/47	45/47	÷
H48 H50 H52 H53 H54 H55 H56	(Extremity) End Tip Edge Rim Lip Leg Stem	48 50 52/54 55 56	48 50 52/54 55/56	48/56
H57 H58 H59 H60 H61 H62 H62a H63	(Prominence) Projection Protuberance Excrescence Pimple Ridge Corrugation Heap Teeth Bump	57 58/65	57/65	57/65
н65 н65а н66 н67 н68	Lobe Obstacle Depression Notch Crack	66/68	66/68	
н68а н69 н70 н71 н72 н73	Groove Hole + Perforation* Perforation Gap Cavity Cutout	69/73	69/73	
H74 H77 H78 H79 H80 H80a	(Connections) Corner Kink Neok Throat Knee Waist	74/77+80a 78 79 80/80a	74/77+80a 78/79 80	74/80a

		First Reduction	Second Reduction	Third Reduction
H95 J1 J1a J1b J2 J3 J4 J5	Spatial relations Spatial Environment Presence Situation Environmental Ambient Surrounding About	95+1/5	95 + 15	95 + 1/10
J6 J8 J9 J10	(With similar conditions) Regime + Domain Belt Region + Field Zone	6+9/10 8	6/10	
J11 J11a J12	Position + Location + Point + Station Located Local	11/11a +83/83b	11/15 +83/83b	11/18 +80/83b
J12 J13 J14 J15 J16 J17 J18	Locally Positioning Spot Line Plane Plateau	15 16 17/18	16 17/18	
J20 J22 J23	Path + Orbit + Trajectory Circuit Course	20/21+23 22	20/21+23 22	20/25
J24 J25	Apogee Perigee	24 25	24/25	
J25a J25b J26 J27 J27a J28	(By proximity, distance) Isolated Alone Proximity Contact Tangency Adjacent	25a 26 27/28a	25 a 26+29 27/28 a	25 a/ 3b
J28a J29 J30 J31 J32 J33	Wetted Vicinity Near Nearest Nearly Close	29 30/34	30/34	
J34 J34a J35 J36	Closely Distant Far Off	34a/35 36	34ª√35 36	
J37 J38	(In vertical plane) Level Altitude	37 38	37 38	37/46
- Ј38а Ј39	Over Above	38 a/40	38a/40	
J40 J41 J42 J42a	Upper Below Lower	41/42a	41/42a	
J42a J43 J44 J45 J46	Lowest Under Submerged Underwater Buried	43/46	43/46	
J47 J48 J49 J49a	(In horizontal plane) Ahead Leading Behind Following	47 48 49/49a	47/48 49/50	47/50

		First Reduction	Second Reduction	Third Reduction
J50	[Following] Trailing	50		
J51 J51n J51b J52 J52a J53 J54	(By Centrality) Central Centrally Centred Middle + Centre Mid Midpoint Midplane	51/54	51/55	51/73
J 56	Axis	56/56a		
J56a J57 J58 J59 J60 J60a	Centreline Bisector Median Inboard Peripheral + Circumferential Fringe	57 58 59 60/60a	59 +61 60/60a	
J61 J62 J63	Outboard Outermost Extremal	61 62/64	62/64	
J64 J65 J65a	Beyond Inner + Internal In	65/66a	65/66a	
J66 J66a J67 J68 J68a	Internally Within External + Outer Externally Outside	67/68a	67/68a	
J70 J71 J72	Boundary Bound Margin	70 71/73	70 71/73	
J73 J74 J75 J75a J75c	Limit Between Interstage Interangular Intermolecular Intramolecular	74/75d	74/75d	74/79
J75d J76 J77 J78 J79	Intersection Junction Node Nodal	76/77 78/79	76/79	
J80 J81 J81a J83a J83a J83b	[Terrestrial] Latitude Meridian Meriddonal New Mexico France U.K.	80 81/81a	80 81/81a	
J84	(Directional relations) Direction	84/85	84/89a	84/892
J85 J86 J87 J88 J89	Orientation Directional Unidirectional Direct Indirect	86/89&		
J89a J90 J91	Non⊷direct Angle Incidence	90/91a+93	90/91a+93	90/93
J91a J92	Attack Slope + Gradient Setting	92	•	
J93 J94	Axial + Axially* Axially	94/94&	94/976	94/102
J94a J95 J96 J97	Axialiy Coaxial Uniaxial Multi axial	95/9 7 a		
J97a J97b J98	Polyaxial Unaxial Radial Radially	97b 98/98a	98/98a	
J98a J99	Centrifugal	99	99/100	
J100 J101 J102	Centripetal Rot ary Clockwise	100 101/102	101/102	
Kl K2 K2a	Lateral + Transverse + Latera Chordwise + Transv Spanning		1/2d	1/5d

		First Reduction	Second Reduction	Third Reduction
K2b	[Spanning] Laterally			
K2c K2d	Transversely Across			
K3 K4	Longitudinal + Lengthwise + Spanwis	e 3/4	3/4	
K5	Longitudinally Edgew i se	5/5a	5/5a	
к 5а кб	Along Forward	6/7	6/7	6/11
к7 к8	Towards Backward	8/9	8/9	,
К9	Rearward		-	
KlO Kll	Reverse Counter	10/11	10/11	
K12 K13	Up Down	12 13/14	12/14	12/24
K14	Incoming	14/16	14/17a	
К15 К16	Inward Into			
Kl.7	Outward	17/17a		
K17a K18	Onto Through	18	18	
K19	Tangential	19	19	
K20	Streamwise	20	20/22	
K21 K22	Upstream Downstream	21 22		
K23	Leeward	23/24	23/24	
K24 K27	Windward Shape	07/70	07/70	07/70
K28	Geometry+Geometrical*+Geometrically	27/30 *	27/30	27/30
K28a K28b	Geometrical Geometrically			
K29	Form			
K29a	Mode			
K30	Shaped	70 /70	70. /70	70 - /742
K30a K31	Symmetry Symmetrical + Symmetric	30a/32	30a/32	30a/34b
K32	Symmetrically			
K33	Axisymmetric + Axisymmetrical	33	33	
K34	Antisymmetric + Antisymmetrical + + Asymmetrical + Non-Symmetric	34/34a	34/34b	
K34a	Unsymmetrical			
K34b	Non-axisymmetric-Non-axisymmetric	al 34b		
	/Z dimensionel)		•	
K35	(3-dimensional) Non-rotational	35/37	35/37	35/38
K35a	Octahedral			
к36	Prismatic			
к37 к38	Pryamidal Wedge	38	38	
K39	Revolution + Rotation	39	39/41	39/64
K41	Rotationally	41		
K42 K43	Cylindrical Tubular	42/43	42/44	
K44	Quasi-cylindrical	44		
K45	Conical + Coned + Conically*	45/45a	45/45b	
K45a K46	Conically Quasi-conical	46		
K47	Spiked	47	47	
к48	Toriconical	48	48+59+63	
K49	Helix Helical	49/51	49/51	
K50 K51	Spiral			
K52	Spherical	52/53	52/ 53	
K53	Spherically	E 4 / E 77	E A / E 7	
K54 K55	Spheroid Oblate	54/57	54/57	
к56	Oblateness			
к57 к58	Prolate Hemispherical , Hamispheric	58/592	58/68h	
	Hemispherical + Hemispheric	58/58b	58/58b	
K58b	Hemispherically	E0.		
K59 K60	Torispherical Ellipsoidal	59 60/61	60/61	8
K61	Semi-ellipsoidal			
K62	Ogival	62	62	•
к63 к64	Toroidal Paraboloidal	63 64	64	
11.04		~ ~	~~	
٠.	(By extremities, edge condition)			

⁽By extremities, edge condition)
K65 Pointed + Tapered

		First Reduction	Second Reduction	Third Reduction
	[Pointed + Tapered]	65+68/69	65+68/69	65/86
K66	Crisp	66/67	66/67	93/00
к67	Crisped			
к63 к69	Taper Tapering			
K70	Sharp + Sharpness*	70/71	70/71	
K70a	Sharpness		•	
K71	Abrupt Blunt + Blunted + Bluntness*	73/75	73/75	
К73 К74	Bluntness	10/10	12/12	
K75	Bluff			
K76 K77	Untapered Unblunted	76/77	76/77	
к73	Hammerhead	78	78÷83	
K79	Round	79/80	79/80	
K79a K80	Roundness Rounded			
K31	Out-of-roundness	81	81	
K93	Headed + Nosed + Capped	82	82	
K33a K34	Based Boattailed	83 8 3 a	83a	
K85	Boattailing	84/85	84/85	
к86	Bevelled	86	86	07/00
к37 к88	Cut Cropped	87/90	87/90	87/99
k39	Clipped			
K90	Truncated	04	0.4	
K91 K92	Non÷truncated Drooped	91 92	91 92	
K93	Extended	93/99	93/99	
K94	Elongated			
к95 к96	Flare Flared			
K97	Skirt			
к93	Skirted			
к99 _	Stretched (2-dimensional)			
Ll	Profile	1/3	1/3	1/3
L2	Contour	å	4+15/19	4/28
L4 L5	Rectilinear Swept	4 5	5/9	4/20
L6	Sweptback	6/7	***	
L3 L10	Sweptforward Unswept	8/9 10	10	
FII	Triangular + Delta + Triangle*	11/12	11/12	
Llla	Triangle			
L12 L13	Caret Rectangular	13/14	13/14	
L13a	Rectangle	12/ 14	.,, .,	
L14	Square	45		
L15 L16	Rhombic Sextic	15 16		
L17	Trapezoidal	17		
L18	Diamond Arrow	18 19		
L19 L20	Polygon	20"24	20/24	
L21	Polygonal			
L22	Hexagonal Octagonal			
L23 L24	Dodecagonal.			
L25	Equilateral	25/26	25/26	
L26 L27	Isosc@les Cruciform	27/28	27/28	
L28	Cross	21/20	2// 20	
L28a	Curvilinear	28a/31	28a/31	284/51
L29 L29a	Circular Quasi-circular			
L30	Circularity			
L31	Supercircular	70	70	
L32 L33	Non-circular Annular	32 33/35	32 33/35	
L34	Annulus))) '	111 11	
L35	Ring	-7	3/ /2m	
L36 L37	Horseshoe Concave	36 37	36/37	
L38	Convex	38/40b	38/40ъ	
L39	Biconvex			
L40 L40a	Lenticular Cambered			
$\Gamma^{l_{1}0p}$	Uncambered			
L42	Fusiform Oval + Elliptic	41 42/45	41 42/45	
L43	Ellipse	46/47	46/47	
-	*			

		First Reduction	Second Reduction	Third Reduction
	[2-dimensional]			
	[Curvilinear] [Ellipse]			
L44	Elliptical			
L45 L45a	Ellipticity Ogee	45a/45b	45a/45b	
L45b	Gothic			
L46 L47	Parabola Parabolic	46/47	46/47	
L 48	Hyperbola	48/49	48/49	
L 49 L50	Hyperbolic	F0	50	
	Hypelliptic	50	50	
L51	Semi-circular	51	5 1	
	Fr + + T			
L52	[Letter] I	52/56	52/56	52/56
L52a	T		3 3-	J/ J-
L54 L55	V W			
L55a	· Z			
L56 L56a	Lamda Theta			
	[One-dimensional]	- 4 -		
L 57 L 58	Straight Straightness	57/58	57/60	57/72
L60 L61	Out-of-straightness Curved + Curve* + Curvature*	60 61/63	61/66a	
L62	Curve Curve + Ourve	01/03	01/008	
L63 L64	Curvative Arc	64/65		
L65	Subarc	64/65		
166 166a	Sinusoidal	66/66a		
L67	Sinusoidally Bent + Bend*	67/68	67/72	
L68 L69	Bend	(0/70		
L69a	Angular Angled	69/72		
L69b	Angularly			
L70 L71	Orthogonal Orthogonally			
L72	Orthogonality			
L73	(Surface features) Flat	73/76	73/76	73/84a
L75	Planar	,	,	12.
L76 L77	Co-planar Non-planar	77	77	
L78	Smooth	78	78	
L79 L80	Rough + Roughness* Roughness	79/80	79/80	
L81	Corrugated	81/820	81/820	
L82 L82a	Wavy Wrinkled			
L82b	Unwrinkled			
L82c L83	Waviness Slotted	83/84a	83/84a	
L84	Perforated	0)/ 044	0)/ 044	
L84a _	Drilled (Cross section area)			
L85	Convergent	85	85	85/86
L86	Divergent	86	86	
L87	(Dimension) Dimension	87	87/91	87/91
L88	Magnitude + Size	88/91	01/).	0,, 5.
L89 L90	Amplitude Bulk			
L91	Scale			
L92 L92a	Length Inch	92/93	92/93	92/100 o
L93	Distance			
L94 L95	Slenderness Thickness	94	94/95+100 o	
L9 6	Thickness Height	95 +1 00 0 96	96/97	
L97	Depth	97		
L98 L99	Perimeter Circumference	98/99	98/99	
L100	Radius	100/100b	100/100ъ	
L100a L100b	Bore Diameter			
L100c	Width	40+4+==	****	404 /
L101	Area Extent	101/102m	101/102a	101/102a
-				

		First Reduction	Second Reduction	Third Reduction
	[Extent]			
L102a L103	Extensive Volume	103/103a	103/103a	103/103&
L103a L104 L105	Volumetric Mass Weight	104 105/106&	104/106a	104/107
L106a L107	Pound Density	107	107	
	[Dimensions (spatial)]			
M8a M9 M10	Moderate Medium Large	8a/9 10/11	8æ/9 10/12	8a/12
M11 M12	Super Macroscopic	12		
M13 M14	Long Short	13 14	13 14	13/14
M15 M16	Slender + Thin Narrow	15/16	15/17	15/20
M17 M18 M19 M19a	Not-so-thin + Not-so-slender Thick + Non-Slender Wide Widely	17 18 19/20	18/20	
M20	Broad			
M26 M27 M27a	[Dimensions] Deep Shallow Shallowness	26 27/27a	26 27/27®	
31777	[Spatial relations] [Arrangement] Horizontal	73&	73a	73a/85
M73a M74 M75	Non-inclined Vertical	74 75/76a	74 75/760	
M76 M76a M76b	Perpendicular Perpendicularly Normal			
M76c M77 M78	Right Inclined + Inclination* Inclination	77/78	77/78+82/84a	
м79 м80	Oblique Skew	79/81	79/81	
M81 M82 M83	Skewed Sloped Canted	82/84m		
M84 M84a	Steepest			
м85	Inverted	85	85	
M86 M87 M87a	Lattice Mesh Network Chessboard	86/876	86/876	86/88
M87b M88 M90 M91	Crossed Opposite + Facing Opposed	88 90/91	88 90/91	90/94a
M92 M93a M93b	Parallel Asymptotic Asymptote Asymptotically	92 93/94	92 93/94a	

		First Reduction	Second Reduction	Third Reduction
	Spatial relations		,	
	Arrangement			
Moli	[Asymptotically]			
M94 M94a	Pre-asymptotic Non-parallel	94a		
M95	Series	95 /9 5a	95/95a	95/108
M95a	Cycle	0/-/07	+100/102	
м96 м97	Spaced Spacing	96/9 7	96/97	
M98	Stagger	98/99ъ	98/99 b	
м99	Staggered			
M99a Moob	Staggering			
M99b M100	Stepwise Row	100/102		
M101	Bank			
M105	Rake			
M103 M104	Sequential Aligned	103 104/105	103 104/108a	
M1.05	Alignment	104/ 10)	104/ 1008	
M106	Misaligned	106/107		
M1.07	Misalignment	100/100-		
M108 M108a	Non-aligned Crookedness	108/108a		
M109	Closed	109/110	109/110	109/116
Mllo	Enclosed			
Mlll	Bounded	111/112	111/112	
M112 M113	Bounding Open	113/115	113/116	
M114	Exposed	11,5/ 11.5	119/114	
М115	Bare			
M116	Unbounded	116	140	117/120a
M117 M118	Concentric Heliocentric	1 1 7 118/119	117 118/119	11// 1208
M119	Planetcentric	110/11/	110/11/	
M120	Eccentric	120/120a	120/120a	
M120a	Eccentrically			
N1	(Temporal relations) Time + Timewise	1	1/5	1/15
N2	Duration	2/5	","	
N2a	During			
N2b N3	Running History			
N3a	Historical			
N_{4}	Li f e			
N5	Lifetime	,	6/471	6/45-
n6 n7	Period Millesecond	6 7	6/ 13 b	6/15a
N8	Day	8/9		
N9	Diurnal			
NIO	Night Nocturnal	10/11		
N11 N12	Season	12/13		
N13	Seasonal	,		
N13b	Interval	13b	4.4	
N14 N15	Permanent Transient	14	14	
N15a	Momentary	15/15a	15/15a	
N16	Periodic	16/19	16/22	16/24
Nl.7	Periodically	-		
NIS NIO	Periodicity			
N19	Intermittent Isochronous	20/20a		
N20a	Isochrone	20/200		
N21	Cyclic	21		
N22	Successive	22	07/04	
N23 N24	Simultaneous Simultaneously	23/24	23/24	
N25	Rate	25	25	25/33
N26	Frequency	26	26	-5, 25
NOV.	Slow	27/28a	27/33	
N27a N28	Slowly Retarded			
N2Sa	Gradual			
N29	Fast + Quick + Rapid	29/30		
N30	Accelerated			
N31 N32	Sudden Immediately	31/33		
N33	1nstantaneous			
N34	Delay	34/35	34/35	34/35
N35	Lag			
N36	After	36/37	36/38	3 6/38

N105 Descending 103/105 103/107 N104 Descent N105 Glide			First Reduction	Second Reduction	Third
Fost		(After)	Wedda erou	Hed ac \$10H	Vennc #1011
Flying operations Performance		Post	70		
No	-		58		
Many Sange So/51			40+42/46	40/46	40/51
Holding Holding Natarapsed Atrapsed	N41		+50/51		
Mile			41		
Peyload Peyload					
Propulation					
Nig				47/49	
Consumption (of Neal) Expenditure (of power) Flying operations, processed Flight + Flying 52/55 52	N48	Thrust	48	+62/63	
Excenditure (of power)					
Flight + Flying 52/55 52		Expenditure (of power)			
### ### ##############################	N 5 2		52/55	52/55	52/55
No.	N54.	Voyage			
No.	N55	Mission			
NSS			56/58	56/61	56/61
NSO Schip (trajectory) 59					
No. Skipping No.	N59	Semi-ballistic (trajectory)	59		
No.			60/61		
M63 Unpowered M64 Taxiing 64/68 64/68 64/74 M65 Take-off M66 Boosting M67 Traversal M68 Traversing M69 Ascent M69 Ascent M70 Ascending M71 Climb M72 Zoom M75 Escape (from a force field, e.g. gravity) M74 Capture (by a force field, e.g. gravity) M75 Cruising M76 Cruise M77 Manoeuvres + Manoeuvring* M77 Manoeuvres + Manoeuvring* M77 Manoeuvring M78 Hovering M79 Hover M79 Hover M70 Turn M72 Roll M73 Roll M74 Rolling M75 Roll M76 Stawing M77 Rolling M77 Rolling M78 Roll M79 Roll M70 Graze M70 Roll	WOT .	- SALENDAME			
NGL Taxing NGS Take-off NGS Boosting NG7 Traversal NG8 Traversing NG9 Ascent Ascent Scom NT0 Ascending NT1 Climb NT2 Zoom NT3 Escape (from a force field, e.g. gravity) 73/74 Capture (by a force field, e.g. gravity) NT5 Cruising T5/76 Cruise NT6 Cruise NT7 Manoeuvrics + Manoeuvring* T7/71a T7/82+87 NT7 Manoeuvring H80/82 Turn NT8 H0vering NT9 H0ver N80 Turn N82 Circling N83 Roll N84 Rolling N85 Yaw N86 Yaw N86 Yaw N86 Yaw N86 Yaw N86 Yaw N87 N00-yawing N87 N00-yawing N89 Pitcht N89 Pitcht N89 Pitcht N89 Pitcht N89 Pitching N90 Diving N91 Plunging N92 Tumbling N94 Fall N95 Sideslip N99 Sideslip N99 Sideslip N99 Sideslip N99 Sideslip N99 Sideslip N99 Sideslip N90 Descenting N106 Braking N106 Braking N107 Graze N106 Braking N107 Graze N106 N107 Roze N109 Re-entering N111 N114 Approaching N115 N114 Approaching					
M65 Take-off M66 Boosting M67 Traversing M68 Traversing M69 Ascent M69 Ascent M70 Ascending M71 Climb M72 Zoom M75 Escape (from a force field, e.g. gravity) M74 Capture (by a force field, e.g. gravity) M75 Cruising M76 Cruise M77 Manoeuvres + Manoeuvring* M77 Manoeuvres + Manoeuvring* M77 Manoeuvres + Manoeuvring* M78 Hovering M79 Hover M70 Turn M71 Manoeuvring M70 Hover M71 Hover M71 Hover M72 Manoeuvring M73 Roll M74 Rolling M75 Roll M75 Roll M76 Rolling M77 Ron-yawing M77 Ron-yawing M78 Roll M79 Ron-yawing M79 Ron-yawing M79 Ron-yawing M79 Pitch M79 Ron-yawing M79 Pitch M79 Ron-yawing M79 Pitch M79 Ron-yawing M79 Ron-yawing M79 Ron-yawing M79 Pitch M79 Ron-yawing M79 Pitch M79 Ron-yawing M79 Pitch M79 Ron-yawing M70 Diving M70 Diving M70 Diving M70 Rowe M70 Diving M70 Rowe M70 Descending M70 Rowe M70 Descent M70 Graze M70 Rowe-entry M70 Rowe-entr			64/68	64/68	64/74
N67 Traversing 69/71 69/72 N69 Ascent 69/71 69/72 N70 Ascending N71 Climb N72 Zoom N75 Escape (from a force field, e.g. gravity) 73/74 73/74 N74 Capture (by a force field, e.g. gravity) 73/74 73/74 75/76 75/76 75/76 75/79 N75 Cruise 77/77a 77/82+87 77/77a 77/82+87 75/79 75/79 75/79 75/79 75/79 75/79 75/79 75/79 75/79 75/79 75/79 75/79 75/74 75/74 75/79 75/79 75/79 75/79 75/79 75/74 75/74 75/79	N 65		5.4, 55	- 1,	
N68 Traversing 69/71 69/72 N69 Ascent 69/71 69/72 N70 Ascending 172 Comm N71 Climb Zoom 73/74 73/74 N75 Escape (from a force field, e.g. gravity) 73/74 73/74 N74 Capture (by a force field, e.g. gravity) 73/74 73/74 N75 Cruising 75/76 75/76 75/99 N76 Cruise 77/76 75/76 75/99 N77 Manoeuvring 78/79 77/72 77/82+87 N77 Manoeuvring 78/79 77/72 77/82+87 N77 Manoeuvring 78/79 78/79 N79 Hover 78/79 78/79 N79 Hover 88/79 88/80 N80 Turn 83/84 83/84 83/84 N81 Rolling 87 88/89 88/89 N83 Roll 88/89 88/89 88/89 N83 Pitching 88/89 88/89 88/89 N89 Pitching 90/95 90/97 N91 Plunging 90/95 90/97 N92 Tumbling 96/97 <td></td> <td></td> <td></td> <td></td> <td></td>					
N70					
N71	N69	Ascent	69/71	69/72	
N72					
N74 Capture (by a force field, e.g. gravity) N75 Cruising 75/76 75/76 75/99 N76 Cruise N77 Manoeuvres + Manoeuvring* 77/77a 77/82+87 N77 Manoeuvring +80/82 +98/101 N78 Hover 18/79 N79 Hover N80 Turn N82 Circling N83 Roll 83/84 83/84 N84 Rolling N85 Yaw 85/86 85/86 N86 Yawing N87 Non-yawing 87 N88 Pitch 88/89 88/89 N89 Pitching N90 Diving 90/95 90/97 N91 Plunging N92 Tumbling N95 Falling N96 Heaving 96/97 Heave N98 Spin 98 Spin 98 Spin 98 Spin 98 Spin 98 Spin 98 N99 Sideslip 99 N100 Return 102 102+113/117a 102/117a N105 Descending 103/105 103/107 N106 Braking 106/107 N106 Entry 108/112 108/112 N109 Entering N110 Re-entering N111 Re-entering N111 Re-entering N111 Approach 113/114 Approach 111/114	N72			73/74	
N76 Cruise 77 77/82+87 N77 Manoeuvring +80/82 +98/101 N78 Hovering 78/79 N79 Hover 78/79 N80 Turn 83/84 83/84 N81 Roll 85/86 85/86 N84 Rolling 85/86 85/86 N85 Yaw 85/86 85/86 N86 Yawing 87 88 N87 Non-yawing 87 88/89 88/89 N89 Pitching 90/95 90/97 90/97 N89 Pitching 90/95 90/97 </td <td></td> <td></td> <td></td> <td>13/14</td> <td></td>				13/14	
N77 Manoeuvring + Manoeuvring* 17/77a 77/82+87 N77a Manoeuvring +80/82 +98/101 N7B Hover N7B Hover N80 Turn N82 Circling N83 Roll 83/84 83/84 N86 Yawing N87 Non-yawing 87 N88 Pitch 88/89 88/89 N89 Pitching N90 Diving 90/95 90/97 N91 Plunging N92 Tumbling N92 Tumbling N94 Fall N95 Sinking N96 Heave N98 Spin 98 N99 Sidealip 99 N102 Return 102 102+113/117a 102/117a N105 Descending 103/105 103/107 N106 Braking 106/107 N107 Graze N108 Re-entering N110 Re-entering N111 Re-entering N111 Re-entering N111 Approaching			7 5/76	75/76	75/99
N77e Manoeuvring +80/82 +98/101 N73			77/77a	77/82+87	
N79	N77a	Manoeuvring	+80/82		
N80 Turn N82 Circling N83 Roll N84 Rolling N85 Yaw N86 Yawing N87 Non-yawing N89 Pitch N89 Pitch N89 Pitching N90 Diving N91 Plunging N92 Tumbling N94 Fall N95 Sinking N96 Heaving N97 Heave N98 Spin N99 Sideslip N90 Return N103 Descending N104 Descent N105 Glide N106 Braking N106 Braking N107 Graze N108 Entry N109 Re-entry N110 Re-entring N111 Re-entering N111 Re-entering N111 Re-entering N111 Re-entering N111 Approach Inl1/114			78/79		
N83 Roll 83/84 83/84 83/84 N84 Rolling 85/86 85/86 85/86 N85 Yaw 85/86 85/86 85/86 N86 Yawing 87 88/89 88/89 N87 Non-yawing 87 88/89 88/89 N89 Pitching 90/95 90/97 N90 Diving 90/95 90/97 N91 Plunging 90/97 N92 Tumbling 96/97 N95 Sinking 96/97 N96 Heaving 96/97 N97 Heave 98 N99 Sideslip 99 N102 Return 102 102+113/117a 102/117a N103 Descending 103/105 103/107 103/107 N104 Descent 106/107 103/107 108/112 108/112 N106 Braking 106/107 108/112 108/112 108/112 N108 Entry 108/112 108/112 108/112 108/112 <t< td=""><td></td><td></td><td></td><td></td><td></td></t<>					
N84 Rolling N85 Yaw 85/86 85/86 N86 Yawing 87 N87 Non-yawing 87 N88 Pitch 88/89 88/89 N89 Pitch 88/89 88/89 N89 Pitch 88/89 88/89 N89 Pitching 90/95 90/97 N91 Plunging 90/97 N92 Tumbling 96/97 N94 Fall ng 96/97 N97 Heave 98 N98 Spin 98 N99 Sideslip 99 N102 Return 102 102+113/117a 102/117a N103 Descending 103/105 103/107 N104 Descent 100/107 103/107 N105 Glide 106/107 108/112 108/112 N100 Entry 108/112 108/112 108/112 N10 Re-entring 113/114 N114 Approach 113/114 N114 Approaching <			92/91	49/29	
N85 Yaw				0)/04	
N87 Non-yawing 87 N88 Pitch 88/89 88/89 N89 Pitching 90/95 90/97 N90 Diving 90/95 90/97 N91 Plunging 90/95 90/97 N91 Plunging 90/95 90/97 N92 Tumbling 99/97 N93 Falling 96/97 N94 Fall 98 N95 Sinking 96/97 N97 Heave 98 N98 Spin 98 N99 Sideslip 99 N102 Return 102 102+113/117a 102/117a N103 Descending 103/105 103/107 N104 Descent 100/107 N105 Glide N106 Braking 106/107 N107 Graze N108 Entry 108/112 108/112 N109 Entering 113/114 N110 Re-entering 113/114 N111 Re-entering 113/114 N114 Approaching		Yaw	85/86	85/86	
N88			87	1	
N90 Diving 90/95 90/97 N91 Plunging N92 Tumbling N93 Falling N94 Fall N95 Sinking N96 Heaving 96/97 N97 Heave N98 Spin 98 N99 Sideslip 99 N102 Return 102 102+113/117a 102/117a N103 Descending 103/105 103/107 N104 Descent N105 Glide N106 Braking 106/107 N107 Graze N108 Entry 108/112 108/112 N109 Entering N110 Re=entry N111 Re=entering N113 Approaching	N88	Pitch		88/89	
N91 Plunging N92 Tumbling N93 Falling N94 Fall N95 Sinking N96 Heaving 96/97 N97 Heave N98 Spin 98 N99 Sideslip 99 N102 Return 102 102+113/117a 102/117a N103 Descending 103/105 103/107 N105 Glide N1.06 Braking 106/107 N1.07 Graze N1.08 Entry 108/112 108/112 N1.09 Entering N1.10 Re-entry N1.11 Re-entering N1.12 Approaching			90/95	90/97	
N93 Falling N94 Fall N95 Sinking N96 Heaving 96/97 N97 Heave N98 Spin 98 N99 Sideslip 99 N102 Return 102 102+113/117a 102/117a N103 Descending 103/105 103/107 N104 Descent N105 Glide N106 Braking 106/107 N107 Graze N108 Entry 108/112 108/112 N109 Entering N110 Re=entry N111 Re=entering N113 Approach 113/114 N114 Approaching	N91	Plunging	70, 77	3-7-51	
N94 Fall N95 Sinking N96 Heaving 96/97 N97 Heave N98 Spin 98 N99 Sideslip 99 N102 Return 102 102+113/117a 102/117a N103 Descending 103/105 103/107 N104 Descent N105 Glide N106 Braking 106/107 N107 Graze N108 Entry 108/112 108/112 N109 Entering N110 Re=entering N111 Re=entering N111 Approaching					
N96 Heaving 96/97 N97 Heave N98 Spin 98 N99 Sideslip 99 N102 Return 102 102+113/117a 102/117a N103 Descending 103/105 103/107 N105 Glide N106 Braking 106/107 N107 Graze N108 Entry 108/112 108/112 N109 Entering N110 Re-entry N111 Re-entering N113 Approach 113/114 N114 Approaching					
N97 Heave N98 Spin 98 N99 Sideslip 99 N102 Return 102 102+113/117a 102/117a N103 Descending 103/105 103/107 N104 Descent N105 Glide N106 Braking 106/107 N107 Graze N108 Entry 108/112 108/112 N109 Entering N110 Re=entry N111 Re=entering N113 Approach N114 Approaching			06/07		
N98			30/31		
N1.02 Return 102 102+113/117a 102/117a 103/105 N1.03 Descending 103/105 103/107 N1.04 Descent N1.05 Glide N1.06 Braking 106/107 N1.07 Graze N1.08 Entry 108/112 108/112 N1.09 Entering N1.10 Re-entry N1.11 Re-entering N1.13 Approach 113/114 Approaching					
N103 Descending 103/105 103/107 N104 Descent N105 Glide				102+113/117a	102/117a
N1.05 Glide N1.06 Braking 106/107 N1.07 Graze N1.08 Entry 108/112 108/112 N1.09 Entering N1.10 Re-entry N1.11 Re-entering N1.13 Approach N1.14 Approaching	N103	Descending			
N107 Graze N108 Entry 108/112 108/112 N109 Entering N110 Re-entry N111 Re-entering N113 Approach N114 Approaching					
N107 Graze N108 Entry 108/112 108/112 N109 Entering N110 Re-entry N111 Re-entering N113 Approach N114 Approaching	NJ 06	Braking	106/107		
N109 Entering N110 Re=entry N111 Re=entering N113 Approach N114 Approaching	N107	Graze	·	د د د د د د د د د د	
Nl10 Re-entry Nl11 Re-entering Nl13 Approach Nl14 Approaching			108/112	108/112	
Nlll Re-entering Nll3 Approach Nll4 Approaching					
N114 Approaching	ננות	Re-entering	447/444		
			112/114		
	N115	Landing	115/117a		

		First Reduction	Second Reduction	Third Reduction
	[Landing]	115/117a		
N116 N117 N117a	Overshoot Undershoot Miss	2,		
8נבת	(Miscellaneous activities) Navigation	118/119	118/119	118/119
N119 N120 N121	Piloting Guidance (of space vehicles) Telemetry	120/122	120/122	120/122
N122	Telemetering			
N123 N124	(Agents) Human Pilot	123/124a	123/124a	123/ 1 24a
N124a	Operator			
N125	(Properties of flight, agents of f	light) 125/126	125/129	125/129
N126 N127 N128	Visibility Tolerance (to acceleration) Hazard	127/129		
N129 =	Safety Aerodynamics	0a/0b	Oa/Ob	0a/02
0b 01 01a	Aerodynamically Flow + Stream + Flux + Airflow* + Airstream* + Flowing	1/2+28a/28b	1/2+28a/28b	+28/28b +32/34a
01b 02	Airstream Flowing (By speed)			
03 04	Subsonic Subcritical	3/4	3/4	3/4
05 06	Sonic Transonic	5	5 6	5/13
07	Supersonic	6 7/9	7/9	
08 010	Supersonically Hypersonic			
011	Hypervelocity	10/13	10/13	
012 013	Supersatellite Superfast			
014	(By viscosity effects) Frictionless	14/15	44/45	14/15
015	Inviscid + Non-viscous	•	14/15	+24/25
016 017	Viscous + Viscid Creeping (motion regime)	16/17	16/17	16/21
018 019	Laminar Non≖turbulent	18/19	18/19	
020 021	Transitional Turbulent	20 21	20 21	
	(By degree of vorticity)	21	21	
022	Rotational	22	22	22
024 025	Irrotational + Non-rotating Potential	24/25	24/25	
028 028	Isobaric Slug	28/28ъ	28/28b	
028ъ	Mound (Regimes - by density conditions, et		00/14	00/74
029 030	Slip Continuum	29 3 0	29/31	29/31
031 032	Collision Expanded	31 32/34a	32/34a	
033	Overexpanded)27)48.)2/)4 <u>B</u>	
033 a 034	Over-expansion Under expanded			
034a	Underexpansion			
035 036	Separated Unseparated	35 + 37 36 + 38	55 /38	35/38
036a	Reattached	-		
037 038	Stalled Unstalled			
039 040	Incompressible Compressible	39 40	39 40	39/40
2 10	[Flow elements and phenomena]		40	
	[Boundary layer - Processes and ope	erations		

		First Reduction	Second Reduction	Third Reduction
- 1	[Boundary layer = Processes an		A6. A7/AE-	11/16
041 042	Laminarization Transition	41+43/45a 42	41+43/45a 42	41/46a
043	Breakdown	700	44	
Olil	Blowoff			
045	Tripped Initlation			
045a 046	Separation	46	46	
046 046a	Reattaching	46a/46b	46a/46b	
046ъ	Reattachment			
	(B.L. Control)	48/50.57	15/57	
047	Thickening	47/50+53	47/53	47/53
048 049	Avoidance Inhibition			
050	Retardation			
051	Promotion	51/52a		
052	Trip			
052a	Tripping			
053	Re-establishment			
225	(-6.4	54/67	E 4 / 67	CA 167 .
054	Promoters (of transition) Sandpaper	54/63	54/63	54/63
055 056	Carborundum			
057	Bead			
058	Ballotini			
059	Deposit			
059a	Deposited (dirt, etc.)			
060 061	Insect Fruitfly			
062	Drosophila			
063	Housefly			
24	[Flow elements]	64/67	64/68	64/68
064	Streamline	04/01	04/00	04/ 00
065 066	Cocurrent Filament			
067	Dividing			
068	Streamtube	68	(0.80/87-	60/77-
069	Supply (stream)	69+72/73a	69+72/73a	69/73a
070	Sink			
071 072	Doublet Inflow			
073a	Pocket			
074	Vortex + Eddy	7.4	74/79	74/79
075	Vortical	75/79		
076	Swirl			
077 078	Spiralling (fluid) Co-rotating (yortex)			
079	Image [vortex]			
080	Street	80	80/83a	80/83a
081	Shedding	81/83a		
032	Bursting Cancellation			
083 083a	Feeding			
084	Jet	84/84a	84/84a	84/89
084a	Airjet	05/00		
085	Jetlike	85/88		
086	Billowing			
087 088	Plume Curtain			
089	Cushion	89		(00
090	Wave	90	90	90/99
091	N(-wave)	91/92+99	91/92+99	
092	Standing	93/95	93/95+97/98a	
093 094	Blast Explosion	72172	75/75-51/7	
095	Detonation			
096	Shock	96	96	
097	Non-relativistic	97/98a		
. 098	Standoff			
098a	Shockless Riemann			
099 :				
0100 0101	(Flow attributes) Vorticity Circulation	100	100 101	100/101
0101	Circulation	101	101	
12.2	[Similarity parameters]	400	102	102
0102	Mach (number)	102 103	102 103	103
0103 0104	Prandtl (number) Nusselt (number)	104	104	104
0104	Reynolds (number)	105	105	105
		-		
	(Fluid attributes)	4	1	1/2
Pl	Compressibility	1 2	2	-√ ÷
P2	Viscosity	_	7	

		First Reduction	Second Reduction	Third Reduction
	[Viscosity]			
	(Aerodynamic characteristics)			
P3	(Loads) Lift + Lifting*	3	3/5	3 /5
P4 P5	Lifting Indicial	4		
P6	Drag	5	6	6/9
P7	Foredrag	7/9	7/9	
P8 P9	Coulomb Drift			
P9a	Interference	9a .	9 a	9a/17
P10 P11	Stagnation Choking	10 11/12a	10 11/13	+27/29
P11a	Choke	,		
Р11ъ Р12	Choked Blockage			
P12a	Blocking			
P13 P14	Lap Stall	13 14/17	14/17	
P15	Stalling	(4/)	1-7/11	
P16	Destalling			
P17 P18	Unstalling Downwash	18	18/21	18/26
P18a	Upload	18 a/1 9		
P19 P19a	Upwash + Upflow Sideforce	19a/21		
P20	Sidewash	•		
P21 P22	Crossflow Wake + Afterflow	22/23a	22/26	
P23	Wakelike		•	
P23a P24	Isovel Slipstream	24/25		
P24a	Sliplines	24/2)		
P25 P26	Outflow Trail	26		
P27	Turbulence	27/28	27/29	
P27a	Proturbulence			
P2 8 P29	Homologous Surge	29		
	(Aerodynamic properties and pro	(Pappan		
Z12	Gasdynamic properties and pro	12/14	12/14	12/18
Z13 Z14	Aerodynamic Superaerodynamic			
215	Suction	15/15a	15/16+18	
215a 216	Sucking Blowing	16		
Z17	Injection	17	17	
Z18 Z19	Entrainment Hydrodynamic	18 19/23	19/23	19/23
Z20	Spray	13/2)	13/ 2)	17/ 47
Z21	Wetting			
Z22 Z23	Soaking Hydroballistic			
		70/70 70/70	70/70	70/40
P30 P30a	Aeroelasticity Aeroelastic	30/32+35/36	30/32 +35/40	30/40
P31	Thermoaeroelastic		- -	
P32	Aerothermoelastic	4		
P33 P34	Flutter Fluttering	33/34a	33/34a	
P34a	Precipitation			
P35 P36	Ripple Sloshing			
P37	Buzz	57		
P38	Divergence Buffeting	38 39/ 40		
P39 P40	Buffet)3/ 4\/·		
;	(Aerodynamic reference parameter)			
P42	Tiļt	42	42	42/48
P43 P45	Attitude Trim	43 45/46e.	43/48	
P46	Trimming			
P46a P47	Trimmed Banked	47/48		
P48	Unbanked			
P49 P50	Fianform	49 50 / 52	49/52+54	49/69
P51	Sweep Sweepback	JV/ 32		
P52	Sweepforward	C 7	53	
P53 P54	Aspect (ratio) Fineness (ratio)	53 54	53	
-	•	•		

		First Reduction	Second Reduction	Third Reduction
P55	[Fineness (ratio)] Stepdown (height) Twist	55/58		
P57 P58	Warp			
P59	Dihedral	59		
P60	Anhedral	60 6 1/ 64	61/64	
P61 P63	Span Midspan	01/04	01/04	
P64	Semispan			
P65	Chord	65/ 66	65/66	
P66	Midchord			
P67	(Airfoil section) Rooftop			
P68	Peaky			
P69	Camber			
P70	Mechanics (phenomena)	70	70	
	(Statics)	m4./m0	74 /70	71/73
P 71 P 7 2	Moment	71/72	71/72	11/12
P73	Torque Couple	73	73	
P74	Equilibrium	74/77	74/78	74/83b
P 7 5	Quasi-equilibrium			
P75a	Non-vanishing			
P76	Non-equilibrium			
Р 7 7 Р 7 8	Disequilibrium Balance	78		
P79	Stability	79	79	
P80	Transtability	80/83	80/83	
P81	Overstability			
P82	Instability			
P83 P83a	Disturbance + Perturbation Resistance	83a/83b	83a/83b	
183b	Resisting	034 030		
	(Kinematics)			
P84	Motion + Movement* + Moving*	84/88	84/88	84/88
	+ Non-stationary* + Travelling*			
P85 P86	Movement Moving			
287	Non-stationary			
P88	Travelling			
P89	Velocity + Speed	89/90	89/90	89/90
P90	Alfven (speed)	92/93	92/95	92/95
P92	Acceleration Accelerating	74/97	74/3)	34/32
P93 P94	Deceleration	94/95		
P95	Decelerating			
P96	Translation	96/97	96/97	96/Q12
P97	Translating	0/41	0/12	
Q2	Rotating	2/4b	2/4 b +6/10	
Q4 Q4a	Pre-rotation Circulating		10/10	
Q4b	Recirculating	5 6/10	5	
Q6	Gyration	6/10		
Q7	Spinning			
Q8 ~	Whirl Prewhirl			
Q9 Q10	Overswing			
Q11	Gyroscopic	1 1/ 12	11/12	
Q12	Precession			
	(Kinds of motion by direction)		***	Am. 1.5
Q12b	Advance	12b/12c	12b/12c	12b/12c
~ 4 ^	Advancing			
Q12c				
	(Inwardly)	14	14	14/18
Q14		14 15/16	14 15/16	14/18
Q14 Q15	(Inwardly) Intersecting Convergence Converging	15/16	15/16	14/18
Q14 Q15 Q16 Q17	(Inwardly) Intersecting Convergence Converging Penetration		14 15/16 17/18	14/18
Q14 Q15 Q16 Q17	(Inwardly) Intersecting Convergence Converging Penetration Penetrating	15/16	15/16	14/18
Q14 Q15 Q16 Q17 Q18	(Inwardly) Intersecting Convergence Converging Penetration Fenetrating (Outwardly)	15/16 17/18	15/16 17/18	
Q14 Q15 Q16 Q17 Q18	(Inwardly) Intersecting Convergence Converging Penetration Fenetrating (Outwardly) Diverging	15/16 17/18 19	15/16	14/18 19/20
Q14 Q15 Q16 Q17 Q18	(Inwardly) Intersecting Convergence Converging Penetration Penetrating (Outwardly) Diverging Vanishing	15/16 17/18 19 20	15/16 17/18 19/20	19/20
Q14 Q15 Q16 Q17 Q18 Q19 Q20	(Inwardly) Intersecting Convergence Converging Penetration Fenetrating (Outwardly) Diverging	15/16 17/18 19	15/16 17/18	
Q14 Q15 Q16 Q17 Q18 Q19 Q20	(Inwardly) Intersecting Convergence Converging Penetration Penetrating (Outwardly) Diverging Vanishing (Across, past) Transit Crossing	15/16 17/18 19 20 22/23	15/16 17/18 19/20	19/20
Q14 Q15 Q16 Q17 Q18 Q19 Q20 Q22 Q23 Q24	(Inwardly) Intersecting Convergence Converging Penetration Penetrating (Outwardly) Diverging Vanishing (Across, past) Transit Crossing Passing	15/16 17/18 19 20	15/16 17/18 19/20	19/20
Q14 Q15 Q16 Q17 Q18 Q19 Q20 Q22 Q23 Q24 Q24a	(Inwardly) Intersecting Convergence Converging Penetration Penetrating (Outwardly) Diverging Vanishing (Across, past) Transit Crossing Passing Pass	15/16 17/18 19 20 22/23 24/24a	15/16 17/18 19/20 22/24a	19/20 22/24a
Q14 Q15 Q16 Q17 Q18 Q19 Q20 Q22 Q23 Q24 Q24a Q24a	(Inwardly) Intersecting Convergence Converging Penetration Penetrating (Outwardly) Diverging Vanishing (Across, past) Transit Crossing Passing Pass Dynamic + Dynamically*	15/16 17/18 19 20 22/23	15/16 17/18 19/20	19/20
Q14 Q15 Q16 Q17 Q18 Q19 Q20 Q20 Q23 Q24 Q24a Q25 Q25a	(Inwardly) Intersecting Convergence Converging Penetration Penetrating (Outwardly) Diverging Vanishing (Across, past) Transit Crossing Passing Pass Dynamic + Dynamically* Dynamically	15/16 17/18 19 20 22/23 24/24a	15/16 17/18 19/20 22/24a	19/20 22/24a
Q14 Q15 Q16 Q17 Q18 Q19 Q20 Q22 Q23 Q24 Q24 Q25 Q25 Q26	(Inwardly) Intersecting Convergence Converging Penetration Penetrating (Outwardly) Diverging Vanishing (Across, past) Transit Crossing Passing Pass Dynamic + Dynamically*	15/16 17/18 19 20 22/23 24/24a	15/16 17/18 19/20 22/24a	19/20 22/24a
Q24a Q25 Q25a Q26 Q27 Q28	(Inwardly) Intersecting Convergence Converging Penetration Fenetrating (Outwardly) Diverging Vanishing (Across, past) Transit Crossing Passing Pass Dynamic + Dynamically* Dynamically Inertia Momentum Energy	15/16 17/18 19 20 22/23 24/24a 25/26	15/16 17/18 19/20 22/24a 25/26	19/20 22/24æ
Q14 Q15 Q16 Q17 Q18 Q19 Q20 Q22 Q23 Q24 Q24 Q25 Q25 Q26 Q27 Q28 Q28a	(Inwardly) Intersecting Convergence Converging Penetration Penetrating (Outwardly) Diverging Vanishing (Across, past) Transit Crossing Passing Pass Dynamic + Dynamically* Dynamically Inertia Momentum Energy Activation (Energy)	15/16 17/18 19 20 22/23 24/24a 25/26	15/16 17/18 19/20 22/24a 25/26	19/20 22/24æ
Q14 Q15 Q16 Q17 Q18 Q19 Q20 Q22 Q23 Q24 Q24a Q25a Q26 Q27 Q28 Q28a Q28b	(Inwardly) Intersecting Convergence Converging Penetration Penetrating (Outwardly) Diverging Vanishing (Across, past) Transit Crossing Passing Pass Dynamic + Dynamically* Dynamically Inertia Momentum Energy Activation (Energy) Isoerg	15/16 17/18 19 20 22/23 24/24a 25/26	15/16 17/18 19/20 22/24a 25/26	19/20 22/24æ
Q14 Q15 Q16 Q17 Q18 Q19 Q20 Q22 Q23 Q24 Q25 Q25 Q27 Q28 Q28 Q28a Q28a Q28a Q28c	(Inwardly) Intersecting Convergence Converging Penetration Penetration Penetrating (Outwardly) Diverging Vanishing (Across, past) Transit Crossing Passing Pass Dynamic + Dynamically* Dynamically Inertia Momentum Energy Activation (Energy) Isoerg Kinetic	15/16 17/18 19 20 22/23 24/24a 25/26 27 28 28a/28o	15/16 17/18 19/20 22/24a 25/26 27 28/280	19/20 22/24æ
Q14 Q15 Q16 Q17 Q18 Q19 Q20 Q22 Q23 Q24 Q25 Q25 Q25 Q26 Q27 Q28 Q28 Q28b Q28b Q28b	(Inwardly) Intersecting Convergence Converging Penetration Fenetrating (Outwardly) Diverging Vanishing (Across, past) Transit Crossing Passing Pass Dynamic + Dynamically* Dynamically Inertia Momentum Energy Activation (Energy) Isoerg Kinetic Work	15/16 17/18 19 20 22/23 24/24a 25/26 27 28 28a/28o	15/16 17/18 19/20 22/24a 25/26	19/20 22/24æ
Q14 Q15 Q16 Q17 Q18 Q19 Q20 Q22 Q23 Q24 Q24 Q25 Q25 Q26 Q27 Q26 Q27 Q28 Q28 Q28a Q28a Q28a Q28c	(Inwardly) Intersecting Convergence Converging Penetration Penetration Penetrating (Outwardly) Diverging Vanishing (Across, past) Transit Crossing Passing Pass Dynamic + Dynamically* Dynamically Inertia Momentum Energy Activation (Energy) Isoerg Kinetic	15/16 17/18 19 20 22/23 24/24a 25/26 27 28 28a/28o	15/16 17/18 19/20 22/24a 25/26 27 28/280	19/20 22/24æ

		First Reduction	Second Reduction	Third Reduction
	[Force]	32/35	32/35a	
Q33	Lorenz (force)			
Q34 Q35	Pondero (force) Restoring (force)			
Q35a	Gravity	35a		
Q36	Pressure	36/37	36/37	
938 938a	Compression Compressing	38/39	38/39	
Q38b	Compressively			
Q39	Recompression			
ର୍40 ର୍40a	Overpressure Pressuriza t ion	40/40a	40/40a	
Q41 .	Striking	41+48	41/47a	41/49a
Q41a	Jog	41a/44		
Q42 Q42a	Impulse Impulsive			
Q42b	Impulsively			
Q43	Impact			
Q44 Q45	Embedded Impinging	AE / AT		
Q46	Impingement	45/47a		
Q47	Re-impingement			
Q47a	Glancing	10/10	10/10	
Q49 Q49a	Driving Drive	49/49a	49/49a	
Q50	Emission	50	50/58ъ	50/60
Q51	Issuing	51/53	-	•
ୟ52 ୟ53	Effusion Efflux			
Q54	Discharge	54/57b		
€54a	Blowout	J J ! -		
Q55	Exiting			
Q56 Q57	Exhaust Exhausting			
Q57a	Scavenging			
Q57b	Venting	50/50		
Q58 Q58a	Tapping Bleeding	58/58b		
Q58b	Bleed			
Q60	Burst	60	60	2.465
Q61 Q62	Breathing Swallowing	61/62	61/62	61/62
402	The state of the s			
	(Behaviour of defermable bodies)	1/0 = 06	4 04	1.107
Rl Rla	Load + Loading + Stress Airload	1/2+5+26a	1 + 26&	1/26a
R2	Stressing			
R3	Allowable	3/4		
R3a R4	Permissible Failsafe			
R5	Shocking			
R6	Exerted	6/7		
R7 R9	Pre=loading Tension	9/10	9/10	
RlO	Tensile	3/ 10	3/ 10	
R1.1	Shear	11/12	11/12	
R12	Shearing	47/44	47/45-	
R13 R14	Torsion + Torsional* Torsional	13/14	13/15a	
R15	Twisting	15/15a		
R15a	Warpage	16/10	46 107	
R16 R17	Bending Flexural	16/18	16/23	
R18	Flexure			
R19	Inflection	19/23		
R20 R21	Bowing Curling			
R22	Anti-elastic (effect)			
R23	Hogging			
R24 R26	Folding Unfolding	24/26	24/26	
R26a	Unfolding Unloading			
R27	Rheological (behaviour)	27/29	27/29+60/63	27/66
R28	Deformation + Strain			
R29 R30	Deforming Deflection + Displacement	30/30a	30/32	
R30a	Droop)U/)UG	JU/ JE	
R31	Deflecting	31/32		
R31a R32	Deflect Traction + [edge traction]			
R33	Stretching	33/34	33/34	
R33a	Extending		-	
R33b R33c	Extension Extensional			
R33e	TW OCHO TOHOT			

				m, r ,
	R	First eduction	Second Reduction	Third Reduction
	[Extensional]			
R34	Attenuating			
R35	Shortening	35/36	35/36	
R36 R37	Truncation Blunting	37/38	37/38	
R38	Flattening			
R40 R40a	Cavitation Drilling	40/41a	40/41a	
R40a R41	Cracking			
R41a	Intercrystalline			
R42	Distortion	42/43	42/43	
R43 R44	Distorting Dislocation	44	44/50	
R46	Damage	46/48	1175	
R47	Fracture			
r48 r49	Rupture Collapse	49/50		
R50	Collapsing	4)/)0		
R51	Buckling + Buckle*	51/52	51/52	
R52	Buckle Forcible	53/59	53/59	
R53 R54	Explosive	7717 7	331 33	
R55	Monocell			
R56	Snap			
R57 R58	Threshold Pre-buckling			
R59	Post-buckling			
R60	Snapping			
R61. R62	Wrinkling Dimpling			
R63	Kinking			
R64	Creep	64	64	
R64a. R64b	Endurance	64a/65	64a/65	
R65	Straintime Fatigue			
R66	Failure	66	66	
:	(Mechanical properties) [Flexibility]			
R67	[Flexibility] Elasticity + Elastic* +Elastically	* 67/69+73/74	67/74	67/77
R68	Elastic + Flexibility*+Flexible	*+Flexibly*		
R69	Elastically	20/21		
R70 R71	Viscoelasticity Viscoelastic	70/71		
** !	12000333000			
R73	Flexible			
R74 R76	Flexibly Rigidity	76/77	76/77	
R77	Stiffness			
R78	Plasticity	78/80	78/80	78/83&
R79 R80	Plastic Viscoplastic			
R81	Strength	81+83a	81+83a	
R82	Soft	82/83	82/83	
R83 R83a	Hard Endgrain	83a.		
moja.	DINES COLLE			
-0-	(Surface condition)	or for	05 108	as lon
r85 r87	Smoothness Polished	85/87	85/87	85/87
r88	Porous	88/89	88/90a	88/90a
R89′	Permeable	90/90a		
R90 R90a	Impermeable			
v20g	Non-porous			
	(Processes)	* 0		
R91	Vibration + Oscillation + Vibrating Vibrating + Oscillating	91/92a	91/92a	91/97
R92 R92a	Flapping	, , , , , , , ,	7.7,7==	3.731
R93	Harmonic	93/95	93/97	
R93a	Harmonically Resonant			
R94 R9 5	Libration			
R95a	Pulse	95a		
R95b	Non-oscillatory	95b		
R96 R97	Modal Unimodal	96/97		
R98	Sound + Acoustic + Acoustically*	98/100	98/100	98/105a
R99	Acoustically			
R100 R101	Panting Noise	101/105a	101/105a	
RlOle		.0., .0,	,,	
R102	Boom			

		First Reduction	Second Reduction	Third Reduction
	[Boom]			
R103	Siren			
R104	Bomb			
R105 R105a	Exploded Quadrupole			
20070	(Properties and processes of	vibration)		
_	(Frequencies)	40//400	406/444	106/111
R106	Beat	106/108	106/111	106/111
R107 R108	L(~band) RF			
R109	Aural	109/111		
R110	Audio			
Rlll	Subaudio	444-/444-	111a/111f	111a/111f
Rilla	Spectra	111a/111c	iiidy iiid	11164/1114
Rlllb Rlllc	Spectrum Spectral			
Rllld	Wavelength	111d/111£		
Rllle	Waveform			
Rillf	Bandwidth	112	112	112/120
R112 R112a	Alternating Refraction	112a/112b	112a/112b	112/120
R112b	Diffraction			
R113	Impedance	113		
	(Amplification)	444	114	
R114 R115	Resonance Attenuation	114 115	114 115	
R115	Demping	116/119	116/119	
RL17	Isolation	,	, •	
R1.18	Hysteretic			
R119	Phugoid	400	120	
R120 R122	Hysteresis Luminosity	120 122/124	122/125	122/125a
R1.23	Luminous	122/ 124	1664 165	, ,,
R124	Transparent			
R125	Sensitivity	125		
R125a	Monotonic	125a	125a	
Sl	Electromagnetic (wave) +Ele	actromagnetically*		
S2 /	Electromagnetically	1/2	1/5	1/5
83	Ray	3/5		
84	Ultraviolet			
S5	Microwave	6	6	6/220
s 6	Radioactive	•	Ū	3, 44
	(Electric and magnetic phe	enomena)		
s 8	Electrical	8/10+16a+21/2		20 -
S9	Electric		16/17+21/2	220
S10 S11	Electrically Magnetic	11		
812	Electronic			
S13	Magnetofluid mechanics	Magnetohydrodynam	ics	
1	+ Hydromagnetics	13/15	13/15	
S14	Electrohydrodynamics Magnetogasdynamics + 1	degretoserodynamic		
S15	+ Electrogasdynamics	100110 00000 0 -V		
s 16	Photoelectric	16		
S16a	Photoelastic	16a.		
S17	Electrostatic	17		
0		40.10-	40.140	
S18	Current	18/19	18/19	
S19 S20	D-C Charge	20	20	
521	Inductance			
S22 S22a	Dielectric			
322a S22b	Capacitance Magnetostriction			
S220	Pole			
\$23	Thermodynamic behaviour, prince Heat	roperties	23/29	23/29
S24	Thermodynamic	24/29		
824 a	Aerothermodynamic			
S25	Thermodynamically			
826 827	Thermally Caloric			
s2 8	Calorically			
\$29	Aerothermochemistry			
	A COL			
023	(Change of state)	31/31a	31/38	31/38
531 531a	Melting Liquefaction	J1/J18	7.1 30	2.4.7-
832	Resolidification	32/33+38		
S33	Sublimation			
834	Vaporisation	34/37		
S35	Evaporation			

		First Reduction	Second Reduction	Third Reduction
S36 S37	[Evaporation] Evaporating Boiloff			
\$3 8	Condensation			
S39 S40	(Heat transfer processes) Replenishment Heating	39+56a 40/41	39+51/56a 40/41	39/56a
541 541c	Joule Insulation	41c/41d	410/41d	
S41d	Insulating	40.40/40	42/49	
542	Cooling (By mass transfer)	42+48/49	46/ 47	
543 544 545 546	Soret (effect) Ablation Ablating Transpiration	43/45 46/47		
\$47 \$48 \$50 \$51	Sweat Regression (of surface) Convection Radiant (heat transfer)	50 51/54	50	
852 853 854 855	Non-radiating Losing Black (body radiation) Accommodation	55		
856 856a	Exothermic (reaction) Cycling	55 56		
-	(Thermal properties and proce	esses)	-7:1-7	~() /(**
8566 8560 856d	Thermomechanical Thermomelastic Photothermomelastic	56 b/ 56s	56b/56a	56b/65
856e 857	Photothermoelasticity Soaked	57	57+62/65	
858	Enthalpy	58	60/61	
s60 s61	Entropy Residue (of heat energy)	60/61	60/61	
s62	Adiabatic + Non-conductive			
863	Isothermal	62 63/64		
864 865	Non-isothermal Isentropic	65		
s66	Temperature	66/67+74	66/ 7 6a	66/76a
867 868 869	Room Cold Cooled	68/70		
570 571	Cool. Uncooled	71/72		
872	Unheated	73		
S73 S74	Heated Equicohesive (Measures)	73		
875 876 876a	Kelvin Fahrenheit DTU	75/76a		
877	Oxidation	77/80		77/87
878 879 880	Combustion + Burning Afterburning Deflagration			
s 81	Flome	81/82		
s82 s83 s84	Fire Ignition Autoignition	83/85	83/87	
\$85 \$86	Firing Extinction Quenching	86/87	•	
s 87	programmer of the control of the con			
590 590a 590b	(General processes, physical-chemical Action Acting Activity	il processes) 90/92	90/92	90/944
891 891a 891b 891c	Event Occurrence Encounter Case			
592 593 594 594a	Behaviour Interaction Interacting Undergoing	93/94	93/944	
~/~~				

		First Reduction	Second Reduction	Third Reduction
8946 894c 894d	[Undergoing] Subject (to) Counteraction Competition	94a/94b 94o/94d		
895 896 897	Physical Chemical + Chemically* Chemically	95 96/9 7 + T13 98/99	95 96/97+7/7 a +13 98/99+12	96/99 +1/13
598 599 T1 T2	Reaction + Reacting* Reacting Decomposition Dissociation	96/99 1 2/3	1/6b	
T3 T4 T5 T6	Dissociating Ionization Charging Freezing	4/6a		
T6a T6b T7 T7a	Freeze Lagging Catalytic Non-catalytic	6b 7/7a 8/10	8/11	
T8 T9 T10 T11	Absorption Adsorption Chemisorption Saturation	11 12	-,	
T12 T13	Erosion Thermochemical (General processes, activities, operate	13	14+16/22 a	14/29
T14 T15 T15a T15b	Influence +Effect*+Contribution* Effect +Consideration* Contribution Consideration	15/15b 16/18+29	15/15b+29	14/ 23
T16 T17 T18 T19	Change Changing Alteration Modification	19/20		
T20 T21 T22	Modifying Improvement Refinement Tailoring	21/22a		
T22a T23 T24 T25	Variation Varying Fluctuation	23/2 4 25/26	23/28	
T26 T27 T28 T29	Peak to peak Fluctuating Modulation Modulating Conversion	27/28		
T31 T33 T34	(Altering composition) Recombination Mixing Isoenergetic	31 33/34	31+35/38	31/39a
Т35 Т36 Т37	Confluence Merging Merger	35/38		
T39 T39a	Contamination Mon-concurrence	39 39a		
T40 T42 T43 T44	(Altering state) Division Partition Bifurcation Breaking	40/44	40/44	40/44
T46 T47 T47a T48	(Altering state) Hardening Plating Electroplating Chroming	46 47/48	46 47/48	46/48
T50 T52	(Altering location) Transport + Removal* Removal	50/52	50/52 53/54a	50/52 +70/80 53/66a
T53 T54 T54a T55 T57	Distribution Maldistribution Redistribution Spreading Scattering	53/54 a 55/58+61	55/61	المراجع المراجع

		First	Second	Third
		Reduction	Reduction	Reduction
mc0	[Scattering]			
T58 T59	Dispersion Dissipation	59/59a		
T59a T60	Non≖dissipation Diffusion	60		
T61	Rarefaction			
т62 т63	Separating Detachment	62/63	62/63	
T 66	Disturbing	66/66a	66/66a	
T66a T68	Cavitating Concentration	68	68	68
T 70	Transfer	70	70	
T71 T72	Exchange Shuffle	71/72+80	71/72+80	
T 73	Transmission	73	73/79	
T74 T75	Conducting Conduction	74/75		
T 76	Convecting	76		
1778 1779	Radiation Radiating	78/79		
T80	Release			
	(Changing direction)			
T81	Redirection + Redirecting*	81/84	81/88	81/88
T81a T82	Redirecting Switching			
T 83	Turning			
T83a T84	Rementrant Deviation			
T 85	Reflection	85/86		
Т8 5 ъ Т85с	Specular Echoes			
T 86	Reflecting	0.5		
т87 т88	Reversal Inversion	87 88		
	(Date de Lee			
T88a	(Retaining) Continuation	882+89/92	88a/94a	88a/105a
T 89	Holding			
T90 T91	Arrest Retarding			
T 93	Constraint	93/94a		
Т94 Т 94а	Restraint Restraining			
T97	Fixing	97	97/100	
T98 T99	Joining Clamping	98/100		
T100	Coupling	104/407	404 /405	
TlOl TlOla	Stabilizing Destabilizing	101/103	101/105a	
T102	Stabilization			
T103 T104	Equilibration Relaxation	104/105&		
T10 5	Relaxing			
T105a	Shakedown			
	(Influencing dimension)	4/5-	1/5&+8/12&	1/12
Ul Ula	Increasing Rise	1/5a	1/ 78+0/ 128	1/12
U 2	Increase			
U2a U3	Jump Increment			
Մվ+	Augmentation			
U5 U5a	Addition Gain			
U6	Expansion	6/7	6/7	
ช7 ช8	Expanding Accumulation	8/10	-	
U9	Collecting	-,		
U10 U11	Scooping Amplification	11/12a		
U1 2	Magnification	,		
U12a U13	Filling Decreasing	13/15&+20	13/23a	13/23a
ՄՂ [4	Decrease	·)/ ·) @ 1 E U	· // -/-	.),)***
บ15 U15a	Decrement Drop			
015a	Reduction	16		
U16a	Deficit Loss	16a/16b		
V1.6b V1.8	Contraction	18/19		
U19	Contracting			

		First	Second	Third
		Reduction	Reduction	Reduction
U20	[Contracting] Constriction			
U21	Minimisation	21/22		
U22 U23	Minimising Elimination	23/23a		
U23a	Omission		04/05	04/05
U24 U25	Limitation Limiting	24/25	24/25	24/25
	(Development)			
U 26	Beginning	26/29,	26/29	26/30ъ
U27 U28	Starting + S tart* Start			
U28a U29	Onset Re≕start			
U30	Stopping + Cessation	30/30b	30/30ъ	
U30a U30b	Completion Rounding (off)			
U31	Source	31/31a	31/31a	31/46
U31a U32	Providing Derivation	32/34+38	32/38	
U33	Originating	7-, 74.70	<i>y=</i> , <i>y</i> -,	
U34 U35	Initiating Induction	35		
U 36	Generation	36/37		
U36a U37	Generating Formation	•		
U38 U40	Appearance Propagation	40/41	40/45	
U^{1+1}	Propagating		40/45	
U42 U43	Excitation Parasitic	42/45		
Ոֈֈֈ	Exciting			
บ45 บ46	De⇒excitation Growth	46/48	46/48	46/51d
U48	Development			+55
U49 U49a	Conservation Sustaining	49/51	49/51	
บ50	Maintenance	E4-/E43	ra-/ras	
U51 U51a	Lubrication Protection	51a/51d	51a/51d	
U51.b	Waterproofing			
U 51c U51d	Sealing Shielding			
ช52 ช53	Decay Degeneration	51/53d	51/53d	52/5 3 d
U53a	Decaying			
U53b U53c	Weakening Yielding			
U53a	Failing			
U 55	Recovery	55	55	
U 59	(Technical operations and perocess	processes (general)) 59/61	59/61+67/69	59/63a
U60	Operation + Operating*	77/01	29/01+01/09	+67/69
U62 U62	Operating Method + Technique	62/63a	62/63a	
U62a	Treatment	, -, -, -	,,	
U63 U63a	Routine P r ocedure			
U64	Design	64/65	64/65	64/66a
U65 U66	Matching Construction	66/66a	66/66a	
U66a U67	Output Storage	67/68		
u68	Storing			
U69 U70	Supplying Control	69 70/72	70/72+74/75	70/77
U71	Controlling	10/12	107 (24747)	10/11
U71a U72	Regulation Optimisation			
U 73	Correction	73	73	
ช74 ช75	Prevention Suppression	74/75		
U 76	Response	76/77	76/77	
U 77	Feedback			
U 78	Economics	78/80	78/82	78/83
U79 U79a	Cost Penalty			
0 1 7 Q	i circui oj			

		First Reduction	Second Reduction 1	Third Reduction
•	[Penalty]			
U80 U81	Saving Application			
U 82	Use Code			
U83 U85	Research + Investigation + Study	85	85	85
U86	(Operations) Classification	86/87+92/922	86/87+92/920	86/920
U87 U88	Comparison Analysis	+96 88/91	+96 88/91	
บ89	Analytic [operations]	20, 7.	, ,	
U90 U91	Analyti cal Sermentation			
บ92	Deduction			
U92a U92b	Synthesis Idealization	92b/92c		
U92c	Simplification	93/95	93/95	93/97
. U93	Estimation Estimate	70/7 0	72/72	<i>33/3</i> !
U95	Assessment			
บ96 บ97	Evaluation Prediction	97 98/101	987101	98/106
U98	Determination Tracing	98/101	96/101	30/ 100
U99 U100	Traverse			
U102	Resolution Measurement	102/104	102/104	
U103	Quantitative			
U104 U105	Qualitative Verification	105/106	105/106	
0106	Proof (test)			
Vl	Experiment + Experimental*	1/2	1/2	1/9
V 2	Experimental	z /n	3/9	
V3 V4	Observation Detection	3/7	וו	
V 5	Recording			
v 6 v 7	Counting Reading			
V 7a	Simulating	7≈/9		
V3 V9	Simulation Representation			
	Visual (methods) + Visualization*	9a/13	9a/25b	9a/25b
V 9a V10	Visualization	+25a/25b	2-4 - 2 -	, .
V1.1	Spectrography			
V12 V13	Stroboscopic Shadowgraph			
V14	Photography Photorecording	14/16		
V16 V17	Schlieren	17/18		
A18	Spark Interferometry	19/25		
A50	Interferometric			
V21	Interferogram Isopycral (chart)			
V23	(fringe) Shift			
V24 V25	Strioscopy • Interferential			
V25a	Clay			
V 25b	China (Model tests)			
V 25e	Focal (point method)	25c/28	25c/28 ·	25c/28
V27 V28	Lockheed Tuft			
WOOD	(Operations) Cali b ration	29b/29ga	29b/29ga	29b/29ga
V29B V29C	Lighting	->->->-		
V29E	Supporting + Carrying Jacking			
V29F V2 9 G	Tap			
V29GA	Foreing			
V29H	Theory +Theoretical*+Philosophy*+D	escriptive*	00h/00h-	29h/29r
V29HA V29HB	Theoretical Philosophy	29h/29ho	29h/29ho	~ 211/ « 21.
V29HC	Descriptive	00: /02:	003 /003	
V29I V29J	Generalized Unified	291/291	291/291	
V29K	Linearized			
V29L	Lineariz ati on			

		First Reduction	Second Reduction	Third Reduction
	[Linearization]			
V29M V290 V290 V290 V29P V29Q V29R	Rule + Law + Principle Hypothesis Assumption Assuming Expression Analogy Analogue	29M/29R	29M/29R	30/57 +87
V30 V31 V31a V31b V32 V33 V34 V34a	Mathematical (operations) Calculation Calculating Calculated Compute Computation Computational Formulation	30/34a+43 +87	30/34&+43 +46/51 +54/56 +87	
v36 v37 v38	(Methods) (Analytical) Exact Inverse Semi-inverse	36/38+42	3 6/ 38+42	
V41 V41	(Approximate) Variational Semi-analytical	41	41	
V43 V44	Graphical Numerical	44/45	44/45	
V45 V46 V47 V47a	Semi⇒numerical Interpolation Interpolative Extrapolated	46/48+51+56		
v48 v49 v50	(Curve) Fitting Plotting Plot	49 /50		
V51 V52	Bestfit (curve) Iteration	52/53	52/53	
V53 V54	Iterative + Step-by-step Relaxational Over-relaxational	54/55		
V55 V56 V57	Smoothing Integration	57	57	
V58 V59 V596 V596 V596 V596 V596 V60 V61 V61a	Agents in mathematical operations Computer Digital Dence Eniac Univac Pace NUTE (Analogue) Correlator Simulator Analyzer		58/64	58/64
v62 v63 v64	(Operations) Programming Programme Algorithm	62/64	,	
V65 V66 V67 V68 V68a V69	(Elements, etc. in mathematics) Problem Parameter Co-ordinate Ignorable Ordinate Replacement Term	65 66 67/68& 69/72+74/76	65 66/72+74/7 6	65/84
V71 V72 V73 V74 V74a	Termwise Root Value Pivotal (point) Exponent Formula	73	73	
V75 V76 V77 V78 V79 V79a V79b	Quadrature Solution Approximate Approximating Tending Approximation Tnscribed	77+82/83 78/79d	77+82/8 3a 78/79 d	
V79c V79d V80	Circumscribed Similar	80/81	80/81	

		First Reduction	Second Reduction	Third Reduction
	[Similar]			
A87	Similarity			
V82	Integrable Integrating			
v83 v83a	Integrative	•		
v84	Theorem	84	84	
	Division I (one liveric)	85/86	85/86	85/86
v85 v86	Dimensional (analysis) Scaling (law)	65/ 66	65/60	0)/00
v87	Multiplication			
v88	Algebraic	88+90/92	88+90/94	88/98a
V89	Equation	89	89	
V90	Biharmonic Compatibility			
V91 V91a	Transcendental			
V92	Telegraph			
V93	Polynomial	93 94		
V94	Matrix	95/98s	95/98a	
V95 V96	Transformation . Transform	, , , , ,	23/2	
V97	Conformal			
v98	Mapping			
V98a	Transversality (condition)			
3200	(Analysis) Calculus	99/103	99/103	99/103
V1.00	Tensor			
V100a	Operational			
V100b	Transcendant			
V102	Eigenvalue Eigenfunction			
V1.03 V1.04	Function	104/109	104/109	104/109
V104a	Logarithmic			
VlO4b	Exponential			
V105	Hypergeometric Confluent			
V106 V106a	Sine (law)			
V107	Bei			
V1.08	Ber			
V109	(Joint) Acceptance Trigonometric	110	110	110
V110 V111	Statistics	111/115+119	111/115+119	111/119
Vllla	Probability		,	
Vlllb	Coverage			
V112	Statistical Averaging			
V112a V113	Sampling			
V114	Weighting	446/440	116/118	
V115	Superposition Correlation + Relation + Rela	116/118 stionship	110/110	
V116 V116a	Correlation + held ton + held Correlation			
V110a	Autocorrelation			
V118	Autocorrelogram			
A173	Stochastic			
	(Common properties)	a/b	a/b	a/20a
Wa	Property + characteristic + Quality*	e, 5	~ ~	
Wl Wb	Quality Structural	1 2/5	1_	
MS	Isotropic	2/5	2/5	
W3	Orthotropic			
Wh Wha	Anisotropic Anisotropy			
W462 W5	Aelotropic			
	#Trapparation Control of Committee and State Control of	6+7	6/10+13/17	
w6	Composition	7/10	U7 107 13/ 1 (
W7 W8	Complex Composite			
W9	Compound			
MJO	Multi-component	11/12	11/12	
Wll	Mixture Stoichiometric	,		
W13	Homogeneous	13		
WIL	Inhomogeneous	14/16		
W15	Non-homogeneous			
W16	Heterogeneous Multiphase			
W17 W19	Integral	19	19	
W20	Sectional	20/20a	20/20&	
W20a	' Multi-sectional			

	(v	First Reduction	Second Reduction	Third Reduction
	[Multi-sectional]			
	(Results of action)			
W2l	(Influenced, altered) Applied	21/21b	21/21b	21/37
W2la	Affected	£1/ £10	21/210	21/)!
W21b	Processed			
W23	Modulated			
W24	Modified	05/06		
W25 W26	Revised Improved	25/26		
W27	Transformed	27	27/32a	
w28	Combined + Merged	28/29		
W29	Integrated	70/20-		
W30 W31	Mixed Premixed	30/32a		
W32a.	Unchanged			
W33	Bifurcated	33	33+ 3 7	
W35	Ablated	35/36	35/36	
₩36 ₩37	Non≈ablating Filter e d	57		
"")	(State altered)	21		
w38	Fluidized	38/40	38/40	38/48
W39	Dissolved			
W40	Absorbed	42/43	42/43	
W42	Ignited Burned	48 <i>43 43</i>	42/4)	
W44	Dissociated	44	44/48	
W46	Undissociated	46		
W47	Ionized + Changed	47		
W48	Frozen	48		
	(Dimensions altered)			
W52	Demped	52/54	52/54	52/58
₩53	Impeded			
W54	Obliterated Increased	55/57	55/57	
₩55 ₩56	Boosted	ו כי עביב	ול זכל	
₩57	Scooped			
	And the second s			
1150	(Position, etc., altered)	#0164	E0/60	E0/72*
W59 W60	Removed Transferred	59/61	59/69	59/72a
W61	Convected			
W62	Distributed + Spread	62/64a		
w63	Dispersed			
W64 W64a	Displaced Diffuse			
w65	Detached	65		
W66	Attached	66		
w67	Disturbed	67/68a		
W68 W68a	Perturbed Washed			
w69	Undisturbed	69		
W70	Concentrated	70/728	70/722	
W70a	Imposed			
W71 W72	Superimposed Piled			
W72a	Lumped			
	(Direction altered)	m 7	77/70	77/70
W73 W74	Yawed Unyawed	73 74	73/78	73/78
w 74 w 76	Reflected	76		
W 77	Reversed	77		
	(
W78a	(Held, retained) Prevented	78a/84	78a/84	78 s/84
W79	Restrained	1000 04	1000	, 0 3 0 4
W79a	Restricted.			
w81	Stabilized			
W83	Unretarded Resisted			
W84	nesisted			
	(Result of various forces)			
w85	Loaded	85/89a	85/89a	85/89a
w 86	Forced			
W87	Subjected Controlled			
w88 w89	Controlled Directed			
₩89a	Governed			
,				
	(By direction)			

		First Reduction	Second Reduction	Third Reduction
W90 W91 W92 W93	[(By direction)] Deflected Sheared Inflected Twisted	90/93	90/93	90/93
w94 w96 w97 w98	Compressed Driven Propelled Launched	94 96/98a	94 96/102	94/102
W98a W99 W100	Projected Injected Blown Emitted	99/100 101/102		
ATOS ATOTE ATOT	Tapped Extruded	10 17 10 2		
W1.03a W1.04	(Shape altered) Deformable Deformed	103a/105	103a/109	103e/X1
W105 W106 W107 W108	Warped Buckled Unbuckled Prebuckled	106/108		
Wille Xl	Rolled Thickened	111 <u>8</u> 1	1112	
X6 X7 X8	(Chemical/Physical properties) Combustible Non-flammable Non-toxic	6/8	6/11	6/11
XII XIO XQ	Dry Sumidity Dewpeint	9/11		41/41
XT/tp XT/t	Strong Wesk	14/14b	14/14b	14/145
X14c X15 X15a X16 X16a	(Origin, source) Accidental. Artificial Artificially Natural Nature	140/16a	140/16a	140/16a
X17a X17b X17c X17d X18 X18 X18a X19	Introduced Obtained Caused Due Generated Initiated Formed	17a/20+23a/23b	17e/20 +22/23b	18/23b
X19a X20 X21 X22 X23 X23a X23b	Composed Developed Induced Excited Simulated Terminated Ended	21 22 23	21	
X24 X25 X26	Frictional Mechanical Hydrostatic	24 25/26a	24 25/26a	24/28
X26a X28	Preumatic (Prayitational (Properties of processes)	28	28	
X30 X32 X32a	Eccentricity Circulatory Rotatory	30 32/32a	30 32/35	30/35
X33 X35 X36	Non-circula tory Translational Oscillatory	33/35 36 37/39	36/39	36/39
X37 X38 X39	Vibrational Vibrationally Vibratory	21/ 22		
X41a X41a X41b	Automatic Self-adjusting Self Auto	40/416	40/42	40/42
X/5	Corrective	42		

		First Reduction	Second Reduction	Third Reduction
	[Corrective]			
Х44 Х45 Х47	Dissipative Non-dissipative Diffusivity	44/45+53 47	44/47+53	44/53
X48 X48a X51	Conductive Conductivity Non-conducting	48/51+52a	48/52a	
X52 X52a X53	Convective Radiative Emissivity	52		
X55	Compressive	55	55	55
Ж56 Ж56а	Incipient Embryonic	56/56a	56	56/67a
X57 X58	Progressive Cumulative	57/58	57/58+63/65	
X59	Reversibility Irreversible	59/60	59/60	
X60 X63	Extensible	63/64		
X63a X64 X65 X66	Extendible Developable Incremental Redundant	6 5 66/67a	66/67a	
X67 X67а	Residual Remain	68/60	68/75	68/75
x68 x69	Steady Quasi-steady	68/69	00/15	00/15
X70 X71 X72 X72a	Continuous Continuously Continuity Sustained	70/72a		
X73 X74	Constant Uniform	73+75 74/74a		
X74a X74b X74c X74d X75	Uniformity Uniformly Regular Unperturbed Invariant			
X752 X75b X76 X76a X77 X78	Non⇒uniformly Non⇒uniformity Unsteady Quasi⇒unsteady Unsteadily Unsteadiness	75 √ 79	75a/88	75a/88
X79 X80 X81 X82	Nonsteady Discontinuous Discrete Discontinuity	80/82		
x83 x84 x85	Irregularity Uneveness Erratic Variable	83/85 86		
x86 x88	Non-monotonic	88		
x89 x89a x89b x90	Negatively No Not Negative	8 9/ 90	89/92	89/92
X90a X91	Non-negative Positive	901/91		
X92 X92a X92b X92c	Neutral Adverse Favourable Good	9 2/ 920		
X93 X93a	Rest Resting	93/94+96/97	93/97	93/107
x94 x95 x96 x97	Stagnant Stationary Dead Still	95		
x98 x98a	Static Statically	98/98a	98/98a	
X999 X998 X100 X101 X101a X102	Immobile Motionless Passive Non-reactive Non-reacting Inert	99/102	99/102	

		First Reduction	Second Reduction	Third Reduct <u>ion</u>
	F "	MARIO ETOTI	110000 02000	Committee Anna Committee C
X103 '	[Inert] Stable	103/104a	103/104a	
X103	Bistable	104b/104e	104b/104e	
X104b	Movable			
X1040	Allmovable Immovable		·	
X1046 X1040	Allmoving			
25 1090	[Aotive]	40//000	106/107	
X106	Active Reactive	106/107 106/107	106/107	
X107	NOSCILVE	100, 101	,	
	[General relations]		4 - 4 -	
X110	Absolute	110	110/112a	110/119
X111	Relative Relativistio	111/112&		
X112 X112s	Relatively			
X113	Containment	113	113/119	
X115	Correlation	115/119		
X115a X115b	With Corresponding			
X1150	Applicability			
X116	Correlated			
X117	Connected Associated			
X118 X119	Countai	.7		
~~,	[Sequence and Rank (General relation	(anc	81+84a/90	81/92
H81	Order Preliminary	81a/84	81a/84	- ")-
H81a H82	First + Initial			
H82a	Initially			
Н82Ъ	Originally			
H83	Primary + Major + Main Semi-major			
H84 H84a	Minor	84a/85a		
H85	Second + Secondary	86/87		
H86	Third Cubic	30/01		
H87 H88	Fourth	88/9		
1169	Quartic	89a		4
H89a	Fifth Sixth	89b		
H89b H90	Seventh	90		
H90a	Eighth	90a 91/92	91/92	
H91	Final Terminal	J 1/ JC	71174	_
H92 H93	Auxiliary	93	93	93/94
H94	Intermediate	94	94	
70.0	(Dimensional properties) Dimensional	1/28	1/28+8	1/8
M2 M1	Geometric	•		
M2a	Dimensionless	7/11	3/3b	
м3	One-dimensional + Linear	3/3b	J/ J6	
M3a M3b	Linearly Quasi-one-dimensional			
MA	Non-linear	4/5	4/5	
М5	Non-linearly	6/6a	6/6a	
m6 m6a	Two-dimensional Two-dimensions			
M7	Three-dimensional	7/72	7/7a	
M7a	Three-dimensionality	8	8	
M8 M21	Non-dimensio nal High + Highly*+Higher*+Elevated*	21/220	21/24	21/27 a
M22	Highly			
M22a	Higher			
M22b	Elevated Ultrahigh			
M220 M23	Critical	23/24		
M24	Superoritical	25	25	
M25	Low + Small + Reduced	~ /	-,	
	[Magni tudo]			00 1003
M28	Dense	28	28/28b +28g	28/28b +28g
M28a	Lightly	28a/28g 28b	420g	1 6 9 9
M28b M28c	Heavy Intensi ty	28o/28d	28o/28f	28¢/28£
M28d	Intense			
M28e	Severe	28e/28£		
M281	Severely Light + Lightweight			
M28g	market from the second of the person of the person of the second of the	29/30	29/30	29/31b
м29 м30	Quantity + Amount Content	- J/ JU	2/ 2-	+65e/65b
m JO	O 202 A 200 A			

			First Reduction	Second Reduction	Third Reduction
	[General relations] [Magnitude] [Content]				
N31 N31a	Dogree Percentage		31/316	31/31b	
M316 M32 M33	Semi Single Singly		32/34	32/34	32/29b +57/65
M34 M35 M36	Singular Multi + Multiple Double Dual		35 36/39b	35/39b	
M37 M38 M39 M39a	Binery Pair Twin				
M39b M40 M41	Conjugate Finite Limited		40/41	40/41	40/43a
N42 M43 M43a	Semi-infinite Infinite Infinity		42 43/43d	42 43/43d	
M43b M430 M43d M44	Infinitely Infinitesimal Infinitesimally Minimum		44	44	44/56
и45 N46 N46ћ	Mean Maximum + Maxima Optimal		45 46 46b	45 46/46b	, , , ,
M47a M47a M47b	Least Less-than Negligible		47/47d	47/47å	
M470 M47d M48	Vanishingly Slightly Partially + Partly		48/48a	48/48a	
M48a M49 M50	Partial Fully Full		49/52	49/56	
M51. M51s M51b M52	Completely Entirely Very Complete				
M53 M54 M55	Total Overall Collective	4	53/55&		
M55a M56	Excess Extreme		56		
M57 M57a M57b	Number Zero Non-zero		57/57b +60/65	57/65	
M58 M59 M58a M60	One Unit Unity Two		58 / 59&		
M61 M63 M64	Three Four Five				
M65 M65a	Six		65 a/ 65b	65s/65b	
M65b M65c	GYO 88		650/68	650/68	65c/71a
M66 M67 M68	Arrangement Configuration Layout Pattern		0,0,00		, , , ,
M71 M71a	Combination Collocation		71/718	71/71 a	
M72 M72a M72b	Ordered System atic Graded		72/ 72b	72/72b	72/73
M720 M72d M73	Arbitrary Arbitrarily Random		720/73	720/73	

		First Reduction	Second Reduction	Third Reduction
Y1 Y2 Y3 Y3a Y3b	[Random] Equivalence Equivalent Equally Equal Equal	1/3b	1/36	1/36
Y5 Y6 Y6a	Complementary Mutual Reciprocity	5/6b	5/6b	5/17a
Y6d Y7 Y8 Y9	Reciprocal Dependence Dependent Dependant	7/9	7/9Ъ	
Y9a Y9b Y10 Y11	Incident Coincident Proportional Proportion	9a/9b 10/11	10/11+14/16	
Y13 Y14 Y15	Ratio Modulus Secant	13 14/16	13	
Y16 Y17 Y17a	Tangent Coefficient Derivative	17 17a	17/17a	
Y19 Y19a Y19b Y19c	Similitude Analogous Alternative Substitute	19/23	19/24b	19/24b
Y22 Y23 Y24 Y24	Quasi Pseudo Consistent Valid	2 4/ 24b		
¥24b ¥25	True Difference	25/26+28/29	25/26+28/31	25/31b
Y26 Y27 Y28	Differential Differential Differentially	27	27	
Y29 Y30	Differing Unequal	30/31		
Y31 Y31a	Inequality Foreign	31a/31b	31a/31b	
Y31b Y32 Y33 Y33a	Non-similar Independence Independent + Free + Uncoupled Unsupported	32/33a	32/33a+37/37a	32/39
Y35 Y36 Y36a	Freely Freedom Buoyanoy	35/36a	35/36a	
Y37 Y37a	Singularity Individual	37/37a		
¥38 ¥39	Without Absence	38/39	38/39	
Y40	(Abstract concepts) Phenomena	40/40a	40/40a	40/40a +43/48
Y40a Y41 Y42 Y42a Y42b Y420	Phenomenological System Group Type Species Set	41/42d	41/42d	41/42d
Y42d Y43	Member Concept	43		
Y44 Y45 Y47	Criterion Reference Ultimate	44/47	44/47	
Y48	Best	48	48	
Y48a Y48b Y48c	Input Output Yield	48a/48d	48a/48d	48a/48d
Y48d Y49 Y49a Y50 Y51 Y51a	Target Dåta Product Result Resultant Resulting	49/52	49/52	49/52
<i>₽</i> · ==	CONTRIBUTION OF A MEDICAL MEDICAL CONTRIBUTION OF THE PROPERTY			

		First Reduction	Second Reduction	Third Reduction
Y52	[Resulting] (Form of information) Record			
Y52a Y53 Y55	Description Diagram Chart	5 2a+53 /58b	52a/58b	52&/60
Y55a Y55b Y57	Table Tabulation Survey			
Y57a Y58 Y58a	Art (state of) Review Bibliography			
Y58Ъ Y59 Y60	Literature Index Trace	59 60	59/60	
Y60a	(Curves, etc.) Polar	60a	60 a/6 5	602/65
161 162	Hodograph Isobar Vector	61/65		
Y63 Y64 Y65	Isogon U-R (velocity-distance)		•	
Y68	[Abstract concepts] Empirical	68/68a	68/68a	68/68a +80/86a
Y68a Y70 Y71	Semi-empirical Hypothetical Ideal	70/73a+79/79b	70/79ъ	70/79b
¥72 ¥73	Idealized Perfect			
Y73a Y74	Perfectly Simplified	74/768		
Y75 Y76	Simply Simple			
Y76a Y77 Y78	Elementary Pure Optimum	77 78		
Y79 Y79a	Apparent Assumed + Given	•		
Y79Ъ Y80	Expressible Actual	80/83	80/86&	
Y81 Y82	Practical Imperfect			
Y83 Y84 Y85	Non-perfect Working Real	84/85		
186 186a	Effective Ordinary	86/ 86a		
186b 1860	General Normally	86b/88	86b/91a	861/99
Y87 Y88	Standard Average	90/90-		
Y89 Y89a	Classical Conservativeness Basic	89/89& 91/91&		
Y91 Y91a Y93	Plain Required	93/94	93/94	
Y93a Y94	Appropriate Requirement			
195 196	Preferential Selected	95/96	95/96 98/98a	98/100
Y98 Y98a	Specific Particular	98/98a 99/100	99/100	90/100
Y99 Y99a Y100	Measured Valued Calibrated	937, 100	<i>yy</i> , ,	
Z1 Z1a	Efficiency Efficient	1/2		
Z2 Z3	Effectiveness Accuracy	3/3a		
23a 24	Accurate Capability	4/5		
25 26	Capacity Deficiency	6/8a		
Z7 Z8 Z8a	Defect Imperfection Inclusion [in crystal]			
29 210	Error Aberration	9/10		

APPENDIX 5.4

SIMPLE CONCEPT HIERARCHIES

The schedules include the concepts used in the indexing of the sub-set of 200 documents as given in Appendix JE.

The concepts are in natural uncontrolled language.

Notation is purely ordinal, and in no way reflects the relations between the concepts represented.

Plus signs indicate synonyms.

Certain concepts appeared in the sub-set of test questions, but did not occur in the indexing of the documents. Such concepts have been included and are followed by the letter 'Q'.

Summary of the main sections

```
[Bodies, Structures]
          Aircraft types
[Parts of Aircraft]
A2
           Airfoils; control surfaces
A79
          Other parts: Fuselage, panels, skin, forebody, etc.
C57
             Propulsion systems, etc.
             Compressors
E64
             Ducts: pipes, nozzles
G5
           Rodies
H17x
             By shape, etc.
H43
           Non aerodynamic structures: pivots, beams, etc.
J35
J46
         Flight
           Interplanetary flight
J48
           Performance
к6
           Flying operations
K43
         Fluids
L5
L25
           Gases
L86
              Atmosphere
м34
           Flow
 M42
              By speed
              By various other characteristics
Boundary layer
 M69
 N64
              Flow elements: streamline, vortices, jets, etc.
 064
                Jets
 P27
                Waves, shock waves
              Aerodynamic forces and loads: lift, drag, moments, pressure, etc.
 Q33
              Aerodynamic processes and properties: control, stability, interference, wake, velocity, flow parameters
 R22
 T27
            Fluid properties: density, viscosity, etc.
 U77
            Aeroelasticity, flutter
            Aerodynamic reference parameters: angle of attack, planform, aspect ratio,
 U90
 V94
                                                  section, camber, etc.
          Mechanics: motion, force, stress-strain phenomena
 ₩50
 X4
            Heat
              Heat transfer
 X80
            Kinetic theory: dissociation
 Y35p
          Research and investigation
  ¥76
            Experiment
 ¥82
              Wind tunnel
 Z1
             Theory
  Z98
               Mathematics
  ZA9
               Automatic control theory
  ZB27
          General properties
General processes
  ZB45
ZB73
```

```
Magnetoplasma [r.p.]
                                                          A74
      Bodies
                                                                          Ion [r.p.]
A2
        [Airplane] Q
                                                          A75
                                                                           Solar heating r.
A3
           High speed aircraft
                                                          476
                                                                          Midcourse r.
                                                          A77
           Avro-707A
                                                                         Light wave propulsion +
            Airoraft structures
                                                          A78
A5
                                                                         Solar sail
           Models of complete aircraft
           Tailless a.
A7
                                                          A79
AB
           Tail-boom a.
                                                          A80
                                                                      Lifting element
A9
            Aircraft shapes
                                                                       Non-1. airfoil + Non-1. body
                                                          A81
A10
         Vahioles
                                                          A82
                                                                       Airfoil equation
            Ballistic v. + B. Missile
A11
                                                           A83
                                                                       Two-dimensional &.
            Winged missils
A12
                                                          A84
                                                                       Three-d. a.
            Long range m.
A13
                                                                       A. shape
                                                           A85
            Missile oscillating in roll
ATA
                                                                         Wedge a.
                                                          486
A15
A16
            Space v.
                                                                           Single w.s.
              Satellite + S.v. + Artificial s.
                                                           487
                                                                           Double w.a.
                                                           A88
                Earth satellite
A17
                                                                         Circular arc a. + C.arc profile
                                                           A89
A18
                 Atlas
                                                           A89a
                                                                         Biconvex a.
A19
                 Discoverer s.
                                                                         Straightline a.
                                                           A90
                 Explorer 1
A20
                                                                         Blunt nosed a.
                                                           A91
A22
                 Explorer 3
                                                           A92
                                                                         Blunt trailing edge a.
                 Explorer 4
A23
                                                           a93
                                                                         Symmetrical a.
A24
                 Midas 2
                                                           A94
                                                                         Rigid a.
A25
                 Sputnik 1
                                                                         NACA 4412 &.
                                                           A95
A26
                 Sputnik II
                                                                         NACA 64A008 a.
                                                           A95a
                 Sputnik III
A27
                                                                         Semi-infinite plane
                 Sputnik V
                                                           A96
A28
                                                                       Wing body combination +
                 Vanguard I
                                                           A97
 129
                                                                       W.b. arrangement
               Re-entry v. + R. body
A30
                                                                         W.b. interference +
                                                           A98
                 Highly cooled body
 A31
                                                                       W.fuselage i.

[Effect of] Small span
curvature of flowfield
Airplane wing + W. + W.body
                 Graphite r.e. body
 A32
                                                           A99
                 Lifting r.e. configuration
 A33
                 Delta wing r.e. configuration
 ASA
                                                           В1
                 Sweptback w. r.e. configuration
 A35
                                                           В2
                                                                         Linearized w. theory
                 Blunt r.e.v. + Blunted r.e.
 A36
                                                                         W. model
                                                           B3
                                 configuration
                                                                         W. shapes + Arbitrary planform
                                                           В4
                 Re-entry glider + R.e. glider
                                                                         w. + W.p.
[Simple planform w.]
 A37
                 configuration + Glide v. +
                                                           B5
                 Glider r.e. configuration +
                                                           В6
                                                                            Swept w.
                                                                              Highly s.w.
                 Glider v.
                                                           B7
                   Winged r.e.g. configuration [with] Wing tip panel
                                                                              W. sweep small
 A38
                                                           RA.
                                                                              Sweptback w.
 A39
                                                           B9
                                                                                septeack w.
5.w. model
45 sweptback w.
5.w. 50
60 sweepback w.
                        Folding
                                                           B10
 AAO
                        Unfolding
 649
                                                           B10a
                  Lifting r.e.g. configuration [Uncontrolled v.] Q
 AA2
                                                           B10b
                                                           B11
 AAS
                  Škip v.
                                                                                S. delta w.
                                                           B12
 AAA
                                                           B13
                                                                              Sweptforward w.
                 R.e.v. surfaces
 A45
                                                                              Sweptforward w. tip
                                                           B14
               Test v.
 AA6
                                                                            Unswept w. + Straight w.
                  Probe v. + P.v. configuration
                                                           B15
 147
                                                                            Triangular w. + Delta w.
                                                           B16
                  + Space p.
                                                           B17
                                                                              Near t.w. Flat t.w.
                    Short p.v.
 A48
                    Sounding p. + S. rooket system
                                                           B18
 A49
                                                                              Clipped d.w.
                                                           B19
                Supersonic v.
  A50
                                                                              T.w. with twist
                                                           B20
  A51
                  S. model
                                                                            Rectangular w. + R.planform w.
                                                           B21
               Hypersonic v. + H. body
  A52
                                                                              Slender r.w.
                                                           B22
                                                                            Circular planform w.
                                                           B23
              Vehicle shapes
                                                                            Curved tip planform w.
                                                           R2A
                Arrowing v. Blunt body of revolution v.
  A53
                                                                            Biconvex section w.
                                                           R24a
  A54
                                                                            Elliptical w. + Elliptic w.
                Blunt noss v.
                                                           B25
  A55
                                                                              [with] Major/minor axis
                                                           B26
                    blunted come v.
  A56
                                                                               ratio 5.1
                Conical afterbody v.
  A57
                                                                            Symmetrical w.
                                                           B27
                Flat base v.
  A58
                                                                              S.w. thickness
                                                           B28
              Propulsion systems
  A59
                                                                            Tapered w.
                                                            B29
                Airbreathing v.
  A60
                                                                              T.w. model
                                                            B30
                Rocket p. + R.p. system +
  A61
                                                            B31
                                                                            Untapered w.
                R. propelled v. + R.v.
                  Multiple r. engine
  462
                                                           B32
                                                                            Cambered w.
                  Five stage r.s.
                                                                            Uncambered w.
  A63
                                                           B33
                  Chemical r.
                                                            B34
  A64
                    Liquid propellant
                                                                              W. with parabolic twist
                                                           B35
  A65
                    Solid propellant + Solid
  A66
                                                                            Warren 12 planform
                                                            B36
                    fuel r.o.
  A67
                  Electrical r.
                                                            B37
                                                                            Rigid w.
                  Nuclear energy source
                                                                            Elastic w.
  A68
                                                            B38
                    Fusion
                                                                            Arbitrary stiffness w.
  A69
                                                            B39
                     Fission
                                                                            Solid construction w.
  A70
                                                            B40
                    Radicactive isotope
  A71
                                                            B41
                                                                            Cantilevered w.
                       Decay
                                                                              C.w. flutter
  A72
                   Arc-heating r.
```

			. F
B43	Gropped w.	C22	Tailplane Li.e. horizontal
B44	Perforated w. + Hole drilled w.		stabilizing surface)
B45	Sharp edge w.	C23	T. setting
B46	Sharp leading edge w.	C24	T. spanload
B47	Subsonic edged w.	C25	Control surface
B48	Supersonic edged w.	C26	All movable c.s.
210	Two-dimensional w.	C27	[Apex located c] Q
B49	Three-d. wing	C28	(Forward located c.] Q
B50	7777 G.C. A. 4. 1770		[Rudder]
B51	Long w.	C29	R. deflection
B52	Wide w.	030	Differential r.d.
1953	Narrow w.		Aileron
B54	Slender w. + Thin w.	C31	_ Oscillatory a. coefficients
B55	S.w. theory		Elevon
B56	Not so slender w. theory	C32	E. deflection
B57	Thick w.	C33	Differential e.d.
	The second making with 1 th 50	C34	Control flap
B58	Low aspect ratio w. + h.a.r.	C35	Hinged f.
1559	Vanishing l.a.r.w.	C36	Blowing f.
360	W. of very small a.r.	C37	Jet f.
B61	High a.r.w. Finite a.r.w.	C38	Thwaites j.f.
362 363	Infinite a.r.w.	C39	Leading edge f.
B63	Finite w.	C40	Trailing edge f. [Slot]
B64	Finite span w.	044	Slot blowing + High
365	Effect Show as	C41	velocity blowing
B66	Infinite w.	040	S, configuration
B67	Infinite s.w.	C42	Control surface tabs
B68	W. with concentrated weights	C43	Oscillatory t. coefficient
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P35	Supersonic rocket	Q10	J. interference + J.i.phenomena
P36 .	Propulsive J.e.	Q11	Rocket j.1.
P37	Rocket j. + Rocket [producing j.]	Q12	Jet exhaust i.
P39	Supersonic retrorocket		
	Simulated r	Q13	J.i. effects
P40		Q14	Exit shock
P41	Jet engine [producing j.]	Q15	E.s. wave angle
	•	• •	

Q16	J.s.w. angle	Q88	Terminal position
Q17	J.s.w. apex location	Q89	S.w. position divergence Sonic point location +
Q18	J. wake	290	Location of s.p. + Sonic p.
Q19	Propulsive j.w.		+ Position
Q20	J. mixing	Q91	Stand-off distance +
Q21	Compressible j.m. Isoenergetic j.m.	• .	s.s-o.d. + S.detachment
Q22	Turbulent j.m.		distance
Q23 Q24	Production of species	Q92	Shock Mach no.
W. 4	[in reacting gases]	Q93	Range of primary s.m.n.
Q25	Net mass rate	Q94	High s.m.no.
Q26	Continuity of species	Q95	Low s.m.no.
Q27	Chemical reaction	Q96	Air s. velocity S. equilibrium temperature
Q28	Concentration profiles	97	Shock layer] Q
Q29	Effect of Rocket thrust	Q98	Entropy layer
	[Study]	Q99 R1	Detached s.1.
Q30	[Use of] Sting balance	R2	Thin s.l.
0.74	measurement Jet simulation	R3	Thin s.l. theory
Q31	By Helium s.	R4	Shock wave surface
Q32	- 7		[Properties]
	Waves	R5	S. layer luminosity
933	Kelvin Helmholtz w.	R6	S.1. luminous gas region
Q34	Riemann w. Tollmien Schlichting w.	R7	Equilibrium s.l.property
Q35	Reflected w. + Waves reflected		Processes
Q36	Reflected from contact surface	DO.	S.w. formation
Q37 Q38	Wave reflection interference	R8	Time of appearance
Q39	Flexural w.	R9 R10	S.w. development
Q40	Longitudinal w.	R11	S.w. movement
Q42	Compression w.	R12	S. reflection
Q43	C.w. resonances	R13	R. of primary s.
Q44	Expansion w.	R14	Attenuation of reflected s.
Q45	Reflected head	R15	Attenuation of s.w.
Q46	Originating	R16	[Analysis by] Conservation
Q47	<pre>[at]Main diaphragm [of shock tunnel]</pre>		of mass equation
0.40		R17	Transonic flow attachment
Q48	Shock w. + S. Strong s.w.	R18	S. displacement
Q49	Mach w. + M. line	R19	Conical s. separation S. transition zone
Q50	Infinitesimally thin s.w.	R20	S.w. interaction
Q51	Forward s.w.	R21 R22	Aerodynamic forces +[Airforces] Q
Q52 Q53	Rear s.w.	R24	Aerodynamic inertia
Q54	Terminal s.w.	R25	A. normal f. + Normal f.
Q55	Rate of movement	R26	Reaction n.f.
Q56	Outboard s.w.	R27	Incremental n.f.
Q57	Normal s.w.	R28	Spanwise coordinate
Q58	Oblique s.w. + O.s.w.system	R29	Axial f.
Q59	Detached s.w.	R30	A. body f.
Q60	Attached s.w. system	R31	Centrifugal f.
Q61	Hypersonic s.w.	R33	Side f.
Q62	Incident s.	R34	Tangential f.
Q63	Oxygen s.w.	R35	Retarding f.
Q64	O.s. front Nitrogen s.w.	R36	Deceleration f.
Q65	N.s. front	R37	Newtonian impact f.
Q66	Argon s.w.	R39	Body f.
Q67 Q68	Spherical s.		F. coefficient
Q69	Cylindrical s.	R40 R41	Static aerodynamic
Q69a	Mach cone	1124 1	characteristics
Q70	Primary s.	R42	Longitudinal aerodynamic
Q71	Reflected s.		characteristics + Longitud-
••	- 7		inal aerodynamic coefficients
	Shock wave phenomena	R43	Axial f.c.
Q72	Rankine-Hugoniot s. relations	R44	Chord f.c.
Q73	Mach angle	R45	Normal f.c.
Q74	Deflexion angle	R46	Incremental n.f.c.
Q75	Expansion zone + E. region	R47	Force determination
Q76	Supersonic e.r. Initial e.r. + Prandtl-	R48	Force measurement
Q77		R49	Loading + Aerodynamic load +
	Meyer region Flow behind a.s.w. +		A. loading Chordwise load + C.loading
Q78	Flowfield behind + Region	R50	Spanwise loading
	behind shock + Upstream of s.w.	R51	Uniform s.1.
040	Behind normal s.	R52	Subsonic span loads
Q79 080	Ahead of s.w.	R53	Total span loads
ଭୃ 80 ଭୃ81	S.w. pattern	R54 R55	Symmetrical loading
Q82	S.w. structure	R56	Model louding
Q83	S.w. shape	R57	Induced load
Q84	S. curvature	R58	Continuous load + C. loading
Q85	S.w. thickness	R59	Representation (by)
Q86	S.w. strength + S.s.	R60	Patterns of horseshoe
Q87	S.w. location + S. positions		vortices

R61	Unsteady loads	076	[Drag phenomena] D. coefficient
R62	Vanishing load see also Flow attachment	S36 S37	Profile d.c.
R63	Load distributions	S38	Total d.c.
R65	Lift	S39	D. reduction
R66	Starting 1.	S40	Minimum-d.design
R67	Transient 1.	S41	D. rise
R68	Final 1.	S42	D. reversibility theorem
R69	Maximum 1.		[Other loads]
R70	Zero 1. Unsteady 1.	344	(Blast + Gust) Q
R71 R72	[U.1. distribution] Q	S45	Sinusoidal gust
R73	Indicial 1.	S46	Gust penetration
R74	Non-linear 1.	S47	P. of sharp edged g.
R75	Circulatory 1.	S48	P. of s.a. normal g.
R76	Compressible 1.	S49	Moments + Aerodynamic m.
277	Incompressible 1.	S50	Pitching m.
R78	Induced 1.	S51	Aerodynamic centre A.c. characteristics
R79	Incremental 1.	552 552a	Local s.c.
R80 R81	Rody 1. Section 1.	S53	P.m. coefficients
October 1		854	P.m. curves
	Lift phenomena	S55	P.m. derivatives
R82	Lift effect [see also Vortex	S56	P.m. reversal
	system 092	S57	Rolling m.
R95	L. slope	S58	Frequency response in
R84	Sectional 1.s. + Section 1.	cen	moment Indicial response in moment
TOP	curve	S59 S60	M. function
R85	Distribution of s. Lift-curve slope	S61	Indicial m.f.
R86 R87	Lift-drag ratio	\$62	M, vector
RSS	Naximum 1.d.r.	S63	M. coefficient
R89	L. vector	S64	Aerodynamic balance
R90	Phase angle	S65	Pressure + P. load
R91	Magni tude	S66	Total p.
R92	L. distribution	S67	T. head measurement
R93	[Calculation by] Rounding off ru	S69	T.p. ratio Static p.
R94	Spanwise 1.d. D. of pivoted points	S70	S.p. survey
R95 R96	L. interference	571	S.p. ratio
. · R97	Total 1. response	S72	S.p. measurements
R98	Frequency response in 1.	S73	S.p. gradient
R99	Indicial response in 1.	S74	S.p. distribution
51	Apparent mass effects	S75	Dynamic p.
318	Vertical acceleration	S76	Impact p. survey
52	Lift coefficient + Lifting C.	S77 S78	Base p. Two-dimensional
C. 2	+ C. of l. L.c. divergence	579	B.p. measurement
S5 S4	Oscillatory 1.c.	S80	Kink p.
\$5	Asymptotic behaviour	S81	Stagnation p. + S. point p.
sé	Total l.c. + Coefficient of t.l.	S82	S. pressure coefficient
57	Trim 1.0.	883	Flight s.p.
\$8	No-lift moment coefficient	S84	Centre of p.) <u>see</u> Spanwise c.o.p.) P. char-
\$9 \$10	Indicial 1. function + Unsteady 1.	2. 367 S86	Chordwise location) acteri-
S10 S11	Wagner i.l.f, Theodorsen alternating l.f.	300	atics (S92)
S12	Kussner function	S87	Induced p.
· · · · · · · · · · · · · · · · · · ·	Determination, etc.	588	Constant p.
513	Downwash integral equation	589	Oscillating p. field
514	Classical wave equation solution	S90	Oscillatory p. decay
S15	Kirchoff formula	S91	Oscillatory p. plotting [Pressure characteristics]
S16	Drag + Retarding force		<u>see also</u> 385/86
517	Friction effects Model d.	892	P. field
S18 S19	Total d.	S93	P. coefficient
S20	Total d. coefficient	894	P. ratio
S21	Profile d. coefficient	897	P. distribution + P.profile
522	Form d.	S98	P. gradient
523	Skin friction + S.f.d.	S99	Self-induced
S24	Average s.f. coefficient	T1	Streanwise Body p.d.
S25	Distribution	T2 T3	Chordwise p.d.
S26	Viscous d. + V. force Pressure d. (<u>see also</u> Form d.)	19 14	Longitudinal p.d.
527 528	Normal p.d.	T5	Radial p.d.
S29	Wave d.	T6	Sinusoidal p.d.
S30	Lift dependent d. + Induced d.	17	Deflexion
S31	L.d.d. component	T8	Travelling p.d.
532	Induced d. correction	T9	Prescribed p.d. Tangential p. gradient
S33	Resultant d.	T10	Vanishing p. gradient
S34	D. penalty	T11 T12	Zero p. gradient
S35	Excess d.	4. 1 Co	

713	P. variation + P. Fluctuation	T78	Discrete horseshoe vortex
	→ P. difference		[method] + Finite step
214		T79	Shock expansion method
215		Т79а	Blasius equation
T16	Base bleed	moo	[Interference parameters]
217	• • • • • • •	T80	Nonsteady perturbation
1118	P.r. reduction	T81	potential
719	P. rise Pressure de termination	101	Perturbation velocity potential
T20 T21		T82	Relaxation time effects
T22		T83	Interaction force
123		T84	I.f. ratio
124		T85	Mixing flow + Fluid m. +
425	Photorecording manometer	-	M. process
T26		T86	Constant pressure m.
	Aerodynamic processes and properties	T87	Laminar m.
927	Flow development	T88	Turbulent m.
728		T89	Stagnation flow
729		T90	S. air f.
430	*	T91	S. region
231		T92	Axisymmetric s.r.
T32	jer va	T93	S.r. regime I
2000	for C	T93a	S.r. regime II
1535		T94	S. point + S.p. flow Rear s.p.
T34		T95	near paba
N1 TE C	type j.C.	T96	S. conditions
T35	Stability + Aerodynamic s. Flow s.	T97	S. gas state
T36 T37	State a	T98	S, temperature
797 738	S.s. characteristics	T99	High s.t.
139 139	S.s. derivatives	Т99а	S.t. from 2400-3800°F
-37	Dynamic s.]	T100	S.p. heating
TAO	D.s. characteristics	T101	Flight s. enthalpy
241	D.s. derivatives	T102	Choking
742	Longitudinal s.	U1	Choking Mach no. [Choked tunnel] Q
T43	Lateral s.	U4	C. wind t.
T44	Directional s.	บ5 บ6	C.w.t. tests
245	[Aerodynamic derivatives] Q	U7	Downwash
	+ Stability d.	บ่อ	Induced d.
T46	Direct a.d.	U9	Total d.
247	Indirect a.d.	U10	Centre line d.
248	Rotary d.	U11	Steady d. conditions
T49	S, control	U12	D. distribution
2498	[Stabilization] Q	013	Periodio d.d.
T50	Destabilizing effect		[Upwash]
751	Disturbance + D. to flow	U15	. field
152	Distortion	U16	U. distribution
253	Downstream d.	U17	Sidewash
254	Small flow d.	U18	Spanwise outflow
155 1156	Infinitely small d. Small d. theory + Small	U19	Wake + W. flow
1156	perturbation theory	U20	W. width divergence
257	Inviscid s.d.t. see also	U21	Development of circulating w.
غ م _{ال} شا	B.L. theory N65	U22	External slipstream
758	Hypersonic s.d.t. see also	U23	Turbulent
A. 100 111	B.L. theory N65	U24	Surge
1259	Interference + Interaction		[Flow attributes]
T60	Interference load	U25	Velocity
261	Aerodynamic interaction effect	1127	Induced v.
	+ Interaction effects + Interference	U28	Vertical i.v.
	effects	U29	Vertical v.
762	Second order interference	U32	Gas flow v.
763	Wake interference effects	U33	V. potential
		U34	Symmetric v.p.
T64	Viscous interaction	U35	Anti-s.v.p.
165	Support system interference	U36	Acceleration p.
166	[Interaction between] Parallel streams	U 37	V. distribution + V. variation
m ರ ಅ		U38	V. gradient
T67 T68	Blockage interference B. effects	U39	V. profile
	Solid b.	U40	Linear v.p.
T69 T70	Wake b.	U40a	
		U41	Quasi-conical v. field
771	B. ratios	U42	V. measurement
272	B. correction	U44	M.f. fluctuation see also
773	Mass flow method	11 45	Sound waves
274	Reflection + Reflected	U45	Mass flow profile
275	Multiple reflection	U46	Flow purameters Mach number
176	Ground effect + Proximity	U47	Stream L.n.
275 p. 10 h h h	te ground	U48	Local E.n.
T77	Adverse g.e.	U49 U50	Increasing M.n.
	<pre>[Methods of determination, etc., of interference]</pre>	053	E.n. profile
	+	~ 10	******

U54	Prandtl number	V21	Aerodynamic f. derivatives
บรีร์	Reynolds number	A55	Two-dimensional f.d.
U56	Low R.n.	V23	Subsonic f.d.
U56a	R.n. less than 40	V24	F. frequency F. speed
U56b	R.n. 40	V25 V26	Critical f.s.
U56a	R.n. 44 Local mesh R.n.	V27	F. boundaries
V57 V58	Hartmann number	V28	Margins of safety
U59	Flow coefficient	V29	F. calculation
- 20	[Circulation]	V30	F. testing + F. tests
U60	C. function	V31	Variable medium
U61	C. control	V32	F. model
U62	by Leminarization	V33	Construction M. density ratio
U63	Reduction of c.	V34 V35	M. stiffness[=bending s.]
U64	Vorticity	V36	M. weight ratio
	[Injection] [Characteristics]	V37	Reduction m. elasticity
U66	Blowing pressure 037f	V38	Mass loading
U67	Injected gas + Injected	V39	Increased
	gas stream 037g	V40	Modal analysis
U68	Molecular weight 037h	V41	F. precipitation
U69	Specific heat ratio 0371	V42	Static testing
U70	Accommodation height 037j	743	Modified strip analysis
U71	Injected liquid 037k	V44	+ Rayleigh type a. Dietze's method
U72	Primary gas 037m	V45	Vibrational modes + Oscillation m.
873	Exothermic reaction 037n	V46	Uncoupled v.m.
U74	Magnification factor 037p	V47	Normal m. + N.m. of v.
U75	Small rate of heat	V48	Oscillating in elastic m.
	addition 037q	V49	Oscillating in rigid m.
U76	Specific impulse 037s	750	Body m.
7.7 mm out	[Fluid properties]	V51	Rayleigh oscillation m.
U77	Density	V52	Selected m.
U78	D. distribution	V53	Sound waves see also Mass flow
U79 U80	D. variation + Change of d. D. ratio	170° a	fluctuation Aerodynamic noise
U81	Low d.	V54	Subjected to + Exposed to +
U82	Compressibility	V55	Engine noise
U83	C. effects	V56	Jet e.n.
U84	Viscosity	V57	Sound propagation
	[Characteristics]	V58	Acoustic radiator
U85	[VTemperature law] Q	-	
บอ6	[Linear vt. solution] Q		[Vibration characteristics]
U87	Viscous effects		[see also Stability]
U87 U88	Viscous effects Shear	V59	<pre>(see also Stability] Oscillatory coefficients</pre>
U87	Viscous effects Shear Coefficient of v.	V60	[see also Stability] Oscillatory coefficients 0. aerodynamic derivatives
U87 U88 U89	Viscous effects Shear Coefficient of v. [Aeroelasticity]	V60 V61	[see also Stability] Oscillatory coefficients O. aerodynamic derivatives Yeasurement of reaction
U87 U88 U89 U90	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a.	V60	[see also Stability] Oscillatory coefficients O. aerodynamic derivatives Measurement of reaction [Neasurement by]
U87 U88 U89	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Cacillatory motion + Oscillating	V60 V61 V62	[see also Stability] Oscillatory coefficients O. aerodynamic derivatives Measurement of reaction [Measurement by] Logarithmic circuit
U87 U88 U89 U90 U91	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a.	V60 V61	[see also Stability] Oscillatory coefficients O. aerodynamic derivatives Measurement of reaction [Measurement by] Logarithmic circuit Aerodynamic stiffness
U87 U88 U89 U90	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Oscillatory motion + Oscillating [see also] W50	V60 V61 V62	[see also Stability] Oscillatory coefficients O. aerodynamic derivatives Measurement of reaction [Measurement by] Logarithmic circuit
U87 U88 U89 U90 U91	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Oscillatory motion + Oscillating [see also] %50 Electrical excitation + Electrically excited oscillations	v60 v61 v62 v63	[see also Stability] Oscillatory coefficients O. aerodynamic derivatives Measurement of reaction [Measurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d.
U87 U88 U89 U90 U91 U92	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Oscillatory motion + Oscillating [see also] W50 Electrical excitation + Electrically excited oscillations [at] Phase resonance	v60 v61 v62 v63	[see also Stability] Oscillatory coefficients O. aerodynamic derivatives Measurement of reaction [Neasurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom
U87 U88 U89 U90 U91 U92 U93 U94	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Oscillatory motion + Oscillating [see also] \(\) 50 Electrical excitation + Electrically excited oscillations [at] Fhase resonance Electrical self-excitation	V60 V61 V62 V63 V64 V65 V66	[see also Stability] Oscillatory coefficients O. aerodynamic derivatives Measurement of reaction [Measurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency
U87 U88 U89 U90 U91 U92 U93 U94 U95	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Coscillatory motion + Oscillating [see also] W50 Electrical excitation + Electrically excited oscillations [at] Phase resonance Electrical self-excitation [with] Coupled freedoms	v60 v61 v62 v63 v64 v65 v66 v67	[see also Stability] Oscillatory coefficients O. aerodynamic derivatives Measurement of reaction [Neasurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency Frequency parameters
U87 U88 U89 U90 U91 U92 U93 U94 U95 U96	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Coscillatory motion + Oscillating [see also] W50 Electrical excitation + Electrically excited oscillations [at] Phase resonance Electrical self-excitation [with] Coupled freedoms Elastic excitation	V60 V61 V62 V63 V64 V65 V66 V67 V67	[see also Stability] Oscillatory coefficients O. aerodynamic derivatives Measurement of reaction [Neasurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency Frequency parameters F.p. less than .02
U87 U88 U89 U90 U91 U92 U93 U94 U95 U96 U97	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Oscillatory motion + Oscillating [see also] %50 Electrical excitation + Electrically excited oscillations [at] Phase resonance Electrical self-excitation [with] Coupled freedoms Elastic excitation Slowly propagating c.	v60 v61 v62 v63 v64 v65 v66 v67	See also Stability Oscillatory coefficients O. aerodynamic derivatives Measurement of reaction [Measurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency Frequency parameters F.D. less than .02 F. measurement
U87 U88 U89 U90 U91 U92 U93 U94 U95 U96 U97 U98	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Oscillatory motion + Oscillating [see also] W50 Electrical excitation + Electrically excited oscillations [at] Fhase resonance Electrical self-excitation [with] Coupled freedoms Elastic excitation Slowly propagating o. Fast propagating o.	V60 V61 V62 V63 V64 V65 V66 V67 V67 V68	[see also Stability] Oscillatory coefficients Oscillatory coefficients Oscillatory coefficients Oscillatory coefficients Measurement of reaction [Measurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency Frequency parameters Fsp. less than .02 F. measurement [Phase]
U87 U88 U89 U90 U91 U92 U93 U94 U95 U97 U97 U98 U99	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Coscillatory motion + Oscillating [see also] W50 Electrical excitation + Electrically excited oscillations [at] Phase resonance Electrical self-excitation [with] Coupled freedoms Electric excitation Slowly propagating o. Fast propagating o. Decaying o.	V60 V61 V62 V63 V64 V65 V66 V67 V67	[see also Stability] Oscillatory coefficients O. aerodynamic derivatives Measurement of reaction [Measurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency Frequency parameters F.p. less than .02 F. measurement [Phase] P. measurement
U87 U88 U89 U90 U91 U92 U93 U94 U95 U96 U97 U98	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Oscillatory motion + Oscillating [see also] W50 Electrical excitation + Electrically excited oscillations [at] Fhase resonance Electrical self-excitation [with] Coupled freedoms Elastic excitation Slowly propagating o. Fast propagating o.	V60 V61 V62 V63 V64 V65 V66 V67 V67 V68	See also Stability Oscillatory coefficients O. aerodynamic derivatives Measurement of reaction [Neasurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency Frequency parameters F.p. less than .02 F. measurement [Phase] P. measurement [Amplitude]
U87 U88 U89 U90 U91 U92 U93 U94 U95 U96 U97 U99 U99 U99	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Coscillatory motion + Oscillating [see also] W50 Electrical excitation + Electrically excited oscillations [at] Phase resonance Electrical self-excitation [with] Coupled freedoms Electric excitation Slowly propagating o. Fast propagating o. Fast propagating o. Sinuscidal sinking o.	V60 V61 V62 V63 V64 V65 V66 V67 V67 V68 V69	See also Stability Oscillatory coefficients O. aerodynamic derivatives Measurement of reaction [Measurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Matural frequencies Reduced frequency Frequency parameters F.p. less than .02 F. measurement [Phase] P. measurement [Amplitude] A. measurement
U87 U88 U89 U90 U91 U92 U93 U94 U95 U96 U97 U98 U97 U98	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Oscillatory motion + Oscillating [see also] %50 Electrical excitation + Electrically excited oscillations [at] Phase resonance Electrical self-excitation [with] Coupled freedoms Elastic excitation Slowly propagating o. Fact propagating o. Decaying o. Sinusoidal sinking o. Continuous o.	V60 V61 V62 V63 V64 V65 V66 V67 V68 V69 V70	See also Stability Oscillatory coefficients O. aerodynamic derivatives Measurement of reaction [Neasurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency Frequency parameters F.p. less than .02 F. measurement [Phase] P. measurement [Amplitude]
U87 U88 U89 U90 U91 U92 U93 U94 U95 U96 U97 U99 V1 V2 V3 V4 V5	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Coscillatory motion + Oscillating [See also] W50 Electrical excitation + Electrically excited oscillations [at] Phase resonance Electrical self-excitation [with] Coupled freedoms Electrical self-excitation Slowly propagating o. Fast propagating o. Fast propagating o. Jecaying o. Sinuscidal sinking o. Continuous o. Oscillating harmonically Sinuscidal vertical o. Asymmetrical o.	V60 V61 V62 V63 V64 V65 V66 V67 V67 V68 V69	Gee also Stability Oscillatory coefficients Oscillatory coefficients Oscillatory coefficients Measurement of reaction [Measurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency Frequency parameters F.p. less than .02 F. measurement [Phase] P. measurement [Amplitude] A. measurement Excitation
U87 U88 U89 U90 U91 U92 U93 U94 U95 U96 U97 U96 U97 U98 V2 V2 V3 V4 V5 V6	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Coscillatory motion + Oscillating [see also] W50 Electrical excitation + Electrically excited oscillations [at] Phase resonance Electrical self-excitation [with] Coupled freedoms Elastic excitation Slowly propagating c. Fact propagating c. Fact propagating c. Decaying c. Sinuscidal sinking c. Continuous c. Oscillating harmonically Sinuscidal vertical c. Asymmetrical c. Flapping c.	V60 V61 V62 V63 V64 V65 V66 V67 V68 V69 V70 V71 V72	See also Stability Oscillatory coefficients O. aerodynamic derivatives Measurement of reaction [Measurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency Frequency parameters F.p. less than .02 F. measurement [Phase] P. measurement [Amplitude] A. measurement Excitation Damping Internal d. Structural d.
U87 U88 U89 U90 U91 U92 U93 U94 U95 U96 U97 U98 U99 V19 V2 V3 V4 V5 V6 V7	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Oscillatory motion + Oscillating [see also] #50 Electrical excitation + Electrically excited oscillations [at] Phase resonance Electrical self-excitation [with] Coupled freedoms Elastic excitation Slowly propagating o. Fact propagating o. Decaying o. Sinuscidal sinking o. Continuous o. Oscillating harmonically Sinuscidal vertical o. Asymmetrical o. Flapping o. Torsional o.	V60 V61 V62 V63 V64 V65 V66 V67 V68 V69 V70 V71 V72 V73 V75	Gee also Stability Oscillatory coefficients Oscillatory coefficients Oscillatory coefficients Measurement of reaction [Measurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency Frequency parameters F.D. less than .02 F. measurement [Amplitude] A. measurement Excitation Damping Internal d. Structural d. Aerodynamic d.
U87 U88 U89 U90 U91 U92 U93 U94 U95 U96 U97 U98 U99 V1 V2 V3 V4 V5 V6 V7 V8	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Coscillatory motion + Oscillating [see also] W50 Electrical excitation + Electrically excited oscillations [at] Phase resonance Electrical self-excitation [with] Coupled freedoms Elestic excitation Slowly propagating o. Fast propagating o. Decaying o. Sinusoidal sinking o. Continuous o. Oscillating harmonically Sinusoidal vertical o. Asymmetrical o. Flapping o. Torsional o. Bending o.	V60 V61 V62 V63 V64 V65 V66 V67 V68 V69 V70 V71 V72 V73 V74 V75 V76	Gee also Stability Oscillatory coefficients Oscillatory coefficients Oscillatory coefficients Oscillatory coefficients Measurement of reaction [Neasurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency Frequency parameters F.p. less than .02 F. measurement [Phase] P. measurement [Amplitude] A. measurement Excitation Damping Internal d. Structural d. Aerodynamic d. Phugoid d.
U87 U88 U89 U90 U91 U92 U93 U94 U95 U96 U97 U98 U99 V1 V2 V3 V4 V5 V6 V7 V8 V9	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Coscillatory motion + Oscillating [See also] W50 Electrical excitation + Electrically excited oscillations [at] Phase resonance Electrical self-excitation [with] Coupled freedoms Electrical self-excitation Slowly propagating o. Fast propagating o. Fast propagating o. Decaying o. Sinusoidal sinking o. Continuous o. Oscillating harmonically Sinusoidal vertical o. Asymmetrical o. Flapping o. Torsional o. Bending o. Sinusoidal o. displacement	V60 V61 V62 V63 V64 V65 V66 V67 V68 V69 V70 V71 V72 V73 V74 V75 V76 V77	See also Stability Oscillatory coefficients O. aerodynamic derivatives Measurement of reaction [Neasurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency Frequency parameters F.p. less than .02 F. measurement [Phase] P. measurement [Amplitude] A. measurement Excitation Damping Internal d. Structural d. Aerodynamic d. Phugoid d. A.d. coefficient
U87 U88 U89 U90 U91 U92 U93 U94 U95 U96 U97 U99 V10 V2 V3 V4 V5 V6 V7 V8 V9 V10	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Oscillatory motion + Oscillating [see also] W50 Electrical excitation + Electrically excited oscillations [at] Fhase resonance Electrical self-excitation [with] Coupled freedoms Elastic excitation Slowly propagating o. Fast propagating o. Decaying o. Sinuscidal sinking o. Continuous o. Oscillating harmonically Sinuscidal vertical o. Asymmetrical o. Flapping o. Torsional o. Bending o. Sinuscidal o. displacement Neutral o.	V60 V61 V62 V63 V64 V65 V67 V67 V68 V69 V70 V71 V72 V73 V74 V75 V76 V77	See also Stability Oscillatory coefficients O. aerodynamic derivatives Measurement of reaction [Neasurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency Frequency parameters F.p. less than .02 F. measurement [Phase] P. measurement [Amplitude] A. measurement Excitation Damping Internal d. Structural d. Aerodynamic d. Phugoid d. A.d. coefficient A.d. derivatives + D.d.
U87 U88 U89 U90 U91 U92 U93 U94 U95 U96 U97 U98 U99 V1 V2 V3 V4 V5 V7 V8 V9 V10 V11	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Oscillatory motion + Oscillating [see also] #50 Electrical excitation + Electrically excited oscillations [at] Phase resonance Electrical self-excitation [with] Coupled freedoms Elastic excitation Slowly propagating o. Fast propagating o. Jecaying o. Sinuscidal sinking o. Continuous o. Oscillating harmonically Sinuscidal vertical o. Asymmetrical o. Flapping o. Torsional o. Bending o. Sinuscidal o. displacement Neutral o. Free o. + Free vibration	V60 V61 V62 V63 V64 V65 V66 V67 V68 V69 V70 V71 V72 V73 V74 V75 V76 V77	Gee also Stability Oscillatory coefficients O. aerodynamic derivatives Measurement of reaction [Measurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency Frequency parameters F.p. less than .02 F. measurement [Amplitude] A. measurement Excitation Damping Internal d. Structural d. Aerodynamic d. Phugoid d. A.d. coefficient A.d. derivatives + D.d. Buzz
U87 U88 U89 U90 U91 U92 U93 U94 U95 U96 U97 U98 U99 V11 V2 V3 V4 V5 V6 V7 V8 V9 V11 V11 V12	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Oscillatory motion + Oscillating [see also] W50 Electrical excitation + Electrically excited oscillations [at] Fhase resonance Electrical self-excitation [with] Coupled freedoms Elastic excitation Slowly propagating o. Fast propagating o. Decaying o. Sinuscidal sinking o. Continuous o. Oscillating harmonically Sinuscidal vertical o. Asymmetrical o. Flapping o. Torsional o. Bending o. Sinuscidal o. displacement Neutral o.	V60 V61 V62 V63 V64 V65 V66 V67 V68 V69 V70 V71 V72 V73 V74 V75 V76 V77 V76 V77	Gee also Stability Oscillatory coefficients O. aerodynamic derivatives Measurement of reaction [Neasurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency Frequency parameters F.p. less than .02 F. measurement [Phase] P. measurement [Amplitude] A. measurement Excitation Damping Internal d. Structural d. Aerodynamic d. Phugoid d. A.d. coefficient A.d. derivatives + D.d. Buzz [Divergence]
U87 U88 U89 U90 U91 U92 U93 U94 U95 U96 U97 U98 U99 V1 V2 V3 V4 V5 V7 V8 V9 V10 V11	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Oscillatory motion + Oscillating [see also] W50 Electrical excitation + Electrically excited oscillations [at] Phase resonance Electrical self-excitation [with] Coupled freedoms Elestic excitation Slowly propagating o. Fast propagating o. Decaying o. Sinusoidal sinking o. Continuous o. Oscillating harmonically Sinusoidal vertical o. Asymmetrical o. Flapping o. Torsional o. Bending o. Sinusoidal o. displacement Neutral o. Free o. + Free vibration Angle of attack o.	V60 V61 V62 V63 V64 V65 V66 V67 V68 V69 V70 V71 V72 V73 V74 V75 V76 V77 V78 V79	Gee also Stability Oscillatory coefficients O. aerodynamic derivatives Measurement of reaction [Measurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency Frequency parameters F.p. less than .02 F. measurement [Amplitude] A. measurement Excitation Damping Internal d. Structural d. Aerodynamic d. Phugoid d. A.d. coefficient A.d. derivatives + D.d. Buzz
U87 U88 U89 U90 U91 U92 U93 U94 U95 U96 U97 U98 V9 V1 V2 V3 V4 V5 V6 V7 V8 V9 V10 V11 V12 V13	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Coscillatory motion + Oscillating [See also] W50 Electrical excitation + Electrically excited oscillations [at] Phase resonance Electrical self-excitation [with] Coupled freedoms Electrical self-excitation Slowly propagating o. Fast propagating o. Decaying o. Sinusoidal sinking o. Continuous o. Oscillating harmonically Sinusoidal vertical o. Asymmetrical o. Flapping o. Torsional o. Bending o. Sinusoidal o. displacement Neutral o. Free o. + Free vibration Angle of attack o. Resonant vibration	V60 V61 V62 V63 V64 V65 V66 V67 V68 V69 V70 V71 V72 V73 V74 V75 V76 V77 V76 V77	Gee also Stability Oscillatory coefficients Oscillatory coefficients Oscillatory coefficients Oscillatory coefficients Measurement of reaction [Neasurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency Frequency parameters F.p. less than .02 F. measurement [Phase] P. measurement Excitation Damping Internal d. Aerodynamic d. Phugoid d. A.d. coefficient A.d. derivatives + D.d. Buzz [Divergence] D. dynamic pressure
U87 U88 U89 U90 U91 U92 U93 U94 U95 U97 U98 U97 U98 V97 V8 V7 V8 V7 V8 V7 V8 V9 V10 V11 V11 V11 V11 V11 V11 V11 V11 V11	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Coscillatory motion + Oscillating [See also] W50 Electrical excitation + Electrically excited oscillations [at] Phase resonance Electrical self-excitation [with] Coupled freedoms Electric excitation Slowly propagating o. Fast propagating o. Decaying o. Sinusoidal sinking o. Continuous o. Oscillating harmonically Sinusoidal vertical o. Asymmetrical o. Flapping o. Torsional o. Bending o. Sinusoidal o. displacement Neutral o. Free o. + Free vibration Angle of attack o. Resonant vibration Inexorable forcing External rigid drive Internal rigid drive	V60 V61 V62 V63 V64 V65 V66 V67 V68 V69 V70 V71 V72 V73 V74 V75 V76 V77 V78 V79 V81 V82	Gee also Stability Oscillatory coefficients Oscillatory coefficients Oscillatory coefficients Measurement of reaction [Measurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency Frequency parameters F.p. less than .02 F. measurement [Amplitude] A. measurement Excitation Damping Internal d. Structural d. Aerodynamic d. Phugoid d. A.d. coefficient A.d. derivatives + D.d. Buzz [Divergence] D. dynamic pressure Buffeting B. boundary [Research]
U87 U88 U89 U90 U91 U92 U93 U94 U95 U96 U97 U98 V9 V10 V11 V12 V13 V14 V15	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Coscillatory motion + Oscillating [See also] W50 Electrical excitation + Electrically excited oscillations [at] Phase resonance Electrical self-excitation [with] Coupled freedoms Electrical self-excitation Slowly propagating o. Fast propagating o. Decaying o. Sinusoidal sinking o. Continuous o. Oscillating harmonically Sinusoidal vertical o. Asymmetrical o. Flapping o. Torsional o. Bending o. Sinusoidal o. displacement Neutral o. Free o. + Free vibration Angle of attack o. Resonant vibration Inexorable forcing External rigid drive Internal rigid drive [Flutter]	V60 V61 V62 V63 V64 V65 V66 V67 V68 V69 V70 V71 V72 V73 V74 V75 V76 V77 V78 V79 V81 V82	Gee also Stability Oscillatory coefficients O. aerodynamic derivatives Measurement of reaction [Neasurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency Frequency parameters F.p. less than .02 F. measurement [Phase] P. measurement Excitation Damping Internal d. Structural d. Aerodynamic d. Phugoid d. A.d. coefficient A.d. derivatives + D.d. Buzz [Divergence] D. dynamic pressure Buffeting B. boundary [Research] Systematic procedure
U87 U88 U89 U90 U91 U92 U93 U94 U95 U97 U98 U99 V1 V13 V4 V5 V7 V8 V9 V10 V11 V12 V13 V14 V15 V16 V17	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Oscillatory motion + Oscillating [see also] W50 Electrical excitation + Electrically excited oscillations [at] Phase resonance Electrical self-excitation [with] Coupled freedoms Elastic excitation Slowly propagating o. Decaying o. Sinuscidal sinking o. Continuous o. Oscillating harmonically Sinuscidal vertical o. Asymmetrical o. Flapping o. Torsional o. Bending o. Sinuscidal o. displacement Neutral o. Free o. + Free vibration Angle of attack o. Resonant vibration Inexorable forcing External rigid drive [Flutter] F. oscillations + Flutter	V60 V61 V62 V63 V64 V65 V66 V67 V68 V69 V70 V71 V72 V73 V74 V75 V76 V77 V78 V79 V81 V82 V83 V84 V85	Gee also Stability Oscillatory coefficients Oscillatory coefficients Oscillatory coefficients Measurement of reaction [Measurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency Frequency parameters F.p. less than .02 F. measurement [Phase] P. measurement [Amplitude] A. measurement Excitation Damping Internal d. Structural d. Aerodynamic d. Phugoid d. A.d. coefficient A.d. derivatives + D.d. Buzz [Divergence] D. dynamic pressure Buffeting B. boundary [Research] Systematic procedure Kernel function
U87 U88 U89 U90 U91 U92 U93 U95 U95 U97 U98 U97 V98 V97 V8 V7 V8 V9 V10 V11 V12 V15 V16 V17 V16 V17 V18	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Oscillatory motion + Oscillating [see also] #50 Electrical excitation + Electrically excited oscillations [at] Phase resonance Electrical self-excitation [with] Coupled freedoms Elastic excitation Slowly propagating c. Fast propagating c. Fast propagating c. Oscillating harmonically Sinuscidal sinking o. Continuous c. Oscillating harmonically Sinuscidal vertical c. Asymmetrical c. Flapping c. Torsical c. Bending c. Sinuscidal c. displacement Neutral c. Free c. + Free vibration Angle of attack c. Resonant vibration Inexorable forcing External rigid drive [Internal rigid drive [Flutter] F. oscillations + Flutter Eending type f.	V60 V61 V62 V63 V64 V65 V66 V67 V68 V69 V70 V71 V72 V73 V75 V76 V77 V78 V79 V81 V82 V83 V84 V85 V86	Gee also Stability Oscillatory coefficients O. aerodynamic derivatives Measurement of reaction [Measurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency Frequency parameters F.p. less than .02 F. measurement [Amplitude] A. measurement Excitation Damping Internal d. Structural d. Aerodynamic d. Phugoid d. A.d. coefficient A.d. derivatives + D.d. Buzz [Divergence] D. dynamic pressure Buffeting B. boundary [Research] Systematic procedure Kernel function Piston theory
U87 U88 U89 U90 U91 U92 U93 U94 U95 U97 U98 U99 V1 V13 V4 V5 V7 V8 V9 V10 V11 V12 V13 V14 V15 V16 V17	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Oscillatory motion + Oscillating [See also] W50 Electrical excitation + Electrically excited oscillations [at] Phase resonance Electrical self-excitation [with] Coupled freedoms Electric excitation Slowly propagating o. Fast propagating o. Decaying o. Sinusoidal sinking o. Continuous o. Oscillating harmonically Sinusoidal vertical o. Asymmetrical o. Flapping o. Torsional o. Bending o. Sinusoidal o. displacement Neutral o. Free o. + Free vibration Angle of attack o. Resonant vibration Inexorable forcing External rigid drive Internal rigid drive [Flutter] F. oscillations + Flutter Bending type f. High speed f.	V60 V61 V62 V63 V64 V65 V66 V67 V68 V69 V70 V71 V72 V73 V74 V75 V76 V77 V78 V79 V81 V82 V83 V84 V85 V86 V87	Gee also Stability Oscillatory coefficients Oscillatory coefficients Oscillatory coefficients Measurement of reaction [Measurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency Frequency parameters F.p. less than .02 F. measurement [Amplitude] A. measurement Excitation Damping Internal d. Structural d. Aerodynamic d. Phugoid d. A.d. coefficient A.d. derivatives + D.d. Buzz [Divergence] D. dynamic pressure Buffeting B. boundary [Research] Systematic procedure Kernel function Piston theory Reciprocal relations [method]
U87 U88 U89 U90 U91 U92 U93 U95 U95 U97 U98 U97 V98 V97 V8 V7 V8 V9 V10 V11 V12 V15 V16 V17 V16 V17 V18	Viscous effects Shear Coefficient of v. [Aeroelasticity] Static a. Oscillatory motion + Oscillating [see also] #50 Electrical excitation + Electrically excited oscillations [at] Phase resonance Electrical self-excitation [with] Coupled freedoms Elastic excitation Slowly propagating c. Fast propagating c. Fast propagating c. Oscillating harmonically Sinuscidal sinking o. Continuous c. Oscillating harmonically Sinuscidal vertical c. Asymmetrical c. Flapping c. Torsical c. Bending c. Sinuscidal c. displacement Neutral c. Free c. + Free vibration Angle of attack c. Resonant vibration Inexorable forcing External rigid drive [Internal rigid drive [Flutter] F. oscillations + Flutter Eending type f.	V60 V61 V62 V63 V64 V65 V66 V67 V68 V69 V70 V71 V72 V73 V75 V76 V77 V78 V79 V81 V82 V83 V84 V85 V86	Gee also Stability Oscillatory coefficients O. aerodynamic derivatives Measurement of reaction [Measurement by] Logarithmic circuit Aerodynamic stiffness derivatives + S.d. Degrees of freedom [Frequency] Natural frequencies Reduced frequency Frequency parameters F.p. less than .02 F. measurement [Amplitude] A. measurement Excitation Damping Internal d. Structural d. Aerodynamic d. Phugoid d. A.d. coefficient A.d. derivatives + D.d. Buzz [Divergence] D. dynamic pressure Buffeting B. boundary [Research] Systematic procedure Kernel function Piston theory

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9% RAE 101 streamwise s.
                                                         W47
V90
               Dampometer
                                                                       5% RAE 101 streamwise s.
                Displacement pickups
                                                         W48
V91
                Force pickups
                                                         W49
                                                                       NACA 65A 004 s.
V92
                  Pressure pickups
V93
                                                                Mechanics
         [Aerodynamic reference parameters]
[Angle of incidence] Q + Angle of attack + Arbitrary a.of.a +
                                                         W50
                                                                  Motion
                                                         W51
                                                                     Transient m.
V94
                                                         W52
                                                                     Translation
                                                         753
                                                                     Rotation
           Incidence
             A. of a. correction + A.of i.c.
                                                                     Rotational modes
                                                         W54
V95
                                                                   [Oscillation, vibration see U91] Komentum deficiency profile
              Zero a. of a.
V96
                                                         ₩56
                Zero model incidence
V97
                                                         W57
                                                                   Gravity
              Moderate a. of a.
งจ์ล
             Angle of a. range 0 - 65°
Incidence range 0 - 10°
A. of a. range 4 - 18
                                                          ₩57a
                                                                   Influence of gravity
V98a
                                                         W58
                                                                   Forces
V99
                                                                     Load + Applied f.
                                                         ₩59
V99a
Wi
              High a. of a. + High incidences +
                                                                     Random p. loading
                                                          W60
              Large a. o. a. see also F54
                                                          W61
                                                                     Compression + C. load
₩2
              Uniform a. of a.
                                                                     Tension + T. load
                                                          362
              Increasing i.
W3
                                                                     Bending + B. load
                                                          W63
                                                                     Thermal stress
              A. of a. distribution
                                                          W64
W4
                                                          W65
                                                                       Elastic range
            Vertical position [of tail]
₩5
                                                                       Plastic range
                                                          W66
W6
           Planform
                                                          W67
                                                                     Maximum stresses
              Sweep angle + Sweep
                                                                     Loads, Longitudinal
Loads, Concentric axial
Loads, One-dimensional
W7
                                                          ₩68
                Angle of sweep range of
W8
                                                          W69
                0-75 degrees
              Angle of sweepback

S. a. 55

S. a. 60

79.5 s.
                                                          W70
WO
                                                                   [Stress-strain phenomena]
W9a
                                                                     Weakening
W10
                                                                     Compression buckling
                                                          W72
W11
                                                                     Post-buckling
                                                          W73
            Aspect ratio
W12
                                                          W74
                                                                      Torque
W13
              Zero a.r.
                                                                     Rotational stiffness
                                                          W75
₩13a
              Aspect ratio
                                                                      Bending stiffness
                                                          ₩76
                            2
W13b
                                                                      Torsional stiffness
                                                          W77
                            2.823
₩130
                                                                      Torsional rigidity
                                                          W78
                01
                             2.828
W13d
                                                                      Modules of elasticity
                                                          W79
                89
                             6.1
₩130
                                                                      Time dependent
                                                          W80
W1A
            Taper ratio
                                                                     Mode shapes
                                                          W81
₩14a
              T.r. .333
                                                                   Stress distribution
                                                          W82
            Thickness
₩15
                                                                      S. frequency d.
                                                          WAK
            Thickness/chord ratio
W16
                                                                      Spanload d.
                                                          W84
              T.c.ratio about 5%
₩16c
                                                                      Steady state d.
                                                          W85
                        6%
16%
                                                                      [Nethod of determining s.d.]
W16d
                  19
W16e
                                                                        Weissinger method
                                                          WA6
                  19
W16f
                        24%
                                                                        Downwash matrix
                                                          W87
            Thickness distribution
W17
                                                                        Elasticity matrices
                                                          WAA
₩18
            Twist
                                                                        Aerodynamic twist matrices
                                                          W89
W19
            Dihedral angle
                                                                        Structural twist matrices
              A. of d. range 0 - 30 degrees 4.5° d.a.
                                                          W90
₩20
                                                          W91
                                                                        Plastic analysis
₩20a
                                                                        Stress power spectra
                                                          W92
            [Span]
                                                                           [Prediction by]
              Spanwise
W21
                                                          W93
                                                                           Model response
₩22
              Finite s.
                                                                             Statistical superposition
                                                          W94
₩23
              Half-s. model
                                                                              Powell theory
                                                          W95
W24
              Part s.
                                                                        Test specimens
                                                          W96
W25
            Body size
                                                                        Instrumentation
                                                          W97
W26
               B. diameter
                                                                        Material control
                                                          ₩98
                B.d. ratio
 W27
                                                                          Coupon test
                                                          ₩99
             Chord
                                                                    Magneto-fluid mechanics
                                                          Х1
              Chordwise
 ₩28
                                                                      Electric body force
                                                          Х2
 ₩29
                C. coordinate
                                                                      Electro-magnetical induced
                                                          Х3
 W 30
              Model c.
                                                                      forces
 ₩31
              C. ratio
                                                                    [Thermal phenomena]
 W32
            Camber
                                                           X4
                                                                      [Heat equation] Q
 ₩33
               Conical c.
                                                                      Thermal equilibrium
                                                           Х5
 W34
               C. correction
                                                           Х6
                                                                      Temperature
 ₩35
               C. distribution
                                                                        Ambient t. + Local stream t.
                                                           Х7
             Centre of gravity location
 W36
                                                                        Flame t.
             Section
                                                           х9
                                                                        Room t.
               Aerodynamic design
 ₩38
                                                           X10
                                                                        Transient t.
               Cross section + Body c.s.
                                                                        Constant t. + T. uniform
 ₩39
                                                           X11
               + Arbitrary c.s.
                                                           X12
                                                                        Recovery t.
                Percentage of solid c.s.
 W40
                                                           X13
                                                                        Elevated t.
 ₩41
               Peaky s.
                                                           X14
                                                                        Maximum t.
 W42
               Rooftop s.
                                                                        Reduced t.
 W43
               Midspan s.
                                                                         T. dependent values
 W44
               Sectional shape + S. shape
                                                           X16
                                                                         T. inversion
                                                           X17
                S. thickness
 W45
                                                           X18
                                                                         T. variation
               R.A.E. 101 s.
 W46
                                                                           Unit triangle v.
               10% RAE 101 s.
                                                           X19
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X20
                T. distribution + T.profiles + T. history
                                                                           Quartz protection shield
                                                        X95
                                                                         Heat sink
 X21
                  Transient t.d.
                                                        X96
                                                                           Thicker outer skin
 X22
                One-dimensional t. gradient
                                                        X97
                                                                             Copper
 X23
                Zero t. gradient
T. gradients, Transverse
                                                        YOA
                                                                             Graphite
 X24
                                                        X99
                                                                             Molybdenum
              T. Linear gradient
Equilibrium t.
 X25
                                                                             Tungsten
 x26
                                                                       [Heating]
 X27
              T. measurement
                                                        ¥2
                                                                         Convection h. + Convective
 X28
                Spectrum line reversal
                                                                         h. + Convection + C. of
 X29
                  Double light beam
                                                                         heat + Convective heat
 X30
                  Single light beam
                                                                         transfer
 X31
                  Carbon arc
                                                                           Forced c.h.
 X32
                Sodium line reversal + S.l.r. method Y4
                                                                           Convective heat transfer
 X33
                  Oscillograph
                                                                           rate
 X34
                  Interference filters
                                                                         Frictional h.
                  Sodium chloride + Salt
 X35
                                                                           Aerodynamic h.
 X36
                  Sodium iodide
                                                        ¥7
                                                                             A. laminar h.
 X37
              Specific heat
                                                                             A.h. rate + H.r.
 X38
                S.h. ratio parameters 1.10-1.67
                                                       ¥9
                                                                               Laminar a.h.r.
 X40
                S.h. variation
                                                       Y9a
                                                                                 Green function
 XA1
              Conductivity + Thermal c.
                                                                                 [integration with]
 XA2
                                                       Y10
                Variable c.
                                                                         H. at edge
 X43
                Coefficient of t.c.
                                                                           Local h. rate
                                                       Y11
 XAA
              Emissivity
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Y95 Y96 Y97 Y98 Y99 Z1 Z2 Z3 Z4 Z5 Z6 Z7 Z8 Z9 Z10 Z11 Z12 Z13 Z14 Z15 Z16 Z17 Z18 Z19 Z10 Z12 Z13 Z14 Z15 Z16 Z17 Z18	Experimental data + E. results + Results + Test data Wind tunnel data Tables of results Flight test data [Facilities] Q Wind tunnel W.t. tests + W.t. experiment + W.t. investigation W.t. determination + W.t. measurement Transonic w.t. T.w.t. results N.P.L: 18 x 14 w.t. Supersonic w.t. Hypersonic w.t. + Hypervelocity w.t. Hypersonic shock t. + H.s.tube H.s.tunnel tests Driven [by] Ignited mixture Stoichiometric m. Oxygen and hydrogen m. M. with excess of helium Hypersonic gun tunnel Shock tube + Tube S.t. experiment + S. tunnel experiment Straight through s. tunnel Reflected s. tunnel [driven by] Unheated hydrogen driving air	264 265 266 267 268 269 270 271 272 273 274 275 276 277 278 279 280 281 282 283 284	Model dimensions + k. size [Nounted off-centre] Q [Location in tunnel] End of tube Open area of walls Near model base Along model [Characteristics] Cross sectional size Cross sectional size Cross sectional shape Length of tube Tunnel weight Tunnel speed Power requirements Running time + Flow duration [Short running time] Q Experimental testing time + Testing time Theoretical t.t. + Ideal time Actual time Tunnel wall pressures Bursting pressure ratio [with] Removal of damping plate Reservoir temperature [W.T. interference etc. see Walls] Corrections [Visualisation methods] Oil flow pattern Surface O.f.p.
Y95 Y96 Y97 Y98 Y99 Z1 Z2 Z3 Z4 Z5 Z6 Z7 Z8 Z9 Z10 Z11 Z12 Z13 Z14 Z15 Z16 Z17 Z18 Z19 Z20 Z21	Experimental data + E. results + Results + Test data Wind tunnel data Tables of results Flight test data [Facilities] Q Wind tunnel W.t. tests + W.t. experiment + W.t. investigation W.t. determination + W.t. measurement Transonic w.t. T.w.t. results N.P.L: 18 x 14 w.t. Supersonic w.t. Hypersonic shock t. + H.s.tube H.s.tunnel tests Driven [by] Ignited mixture Stoichiometric m. Oxygen and hydrogen m. M. with excess of helium Hypersonic gun tunnel Shock tube + Tube S.t. experiment + S. tunnel experiment Straight through s. tunnel Reflected s. tunnel [driven by] Unheated hydrogen driving air Constant diameter s.t.	264 265 266 267 268 269 270 271 272 273 274 275 276 277 278 279 280 281 282 283 284 285	Model dimensions + M. size [Nounted off-centre] Q [Location in tunnel] End of tube Open area of walls Near model base Along model [Characteristics] Cross sectional size Cross sectional shape Length of tube Tunnel weight Tunnel speed Power requirements Running time + Flow duration [Short running time] Q Experimental testing time + Testing time Theoretical t.t. + Ideal time Actual time Tunnel wall pressures Bursting pressure ratio [with] Removal of damping plate Reservoir temperature [W.T. interference etc. see Walls] Corrections [Visualisation methods] Oil flow study Oil flow pattern Surface o.f.p. Photographs + Photographio
Y95 Y96 Y97 Y98 Y99 Z1 Z2 Z3 Z4 Z5 Z6 Z7 Z8 Z9 Z10 Z11 Z12 Z13 Z14 Z15 Z16 Z17 Z18 Z19 Z10 Z11 Z12 Z13 Z14 Z15 Z14 Z15 Z16 Z17 Z18 Z19 Z20 Z21	Experimental data + E. results + Results + Test data Wind tunnel data Tables of results Flight test data [Facilities] Q Wind tunnel W.t. tests + W.t. experiment + W.t. investigation W.t. determination + W.t. measurement Transonic w.t. T.w.t. results N.P.L: 18 x 14 w.t. Supersonic w.t. Hypersonic shock t. + H.s.tube H.s.tunnel tests Driven [by] Ignited mixture Stoichiometric m. Oxygen and hydrogen m. M. with excess of helium Hypersonic gun tunnel Shock tube + Tube S.t. experiment + S. tunnel experiment Straight through s. tunnel Reflected s. tunnel [driven by] Unheated hydrogen driving air Constant diameter s.t. [High enthalpy w.t.] Q	264 265 266 267 268 269 270 271 272 273 274 275 276 277 278 279 280 281 282 283 284 285 286 287 288 289	Model dimensions + k. size [Nounted off-centre] Q [Location in tunnel] End of tube Open area of walls Near model base Along model [Characteristics] Cross sectional size Cross sectional size Cross sectional shape Length of tube Tunnel weight Tunnel speed Power requirements Running time + Flow duration [Short running time] Q Experimental testing time + Testing time Theoretical t.t. + Ideal time Actual time Tunnel wall pressures Bursting pressure ratio [with] Removal of damping plate Reservoir temperature [W.T. interference etc. see Walls] Corrections [Visualisation methods] Oil flow study Oil flow pattern Surface o.f.p. Photographs + Photographio record + Photographic study
Y95 Y96 Y97 Y98 Y99 Z1 Z2 Z3 Z4 Z5 Z6 Z7 Z8 Z9 Z10 Z11 Z12 Z13 Z14 Z15 Z16 Z17 Z18 Z19 Z10 Z11 Z12 Z13 Z14 Z15 Z17 Z18 Z17 Z18 Z19 Z20 Z21 Z22 Z23 Z24	Experimental data + E. results + Results + Test data Wind tunnel data Tables of results Flight test data [Facilities] Q Wind tunnel W.t. tests + W.t. experiment + W.t. investigation W.t. determination + W.t. measurement Transonic w.t. T.w.t. results N.P.L: 18 x 14 w.t. Supersonic w.t. Hypersonic w.t. + Hypervelocity w.t. Hypersonic shock t. + H.s.tube H.s.tunnel tests Driven [by] Ignited mixture Stoichiometric m. Oxygen and hydrogen m. M. with excess of helium Hypersonic gun tunnel Shock tube + Tube S.t. experiment + S. tunnel experiment Straight through s. tunnel Reflected s. tunnel [driven by] Unheated hydrogen driving air Constant diameter s.t. [High enthalpy w.t.] Q Elowdown tunnel tests	264 265 266 267 268 269 270 271 272 273 274 275 276 277 278 279 280 281 282 283 284 285 285 286 287 288 289 290	Model dimensions + k. size [Nounted off-centre] Q [Location in tunnel] End of tube Open area of walls Near model base Along model [Characteristics] Cross sectional size Cross sectional size Cross sectional shape Length of tube Tunnel weight Tunnel speed Power requirements Running time + Flow duration [Short running time] Q Experimental testing time + Testing time Theoretical t.t. + Ideal time Actual time Tunnel wall pressures Bursting pressure ratio [with] Removal of damping plate Reservoir temperature [W.T. interference etc. see Walls] Corrections [Visualisation methods] Oil flow study Oil flow pattern Surface o.f.p. Photographic study Photomultiplier
Y95 Y96 Y97 Y98 Y99 Z1 Z2 Z3 Z4 Z5 Z6 Z7 Z8 Z9 Z10 Z11 Z12 Z13 Z14 Z15 Z16 Z17 Z18 Z19 Z10 Z11 Z12 Z13 Z14 Z15 Z14 Z15 Z16 Z17 Z18 Z19 Z20 Z21	Experimental data + E. results + Results + Test data Wind tunnel data Tables of results Flight test data [Facilities] Q Wind tunnel W.t. tests + W.t. experiment + W.t. investigation W.t. determination + W.t. measurement Transonic w.t. T.w.t. results N.P.L: 18 x 14 w.t. Supersonic w.t. Hypersonic shock t. + H.s.tube H.s.tunnel tests Driven [by] Ignited mixture Stoichiometric m. Oxygen and hydrogen m. M. with excess of helium Hypersonic gun tunnel Shock tube + Tube S.t. experiment + S. tunnel experiment Straight through s. tunnel Reflected s. tunnel [driven by] Unheated hydrogen driving air Constant diameter s.t. [High enthalpy w.t.] Q	264 265 266 267 268 269 270 271 272 273 274 275 276 277 278 279 280 281 282 283 284 285 286 287 288 289 290 291	Model dimensions + k. size [Nounted off-centre] Q [Location in tunnel] End of tube Open area of walls Near model base Along model [Characteristics] Cross sectional size Cross sectional size Cross sectional shape Length of tube Tunnel weight Tunnel speed Power requirements Running time + Flow duration [Short running time] Q Experimental testing time + Testing time Theoretical t.t. + Ideal time Actual time Tunnel wall pressures Bursting pressure ratio [with] Removal of damping plate Reservoir temperature [W.T. interference etc. see Walls] Corrections [Visualisation methods] Oil flow study Oil flow pattern Surface o.f.p. Photographs + Photographio record + Photographic study Photomultiplier Schlieren phtographs
Y95 Y96 Y97 Y98 Y99 Z1 Z2 Z3 Z4 Z5 Z6 Z7 Z8 Z9 Z10 Z11 Z12 Z13 Z14 Z15 Z16 Z17 Z18 Z19 Z20 Z21 Z12 Z23 Z24 Z25 Z26	Experimental data + E. results + Results + Test data Wind tunnel data Tables of results Flight test data [Facilities] Q Wind tunnel W.t. tests + W.t. experiment + W.t. investigation W.t. determination + W.t. measurement Transonic w.t. T.w.t. results N.P.L: 18 x 14 w.t. Supersonic w.t. Hypersonic w.t. + Hypervelocity w.t. Hypersonic shock t. + H.s.tube H.s.tunnel tests Driven [by] Ignited mixture Stoichiometric m. Oxygen and hydrogen m. M. with excess of helium Hypersonic gun tunnel Shock tube + Tube S.t. experiment + S. tunnel experiment Straight through s. tunnel Reflected s. tunnel [driven by] Unheated hydrogen driving air Constant diameter s.t. [High enthalpy w.t.] Q Blowdown tunnel tests Arc discharge w.t. Cascade t. test	264 265 266 267 268 269 270 271 272 273 274 275 276 277 278 279 280 281 282 283 284 285 286 287 288 289 290 291 292	Model dimensions + M. size [Nounted off-centre] Q [Location in tunnel] End of tube Open area of walls Near model base Along model [Characteristics] Cross sectional size Cross sectional size Cross sectional shape Length of tube Tunnel weight Tunnel speed Power requirements Running time + Flow duration [Short running time] Q Experimental testing time + Testing time Theoretical t.t. + Ideal time Actual time Tunnel wall pressures Bursting pressure ratio [with] Removal of damping plate Reservoir temperature [W.T. interference etc. see Walls] Corrections [Visualisation methods] Oil flow study Oil flow pattern Surface o.f.p. Photographs + Photographic record + Photographic study Photomultiplier Schlieren phtographs Shadowgraph study S. measurement
Y95 Y96 Y97 Y98 Y99 Z1 Z2 Z3 Z4 Z5 Z6 Z7 Z8 Z9 Z10 Z11 Z12 Z13 Z14 Z15 Z16 Z17 Z18 Z19 Z20 Z21 Z22 Z23 Z24 Z25 Z26 Z27	Experimental data + E. results + Results + Test data Wind tunnel data Tables of results Flight test data [Facilities] Q Wind tunnel W.t. tests + W.t. experiment + W.t. investigation W.t. determination + W.t. measurement Transonic w.t. T.w.t. results N.P.L: 18 x 14 w.t. Supersonic w.t. Hypersonic w.t. + Hypervelocity w.t. Hypersonic shock t. + H.s.tube H.s.tunnel tests Driven [by] Ignited mixture Stoichiometric m. Oxygen and hydrogen m. M. with excess of helium Hypersonic gun tunnel Shock tube + Tube S.t. experiment + S. tunnel experiment Straight through s. tunnel Reflected s. tunnel [driven by] Unheated hydrogen driving air Constant diameter s.t. [High enthalpy w.t.] Q Blowdown tunnel tests Arc discharge w.t.	264 265 266 267 268 269 270 271 272 273 274 275 276 277 278 279 280 281 282 283 284 285 286 287 288 289 290 291	Model dimensions + k. size [Nounted off-centre] Q [Location in tunnel] End of tube Open area of walls Near model base Along model [Characteristics] Cross sectional size Cross sectional size Cross sectional shape Length of tube Tunnel weight Tunnel speed Power requirements Running time + Flow duration [Short running time] Q Experimental testing time + Testing time Theoretical t.t. + Ideal time Actual time Tunnel wall pressures Bursting pressure ratio [with] Removal of damping plate Reservoir temperature [W.T. interference etc. see Walls] Corrections [Visualisation methods] Oil flow study Oil flow pattern Surface o.f.p. Photographs + Photographic record + Photographic study Photomultiplier Schlieren phtographs Shadowgraph study S. measurement Hot wire measurement + H.w.
Y95 Y96 Y97 Y98 Y99 Z1 Z2 Z3 Z4 Z5 Z6 Z7 Z8 Z9 Z10 Z11 Z12 Z13 Z14 Z15 Z16 Z17 Z18 Z19 Z20 Z21 Z12 Z23 Z24 Z25 Z26	Experimental data + E. results + Results + Test data Wind tunnel data Tables of results Flight test data [Facilities] Q Wind tunnel W.t. tests + W.t. experiment + W.t. investigation W.t. determination + W.t. measurement Transonic w.t. T.w.t. results N.P.L: 18 x 14 w.t. Supersonic w.t. Hypersonic w.t. + Hypervelocity w.t. Hypersonic shock t. + H.s.tube H.s.tunnel tests Driven [by] Ignited mixture Stoichiometric m. Oxygen and hydrogen m. M. with excess of helium Hypersonic gun tunnel Shock tube + Tube S.t. experiment + S. tunnel experiment Straight through s. tunnel Reflected s. tunnel [driven by] Unheated hydrogen driving air Constant diameter s.t. [High enthalpy w.t.] Q Elowdown tunnel tests Aro discharge w.t. Cascade t. test	264 265 266 267 268 269 270 271 272 273 274 275 276 277 278 279 280 281 282 283 284 285 286 287 288 289 290 291 292 293	Model dimensions + M. size [Nounted off-centre] Q [Location in tunnel] End of tube Open area of walls Near model base Along model [Characteristics] Cross sectional size Cross sectional size Cross sectional shape Length of tube Tunnel weight Tunnel speed Power requirements Running time + Flow duration [Short running time] Q Experimental testing time + Testing time Theoretical t.t. + Ideal time Actual time Tunnel wall pressures Bursting pressure ratio [with] Removal of damping plate Reservoir temperature [W.T. interference etc. see Walls] Corrections [Visualisation methods] Oil flow study Oil flow pattern Surface o.f.p. Photographs + Photographic record + Photographic study Photomultiplier Schlieren phtographs Shadowgraph study S. measurement

705	Blunt nosed Pitot tube		[Stability]
Z95 Z96	Spherical pendulum	ZA55	Asymptotic stability
Z97	Gyroscopic pendulum	ZA56	Routh-Hurwitz criterion
-21		ZA57	Phase space + State space
Z98	[Theoretical work] Q + T.study +	ZA58	Partial d.e.
•	T. investigation + Theory	ZA59	Difference equation Q
Z99	[Assumption] Q	ZA60	[Navier Stokes d.e.] Q
ZA1	[Comparison] Q + Analogy	ZA61	Finite d.e.
ZA2	Prediction + Theoretical p.	ZA62	F.d. methods
ZA3	[Criterion] Q	ZA63 ZA65	F.d. approximation Eigenvalue problems
ZA3x	Linearized theory	ZA66	Integral equations
ZA3y	Approximate theory	ZA67	Generalized i.e.
ZA4	Zero order theory	ZA68	Abel i.e.
ZA5	Second order t.	ZA69	Birbaum i.e.
ZA6	[Rigorous t.] Q	ZA70	Possio i.e.
ZA7	Potential t.	ZA71	Prandtl i.e.
ZA8	Modified Newtonian t.	ZA72	Finite part
	Mathematics	~	[Series]
ZA9	Tables + Tabulation	ZA73	Time series
ZA10	Table of values	ZA75 ZA76	Convergence C. rates
ZA11	Calculation + C. method +	ZA77	_ Asymptotic s
	[Calculated] Q + [Calculating] Q	2111	[Boundary values]
	+ Methods for calculating +	ZA78	B.v. problems
	Computation	ZA79	Blasius problem
ZA12	Routine method	ZA80	Rayleigh's problem
ZA13	Practical calculation	ZA81	Boundary conditions + Given b.c.
ZA14	Theoretical c.	ZA82	Three point b.c.
ZA15	Estimate + Estimation	ZA83	Time dependent b.c.
ZA16 ZA17	Approximate c. Formula	ZA84	Function
ZA16	Asymptotic expansion f.	ZA85	Partition f.
ZA19	General f.		[Methods, techniques, operations]
ZA20	Solution	ZA86	Method + [Theoretical m.] Q
ZA21	Method of s.	ZA87	Generalized m.
ZA22	Simple s.	ZA88	Analysis + Analytical m. Q
ZA23	Approximate s.		+ Analytical study +
ZA24	Accurate s.		Theoretical a. + Analytic determination + Determination
ZA25	Exact s.		
ZA26	Theoretical s.	ZA89	by a. Analytical estimation
ZA27	Second order s.	ZA90	Analytical results
ZA28	Simularity s.	ZA92	Approximation
ZA29 ZA30	Asymptotic s. Analytical s.	ZA94	Successive a.
ZA31	Approximate a.s.	ZA95	Closed form
45,	,	ZA96	Newtonian a.
7470	[Dimensional analysis]	ZA97	Variational principle
ZA32	Dimensionless form Scale effects	ZA98	V.p. of Biot
ZA33	20019 6114019	ZA99	Perturbation method
	Algebra	ZA99a	Orthogonal functions expansion method
ZA34	Equations	ZB1	Graphical a.
ZA35	Approximate e.	ZB2	Numerical determination +
		ane	N. calculation + N. solution
ZA36	Linear e. Non-1. e.	ZB3	[Iterative method] Q +
ZA37 ZA38	Three dimensional e. Q		Iterative solution +
ZA39	Singular e.		Iterative numerical solution
ZA40	Six degrees of freedom e.	ZB4	[Convergent] Q
ZA41	Elliptic e.	ZB4a	Finite difference
ZA42	Navier Stokes e.		numerical s.
ZA42a	_ [with] No-surface alip	ZB5	Richardson method Second order R.m.
	[Transformation]	ZB6	Liebmann m.
ZA43	Fourier t.	ZB7 ZB8	Extrapolated L.m.
ZA44	Lorenz t.	ZB9	Relaxation methods
	[Analysis]	ZB10	[with] Spherical polar
ZA45	Newtonian a.		coordinate
2104)	[Calculus]	ZB11	Modified
ZA47	Complete elliptic integral	ZB 12	Successive overrelaxation m.
ZA48	Integration + I. methods +	ZB12a	Analytic continuation
•	Integrating solution	ZB12b	[by] Taylor's series
ZA49	Asymptotic i. method	ZB14	Inversion
ZA50	Differential equations	ZB15	Linearization Method of characteristics
ZA51	Linear d.e.	ZB 17	Johannsen and Meyer method
ZA52	L.d.e. with constant	ZB18 ZB19	Polhausen m.
a.c.>	coefficients	ZB 20	Extension of Stratford m.
ZA53	L.d.e. with periodic coefficients + [Periodic		
	coefficient Q	ZB21	Newtonian closed-form
ZA54	[Lyapunov's method] Q	# P20	expression Newtonian coefficient
	(-v	ZB22	1104 0011701 22211232

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ZB23
               Computer
                  [High speed computer] Q
ZB24
ZB25
                   Digital computer Q
                  [Programme] Q
ZB26
           [Automatic control]
             Automatic control theory
See also Stability
ZB27
ZB28
                Dynamic systems
ZB29
                Linear s.
ZB30
                Non-linear s.
                Non-linear mechanics
ZB31
ZB32
                Stationary s.
ZB33
                Non-stationary s.
                Lyapunov's theory
ZB34
                  Second method of L.
ZB35
                Stable regions
ZB36
                Transient behaviour
ZB37
                Limit cycles
ZB38
                Random disturbances
ZB39
                Relay servos
ZB40
                Nonlinear autopilot
ZB41
                Autopilot with feedback
ZB42
                Control system optimization
ZB43
                Time-optimum control
ZB44
       [General properties]
[General form] Q
[Nature] Q
ZB45
ZB46
ZB47
         Characteristic
            [Parameters]
ZB48
ZE49
            Variable properties
Z.850
            Material properties
ZB51
            Configuration
ZB52
              Arbitrary shape
ZB53
            Length
ZB54
              L. changes
ZB55
ZB550
            Thickness effect
Stability [non-aerodynamic]
 ZB55e
            Effective
            Combined + Combination
 ZB56
 ZB57
              Combined effect
 2B58
            Attached
 ZB59
            Associated
 ZB60
               Carry
 ZB61
            Relationship
               Non-linear relation
 ZB62
            Ratio
Finite rate
[Quantitative] Q
 ZB63
 ZR64
 2B65
          [Location, position]
 ZB66
          Location
            Streamwise
 2567
 ZB68
             Ahead of body
 ZB69
            Behind
            Passing through
 ZB70
 ZB71
             Downstream influence
 ZB72
            Far downstream
        [General processes]
[Effect] Q + [Influence] Q
 ZB73
 ZB74
           Design
 ZB75
           Controlling
 ZB76
          Elimination
 ZB77
           Variation
             Arbitrary v.
[Large change] Q
 ZE78
 ZB79
 ZB80
           Transition
 ZB81
           Onset
 ZB82
           Deviation
           Production 4
 ZB84
           Derivation
 ZB85
           Performance
 ZB86
           Dynamic behaviour
 ZB87
           Expansion
 ZB88
 ZB89
           Reduction
```

APPENDIX 5.5

ALPHABETICAL INDEX OF CONCEPTS

This Appendix lists all the concept terms used in the indexing of the 200 documents comprising Subset 1, as given in Appendix 3E. For each term in its natural form is given the appropriate notation for its location in the concept term schedules, which appear as Appendix 5.4.

Each concept appears under each key term, so that, for instance, the following entries will be found:

Abel integral equation Integral equation, Abel Equation, Abel integral

In addition, this list includes terms which occurred in the set of search questions but which did not appear in the indexing. All such terms are marked by an asterisk.

			Т35
Abel Integral Equation	ZA68	Aerodynamic Stability	V63
Ablating Materials	X56	Stiffness Derivatives	W89
Ablation	X51	Twist Matrices	
Cooling	X88	Aeroelasticity Static Aerofoils Blunt-Nosed	
Effective Heat of	VCZ	Single Wedge	
-Rate Sensor	X53	Straight Line	
Rates	X52	Technique Complete	
Surface	G21	Technique Half	
Abrupt Stall Absolute Flow	F90	Three Dimensional	
Absorption Ultra-Violet	£ 70	Aerothermodynamic Relations Basic	
Radiation	G56	Afterbody Conical Base	
Accelerated	G56	Cylinder	
Accelerating Flow	N6	Cylindrical	
Acceleration	K48	Drag	D6
During Boost	K50	Surface	D5
Potential	บ36	Truncated	
Variations in	•	Vehicle Conical	
Vertical		Ahead of Body	ZB68
Accommodation Height	037 j	Jet	P75
Accurate Solution	ZA24	Shock Wave	କ୍ଷ
Acoustic Radiator	V58	Aileron Coefficients Oscillatory	T. C. 4
Across Bow Shock	Y67	Air	L54
Active Gas Chemically		Bleed Compressor	250
Activity Solar		Blowing Jet	P52
Actual Time	Z80	Breathing Vehicle	A60
Addition Small Rate of Heat		Burned In	
Adiabatic Flow	м93	Combustion, methane*	L62
Surface Temperature	D63	Components	LOZ
Temperature Rise Maximum	-/-	Compressed	L66
Wall Temperature	D63	Current	M12
Adjacent Flat Surface	D13	Density	J94
Main Blade	F10	Density at Perigee	0)4
Adjustable Blading	F23	" Perigee " Profile	M13
Inlet Guide Vanes	F33	Dissociated	
Adverse Ground Effect	T77 S64	Drag	M7
Aerodynamic Balance		Drag around Orbit	M9d
Centre	551 552	Dry	,
Centre Characteristics	UJE	Energy of	
Centre Local Characteristics	к6	Expansion Cold	
" Longitudina		Flow*	L65
" Static	nation .	" Correlated Non-Equilibrium	
Coefficients Longitudinal		" Equilibrium	
Damping	V75	" Hot*	
Damping Coefficient	V77	" Stagnation	
" Derivatives	V78	Fully Ionized	
Derivatives*	T45	Heated	
" Direct		High Temperature	
" Indirect		Hydrogen Burned	246
" Oscillatory		Jet	P46
Design	W38	" Blowing	
Flutter Derivatives	V21	" Cold	
Forces	R22	" High Energy	
Heating	У6	" Small	
Heating Rate	Y8	Danbore	C47
" Laminar		" Vortex Generators	F50
Inertia	R24	Outlet Angle	Q96
Interaction Effect	T61	Shock Velocity	₩.7 ₩
Laminar Heating	Y7	Still Straam	L66
Load	R49	Stream Unheated Hydrogen Driving	200
Loading	R49	Aircraft High Speed	
Moments	S49	Models Complete	
Noise	V54	Shapes	A9
Normal Forces	R25	Structures	A5
Problems	K7	i sets, strike i den den den den men den gen,	-

Airfoil	A79	Angle of Attack Arbitrary	
Blunt Trailing Edge	MIT	" " Correction	V 95
		001 7 00 47011	
Circular Are		DISCITORCION	W4
Double Wedge		" " High	
Equation	A82	" " Large	
NACA 4412		" " Moderate	
Nonlifting		" " Oscillation	V12
Rigid		" " Positive	
Shape	A85	" " Uniform	
	AOJ	" " Zero	
Symmetrical		2910	77.4
Two-Dimensional		Between Sun and Perigee	K1
Wedge		Boattail	
Airforces*	R22	Body	
Airplane*	A2	Cone 10 Degree Semivertex	
Airplane Wing	B1	Constant Flow	
Tail-Boom		of Corners Number and	
Tailless		Deflexion	
	K46		
Airspeed Holding	• .	Dihedral	7400
All Movable Control Surface	C26	of Dihedral Range of 30 Degrees	W20
Tail	C16	Exit Shock Wave	
Alloy 2024 T4 Aluminium		Flight Path	
Along Hub	F63	Greater than 1777 Re-Entrant	
Model	Z69	of Incidence*	V94
Alternating Lift Function Theodorsen		of Incidence Correction	v95
Altitude	VO4	Inlet Blade	マフン
	K91		
Various heights in KM-L 96/L98		Jet Shock Wave	
Condition High		Mach	
Flight		Nozzle Divergence	
Flight High		Nozzle Exit	
Maximum		Outlet Flow	
of 300 Neutical Miles	K95	Phase	
of 525 Nautical Miles	K96	Sweep	
			8W
Results	M3	of Sweep Range 0-75 Degrees	
Aluminium Alloy 2024 T4	J45	of Sweepback	₩9
Ambient Atom Mass Fraction	Y38	Turning	
Gas	L26	Wedge	
Supersonic Stream	M54₩	Wedge Shock	
Temperature	Х7	60°Sweepback	
Amount	E56	Annulus Area	F87
Amount of Dissociation		Anti-Symmetric Velocity Potential	U35
	Y39		
Amplitude Measurement	V70	Apex Located Controls*	C27
Analogy	ZA1	· Location Jet Shock Wave	
Analysis	ZA88	Apogee Distance Change	J96
Compressible Flow		Apparatus	Z55
Determination		Apparent Mass Effects	SÍ
Linearized Flow		Appearance Time of	
Modal		Applied Forces (=Load)	₩59
			マンプ
Modified Strip		Approach Target Planet	97/
Newtonian		Approaching Chemically Frozen Flow	Y65
Plastio		Vibrationally Frozen Flow	Y66
Rayleigh Type		Approximate Analytic Solutions	ZA31
Stage-Stacking		Calculation	ZA16
Theoretical		Equation	ZA35
	7740-	.	
Analytic Continuation	ZB12a	Solution	ZA23
Determination	ZA88	Theory	ZAJy
Solutions Approximate		Approximation	ZA92
Analytical Estimation	ZA89	Finite Difference	
Method*	ZA88	Graphical	,
Results	ZA90	Hypersonic	
Solution	-	Intermediate Enthalpy	
	ZA30		
Study	ZA88	Newtonian	
Anemometer Hot Wire		Successive	
Angle Air Outlet		Arbitrary Angle of Attack	V94
of Attack	V94	Blade Shape	F36
of Attack Range 0-65 degree	V98a	Cross-Section	W39
" " 4-18 "	V99a	Hub Shroud Contour	F79
7 -10	1776		- 17

Arbitrary Pitching Rate	K68	Attached Flow Fully	
Planform Wing	B4	Sheet Rigidly	
Shape	ZB52	Shock-Wave System	Q60
Stiffness Wing	B39	Attachment Flow	
Variation	ZB78	Line	N82
Arc Airfoil Circular		Transonic Flow	
Carbon	705	Attack of Angle	
Discharge Wind Tunnel Heating Rockets	Z25	Arbitrary Angle of	
Parabolic Body	A73	Correction Angle of	
Profile Circular		Distribution Angle of High Angle	
Ardc 1956 Model Atmosphere	L29	Large Angle	
Area Annulus		Moderate Angle	
Argon Shock Waves	Q67	Oscillation Angle of	
Around Orbit Air Drag		Positive Angle of	
Arrangement Wing Body		Uniform Angle of	
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Vans	F 20	Rocket Propelled	
Adjustable Inlet Guide Splitter		Satellite	
Surface Splitter		Short Probe	
Vortex Generator	C53	Skip	
Vaneless Diffuser	F84	Space	
Vanguard I	A29	Supersonic	
Vanishing Load	R62	Surface Re-Entry	
Low Aspect Ratio Wing	B59	Uncontrolled*	
Pressure Gradient	T11	Velocity	U25
Vaporization	X57	Air Shock	
Vapour Pressure Saturation		Blade Surface	
Variable Capacitance Sensor	X54	Blowing High	
Conductivity	X42	Distribution	U37
Density Wind Tunnel	Z32	Distribution Prescribed	
Medium	V31	Escape	
Property	ZB49	Exhaust	
Stokes Stream Function		Field Quasi-Conical	
Vorticity	ር ሃ የጎነብታ ምን	Flight	
Variation	ZB77	Gas Flow	U38
in Acceleration	K49	Gradient Gradient Severe	<i>□ </i>
Across Channel	G65	Induced	
Arbitrary	М19	Maximum	
From Day to Night	ニリブ	671 @b./b.n.h. 844 \- 68466	

Velocity Maximum Relative Blade Surface		Vortex Discrete Horseshoe	
Measurement	U42	Displacement Main	
Potential	U33	Distribution Elementary Horseshoe	
Anti-Symmetric		Flow	P7
Perturbation		Formation	P8
Symmetric		Generator Air Jet	
Profile	U39	Ramp-Type	
Profile Linear		Twisted Strip Type	
Re-Entry		Vane	
Relative Upstream		Wedge-Type Wing-Type	
Requirement Trajectory		Growth	P11
Sinking Slip Exit Relative		Horseshoe	
Sonic Throat		Individual	
Trailing Surface		Location of Leading Edge	
Uniform		Location of Part Span	
Variation	U37	Main	- 10
Vector	J75	Movement	P12
Vertical		Pair Stationary	non
Vertical Induced		Path	P20 P21
Venturi Rounded-Entry		Path Location Pattern of Horseshoe	£ 65. 1
Vertical Acceleration	S1a	Rectangular	
Axis	D38 U28	Secondary	
Induced Velocity	U20 H6	Sheet Position	P22
Nozzle Positi on Oscillation Sinusoidal	110	Spanwise	
Plane Control	T30	Strength	P25
Position	W5	Strength Main	
Tail	C18	System	092
Translation	N52	Tip	
Velocity	U29	Trailing	00
Very Small Aspect Ratio Wing of		Type Boundary Layer Separation	02 U64
Vibration Free		Vorticity	P24
Mode Uncoupled		Contour Distribution	P17
Normal Mode		Interaction Domain	P15
Resonant	V88	Interaction Theory	P16
Testing	V88	Variable	P23
Test Vibrational Mode	V45	Wagner Indicial Lift Function	S10
Vibrationally Frozen Flow Approaching	. 43	Wake	U19
Viscid Flow	М36	Blockage	T70
Viscosity	U84	Development of Circulating	1240
Coefficient		Flow	U19
Effect Boundary Layer		Interference Effect	T63
Temperature Law*	U85	Propulsive Jet	U20
Temperature Solution Linear*	3377	Width Divergence	D73
Viscous	N36 S26	Wall Boundary Layer	N69
Drag	U87	Channel	•
Effect	N36	Composite	
Flow Fluid	L9	Condition	D91
Fluid Motion	N37	Constraint	D95
Force	S26	Copper	~ 44
Interaction	T64	Divergence*	241
Layer Regime	L50	Finite Thickness	
Motion	N37	Flow Past Parallel	E3
Scavenging Phenomena Jet		Heat Flux History	200
Volume	K11	Heat Transfer	
Von-Karman Source Sink Method	082	Homogeneous Inconel	
Vortex	087	Interference	D93
Blade Form Free		Non-Parallel	*
Blade Form Half Bound		of Open Area	
Chordwise		Plane	
Co-Rotating		Pressure Tunnel	
Counter-Rotating		Streamline	
THE STATE OF THE S			

Wall Temperature	D62	Wave Surface	
Temperature Adiabatio		Terminal Shock	
Temperature Constant		Thickness Shock	
Temperature Gradient	E1	Tip Shock	
Test Section		Tollmien Schlichting	
Thermally Thick		Upstream of Shock	
Thermally Thin		Wavelength Jet	
Thickness	D76	Weakening	W71
of Tube	Z40	Wedge	018
Tunnel	Sub-C	Aerofoil Single	
Tunnel Solid*	•	Airfoil	A86
Wind Tunnel		Airfoil Double	
Wind Tunnel Solid		Angle	J13
Wind Tunnel Test Slotted		Blunted	.,,
Wind Tunnel Test Solid		Method Tangent	
Ward Extension	J32	Shock Angle	Jia
Warren 12 Planform	B36	Type Vortex Generator	C50
warren 12 Flamform Water Flow	L17	Weight Engine	0,0
Wave Ahead of Shook	70.6	Flow	F99
Angle Exit Shock		Flow Characteristic Equivalent	-))
		Flow Maximum	
Angle Jet Shook		Flow Overall	
Apex Location Jet Shock		Flow Range	G4
Argon Shock Attenuation of Shock		Flow Variation	G 5
		Molecular	w.)
Bow Shock		Ratio	K16
Compression		Ratio Model	25.10
Detached Shock			
Development Shock	600	Tunnel	
Drag	329	Wing with Concentrated	W86
Equation Solution Classical		Weissenger Method	WOO
Expansion		Whirl Inlet	
Flexural		Upstream Flow	G46
Flow Behind a Shock		Wide Range	- •
Formation Shook		Wing	B52
Forward Shock		Width Band	
Generated Shock		Divergence Wake	
Hypersonic Shook		Infinite	
Induced Boundary Layer Separation		Jet Slot	7 (9
Shook		Wind Speed	L67
Infinitesimally Thin Shock		Wind Tunnel	21
Instability	L19	Arc Discharge	
Interaction Boundary Layer Shock		Choked	Va6
Interaction Shock		Data	¥96
Kelvin Helmholtz		Determination	Z3
Location Shock		Experiment	Z2
Longi tudinal		High Enthalpy*	
Mach		Hypersonic	
Movement Shock		Hypervelocity	700.4
Nitrogen Shock		Interference Effect	D94
Normal Shock		Investigation	Z2
Oblique Shock		Low Density*	C# *F
Outboard Shock		Measurement	23
Oxygen Shock		Result Transonic	
Pattern Shock		Solid Wall	
Position Divergence Shock		Supersonio	70
Propulsion Light		Test	Z2
Rear Shock		Test Choked	
Reflected	Q36	Test Slotted Wall	
Reflection Interference	Q38	Test Solid Wall	
Resonance Compression	•	Transonic	
Riemann		Variable Density	C 40
Shape Shook		Wall	Z40
Shock		18" x 14" NPL	T'T A
Sound		Windward Body	J34
Strength Shock		Edge	E9
Strong Shock		Wing	B1
Structure Shock		Wing Airplane	

			ma on valden Didmo	
Wing	Arbitrary Planform		Wing Sharp Leading Edge	
	Arbitrary Stiffness		Simple Planform*	
	Beneath	734	Sinking	
	Body	B1	Slender Slender Rectangular	
	Body Arrangement	A97		
	Body Combination	A97	Slope Slowly Oscillating	
	Body Interference	A98	Solid Construction	
	Cambered	200	Steady	
	Chord	B99	Straight	
	Circular Planform	368	Subsonic Edged	
	with Concentrated Weight	.000	Supersonic Edged	
	Curved Tip Planform		Surface	C4
	Cantilevered		Surface Flat	
	Clipped Delta		Sweep Small	B6
	Cropped		Swept	
	Delta		Sweptback	
	Differentially Deflected		Sweptback Delta	
	Elastic		Sweptforward	
	Elliptic		Symmetrical	
	Elliptical		Tapered	
	Finite		Theory "Not-so-Slender"	
	Finite Aspect Ratio		Theory Linearized	
	Finite Span	В69	Theory Slender	
	with Flap	1347	Thick	
	Flat		Thickness	C1
	Flat Plate		Thickness Symmetrical	
	Flat Surface		Thin	
	Flat Triangular	B94	Three-Dimensional	
	Flow Pattern	B94	Tip	C8
	Flow Study	27-4	Tip Deflection	C9
	Flutter Cantilever		Tip Panel	A39
	Fluttering	A98	Tip Panel Unfolding	
	Fuselage Interference	****	Tip Sweptforward	
	Half Harmonically Oscillating		With Twist Triangular	
			Triangular	
	High Aspect Ratio		Twisted	
	Infinite Span		Two-Dimensional	
	Laminar Lift	B97	Type Vortex Generator	C51
	Lifting		Uncambered	
			Unswept	
	Loaded		Untepered	
	Long Low Aspect Ratio		Vanishing Low Aspect Ratio	
	Model	B3	Vehicle Arrow	-/-
	Model Half		of Very Small Aspect Ratio	B60
	Model Sweptback		Wide	
	Model Tapered		Yawed	. 45
	Narioa Narioa		Winged Missile	A12
	Near Triangular		Re-Entry Glider Configuration	A38
	Nonlifting		Wire Anemometer Hot	
	Oscillating		Measurement Hot	
	With Parabolic Twist	B35	Spanwise	
-	Perforated		Two-Dimensional	
	Pitching		With Feedback Autopilot	
	Plane of		Within Jet Location	
	Planform	B94	Without Corner Cylinder	
	Pressure Distribution	B95	Wood Model Compreg	
	Rectangular		Work Theoretical*	C3 79 A
	Rectangular Planform		Working Section	Z34
	Re-Entry Configuration De	lta	Closed Throat	
	Re-Entry Configuration Sy	reptback	Closed Tunnel	בים ער פין
	Rigid	•	With Open Jet	Z35
	Rolling		With Rigid Boundary	236
	Rotating		Size	
	Shape	B4	Slotted	T# 4% 4%
	Sharp Edge		Oo lem	K77

30° Yaw	к78
Yaw Control	K81
Stability	K79
Yawed Body of Revolution	Н52
Wing	B87
Yawing Derivative	K80
Zero Angle of Attack	. 796
Aspect Ratio	W13
Boundary Constraint	D96
Gravity	к99
Lift	R70
Model Incidence	V97
Order Theory	ZAA
Pressure Gradient	T12
Temperature Gradient	X23
Zone Expansion	
Shook Transition	

APPENDIX 5.6

CONTROLLED TERM VOCABULARY

Terms marked with an asterisk represent terms used in the collection of 200 documents (Document Subset 1 of Appendix 3E), and are therefore comparable with the terms in the concept schedules (Appendix 5.4) and rotated index (Appendix 5.5). The remaining terms were used in the additional 150 documents that make up Document Subset 2.

These terms were based on those which appear in the Thesaurus of Engineering Terms of the Engineers Joint Council. Terms which do not appear in this Thesaurus are preceded by a cross.

- + Ablating materials*
 Ablation*
 Absorption*
 Acceleration*
- + Accommodation height*
 Accuracy*
 Actual time*
 Adiabatic conditions*
 Adjustment*
- + Aerodynamic balance* + Aerodynamic center*
- + Aerodynamic characteristics*
- + Aerodynamic configurations* + Aerodynamic forces*
- Aerodynamic heating*
 + Aerodynamic loads*
- + Aerodynamics*
 Aerothermodynamics*
- + Afterbodies*
 Agents*
 Ahead*
 Ailerons*
 Air*
 Air flow*
 Air flow*
 Airfoils*
 Airframes*
 Airplanes*
 Air resistance*
 Airspeed*
 Altitude*

Aluminium alloys*
Ammonium compounds
Amplitude*
Analogies*
Analogue computers
Analyzing*
Angle of incidence*
Angles*
Annuli*
Antennas
Applying

Applying
Approaches
Approximation*
Arcs*
Area*
Argon*

Attitude control

Arrow wings*
Asbestos
Aspect ratio*
Atmosphere
Atmosphere Extra-terrestial*
Atmosphere Terrestial*
Atmosphere Entry*
Atomic properties*
Atoms*
Attenuation*
Attitude*

Automatic control*
Autopilots*

- + Axial distribution
- + Axial flow + Axial forces*
 - Axial flow pumps*
 Baffles*
 Balances*
 Balancing

Ballistic missiles*
Balsa woods*

Bands*

- + Base pressure*
 Bases*
 Bearing*
 - Beams (Radiation)*

Beaming*

- Bearing (Direction)*
- Behaviour*
 Bending*
 Beveling
 Binary systems
 Blades*
- + Blasius equation*
 Bleeding*
- Blocking*
 Blowers*
 Blowing*
 Blunt bodies*
 Blunt cones*
- Blunt cylinders + Blunt leading edges*
- + Blunt trailing edges*
 + Bluntness*
- + Rluntness* + Boat tails*
- Bodies of revolution*

Bombers Booms* Booster rookets* Boundaries* Boundary layer*

Boundary layer control*
Boundary layer transition*

- Braking
 Bridging*
 Buckling*
 Buffeting*
- Buffeting*
 Buoyancy*
 Bursting*
 Calibration
 Camber*
- + Cantilever structures*
 Cantilever wings*
 Capacity

Caps
Carbon Dioxide*
Carbon arcs*

Carbon tetrachloride

+ Uascades*

Catalysis Catalysts* Cathode-Ray oscilloscopes Centre of Gravity* + Centre of pressure* Centres* Centrifugal compressors* Centrifugal force* Ceramics Chamber* + Channels (Fluid flow)* Characteristics* Charge Charged particles* Chemical equilibrium* + Choking* Chord* Circuits* + Circular aro* + Circular cone* + Circular cross section + Circular cylinder* Circulation* Clamping + Closed throat* Clouds Coefficients*

Cold gases* Combustion* Commercial planes Compatability Components* Composite materials* Compositions® Compressed air* Compressibility* Compressible flow* Compressing* Compressor blades* Compressors* Computation* Computers* Concentrating Concentration Condensing Conduction Conductivity* Conductors Cones* Conics Constraining Constraints* Constrictions Construction* Contingency* Continuity equation*
Continuous beams*

+ Continuous panels* Contraction Control surfaces* Control systems* Convection* Convection heating* Convergence* Conversion Coolants Coolers . Cooling Co-ordinates* Copper*

+ Corners* Correction* Correlation* Corridor

+ Couette flow" Cracking Criteria* Critical Mach No.

Critical velocity* + Crocco's equation Cross flow* Cross sections* Crossings* Curvature* Curves*

Curvilinear co-ordinates Cvcles* Cylinders* Damage Damping* Data* Data reduction Day*

Decay* Deceleration* Deceleration rate* Deflection* Deformation Degrees of Freedom*

Delay time* Delta wings* Density* Density ratio* Depth Derivation* Derivatives* Design* Detaching* Detection Determination* Development*

Diaphragms* + Diatomic gases Dielectrics Differences*

Diameters*

Differential equations* Diffuser*

Diffusion* Dimensionless numbers*

Dimensions*

Discharge* Discontinuity* Disks* Dispersion* Displacement* Dissipation* Dissociation* Distance* Distortion* Distribution* Diurnal variations* Divergence* Dividers* Documents*

+ Downstream* + Downwash* Drag* Driers* Drilling* Drive* Driving* Drum* Dry gas* Ducts

Dynamic characteristics*

Dynamics* Earth* Eccentricity* Eddies* Edges* Efficiency*

Eigen Functions* Elasticity* Electric arcs* Electric arc heaters

Electric fields* Electric heating Electric potential* Electric propulsion* Electrical filters Electrical properties Electricity* Electromagnetic induction* Electromagnetic waves* Electrons Electrons density Electrostatics Slements* Elevators Elevons* Blimination* Ellipses* Slliptic functions* Emission* Emissivity* Emergy* Engine* Engine noise* Enthalpy# Emtropy* + Entry angle + Entry corridor Environment* Equationar Equation of Motion* Equation of State Kauilibrium* Equipment* Equivalence* Equivalent weight* Erosion* Errore* + Escape velocity* Estimation* Ethylene" Evaluation* Evaporation* Exchanging* Excitation* Exhaust gases* Exhausting* Expansion Exparimentation* Exponents Exposure* Expressions Extension* + Faces* Failuro* Peedback* Field Fighters Filling* Filtration* + Fineness ratio Pinite difference theory* + Finite span* Fins Firing* Pixtures* Flames* Flaps* Flat plate models* Flatness* Flaw Flexibility* Floxing Flight* Flight paths Flight testing* Flow#

Flow equations*

Flow measurement

Flow meters* Fluctuation Fluid flow* Fluid mechanics* Flux* Folding* Fog Force* → Forebodies* + Form drag* Formica Forming Formulas: ' Foundation* Fourier transforms* Fractions* + Free flight models + Free molecule flow* + Free stream# Freezing* Freon* Frequencies* Frequency response* Friction* Fuel consumption Fuels* Functions* Gas flow* Gas Injection* Gases Gears* Generators Geometry* Glass Glass Fibres Gliders* Gradiente* Grain size* Graphite* Graphs Gravitation* Green Function* Ground effects* + Ground height + Ground pressure + Ground reflection* + Ground resonance Growth Guidance* Guide vanes* Gust logds# Gyrosoopes* Hall effect Harmonios* Head* Heat" + Heat of ablation* + Heat absorption* Heat shields" Heat sinks Heat transfer* Heat transfer coefficient* Heat Transmission* Heating* Height* Hellum* Hemispheres# High pressure* High temperature* + Hinges* History* Holding Holest Homogeneity* Homogeneous flow + Hot gas Hot wire anemometers*

Humidity*

Lift dependent drag* Humidity control + Lifting* Hydrocarbons* Light* Hydrodynamics* Linear equations* Hydrogen* Linear differential equations* Hyperbolas Linear systems* Typerbolic functions Liners* Hypersonic flight* Lines* Hypersonic flow* Line spectra* Hypersonic planes* Liquefaction* Hysteresis* Liquid flow* Ideal fluid* + Liquid injection* Liquid propellants* Ideal gas* Ideal gas law Ignition* Loading* Impact* Loads* Impedance Impellers* Logarithms* + Longitudinal Jets + Longitudinal stability* Inconel* Increasing* Increments* Longitudinal waves* Indicators* Losses* + Induction (Aerodynamics)* + Low density* + Low Reynolds No. * Inertia* Initiation* Luminosity* + Lyapunov method* Injectant + Injected gases* + Mach cone* + Injected liquids* + Mach Number* Magnesium* Injection Magnetic fields* Injectors Inoculation* Magnetic flux* Inorganic salts Magnetic properties* Magnetohydrodynamics* Instrumentation* Magnification* Insulation* Intake system* Magnitude* Integral equations* + Main stream* Integrals* Man Mars* Integration* Magax Intensity Interference* Mass flow* Interplanetary flight* Mass transfer + Mass transfer cooling Intersection* Matching* Iodides* Ion propulsion* Materials* Ionization* Materials testing* Mathematical analysis* Ionization g ages* Mathematical models Ionosphere* Ions* Matrices* Maxima* Isobars* Isolation Mean* Mean free path* Isotropy* Iteration* Measurement* Measuring instruments* Jet* Mechanics* Jet Engines* Jet nozzles* Melamine Jetstreams* Melting Merged layer* Joining* Meteorites* Joists* Jupiter Methane* Midcourse guidance* Junctions + Karman-Polhausen approximation Minima* + Karman Street Minimization Kinetic energy Missiles* Kinetic theory Mixing* + Kirchoff theory* Mixtures* + Knees* + Modal analysis* Knudsen flow Node* Kussner function* Models* Laminar boundary layer* Modulation Laminar flow* Modulus of elasticity* Laminates* Molecular weight* + Lateral stability* Molecules* Latitude* Molvbdenum* Launching Momen ts* Layers* Momentum* + Leading edges* Momentum transfer Legendre polynomials Motion* Length* Mounting* Life*

+ Multiple*

Lift*

Multiplier phototubes*

+ Nacelles*

+ Navier-Stokes equation*

+ Negative lift

+ Newtonian flow Night*

Ni trogen*

Nitrogen compounds

Nodes* Noise*

+ Non-Equilibrium flow* Non-linear systems* Non-uniform flow

+ Normal force*

+ Noses*

Nozzles# Nuclear fusion*

Nuclear propulsion

Numerical analysis*

Observation*

Ogives*

One-dimensional*

Openings*

+ Open Throat*

Operations*

Optimization*

Orbits#

Ordinates*

Orientation*

Orifices*

Orthogonal functions*

Oscillations*

Oscillographs* Outlets*

Output*

Oxygen*

Panels*

Parabolas#

Parameters*

Partial differential equations*

Particles*

Payloads*

Pendulums* Penetration*

Performance*

Periodic variations*

Permeability*

Perturbation*

Perturbation theory*

Phase*

Phase angle*

Phenolio resins*

Photographs* Physical property*

Pine wood*

Pipe*

Pipe flow*

Pistons*

Pitch*

Pitot tubes*

Pivots*

Plane geometry*

Planets*

+ Planforms*

Plasma*

Plasticity*

Plastics Plates*

Plotting*

+ Pointed bodies*

Points*

+ Polar co-ordinates* Polyamide resins

Polygons*

Polymethyl methorylate

Polystyrene

Polytetra Fluoroethylene*

+ Porous cone

Porous materials

Position#

Potential energy*

+ Potential flow

Power series Prandtl No. *

Predictions*

Preservation*

Pressure*

Pressure distribution*

+ Pressure drag*

Pressure drop

Pressure gradient*

+ Pressure jumps Pressure measurement*

Pressure pumps

Pressure reduction*

Prevention

Procedures*

Production*

+ Profile drag

Projectiles

Projection*

Propagation*

Propellers*

Propertles*

Propulsion systems

Proving

Pumps*

Purity*

Qualitative analysis

Quartz* Radar

Radiation*

Radiation effects*

Radioactive decay* Radioisotopes*

Radii*

Randomization*

Range*

Ratios* + Rayleighs Problem*

Reactions (Chemical)*

Reaction kinetics*

Recombination reactions* Recovery*

Rectangles"

Reduction*

Re-Entry*

Re-entry vehicles* Regression*

Reflection

Refraction

Reinforced plastics*

Relaxation* Relaxation methods

Relaxation time*

Reliability*

Relieving

Removal*

Repeater* Research*

Reservoirs*

Resistance* Resistivity*

Resonant frequency*

Response time* Retarding*

+ Reversal of control

Revisions* Revolution#

Reynolds No.* + Rigid airfoils*

+ Rigid bodies + Rigid wings* Rigidity*

Rocket* + Rocket chambers* Rocket engines* + Rocket gliders* Rocket nozzles* Rocket planes* Rocket propellants* Rocket propulsion* Rods* Roll* Roofs* Rotation* Rotor blades* Rotors* Roughness* Rudders* Safety factor* Satellites* Satellite velocity* Saturation* Scale* Scattering* Scavenging* Screens Seasonal variations* Sensors* Separation* Series* Servomechanisms* Shadow photography Shape* Shear flow* Shearing* Shear stress* Sheet* Shells Shock* Shock tubes* Shock waves* Shroud* Sides* + Sideslip* + Sidewash* Silica* Silicon carbide* + Similarity* Similarity laws* Simulation* + Sinking* Sinusoidal functions* Skewness* Skin* Skin friction* + Skipping* Skip rockets Skip vehicles* Slabs* + Slender bodies* + Slender cones* Slipstream* Slope* Slots* Sodium* Sodium carbonate Sodium chloride* Solar heating* Solar phenomena* Solar radiation* Solar sails* Solid propellants* Solids* Sound waves* Sounding* Sounding rockets* Sound propagation Sources* Spacecraft* Space probes* Spacing*

Span* Specific heat* Specification* Specific impulse* Spectra* Specular reflection* Spheres* Spikes* Spirals Spoilers* Spreading Springs* Squares* Stability* Stages* Stagnation point* Stall* + Stanton No. Starting* Static characteristics* Static pressure* Static tests* Stationary processes* Statistical methods* Stator blades* Steady flow*
+ Steady forces Steady pressure Steady state* Steels* Stiffening* Stiffness* Sting* Stopping* Stoichiometry* Straightness* Strain gages + Stream function* Streamlines* Strength* Stress concentration* Stresses* Stretching Strip* + Strip theory Structures* Subcritical flow* Sublimation* + Subsonic flow* Sun* Supersonic flow* Supersonic planes* Supports* Surfaces* Surface combustion* Surface radiation* Surface temperature* Surges* Surveys* Sweep* Sweepback* Swept wing* Swirling* Symmetry* Systems* Tables* Tabs* Tabulation* Tails* Take off* Tangent* Taper ratio* Tapered wing Taylor's Series* Teflon* Telemetry* Temperature* Temperature control Temperature distribution* Temperature inversions*
Tensions*
Terminal guidance
Test facilities*

+ Test section*
Testing*
Testing equipment*
Testing machines*

Theories*

Theorems*
Thermal conductivity*
Thermal expansion*
Thermal diffusivity
Thermal insulation
Thermal stresses*

Thermodynamics*
Thickening*

Thickness* Thin layer

+ Thin wings*

+ Throats*
Thrust*
Thrust vector control*

Time* Time constant

Time series analysis*

+ Tips*
Titan*
Torque*
Torsion*

+ Total drag"

+ Trailing edges*
Trajectory*
Transferring*
Transformation*
Transition point*

Transition point*
+ Transitional layer*
Transonic flight*
Transonic flow*
Transpiration*
Transverse waves*
Triangles*
Trips*
Trubling*
Trugsten*

Tungsten*
Turbine blades
Turbines*

Turbocompressions* Turbojet engines*

Turbulence*
Turbulent boundary layer*

Turbulent flow*
Twisting*

+ Two-dimensional*
Two-phase flow*
Ultraviolet radiation*

+ Undershoot

Uniform flow*
+ Uniform pressure*

Unsteady flow*
+ Unsteady lift*
Unsteady state*

Unsteady state*
Upper atmosphere*

* Upwash*
Vacuum*
Vanes*
Vaporising*
Vapors
Vapour pressure*
Variability*
Variables*
Variations*
Vectors*
Velocity*
Venturi Meters*
Vanus*

Vibrations* + Vibrational equilibrium Viscous* Viscous flow* Viscous fluids*

+ Visualisation

Volume* Vortex*

+ Vortex filament Vortex generators* Vortex sheet

+ Vorticity*

+ Wake*
Walls*
Water
Water flow*
Wavelength*
Wave propagation
Waves*

+ Wedg**es*** Weight*

Whitham Theory
 Wind*
 Wind tunnel*

Wind tunnel models* Wind tunnel nozzles* Wind velocity*

Windows

Wing-body configurations*

Wing-body-tail configurations* Wings* Wires*

Wood* Yaw*

APPENDIX 6.1

EXAMPLES OF VARIOUS COMPUTER PRINT-OUTS

Original I.B.M. Tape arrangement

The indexing information is arranged in document order, resulting in a linear file of 1,400 documents. The information transferred from the index sheet is recorded in two separate 'blocks' as follows:-

- 1. List of themes and the concepts which make up each theme, as given in the index sheet.
- 2. Lists of the concepts, giving the code letter and terms in each concept.

 The weight is assigned to each term in the concept, but since these weights have been transferred from those assigned to the single terms, the highest weight that any term has received will appear in all its concepts.
 - e.g. Concept F = Turbulent (10) Flow(9) Field(9)
 Concept L = Transonic(8) Flow(9)

The tape was prepared from 80-column punched cards, with the information punched on each line as follows:-

Column 1. Punched with figure 1 or 2. This indicates whether the index information given in columns 6-80 refers to Block 1 (themes) or Block 2 (Concepts)

Columns 2-5 The four-figure document reference number.

Columns 6-80 The index information:

If Block 1, a. A comma indicates the start of each theme.

- b. The themes are identified by a two figure number, and are punched in order 01, 02, 03, etc.
- c. The concepts which make up each theme are listed by their two-letter code.
- d. An asterisk indicates the end of the final theme.
- If Block 2 a. A comma indicates the start of each concept.
 - b. The concepts are identified by a two-letter code and are listed in the following sequence:blank A to blank Z, AA, BB, CC, etc.
 - c. The terms making up each concept are punched in full
 - d. Each term is followed by the weight number, enclosed by slashes //
 - e. An asterisk indicates the end of the final concept.

11419,01 A B C,02 A D E F J,03 A D E G J,04 A D E H J,05 A D E I J* 21419, ASTIFFENED/10/WEB/10/PLATE/10/, BSHEAR/10/LOADING/10/, CINTERMEDIATE/9/ 21419VERTICAL/9/STIFFENER/10/DESIGN/9/, DINTERMEDIATE/9/VERTICAL/9/STIFFENER/10/ 21419, ESHEAR/10/BUCKLING/8/STRESS/8/, FSTIFFENER/10/FLEXURAL/8/RIGIDITY/8/ 21419, GSTIFFENER/10/SPACING/8/, HSINGLE/6/SIDED/6/STIFFENER/10/, IDOUBLE/6/ 21419S10ED/6/STIFFENER/10/, JTEST/7/*
11420,01 A C D E F G,02 A C D H I G,03 A C D J G,04 A C D K G,05 A C D L M 11420,06 A C D N K,07 A C D 0,08 A C D P,09 A C D Q S G,10 A C D Q T G
1420,11 A C D Q U G,12 A C D R S G,13 A C D R T G,14 A C D R U G,15 B C D E F G 11420,16 B C D H I G,17 B C D J G,18 B C D K G,19 B C D L M,20 B C D N K 11420,21 B C D 0,22 B C D P,23 B C D Q S G,24 B C D Q T G,25 B C D Q T G 11420,26 B C D R S G,27 B C D R T G,28 B C D R U G* 21420, ASWEPT/10/WING/10/HIGH/9/ASPECT/9/RATIO/9/, BNARROW/9/DELTA/10/WING/10/ 21420, CSHARP/10/EDGED/10/, DHIGH/9/ANGLE/9/ATTACK/9/, EUPPER/9/SURFACE/9/ 21420, FTURBULENT/10/FLOW/9/FIELD/9/, GWIND/9/TUNNEL/9/EXPERIMENT/9/, HROOT/8/ 21420SECTION, ICONICAL/8/FLOW/9/FIELD/9/, JVORTEX/8/SHEET/8/POSITION/7/ 21420, KYORTEX/8/PATH/8/LOCATION/8/, LTRANSONIC/8/FLOW/9/, MDRAG/8/RISE/8/ 21420, NLEADING/8/EDGE/8/SEPARATION/8/, OVORTICITY/8/, PSTALLED/8/FLOW/8/ 21420, QPRESSURE/8/DISTRIBUTION/8/, RVELOCITY/8/DISTRIBUTION/8/, SSWEEP/8/ 21420ANGLE/9/, TANGLE/9/ATTACK/9/, UREYNOLDS/8/* 11421,01 A D F G H,02 B C D F G H,03 A E F G H,04 B C E F G H,05 A D I G H
11421,06 B C D I G H,07 A E I G H,08 B C E I G H,09 A D F J G H,10 B C D F J G H 11421,11 A F F J G H,12 B C E F J G H,13 A D I J G H,14 B C D I J G H 11421,15 A E I J G H,16 B C E I J G H,17 A D F Q S,18 A E F Q S,19 A D I Q S 11421,20 A E I Q S,21 A D P Q R,22 A E P Q R,23 K C D F G H,24 K C E F G H
11421,25 K C D I G H,26 K C E I G H,27 L C D F G H,28 L C E F G H,29 L C D I G H
11421,30 L C E I G H,31 M C D F G H,32 M C E F G H,33 M C D I G H,34 M C E I G H 11421,35 N C D F G H,36 N C E F G H,37 N C D I G H,38 N C E I G H,39 O C D F G H 11421,40 O C E F G H,41 O C D I G H,42 O C E I G H*
21421, ABLUNT/10/NOSED/10/LONG/9/BODYOFREVOLUTION/10/, BBLUNT/10/NODE/8/ 21421, CLONG/9/CYLINDRICAL/10/AFTERBODY/10/, DHYPERSONIC/9/AIR/9/FLOW/9/ 21421, EHYPERSONIC/9/HELIUM/10/FLOW/9/, FINDUCED/10/PRESSURE/9/, GTHEORETICAL/9/

Tape arrangement at Harvard Computation Laboratory.

The indexing information is arranged in document order, and includes only the indexed concepts, ignoring the themes. Each concept consists of the constituent single terms, terminated by a stop, each concept is treated in the same way as the sentences in the full text normally processed by the SMART system, with the running totals of concepts indicated in a right-hand column. The weights assigned at Cranfield are incorporated by repeating the concepts; three levels are recognised, concepts weighted 10 and 9 are recorded three times, concepts weighted 8 and 7 are recorded twice, and concepts weighted 6 and 5 once.

A copy of the print-out for the same document as shown on the previous page, namely 1420, is printed below.

```
HIGH ASPECT RATIO SWEPT WING . FIGH ASPECT RATIO SWEPT WING . HIGH ASPECT RATIO SWEPT WING . NARROW CELTA WING SHARP EDGED . NARROW DELTA WING SHARP EDGED .

    HIGH ANGLE ATTACK - HIGH ANGLE ATTACK - HIGH ANGLE ATTACK
    UPPER SURFACE FLOW FIELD TURBULENT - UPPER SURFACE FLOW FIELD
    TURBULENT - UPPER SURFACE FLOW FIELD TURBULENT - WIND TUNNEL

                                                                         UPPER SURFACE FLOW FIELD
                                                                                        . WIND TUNNEL EXPERT-
                                                                                                                                       11
MENT . WIND TUNNEL EXPERIMENT . WIND TUNNEL EXPERIMENT . ROOT SECTION, ICONICAL FLOW FIELD . REOT SECTION, ICONICAL FLOW FIELD . POSITION VORTEX PATH LOCATION . POSITION VORTEX
                                                                                                                                       13
                                                                                                                                       16
                                                                                                                                       17
SHEET VORTEX PATH LOCATION . TRANSONIC FLOW
                                                                                        TRANSONIC FLOW
                                                                                                                                       19
 . DRAG RISE LEADING ECGE SEPARATION . CR/G RISE LEADING EDGE SEPARAT-
                                                                                                                                       21
ION . VORTICITY . VORTICITY . STALLED FLOW PRESSURE DISTRIBUTION
                                                                                                                                       23
. STALLED FLCW PRESSURE DISTRIBUTION . VELOCITY DISTRIBUTION . VELOCITY DISTRIBUTION . SWEEP ANGLE . SWEEP ANGLE . ANGLE ATTACK . ANGLE ATTACK . REYNOLDS . REYNOLDS
                                                                                                                                       26
                                                                                                                                       28
                                                                                                                                       32
                                                                                                                                       36
```

Tape arrangement at Cambridge Language Research Unit.

This follows more closely the information on the original tapes, with two sections for each document, the first covering themes, and the second giving the concepts and terms.

An example pf the print-out, for document 1420, is shown below; the description of the print-out is as follows.

```
Section 1
  Line 1.
           Document number; Section 1 digit; half-word count for document.
           Theme count; Half-word count for themes.
  Line 2.
           Theme number; Concept letter count; Half-word count for concept
  Line 3
           letters; Concept letters.
  Line n.
          Section 1 terminator.
Section 2
  Line 1.
           Document number; Section 2 digit.
  Line 2.
           Concept count; Half-word count for concepts.
  Line 3. Concept letter; Word count; Half-word count for words; weight
           number; Half-word count for word; Word.
  . . . . . . .
  Line n. Section 2 terminator.
```

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        7
                     ( 9)+
                            5 ANGLE
                                                     1 91+ 3 ATTACK
                     ( 8)+ 3 REYNOLDS
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